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MATHEMATICAL ANALYSIS OF HEAT CONDUCTION IN A SECTOR OF A HOLLOW CIRCULAR CYLINDER

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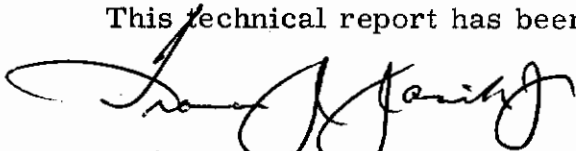
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FOREWORD

This final technical report was prepared by the Martin Company, Orlando Division, Orlando, Florida, under Air Force Contract No. AF 33(615)-2385. The contract, sponsored by Technical Area 750A, Mechanics of Flight, was accomplished under Project 1467, Structural Analysis Methods. It was administered under the Structures Division, Air Force Flight Dynamics Laboratory, Deputy Commander/Technology, RTD, with Mr. T. N. Bernstein, (FDTR), acting as Project Engineer.

This report (contractor's report No. OR 8042) was completed in March 1966 and covers the work performed from March 1965 to January 1966. Mathematical analysis and development were conducted by Mr. J. G. Torian under the supervision of Mr. J. I. Gonzalez, Head of the Fluid Dynamics, Heat, and Mass Transfer Section in the Engineering Mechanics Research Laboratory, Martin Company. The digital programming was performed by Mr. V. T. Murphy under the supervision of Mr. C. J. Gentry of the Scientific Computer Applications Department.

This technical report has been reviewed and is approved.



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ABSTRACT

This study was undertaken to develop an analytical method of determining temperature distribution in a sector of a hollow circular cylinder, with temperature dependent thermophysical properties, based on an exact mathematical solution. The subject non-linear differential equation is reduced to a linear differential equation by an integral transform and with the assumption that only the thermal conductivity is a function of temperature, whereas the thermal diffusivity remains constant. The solution has been programmed in FORTRAN IV for application to an IBM 7094. In addition, a successive transform applicable to the solution of similar problems involving temperature dependent thermal diffusivity is illustrated and discussed.

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SYMBOLS

Mathematical	Program Counterpart	Use
v		real temperature
T		transform temperature
r		radius
ϕ		angle
θ		time
λ		dummy time variable
$K(v)$		thermal conductivity
ρ		density
$C(v)$		specific heat
α	ALPHØ	thermal diffusivity
r_1	ARE(1)	inner radius
r_2	ARE(2)	outer radius
ϕ_1	PHI(1)	lower angular boundary (equal to zero)
ϕ_2	PHI(2)	upper angular boundary
K_o	AZERØ	thermal conductivity at v equal zero
γ	GAMMØS(i)	angular eigenvalue
β	BAYTØS(i)	radial eigenvalue
ϵ	EPILØN(i)	radial eigenvalue
$A_{\gamma\beta}$	ARBA(i,j)	arbitrary coefficient
B_{γ}	BSUB(i)	arbitrary coefficient
$G_{\gamma\beta}$	ARBG(i,j)	arbitrary coefficient

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F_Y	FSUB(i)	arbitrary coefficient
F_ϵ	AF(i)	arbitrary coefficient
M_ϵ	AM(i)	arbitrary coefficient
l_{11}	AL(1,1)	boundary condition indicator
l_{12}	AL(1,2)	boundary condition indicator
l_{21}	AL(2,1)	boundary condition indicator
l_{22}	AL(2,2)	boundary condition indicator
k_{11}	AK(1,1)	boundary condition indicator
k_{12}	AK(1,2)	boundary condition indicator
k_{21}	AK(2,1)	boundary condition indicator
k_{22}	AK(2,2)	boundary condition indicator
b	BEE	constant used in definition of K(v)
c	SEE	constant used in definition of K(v)
$f_{r1}(\phi)$	FARE1(i)	coordinate varying boundary condition in real temperature at r_1
$f_{r2}(\phi)$	FARE2(i)	coordinate varying boundary condition in real temperature at r_2
$f_{\phi1}(r)$	FPHI1(i)	coordinate varying boundary condition in real temperature at ϕ_1
$f_{\phi2}(r)$	FPHI2(i)	coordinate varying boundary condition in real temperature at ϕ_2
$F_{r1}(\phi)$	FARE1(i)	coordinate varying boundary conditions in transform variable at r_1
$F_{r2}(\phi)$	FARE2(i)	coordinate varying boundary conditions in transform variable at r_2
$F_{\phi1}(r)$	FPHI1(i)	coordinate varying boundary conditions in transform variable at ϕ_1
$F_{\phi2}(r)$	FPHI2(i)	coordinate varying boundary conditions in transform variable at ϕ_2

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$h(r,\phi)$	DZERØ(i,j)	initial temperature distribution in real temperature
$f(r,\phi)$	STEADY(i,j)	steady state solution
$\mathcal{T}[v]$		transform of real temperature
$\Lambda [T]$		inversion of transform temperature
$T_i(\theta,r,\phi)$	TERMA	initial temperature distribution solution in transform variable
$T_s(\theta,r,\phi)$	-TERMB+ $g(\theta) f(r,\phi)$	surface effects solution in transform variable
ξ	XI	function as defined in text
$\Xi(v,\theta,r,\phi)$		transform time function
$\Omega(v,\psi,r,\phi)$		inversion of time function
$J_\gamma(\beta r)$		Bessel function of first kind and order γ
$Y_\gamma(\beta r)$		Bessel function of second kind and order γ
$J'_\gamma(\beta r)$		$\frac{\partial}{\partial \beta} J_\gamma(\beta r)$
$Y'_\gamma(\beta r)$		$\frac{\partial}{\partial \beta} Y_\gamma(\beta r)$
$g(\theta)$	GTHETA	time function
ψ		transform time

SECTION I

INTRODUCTION

This study was undertaken to develop an analytical method for determining temperature distribution in a sector of a hollow circular cylinder, with temperature dependent thermophysical properties, based on an exact mathematical solution. The subject non-linear differential equation was reduced to a linear differential equation by an integral transform with the assumption that only the thermal conductivity is a function of temperature, whereas the thermal diffusivity remains constant.

The integral transform method used in linearizing the equations is presented in Reference 1. Carslaw and Jaeger attribute the method, as applied to transient problems, to Van Dusen (1930) and note that the method as applied to steady heat flow dates back to 1894 (Kirchhoff). The Russian authors, Kudryashev and Zhemkov (Reference 2), rediscovered the method in 1963. They present a method of solution to problems involving temperature dependent thermal diffusivity using a successive transform. The formulation of these methods to analysis of two-dimensional transient heat conduction in a segment of a hollow circular cylinder is presented in Section IIA of this report.

The resulting linear differential equation, initial temperature distribution, and coordinate and time varying boundary conditions constitute a Sturm-Liouville System in the transform variable. A series solution of the linear system, as applicable to hollow semicircular cylinders, is presented by Luzzi (Reference 3) and by Torian (Reference 4) as applicable to hollow sectors of cylinders. Section IIB of this report presents a revised solution to the linear system based on Reference 4. The revisions are directed primarily toward greater applicability to calculations on a digital computer.

The solution of the linear system is applicable to either the temperature dependent or constant thermal diffusivity case. A digital computer program was developed based on this system as applicable to the temperature dependent thermal conductivity, constant diffusivity case. Mathematical formulation is given in Section IIB. A discussion of the digital program is given in Section III. The authors' comments on mathematical formulation of the temperature dependent thermal diffusivity case are included in Section IIC and the latter part of Section IIA.

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The utilization of the computer program is given in Section IIIB. A detailed description of the program is provided in Section IIIA, and Section IV shows its application to typical sample problems.

SECTION II

MATHEMATICAL FORMULATION

A. TRANSFORMATION OF THE NON-LINEAR EQUATION

The differential equation to be solved is

$$\rho C(v) \frac{\partial v}{\partial \theta} = K(v) \left[\frac{\partial^2 v}{\partial r^2} + \frac{1}{r} \frac{\partial^2 v}{\partial \theta^2} + \frac{1}{r} \frac{\partial v}{\partial r} \right] + \frac{\partial K(v)}{\partial v} \left[\frac{1}{r^2} \left(\frac{\partial v}{\partial \phi} \right)^2 + \left(\frac{\partial v}{\partial r} \right)^2 \right] \quad (1)$$

where the initial condition is

$$v(0, r, \phi) = h(r, \phi) \quad (2)$$

and the coordinate and time varying surface conditions (Figure 1) are:

1 At ϕ_1

$$-K(v) \frac{\partial v}{\partial \phi} = r f_{\phi_1}(r) g(\theta) \quad (3)$$

(representing a coordinate and time varying heat flux at this surface), or

$$v = f_{\phi_1}(r) g(\theta) \quad (4)$$

(representing a coordinate and time varying temperature at this surface);

2 At ϕ_2

$$K(v) \frac{\partial v}{\partial \phi} = r f_{\phi_2}(r) g(\theta) \quad (5)$$

(representing a coordinate and time varying heat flux at this surface), or

$$v = f_{\phi_2}(r) g(\theta) \quad (6)$$

(representing a coordinate and time varying temperature at this surface);

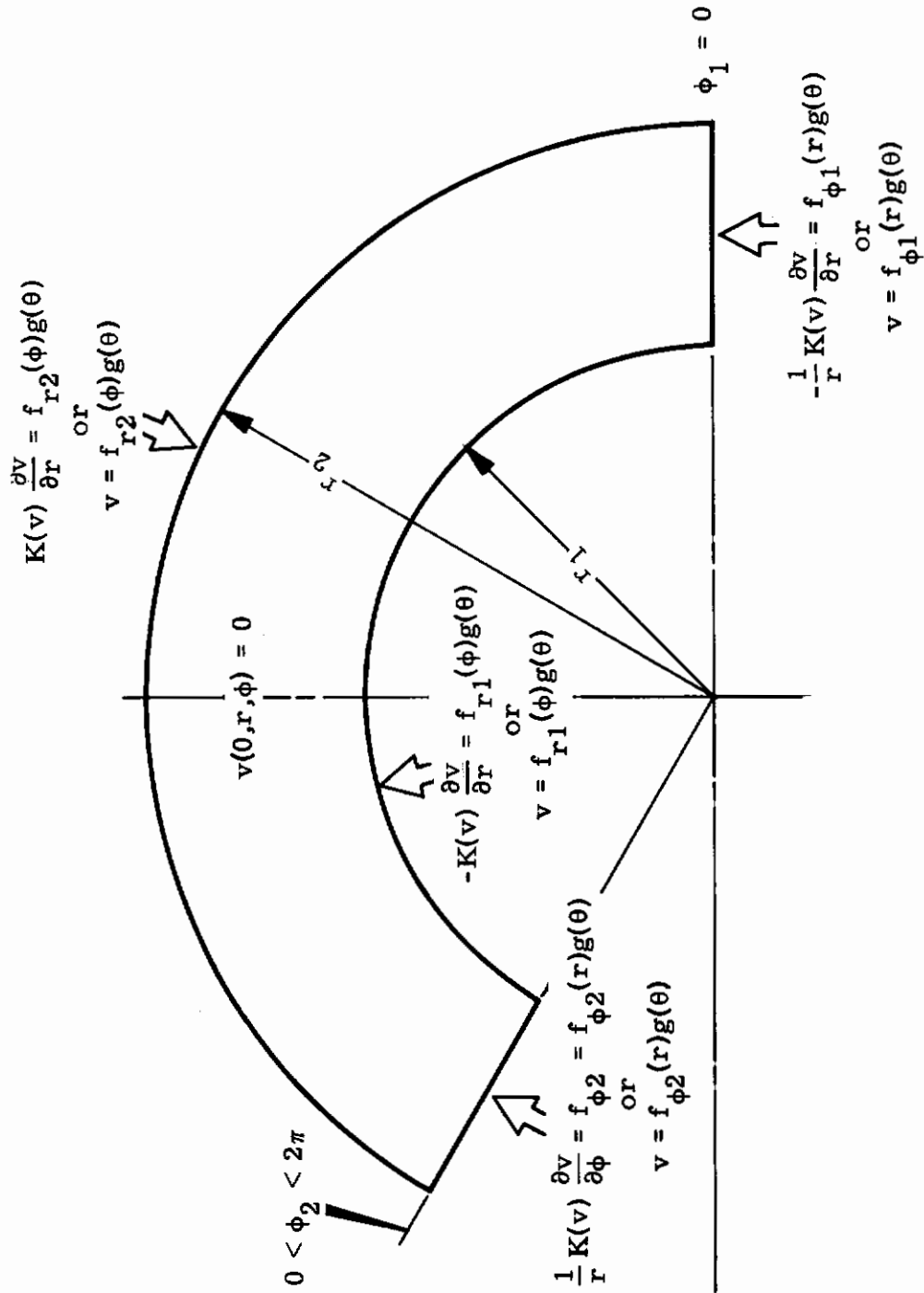


Figure 1. Mathematical Model

3 At r_1

$$-K(v) \frac{\partial v}{\partial r} = f_{r1}(\phi) g(\theta) \quad (7)$$

(representing a coordinate and time varying heat flux at this surface),
or

$$v = f_{r1}(\phi) g(\theta) \quad (8)$$

(representing a coordinate and time varying temperature at this surface);

4 At r_2

$$K(v) \frac{\partial v}{\partial r} = f_{r2}(\phi) g(\theta) \quad (9)$$

(representing a coordinate and time varying heat flux at this surface),
or

$$v = f_{r2}(\phi) g(\theta) \quad (10)$$

(representing a coordinate and time varying temperature at this surface).

By the methods of Reference 1, a new variable is introduced:

$$T = \frac{1}{K_0} \int_0^v K(v) dv, \quad (11)$$

where $K(v)$ is the thermal conductivity expressed as a function of temperature v , and K_0 is the value of $K(v)$ at v equal to zero. We write

$$T = \tau [v] \quad (12)$$

after integrating Equation (11). The inversion may be written as

$$v = \Lambda [T]. \quad (13)$$

When Equation (1) is rewritten in terms of the new variable, we obtain:

$$\frac{\rho C(v)}{K(v)} \frac{\partial T}{\partial \theta} = \frac{\partial^2 T}{\partial r^2} + \frac{1}{r^2} \frac{\partial^2 T}{\partial \phi^2} + \frac{1}{r} \frac{\partial T}{\partial r}. \quad (14)$$

Contrails

If we restrict the variation of $K(v)$ and $C(v)$ such that

$$\frac{\rho C(v)}{K(v)} = \text{constant} = \frac{1}{\alpha},$$

which is indicative of constant thermal diffusivity, Equation (14) becomes

$$\frac{1}{\alpha} \frac{\partial T}{\partial \theta} = \frac{\partial^2 T}{\partial r^2} + \frac{1}{r^2} \frac{\partial^2 T}{\partial \phi^2} + \frac{1}{r} \frac{\partial T}{\partial r} \quad (15)$$

and the initial condition becomes

$$T(0, r, \phi) = T [h(r, \phi)]; \quad (16)$$

the boundary conditions, Equations (3) through (10), become:

1 At ϕ_1

$$-K_o \frac{\partial T}{\partial \phi} = r f_{\phi_1}(r) g(\theta) \quad (17)$$

or

$$T = T [f_{\phi_1}(r) g(\theta)]; \quad (18)$$

2 At ϕ_2

$$K_o \frac{\partial T}{\partial \phi} = r f_{\phi_2}(r) g(\theta) \quad (19)$$

or

$$T = T [f_{\phi_2}(r) g(\phi)]; \quad (20)$$

3 At r_1

$$-K_o \frac{\partial T}{\partial r} = f_{r_1}(\phi) g(\theta) \quad (21)$$

or

$$T = T [f_{r_1}(\phi) g(\theta)]; \quad (22)$$

4 At r_2

$$K_0 \frac{\partial T}{\partial r} = f_{r_2}(\phi) g(\theta) \quad (23)$$

or

$$T = T [f_{r_2}(\phi) g(\theta)]. \quad (24)$$

A solution of the system of Equations (1) through (10) can be obtained for the temperature dependent thermal conductivity (constant diffusivity) case by transforming these equations to a linear system, Equations (15) through (24). When a solution to the linear system for r , ϕ , and θ in the transform temperature (T) is obtained, a solution to the non-linear system can be obtained for r , ϕ , and θ in the real temperature (v) by application of the inversion from Equation (13).

To proceed with solution of the temperature dependent thermal diffusivity case, we introduce the new variable

$$\psi = \int_0^\theta \frac{K(v)}{\rho C(v)} d\theta. \quad (25)$$

Then

$$\frac{\partial T}{\partial \theta} = \frac{\partial T}{\partial \psi} \frac{\partial \psi}{\partial \theta} = \frac{K(v)}{\rho C(v)} \frac{\partial T}{\partial \psi} \quad (26)$$

or

$$\frac{\rho C(v)}{K(v)} \frac{\partial T}{\partial \theta} = \frac{\partial T}{\partial \psi}. \quad (27)$$

When Equation (27) is inserted in Equation (14), we obtain

$$\frac{\partial T}{\partial \psi} = \frac{\partial^2 T}{\partial r^2} + \frac{1}{r^2} \frac{\partial^2 T}{\partial \phi^2} + \frac{1}{r} \frac{\partial T}{\partial r} \quad (28)$$

which is a linear differential equation similar to Equation (15) except in a transform time variable (ψ) rather than real time (θ). (The method of reducing non-linear diffusion Equation (14) to linear diffusion Equation (28) is based on the work of Kudryashev and Zhemkov in Reference 2.)

Equation (25) may be written

$$\psi = \int_0^{\theta} \alpha [\Lambda(T)] d\theta = \Xi(v, \theta, r, \phi) \quad (29)$$

where

$$\frac{K(v)}{\rho C(v)} = \alpha(v) = \alpha[\Lambda(T)]. \quad (30)$$

Then

$$\theta = \int_0^{\psi} \frac{d\psi}{\alpha[\Lambda(T)]} = \Omega(v, \psi, r, \phi) \quad (31)$$

is the inversion to real time in the transform temperature (T).

Although the differential equation in the transform time variable is linear, the real problem can only be transformed into a complete linear system (including the boundary conditions) for certain cases of the boundary conditions. This is a result of the requirements for transformation of the time function $g(\theta)$ into a new function in the transform time variable (ψ). For a prescribed coordinate and time varying temperature at a surface, the transformation, Equation (29), can be accomplished since the transform temperature* is known. A coordinate varying heat flux may be prescribed at a surface, but variation with time is disallowed since the surface temperature is not known and the transformation of the boundary conditions in terms of the new time variable cannot be accomplished.

For a typical coordinate and time varying temperature at a surface in the form

$$v_s = f_1(r, \theta), \quad (32)$$

*The transforms can be made in terms of the real temperature by noting that an equivalent form of Equation (29) is

$$\psi = \int_0^{\theta} \alpha(v) d\theta.$$

the transformation in temperature is

$$T = \mathcal{T}[f_1(r, \theta)]. \quad (33)$$

The succeeding transformation in time is

$$T = \mathcal{T}[f_2(r, \psi)], \quad (34)$$

where the functions f_1 and f_2 are related through Equation (29).

From the preceding, it is concluded that a solution to a system of equations similar to Equations (1) through (10) can also be solved by exact methods (excluding prescribed time varying flux at a surface) for the case of temperature dependent thermal diffusivity. The non-linear system is reduced to a linear system by two successive transformations. Obtaining a solution to the linear system at r and in terms of the transform temperature and time variable, T and ψ , successive inversions are applied to obtain the solution in the real temperature and time.

B. SOLUTION OF THE TEMPERATURE DEPENDENT THERMAL CONDUCTIVITY, CONSTANT DIFFUSIVITY CASE

1. Transformation and Inversion

The solution of the subject problem involving temperature dependent thermal conductivity requires a transformation of the initial and boundary conditions to the transform variable and a corresponding inversion from transform temperature to real temperature. This transformation and inversion, as used in this report, is independent of the coordinate system or the form of the initial and boundary conditions. It depends only on the relationship of thermal conductivity to real temperature. This section discusses various forms of the relationship between thermal conductivity and real temperature and presents the corresponding transformation and inversion.

For various algebraic relations of thermal conductivity to real temperature, Equation (11) can be manipulated to provide an algebraic expression for the transform temperature. This equation can then be solved for real temperature to provide the inversion of Equation (13). Five such cases were prepared for incorporation in the digital computer program. The expression for thermal conductivity and the corresponding transformation and inversion expressions are given in Table I.

Data obtained from a curve of thermal conductivity versus temperature, which cannot be curve fit by one of the expressions on Table I, may be integrated numerically to obtain the relationship between real and transform

TABLE I
Transform and Inversion Relations

K(v)	T v	Λ(T)
Case 1 $K(v) = K_0 \exp(-bv)$	$T = \frac{1}{b} [1 - \exp(-bv)]$	$v = -\frac{1}{b} \ln [1 - bT]$
Case 2 $K(v) = \frac{K_0}{1 + bv + cv^2}$ $q = 4c - b^2$ A. $q < 0, c > 0$ B. $q = 0, c > 0$ C. $q > 0, c > 0$	$T = \frac{1}{\sqrt{-q}} \ln \left[\frac{\frac{2cv}{b - \sqrt{-q}} + 1}{\frac{2cv}{b + \sqrt{-q}} + 1} \right]$ $T = 2 \left[\frac{1}{b} - \frac{1}{2cv + b} \right]$ $T = \frac{2}{\sqrt{q}} \left[\tan^{-1} \frac{2cv + b}{\sqrt{q}} - \tan^{-1} \frac{b}{\sqrt{q}} \right]$	$v = \frac{(b + \sqrt{-q}) X - b + \sqrt{-q}}{2c [1 - X]}$ where $X = \exp(\sqrt{-q} T) \left[\frac{b - \sqrt{-q}}{b + \sqrt{-q}} \right]$ $v = \frac{bT}{2 \left[2 \frac{c}{b} - cT \right]}$ $v = \frac{\sqrt{q}}{2c} Y - \frac{b}{2c}$ where $Y = \frac{\tan \frac{\sqrt{q} T}{2} + \frac{b}{\sqrt{q}}}{1 - \frac{b}{\sqrt{q}} \tan \frac{\sqrt{q} T}{2}}$
Case 3 $K(v) = \frac{K_0}{1 + bv}$	$T = \frac{1}{b} \ln (1 + bv)$	$v = \frac{1}{b} (\exp(bT) - 1)$
Case 4 $K(v) = K_0 (1 + bv)$	$T = v + \frac{b}{2} v^2$	$v = -\frac{1}{b} + \frac{1}{b} \sqrt{1 + 2bT}$
Case 5 $K(v) = K_0$	$T = v$	$v = T$

temperature. The resulting data when plotted can be used for both the transformation and inversion process.

A third form of the relationship between thermal conductivity and real temperature, which combines the input simplicity of the algebraic expressions with the flexibility of the numerical integration of data points, may be obtained by writing n exponential terms in the form

$$K(v) = \sum_{i=1}^n a_i v^{b_i}; \quad (35)$$

then

$$T = \frac{1}{K_0} \left[\sum_{i=1}^n \frac{a_i}{b_i^{-1}} v^{b_i^{-1}} \right]. \quad (36)$$

At this point a curve should be prepared based on Equation (36) and used for the inversion process, since a general expression for the inversion in Equation (13) cannot be written directly from Equation (36).

Regardless of the form of relationship of thermal conductivity to real temperature, the thermal conductivity must be greater than zero over the range of temperatures involved in the particular problem. It is obvious that no one would suggest a negative or zero thermal conductivity in a practical problem. However, when using algebraic relations of higher orders to curve fit given data on thermal conductivity, care should be taken to ensure that the function is positive over the argumented range of temperatures. Abrupt maxima or minima should also be avoided when using forms other than the algebraic cases of Table I. This is not a mathematical restriction in the strict sense; however, maxima or minima in the thermal conductivity curve produce "flat spots" in the curve of transform versus real temperature with attendant inaccuracies in the numerical inversion process.

In the preceding sections, the solution to the subject problem will be solved by the principle of superposition in two parts, T_i and T_s , where T_i is the solution resulting from the initial temperature distribution effects, and T_s is the solution resulting from the coordinate and time varying surface conditions. The inversion process is applied to the sum of T_i and T_s . That is

$$V = \Lambda [T_i + T_s]. \quad (37)$$

2. Solution of the Transformed Equation

The solution of the linear system, Equations (15) through (24), may (by the principle of superposition) be expressed as the sum of the solution resulting from the initial temperature distribution with homogeneous boundary conditions and the solution resulting from the prescribed boundary condition with zero initial temperature distribution. These two solutions are discussed separately in the following paragraphs.

a. Initial Temperature Distribution Effects

The system of equations to be solved is

$$\frac{1}{\alpha} \frac{\partial T_i}{\partial \theta} = \frac{\partial^2 T_i}{\partial r^2} + \frac{1}{r^2} \frac{\partial^2 T_i}{\partial \phi^2} + \frac{1}{r} \frac{\partial T_i}{\partial r} \quad (38)$$

with the initial condition

$$T_i(0, r, \phi) = T[h(r, \phi)] \quad (39)$$

and the boundary conditions

$$k_{11} \frac{\partial T_i}{\partial \phi} + k_{12} T_i = 0 \quad \text{at } \phi_1, \quad (40)$$

$$k_{21} \frac{\partial T_i}{\partial \phi} + k_{22} T_i = 0 \quad \text{at } \phi_2, \quad (41)$$

$$l_{11} \frac{\partial T_i}{\partial r} + l_{12} T_i = 0 \quad \text{at } r_1, \quad (42)$$

and

$$l_{21} \frac{\partial T_i}{\partial r} + l_{22} T_i = 0 \quad \text{at } r_2. \quad (43)$$

Note that the boundary conditions of Equations (40) through (43) are expressed in a general form involving constants k_{ij} and l_{ij} which may be selected as zero or unity to prescribe heat flux or temperature at the subject boundary.

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The solution to the system of Equations (38) through (43) may be written, based on the author's previous work in Reference 4, as

$$T(\theta, r, \phi) = \sum_{\gamma} \sum_{\beta} A_{\gamma\beta} \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \phi \right] \\ \times \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right] e^{-\alpha\beta^2\theta} \quad (44)$$

where γ 's are the real positive roots of

$$\begin{vmatrix} (k_{11} \gamma \cos \gamma\phi_1 + k_{12} \sin \gamma\phi_1) & (k_{12} \cos \gamma\phi_1 - k_{11} \gamma \sin \gamma\phi_1) \\ (k_{21} \gamma \cos \gamma\phi_2 + k_{22} \sin \gamma\phi_2) & (k_{22} \cos \gamma\phi_2 - k_{21} \gamma \sin \gamma\phi_2) \end{vmatrix} = 0. \quad (45)$$

Equation (45) may be solved to obtain an explicit expression for γ under various combinations of boundary conditions.

$$\text{If } k_{12} = k_{22} \neq 0, \quad (46)$$

then

$$\gamma = \frac{n\pi}{\phi_2 - \phi_1}; \quad n = 1, 2, 3, \dots \quad (47)$$

$$\text{If } k_{12} = k_{22} = 0, \quad (48)$$

then

$$\gamma = \frac{n\pi}{\phi_2 - \phi_1}; \quad n = 0, 1, 2, \dots \quad (49)$$

$$\text{If } k_{12} \neq k_{22}, \quad (50)$$

then

$$\gamma = \frac{\pi(2n+1)}{\phi_2 - \phi_1}; \quad n = 0, 1, 2, \dots \quad (51)$$

Contrails

The β 's are the real positive roots of

$$\begin{vmatrix} l_{11} J'_Y(\beta r_1) + l_{12} J_Y(\beta r_1) & l_{11} Y'_Y(\beta r_1) + l_{12} Y_Y(\beta r_1) \\ l_{21} J'_Y(\beta r_2) + l_{22} J_Y(\beta r_2) & l_{21} Y'_Y(\beta r_2) + l_{22} Y_Y(\beta r_2) \end{vmatrix} = 0. \quad (52)$$

Equation (52) can be solved only by numerical techniques. The numerical technique based on Newton's method requires the derivative of Equation (52) with respect to the argument β . The derivative can be obtained by re-writing Equation (52) in the form

$$F = \begin{vmatrix} T_1 & T_2 \\ T_3 & T_4 \end{vmatrix}, \quad (53)$$

then

$$\frac{\partial F}{\partial \beta} = \begin{vmatrix} T_1 & T_2 \\ \frac{\partial T_3}{\partial \beta} & \frac{\partial T_4}{\partial \beta} \end{vmatrix} + \begin{vmatrix} \frac{\partial T_1}{\partial \beta} & \frac{\partial T_2}{\partial \beta} \\ T_3 & T_4 \end{vmatrix}, \quad (54)$$

where

$$\frac{\partial T_1}{\partial \beta} = l_{11} J_Y(\beta r_1) \left[\frac{2}{\beta r_1} - \beta r_1 \right] + \frac{r_1}{\beta} l_{12} J'_Y(\beta r_1), \quad (55)$$

$$\frac{\partial T_2}{\partial \beta} = l_{11} Y_Y(\beta r_1) \left[\frac{2}{\beta r_1} - \beta r_1 \right] + \frac{r_1}{\beta} l_{12} Y'_Y(\beta r_1), \quad (56)$$

$$\frac{\partial T_3}{\partial \beta} = l_{21} J_Y(\beta r_2) \left[\frac{2}{\beta r_2} - \beta r_2 \right] + \frac{r_2}{\beta} l_{22} J'_Y(\beta r_2), \text{ and} \quad (57)$$

$$\frac{\partial T_4}{\partial \beta} = l_{21} Y_Y(\beta r_2) \left[\frac{2}{\beta r_2} - \beta r_2 \right] + \frac{r_2}{\beta} l_{22} Y'_Y(\beta r_2). \quad (58)$$

For γ not equal to zero,

$$F_{\gamma} = 1.0, \quad (59)$$

$$B_{\gamma} = \frac{-k_{12} \cos \gamma \phi_1 + k_{11} \gamma \sin \gamma \phi_1}{k_{11} \gamma \cos \gamma \phi_1 + k_{12} \sin \gamma \phi_1}, \quad (60)$$

$$C_{\gamma\beta} = \frac{-l_{11} Y'_{\gamma}(\beta r_1) - l_{12} Y_{\gamma}(\beta r_1)}{l_{11} J'_{\gamma}(\beta r_1) + l_{12} J_{\gamma}(\beta r_1)}, \quad (61)$$

and

$$A_{\gamma\beta} = \frac{\int_{\phi_1}^{\phi_2} \int_{r_1}^{r_2} r [h(r, \phi)] [B_{\gamma} \sin \gamma \phi + F_{\gamma} \cos \gamma \phi] [C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r)] r dr d\phi}{\int_{\phi_1}^{\phi_2} [B_{\gamma} \sin \gamma \phi + F_{\gamma} \cos \gamma \phi]^2 d\phi \int_{r_1}^{r_2} r [C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r)]^2 dr}, \quad (62)$$

where

$$\int_{\phi_1}^{\phi_2} [B_{\gamma} \sin \gamma \phi + F_{\gamma} \cos \gamma \phi]^2 d\phi = \left[\frac{\phi}{2} (B_{\gamma}^2 + 1) - \frac{1}{4\gamma} (B_{\gamma}^2 - 1) \sin 2\gamma \phi + \frac{B_{\gamma}}{\gamma} \sin^2 \gamma \phi \right]_{\phi_1}^{\phi_2} \quad (63)$$

and

$$\int_{r_1}^{r_2} r [C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r)]^2 dr = \left[\frac{1}{2} r^2 \left(1 - \frac{\gamma^2}{\beta^2 r^2} \right) [C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r)]^2 + \frac{r^2}{2\beta^2} [C_{\gamma\beta} J'_{\gamma}(\beta r) + Y'_{\gamma}(\beta r)]^2 \right]_{r_1}^{r_2}. \quad (64)$$

Under certain combinations of boundary conditions, B_{γ} is infinite for particular γ 's. In this case that particular term in Equation (44) is an indeterminate form. A usable form for this term can be obtained by setting

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$$B_{\gamma} = 1.0, \quad (65)$$

$$F_{\gamma} = 0, \quad (66)$$

and

$$A_{\gamma\beta} = \frac{\int_{\phi_1}^{\phi_2} \int_{r_1}^{r_2} r[h(r,\phi)] \sin \gamma\phi \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right] r dr d\phi}{\left[\frac{\phi}{2} - \frac{1}{4\gamma} \sin 2\gamma\phi \right]_{\phi_1}^{\phi_2} \int_{r_1}^{r_2} r \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right]^2 dr}. \quad (67)$$

The resulting form was obtained by division of the numerator and denominator of Equation (44) by B_{γ}^2 and evaluating as B_{γ} approaches infinity.

If γ is equal to zero the required term can be obtained directly from Equation (44) by setting

$$B_{\gamma} = 0, \quad (68)$$

$$F_{\gamma} = 1.0, \quad (69)$$

and

$$A_{\gamma\beta} = \frac{\int_{\phi_1}^{\phi_2} \int_{r_1}^{r_2} r[h(r,\phi)] \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right] r dr d\phi}{(\phi_2 - \phi_1) \int_{r_1}^{r_2} r \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right]^2 dr}. \quad (70)$$

The series in Equation (44) converges rapidly for values of time greater than zero, since the exponential term involving time (θ) and β^2 approaches zero quite rapidly for successive terms in β . However, for time equal to zero, this exponential term is equal to unity for all terms in Equation (44) and does not play its otherwise dominant role in the rapid convergence of the series. For a similar series, Reference 1 suggests

$$\frac{\alpha\theta}{\frac{2}{r_2}} > 0.02 \quad (71)$$

to ensure suitable convergence.

b. Coordinate and Time Varying Surface Effects

To obtain a solution of the coordinate and time varying surface effect in a form similar to that used in Reference 4, wherein the coordinate and time functions are separable, the following assumption is required. For a typical prescribed temperature at a boundary such as in Equation (18), we assume

$$T[f_{\phi 1}(r) g(\theta)] \cong T[f_{\phi 1}(r)] g(\theta) \quad (72)$$

This is indicative of a linear transformation (constant thermal conductivity) over the range of the time function. This does not place any compromise on exactness of the solution for prescribed heat flux, zero surface temperature, or unit time function at a surface. However, Equation (72) is an approximation for a prescribed surface temperature with $g(\theta)$ not equal to unity. (An example of transformations which do not involve this approximation is discussed in Section IIC). The magnitude of the error introduced by such an approximation will be immediately recognizable in the difference between prescribed surface temperature and the resulting solution at that surface. The magnitude of the error will increase with increased deviations in $g(\theta)$ from unity and large variation in thermal conductivity with temperature.

The system of equations to be solved is

$$\frac{1}{\alpha} \frac{\partial T_s}{\partial \theta} = \frac{\partial^2 T_s}{\partial r^2} + \frac{1}{r^2} \frac{\partial^2 T_s}{\partial \phi^2} + \frac{1}{r} \frac{\partial T_s}{\partial r} \quad (73)$$

with the initial condition

$$T_s(0, r, \phi) = 0 \quad (74)$$

and the boundary conditions

$$k_{11} \frac{\partial T_s}{\partial \theta} + k_{12} T_s = \left\{ \frac{k_{11} r}{K_o} f_{\phi 1}(r) + k_{12} T[f_{\phi 1}(r)] \right\} g(\theta), \quad (75)$$

$$k_{21} \frac{\partial T_s}{\partial \phi} + k_{22} T_s = \left\{ \frac{k_{21} r}{K_o} f_{\phi 2}(r) + k_{22} T[f_{\phi 2}(r)] \right\} g(\theta), \quad (76)$$

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$$l_{11} \frac{\partial T_s}{\partial r} + l_{12} T_s = \left\{ \frac{l_{11}}{K_o} f_{r1}(\phi) + l_{12} T[f_{r1}(\phi)] \right\} g(\theta), \quad (77)$$

and

$$l_{21} \frac{\partial T_s}{\partial r} + l_{22} T_s = \left\{ \frac{l_{21}}{K_o} f_{r2}(\phi) + l_{22} T[f_{r2}(\phi)] \right\} g(\theta). \quad (78)$$

As in the previous section, the boundary conditions have been expressed in the general form involving the constants k_{ij} and l_{ij} which can be selected alternately as zero or unity for either (j) to obtain Equations (17) through (24).

By the methods of Reference 4 the solution to Equations (73) through (78) may be written*

$$T_s(\theta, r, \phi) = \sum_{\gamma} \sum_{\beta} G_{\gamma\beta} \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi \right] \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right] \times \left[\alpha\beta^2 \int_0^{\theta} g(\lambda) e^{-[\alpha\beta^2(\theta-\lambda)]} d\lambda \right] \quad (79)$$

where

$$G_{\gamma\beta} = \frac{\int_{\phi_1}^{\phi_2} \int_{r_1}^{r_2} f(r, \phi) \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi \right] \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right] r dr d\phi}{\int_{\phi_1}^{\phi_2} \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi \right]^2 d\phi \int_{r_1}^{r_2} r \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right]^2 dr} \quad (80)$$

and B_{γ} , F_{γ} , $C_{\gamma\beta}$, γ and β have been defined in the previous section. Here, $f(r, \phi)$ is the steady state solution resulting from the coordinate varying boundary condition functions in Equations (75) through (78). This steady state solution is given in detail in Appendix A of this report. Equation (79) is subject to the same conditions regarding B_{γ} of infinity and γ equal to zero as discussed for Equation (44) in the previous section. Under these conditions $G_{\gamma\beta}$, Equation (80), is expressed in forms similar to Equations (67) and (70).

*A more rapidly converging form will be introduced subsequently.

By adding and subtracting $g(\theta)$ in the latter bracket, Equation (79) may be written

$$T_s(\theta, r, \phi) = \sum_{\gamma} \sum_{\beta} G_{\gamma\beta} \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi \right] \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right] \\ \times \left[g(\theta) - g(\theta) + \alpha\beta^2 \int_0^{\theta} g(\lambda) e^{-\alpha\beta^2(\theta-\lambda)} d\lambda \right] \quad (81)$$

and also noting that

$$f(r, \phi) = \sum_{\gamma} \sum_{\beta} G_{\gamma\beta} \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi \right] \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right]. \quad (82)$$

Equation (81) becomes

$$T_s(\theta, r, \phi) = g(\theta) f(r, \phi) - \sum_{\gamma} \sum_{\beta} G_{\gamma\beta} \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi \right] \\ \times \left[C_{\gamma\beta} J_{\gamma}(\beta r) + Y_{\gamma}(\beta r) \right] \left[g(\theta) - \alpha\beta^2 \int_0^{\theta} g(\lambda) e^{-\alpha\beta^2(\theta-\lambda)} d\lambda \right] \quad (83)$$

Equation (83) offers several advantages in practical computation over Equation (79). The form given by Equation (83) is such that the transient series is converging toward the difference between the steady state solution that results from the boundary condition at that time and the actual temperature. As such, this series has convergence characteristics similar to the initial temperature distribution effects solution in Equation (44) of the previous section, in regards to large values of time. In contrast, the series defined by Equation (79) converges rapidly for small times, but rather slowly for large times. In addition, Equation (83) gives the proper solution at the boundaries whereas Equation (79) characteristically converges to zero at a boundary when temperature is prescribed. The form expressed by Equation (83) is used in the digital computer program.

C. COMMENT ON SOLUTION OF THE TEMPERATURE DEPENDENT THERMAL DIFFUSIVITY CASE

A successive transformation, Equation (25), which linearizes the temperature dependent thermal diffusivity equations was presented in Section IIA. It covers transformation of the non-linear differential equation associated with temperature dependent thermophysical properties. Since

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there are two successive transforms involved, particular attention should be given to combinations of the arbitrary functions that define the boundary conditions and the thermophysical properties in the application of this scheme to practical problems. Therefore, it is advisable to define the algebraic form of all the relationships involved and manipulate them as a set through the successive transformations.

As an example, we consider a specific set of the various arbitrary functions of the form

$$K(v) = K_0 (1 + k_1 v), \quad (84)$$

$$\alpha(v) = \alpha_1 v, \quad (85)$$

$$g(\theta) = g_1 \theta, \text{ and} \quad (86)$$

with a prescribed temperature at the boundary ϕ_1 of

$$v(\theta, r, \phi_1) = g(\theta) [f(r)]^2. \quad (87)$$

Then from case IV of Table I

$$T(\theta, r, \phi_1) = \left\{ g_1 [f(r)]^2 \theta + \frac{k_1}{2} g_1^2 [f(r)]^4 \theta^2 \right\} \quad (88)$$

from Equation (25)

$$\psi = \frac{\alpha_1 g_1}{2} [f(r)]^2 \theta^2 \quad (89)$$

$$\theta = \sqrt{\frac{2\psi}{\alpha_1 g_1}} \frac{1}{f(r)} \quad (90)$$

When Equation (90) is substituted in Equation (88), we obtain the transformed boundary condition

$$T(\psi, r, \phi_1) = \left\{ \sqrt{\frac{2\psi g_1}{\alpha_1}} f(r) + \frac{k_1 \psi g_1}{\alpha_1} [f(r)]^2 \right\}. \quad (91)$$

This equation is the complete transformation of the original boundary conditions in terms of both transform temperature and transform time. The two members in Equation (91) are products of a time function and a coordinate function and can be solved by methods similar to the one used in this report. Note that in this development the transform boundary condition is exact. It does not involve an approximation such as in Equation (72)*.

Since the solution to the linear differential equation is in series form, an algebraic expression for the inversion to real time is not generally obtainable and numerical integration must be used. It is, however, under some combinations possible to obtain usable expressions for the inversion process. As an example, consider the relations

$$K(v) = K_0 \quad (92)$$

and

$$\alpha(v) = \frac{1}{a_0 + a_1 v}. \quad (93)$$

Here the transform temperature is

$$T = v \quad (94)$$

and

$$\alpha(T) = \frac{1}{a_0 + a_1 T}. \quad (95)$$

Consider a typical solution in the form

$$T(x, \psi) = \sum_m X(x) e^{-m^2 \psi} \quad (96)$$

where $X(x)$ is the coordinate part of the product solution.

*In place of Equation (88), the approximation defined by Equation (72) would lead to

$$T(\phi, r, \phi_1) = \left\{ g_1 [f(r)]^2 \theta + \frac{k_1}{2} g_1 [f(r)]^4 \theta \right\}.$$

Then from Equation (31), we obtain real time θ in the form

$$\theta = \left| a_0 \psi + a_1 \left[\sum_m \frac{X(x) e^{-m^2 \psi}}{-m^2} \right] \right|_0^\psi . \quad (97)$$

The summation in Equation (97) is in the same form as the solution to Equation (96) except that each term is divided by minus m^2 . In a practical solution these terms could easily be calculated parallel with the calculation of Equation (96).

SECTION III
PROGRAMMING

A. PROGRAM DESCRIPTION

1. Programming Philosophy

The mathematical formulation presented in the previous section is practical only when solved on a high speed computer. The temperature dependent thermal conductivity, constant thermal diffusivity, case has been developed in FORTRAN IV for application to an IBM 7094.

The computer program was developed with the view toward several objectives not usually considered in such a project. In addition to affording an analytical tool for development and study of thermal structures, the items discussed in the following paragraphs were considered.

Those familiar with mathematical texts on heat conduction such as Reference 1 will recognize that the solutions to diffusion problems have common mathematical expressions and methods of solution independent of the coordinate system. For example, the transformation and inversion used in linearizing the equations, the need for root searching techniques, the application of Duhamel's integral to incorporate time varying boundary conditions, etc., are common to many problems other than the specific problem considered in this task. Accordingly, this program was developed with these anticipated common elements as individual subroutines. This provides the user with building blocks or, as a minimum, specific guidelines for the development of similar computer programs. Detailed discussions of the subroutines are given in Appendix C.

Mathematics required to develop a program that considers temperature dependent thermal diffusivity have been discussed in preceding sections. The solution reduces to a linear system similar to that programmed in this task. Under this effort the solution of the linear system was isolated, thus providing the current program with the revision potential to incorporate temperature dependent thermal diffusivity at a later date.

For those users who are interested in the study of the characteristics of series solutions, either academically or in conjunction with development of solutions to similar problems, two control cards were incorporated as

part of the input. These control cards, discussed subsequently, provide variation in the number of terms in the series, as well as increments used in numerical integration, and control a unique array of intermediate print options.

2. Basic Program Flow

The basic digital relationship between the MAIN routine and the sub-routines of the digital program is given in Figure 2. Detailed discussions of these routines are given in Appendix C. Flow diagrams of the MAIN routine and prime subroutines are given in Appendix D.

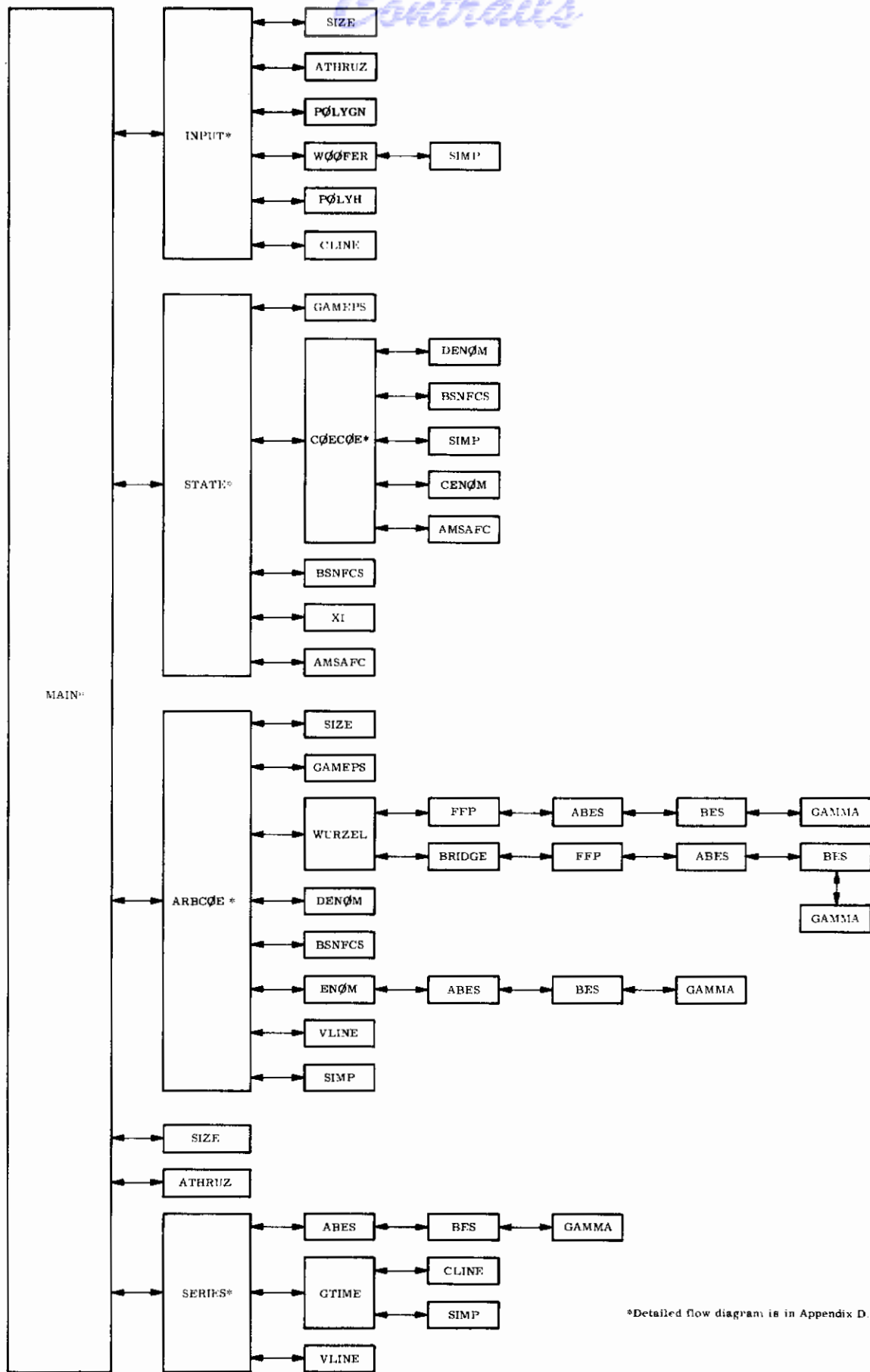
3. Output Data and Format

Certain intermediate print options, as discussed previously, are available for detailed study. The printout for these options is described for the individual subroutines in Appendix C. Independent of the use of these options, the standard program output is:

- 1 The problem title
- 2 All of the input problem description
- 3 The steady state solution
- 4 The times and lattice coordinates where the program is computing temperatures
- 5 The final temperatures at these coordinates.

An illustration of this output is given with the sample problems in Section IV.

In reference to 3 above, the steady state solution printout also includes a printout of the last terms in the series due to boundary condition contribution at each of the faces. The T_1 , T_2 , T_3 , and T_4 refer to the contribution of the boundaries r_1 , r_2 , ϕ_1 , and ϕ_2 , respectively. The IR and IP are program subscripts which locate the solution in the r and ϕ directions, respectively. The latter column is the partial sum of the four series involved. With the exception of corrections at the corners in cases where temperatures are prescribed at the corresponding adjacent faces (Appendix A) this is the steady state solution $f(r,\phi)$ used in evaluating Equation (80).



*Detailed flow diagram is in Appendix D.

Figure 2. Program Subroutine Relations

B. UTILIZATION OF THE COMPUTER PROGRAM

Before reading this section a review of Equations (1) through (10) and Figure 1 in Section IIA is recommended.

This section describes the utilization of the computer program via a general discussion of the input requirements followed by a detailed description of the input and output formats on Table II. Two sample problems are included in Section IV to illustrate utilization of the program.

1. Discussion of Input

It is assumed in the calculation that the physical units input for a particular problem are dimensionally compatible. That is, units of heat length, time, mass, and temperature must be consistent in a given problem. Heat flux at a surface is treated as per unit area.

a. Control (Cards A and B)

A control system, as previously mentioned, was developed to increase the flexibility of the program. Input the numbers given in the input data format, which follows for normal use of the program. See sequence line numbers 000180 through 000280 of the program listing (Appendix B) and detailed discussion of MAIN (Appendix C) for definitions if alternation to number of terms used in the series or number of increments used in numerical integration is desired.

b. Geometry (Cards D and E)

As presently programmed, the system will consider sectors between 0 and 360°, where

$$\phi_1 = 0$$

$$0 < \phi_2 < 360^\circ.$$

The radial boundary restrictions are

$$r_2 > r_1$$

$$r_1 \neq 0$$

c. Thermophysical Properties (Card(s) F)

The variation of thermal conductivity with temperature may be defined by selecting any of the programmed algebraic expressions given in Table I, inputting tabular data, or defining the coefficients and exponents in an expression of the form:

$$K(v) = \sum_{i=1}^n a_i v^{b_i} \quad (35)$$

The latter expression is referred to as PØLY in the utilization instructions. The thermal diffusivity is input as a constant. Refer to Section IIB.1 (page 9) for a discussion of the associated transformation and inversion.

d. Initial Temperature Distribution (Card(s) G)

The initial temperature distribution may be input as either a lattice array of temperatures or in polynomial form similar to Equation (35). The polynomial form is:

$$h(r,\phi) = \sum_{i=1}^n a_i r^{b_i} \phi^{c_i} \quad (98)$$

e. Boundary Conditions (Card(s) H, I, J, K, and optional L)

The coordinate varying portion of the boundary conditions is incorporated in the solution via a steady state solution (see Section IIB.2.b, page 17). This steady state solution will be calculated in response to definition of the coordinate varying functions and mode (prescribed temperature or heat flux) at each surface. The solution is written such that a positive heat flux represents heat into the surface. The coordinate varying portion of a boundary condition may be input either as tabular data or by defining number of terms, coefficients, and exponents in a form similar to Equation (35).

In particular cases, the steady state solution may be known by the user either from a previous problem or because of the simplicity of the particular problem. In such cases, the steady state temperature distribution in either the real or the transform variable may be input either as a lattice array of temperatures or in polynomial form similar to the initial temperature distribution. The boundary condition mode is still required when using this option.

f. Time Function (Card(s) M)

The time function $g(\theta)$ may be input as either tabulated or polynomial data. For a discussion of the approximation associated with application of this function refer to Section IIB.2.b, page 17.

g. Output Requested (Card(s) N)

Temperature distribution data at various times may be requested either at specific values of r and ϕ or in an evenly distributed lattice. It is advisable to request calculations for times in which

$$\frac{\alpha\theta}{r_2^2} > .02. \quad (71)$$

For smaller times an increased number of terms in the series may be required.

2. Input Data Format

All input data must contain a decimal point unless otherwise specified. If data is specified as not containing a decimal point, it must be right adjusted in its field.

When point data is furnished, the independent variable should always be supplied in monotonically increasing order. Table II presents the input data loading procedure.

TABLE II

Input Data Loading Procedure

(Note: Quotation marks indicate Hollerith terms.)

Instruction	Card Designation	Column No.	Description
Control	A	5	6
		10	6
		13-15	100
		18-20	100
		25	6
		29-30	10
		35	20
		39-40	30
		44-45	30
		B	1-35
C	1-72	Problem title	
Geometry	D	1-10	ARE(1) - the value of r_1
		11-20	ARE(2) - the value of r_2
	E	1-10	PHI(1) - the value of ϕ_1 which must be zero
		11-20	PHI(2) - the value of ϕ_2 in degrees and fractions
Thermophysical Properties <ul style="list-style-type: none"> • To input point data for $K(v)$ versus v <ul style="list-style-type: none"> • To input $K(v)$ as a polynomial 	F	1-5	"PØINT"
		11-20	N - the number of point pairs to be input. This number is right adjusted with no decimal point ($N \leq 50$).
		21-30	ALPHØ - the value of α
		31-40	AZERØ - the value of K_0
		F1	1-60
	F	1-4	"PØLY"
		11-20	N - the number of polynomial terms to be input. This number is right adjusted with no decimal point ($N \leq 10$).
		21-30	A - the minimum value at which the polynomial is to be evaluated
		31-40	B - the maximum value at which the polynomial is to be evaluated

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Instruction	Card Designation	Column No.	Description
<ul style="list-style-type: none"> To input special cases of $K(v)$ (see Table I). 	F1	41-50	ALPHØ - the value of α
		51-60	AZERØ - the value of K_0
	F	1-10	A coefficient of the polynomial
		11-20	The corresponding order of the independent variable. This card (F1) is repeated as necessary.
		1-4	"CASE"
		11-20	N - the case number of the transform. This value is right adjusted with no decimal point ($N \leq 7$).
		21-30	BEE - the value of b in the transform equation
		31-40	SEE - the value of c in the transform equation
		41-50	ALPHØ - the value of α
		51-60	AZERØ - the value of K_0
Initial Temperature Distribution			
<ul style="list-style-type: none"> To input $h(r, \phi)$ as point data 	G	1-5	"PØINT"
		11-20	N - the number of points in the r direction. This value is right adjusted with no decimal point ($N \leq 7$).
<ul style="list-style-type: none"> To input $h(r, \phi)$ in polynomial form 	G1	21-30	M - the number of points in the ϕ direction. This value is right adjusted with no decimal point ($M \leq 25$).
		1-70	The values of the dependent variable in ten column fields starting with $h(r_1, \phi_2)$ and varying (r) most rapidly, with all the values for one complete cycle (r) on one card. The values of dependent variable must be evenly spaced in any given direction. This card (G1) is repeated as necessary.
	G	1-4	"PØLY"
		11-20	N - the number of terms in the polynomial. This value is right adjusted with no decimal point ($N \leq 10$).

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Instruction	Card Designation	Column No.	Description
<p>2 <u>Boundary conditions at r_2, ϕ_1, and ϕ_2</u></p> <p>3 <u>Steady state solution input directly*</u></p>	H1	1-10 11-20	A coefficient of the polynomial The corresponding order of the polynomial. This card (H1) is repeated as necessary.
	I, J, K		Cards I, J, and K follow the same form as cards H (and H1) but for the r_2 , ϕ_1 , and ϕ_2 boundaries, respectively. The options of TEMP, FLUX, POINT, and POLY available on card H may be used in any combination at the remaining boundaries except that the option to input or solve for the steady state solution has been fixed with card H and cannot be modified thereafter.
	L, L1		The steady state solution card L and card(s) L1 are input in exactly the same format as the initial temperature distribution (card G and card(s) G1 above) with the same restrictions and options but with the following addition. If the steady state solution is input in the real variable (v), write "TRANSFORM" in columns 31 through 39 of card L. (The program will perform the required transformations before proceeding with the remainder of the solution.) If the steady state solution is input in the transform variable (T) leave columns 31 through 39 blank.
<p>Time Functions</p> <ul style="list-style-type: none"> • To input the time function $g(\theta)$ as point data 	M	1-5 11-20	"POINT" N - the number of point pairs to be input. This value is right adjusted with no decimal point ($N \leq 50$).
	M1	1-60	The point pairs in ten column fields with the time value (θ) given first in each case. This card (M1) is repeated as necessary.

*All L cards are omitted if steady state solution is not input.

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SECTION IV
SAMPLE PROBLEMS

This section presents two typical samples and their solutions as obtained from the digital computer program. These sample problems are primarily to familiarize the reader with input and output format of the program. A sample test problem is discussed in Appendix E.

A. SAMPLE PROBLEM 1

1. Geometry

$$r_1 = 0.7 \text{ inch}$$

$$r_2 = 0.9 \text{ inch}$$

$$\phi_1 = 0 \text{ degrees}$$

$$\phi_2 = 270 \text{ degrees}$$

2. Thermophysical Properties

Based on glass reinforced polyester*

$$K(v) = K_0 [1 + bv] \quad \text{Case 4}$$

$$K_0 = 0.925 \times 10^6 \text{ Btu/sec inch } ^\circ\text{R}$$

$$b = 0.00125$$

$$\alpha = 0.000122 \text{ inch}^2/\text{sec}$$

3. Initial Temperature Distribution

$$h(r, \phi) = 600^\circ\text{R}$$

4. Boundary Conditions

$$\text{at } r_1 \quad v = 1000^\circ\text{R.}$$

$$\text{at } r_2 \quad v = 1000^\circ\text{R.}$$

$$\text{at } \phi_1 \quad v = 1000^\circ\text{R.}$$

and

$$\text{at } \phi_2 \quad v = 1000^\circ\text{R.}$$

* Data per Reference 5.

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In this case the steady state solution can be input directly since it is obviously 1000°R at all points. It can be input in either the real or the transform variable. In this case it will be input in the transform variable. Referring to Table I.

if

$$K(v) = K_0 [1 + bv]$$

then

$$T = v + \frac{bv^2}{2}$$

which for $v = 1000^{\circ}\text{R}$ yields

$$T = 1625^{\circ}\text{R} = f(r, \phi).$$

Since the boundary conditions define a step function, $g(\theta)$ is equal to unity.

The input to obtain the temperature distribution at 40, 60, 80, and 120 seconds for five values in each of the radial and circumferential directions is shown in Table III. Printout is shown in Table IV. Figures 3 through 7 are illustrations of the temperature distribution.

B. SAMPLE PROBLEM 2

1 Geometry

$$r_1 = 0.2 \text{ inch}$$

$$r_2 = 0.8 \text{ inch}$$

$$\phi_1 = 0 \text{ degree}$$

$$\phi_2 = 90 \text{ degrees}$$

2 Thermophysical properties

Same as sample problem 1.

3 Boundary Conditions

$$\text{at } r_1 \quad v = (720 - 25.4\phi)g(\theta),$$

$$\text{at } r_2 \quad \frac{\partial v}{\partial r} = 0,$$

$$\text{at } \phi_1 \quad v = (700 + 100r) g(\theta),$$

$$\text{at } \phi_2 \quad v = (700 - 100r) g(\theta);$$

where

$$g(\theta) = 0.8 + 0.00333\theta.$$

4. Initial Temperature Distribution

$$h(r, \phi) = 600^\circ\text{R}.$$

The input to obtain the temperature distribution at 40, 60, 80, 120 seconds for five points in the radial direction and five points in the circumferential direction is shown in Table V. Printout is shown in Table VI. Figures 8 through 12 illustrate the temperature distribution.

Contrails

TABLE IV

Program Printout, Sample Problem Number 1

SAMPLE PROBLEM 1

INPUT

GEOMETRY

R1= 0.70000000E 00 R2= 0.90000000E 00
P1= 0. P2= 0.47123883E 01 (RADIAN)

THERMO PHYSICAL PROPERTIES

TRANSFORM ACCORDING TO CASE 4, B= 0.12500000E-02 C= 0.
ALPHA= 0.12200000E-03 KZERO= 0.92500000E-06

INITIAL TEMPERATURE DISTRIBUTION

POLYNOMIAL DATA, 1 TERMS

(0.60000000E 03)(R** 0.)(P** 0.)

BOUNDARY CONDITIONS

AT R1, TEMP L11= 0. L12= 1.00
AT R2, TEMP L21= 0. L22= 1.00
AT P1, TEMP K11= 0. K12= 1.00
AT P2, TEMP K21= 0. K22= 1.00

STEADY STATE SOLUTION IS BEING INPUT

POLYNOMIAL DATA, 1 TERMS

(0.16250000E 04)(R** 0.)(P** 0.)

TIME FUNCTION

POLYNOMIAL DATA, 1 TERMS, X=DUMMY VARIABLE

(0.10000000E 01)(X** 0.)

EVALUATE POLYNOMIAL FROM 0. TO 0.12000000E 03

5 LATTICE POINT ANSWERS IN -R-
0.70000000E 00 0.75000000E 00 0.80000000E 00 0.84999999E 00 0.90000000E 00

5 LATTICE POINT ANSWERS IN -PHI-
0. 0.11780971E 01 0.23561941E 01 0.35342912E 01 0.47123883E 01

TIME= 0.40000000E 02 R= 0.70000000E 00 PHI= 0. (RADIAN)
TEMPERATURE= 0.99999999E 03

TIME= 0.40000000E 02 R= 0.75000000E 00 PHI= 0. (RADIAN)
TEMPERATURE= 0.99999999E 03

TIME= 0.40000000E 02 R= 0.80000000E 00 PHI= 0. (RADIAN)
TEMPERATURE= 0.99999999E 03

TIME= 0.40000000E 02 R= 0.84999999E 00 PHI= 0. (RADIAN)
TEMPERATURE= 0.99999999E 03

TIME= 0.40000000E 02 R= 0.90000000E 00 PHI= 0. (RADIAN)
TEMPERATURE= 0.99999999E 03

TIME= 0.40000000E 02 R= 0.70000000E 00 PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.99999999E 03

TIME= 0.40000000E 02 R= 0.75000000E 00 PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.89553053E 03

Contrails

TIME= 0.4000000E 02	R= 0.8000000E 00	PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.85510616E 03		
TIME= 0.4000000E 02	R= 0.8499999E 00	PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.90183670E 03		
TIME= 0.4000000E 02	R= 0.9000000E 00	PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.4000000E 02	R= 0.7000000E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.4000000E 02	R= 0.7500000E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.88835024E 03		
TIME= 0.4000000E 02	R= 0.8000000E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.84493773E 03		
TIME= 0.4000000E 02	R= 0.8499999E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.89500820E 03		
TIME= 0.4000000E 02	R= 0.9000000E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.4000000E 02	R= 0.7000000E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.4000000E 02	R= 0.7500000E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.89539008E 03		
TIME= 0.4000000E 02	R= 0.8000000E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.85490806E 03		
TIME= 0.4000000E 02	R= 0.8499999E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.90170380E 03		
TIME= 0.4000000E 02	R= 0.9000000E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.4000000E 02	R= 0.7000000E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.4000000E 02	R= 0.7500000E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.4000000E 02	R= 0.8000000E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.4000000E 02	R= 0.8499999E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.4000000E 02	R= 0.9000000E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.7000000E 00	PHI= 0. (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.7500000E 00	PHI= 0. (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.8000000E 00	PHI= 0. (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.8499999E 00	PHI= 0. (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.9000000E 00	PHI= 0. (RADIAN)
TEMPERATURE= 0.9999999E 03		

Contrails

TIME= 0.6000000E 02	R= 0.7000000E 00	PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.7500000E 00	PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.94338583E 03		
TIME= 0.6000000E 02	R= 0.8000000E 00	PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.92201011E 03		
TIME= 0.6000000E 02	R= 0.8499999E 00	PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.94686591E 03		
TIME= 0.6000000E 02	R= 0.9000000E 00	PHI= 0.11780971E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.7000000E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.7500000E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.93992796E 03		
TIME= 0.6000000E 02	R= 0.8000000E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.91717767E 03		
TIME= 0.6000000E 02	R= 0.8499999E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.94357659E 03		
TIME= 0.6000000E 02	R= 0.9000000E 00	PHI= 0.23561941E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.7000000E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.7500000E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.94331553E 03		
TIME= 0.6000000E 02	R= 0.8000000E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.92191216E 03		
TIME= 0.6000000E 02	R= 0.8499999E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.94679937E 03		
TIME= 0.6000000E 02	R= 0.9000000E 00	PHI= 0.35342912E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.7000000E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.6000000E 02	R= 0.7500000E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999997E 03		
TIME= 0.6000000E 02	R= 0.8000000E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999995E 03		
TIME= 0.6000000E 02	R= 0.8499999E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999997E 03		
TIME= 0.6000000E 02	R= 0.9000000E 00	PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.8000000E 02	R= 0.7000000E 00	PHI= 0. (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.8000000E 02	R= 0.7500000E 00	PHI= 0. (RADIAN)
TEMPERATURE= 0.9999999E 03		
TIME= 0.8000000E 02	R= 0.8000000E 00	PHI= 0. (RADIAN)
TEMPERATURE= 0.9999999E 03		

Contrails

TIME= 0.8000000E 02	R= 0.84999999E 00	PHI= 0.	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.8000000E 02	R= 0.9000000E 00	PHI= 0.	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.8000000E 02	R= 0.7000000E 00	PHI= 0.11783971E 01	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.8000000E 02	R= 0.7500000E 00	PHI= 0.11783971E 01	(RADIANS)
TEMPERATURE= 0.96917486E 03			
TIME= 0.8000000E 02	R= 0.8000000E 00	PHI= 0.11783971E 01	(RADIANS)
TEMPERATURE= 0.95762274E 03			
TIME= 0.8000000E 02	R= 0.84999999E 00	PHI= 0.11783971E 01	(RADIANS)
TEMPERATURE= 0.97106580E 03			
TIME= 0.8000000E 02	R= 0.9000000E 00	PHI= 0.11783971E 01	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.8000000E 02	R= 0.70000000E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.8000000E 02	R= 0.7500000E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.96749291E 03			
TIME= 0.8000000E 02	R= 0.8000000E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.95528220E 03			
TIME= 0.8000000E 02	R= 0.84999999E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.96946320E 03			
TIME= 0.8000000E 02	R= 0.90000000E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.8000000E 02	R= 0.7000000E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.8000000E 02	R= 0.7500000E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.96913924E 03			
TIME= 0.8000000E 02	R= 0.8000000E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.95757343E 03			
TIME= 0.8000000E 02	R= 0.84999999E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.97103207E 03			
TIME= 0.8000000E 02	R= 0.9000000E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.8000000E 02	R= 0.7000000E 00	PHI= 0.47123883E 01	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.8000000E 02	R= 0.7500000E 00	PHI= 0.47123883E 01	(RADIANS)
TEMPERATURE= 0.9999997E 03			
TIME= 0.8000000E 02	R= 0.8000000E 00	PHI= 0.47123883E 01	(RADIANS)
TEMPERATURE= 0.9999997E 03			
TIME= 0.8000000E 02	R= 0.84999999E 00	PHI= 0.47123883E 01	(RADIANS)
TEMPERATURE= 0.9999997E 03			
TIME= 0.8000000E 02	R= 0.9000000E 00	PHI= 0.47123883E 01	(RADIANS)
TEMPERATURE= 0.9999999E 03			
TIME= 0.1200000E 03	R= 0.7000000E 00	PHI= 0.	(RADIANS)
TEMPERATURE= 0.9999999E 03			

Contrails

TIME= 0.12000000E 03	R= 0.75000000E 00	PHI= 0.	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.80000000E 00	PHI= 0.	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.84999999E 00	PHI= 0.	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.90000000E 00	PHI= 0.	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.70000000E 00	PHI= 0.11780971E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.75000000E 00	PHI= 0.11780971E 01	(RADIANS)
TEMPERATURE= 0.99079582E 03			
TIME= 0.12000000E 03	R= 0.80000000E 00	PHI= 0.11780971E 01	(RADIANS)
TEMPERATURE= 0.98734950E 03			
TIME= 0.12000000E 03	R= 0.84999999E 00	PHI= 0.11780971E 01	(RADIANS)
TEMPERATURE= 0.99135558E 03			
TIME= 0.12000000E 03	R= 0.90000000E 00	PHI= 0.11780971E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.70000000E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.75000000E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.99039390E 03			
TIME= 0.12000000E 03	R= 0.80000000E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.98678927E 03			
TIME= 0.12000000E 03	R= 0.84999999E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.99097077E 03			
TIME= 0.12000000E 03	R= 0.90000000E 00	PHI= 0.23561941E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.70000000E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.75000000E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.99078654E 03			
TIME= 0.12000000E 03	R= 0.80000000E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.98733665E 03			
TIME= 0.12000000E 03	R= 0.84999999E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.99134679E 03			
TIME= 0.12000000E 03	R= 0.90000000E 00	PHI= 0.35342912E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.70000000E 00	PHI= 0.47123883E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.75000000E 00	PHI= 0.47123883E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.80000000E 00	PHI= 0.47123883E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			
TIME= 0.12000000E 03	R= 0.84999999E 00	PHI= 0.47123883E 01	(RADIANS)
TEMPERATURE= 0.99999999E 03			

Contrails

TIME= 0.12000000E 03 R= 0.90000000E 00 PHI= 0.47123883E 01 (RADIAN)
TEMPERATURE= 0.99999999E 03

END-OF-DATA ENCOUNTERED ON SYSTEM INPUT FILE.

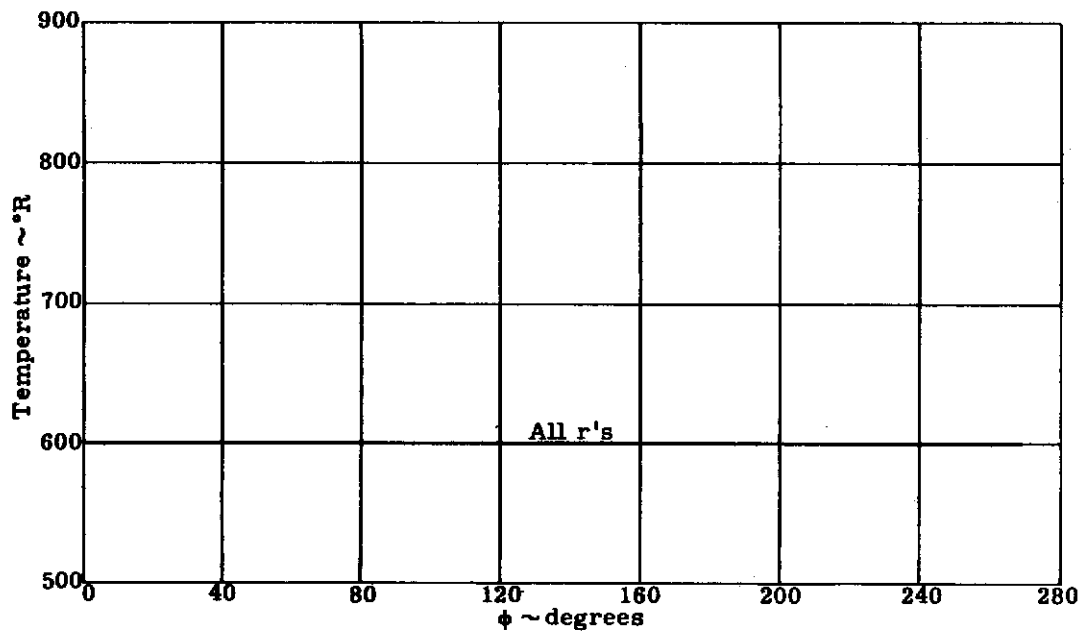


Figure 3. Temperature Distribution, Sample Problem No. 1,
 $\theta = 0$ Second

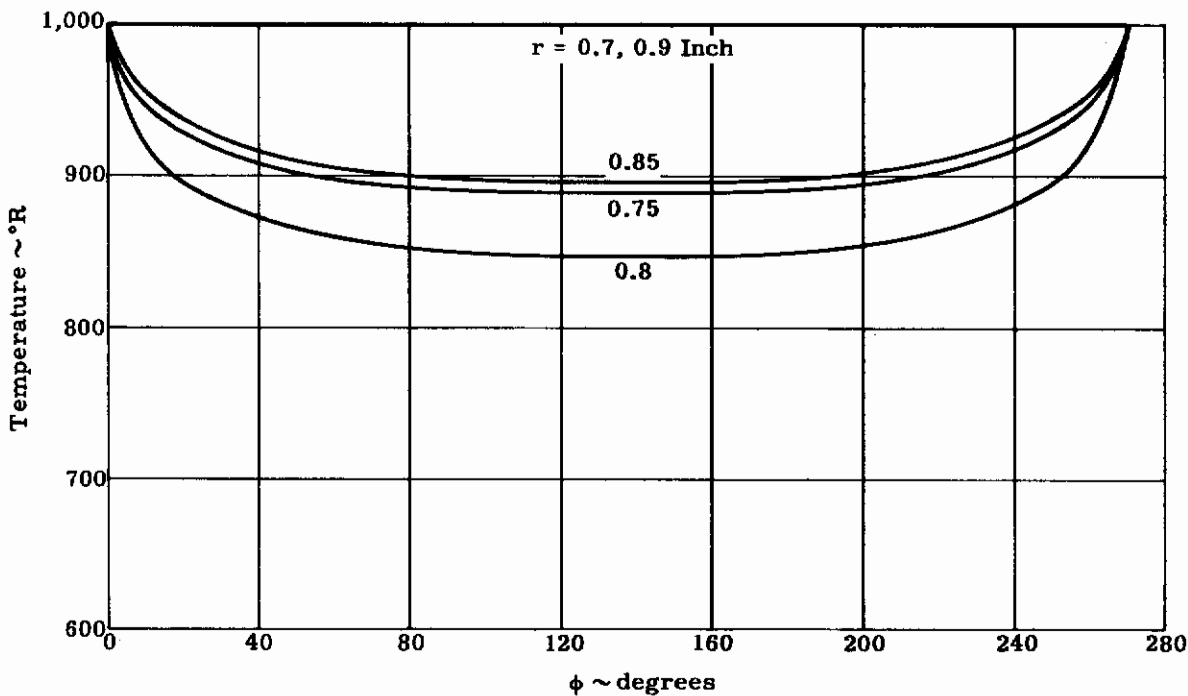


Figure 4. Temperature Distribution, Sample Problem No. 1, $\theta = 40$ Seconds

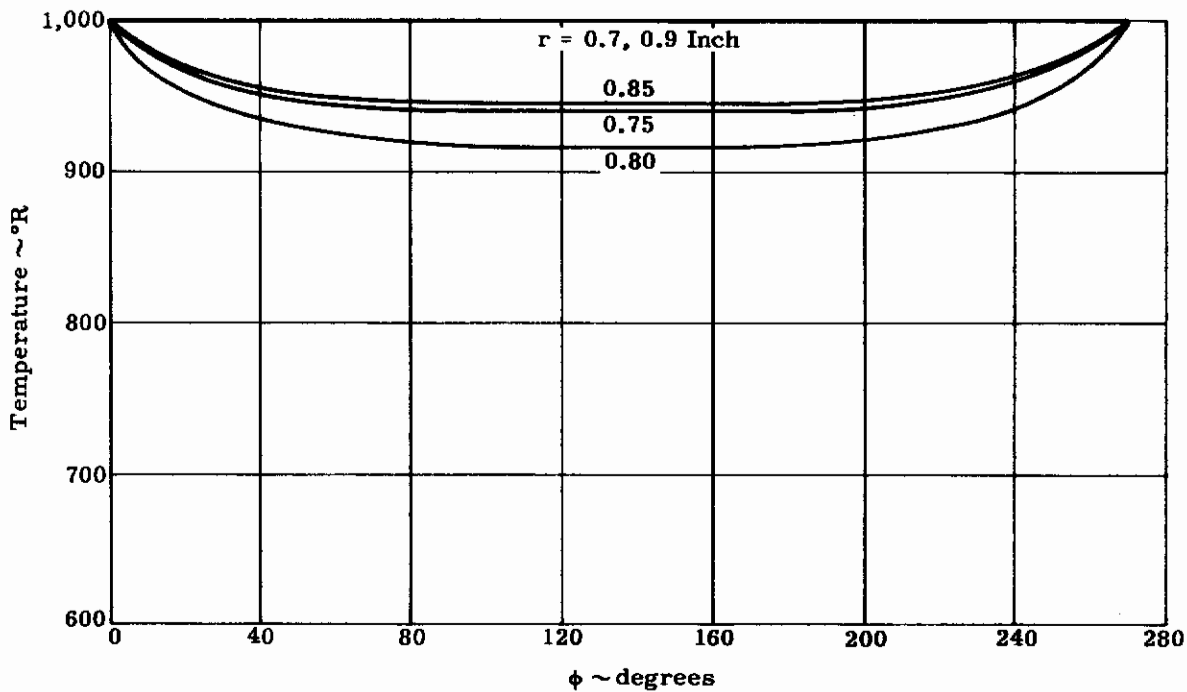


Figure 5. Temperature Distribution, Sample Problem No. 1, $\theta = 60$ Seconds

Contrails

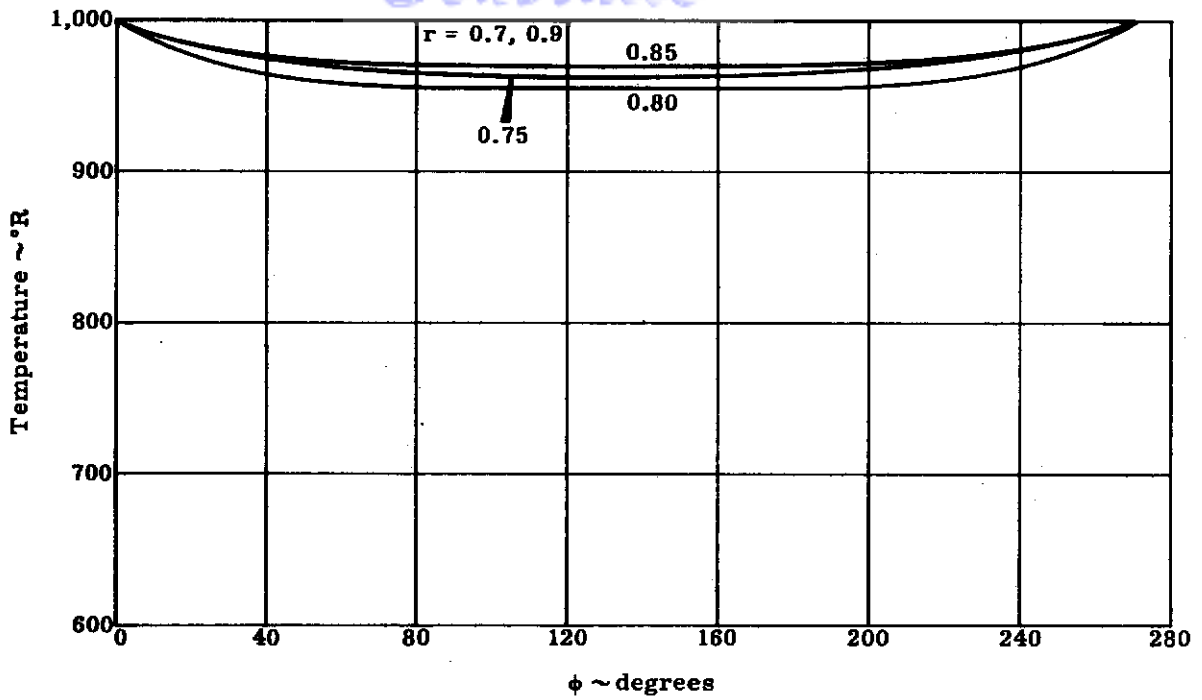


Figure 6. Temperature Distribution, Sample Problem No. 1,
 $\theta = 80$ Seconds

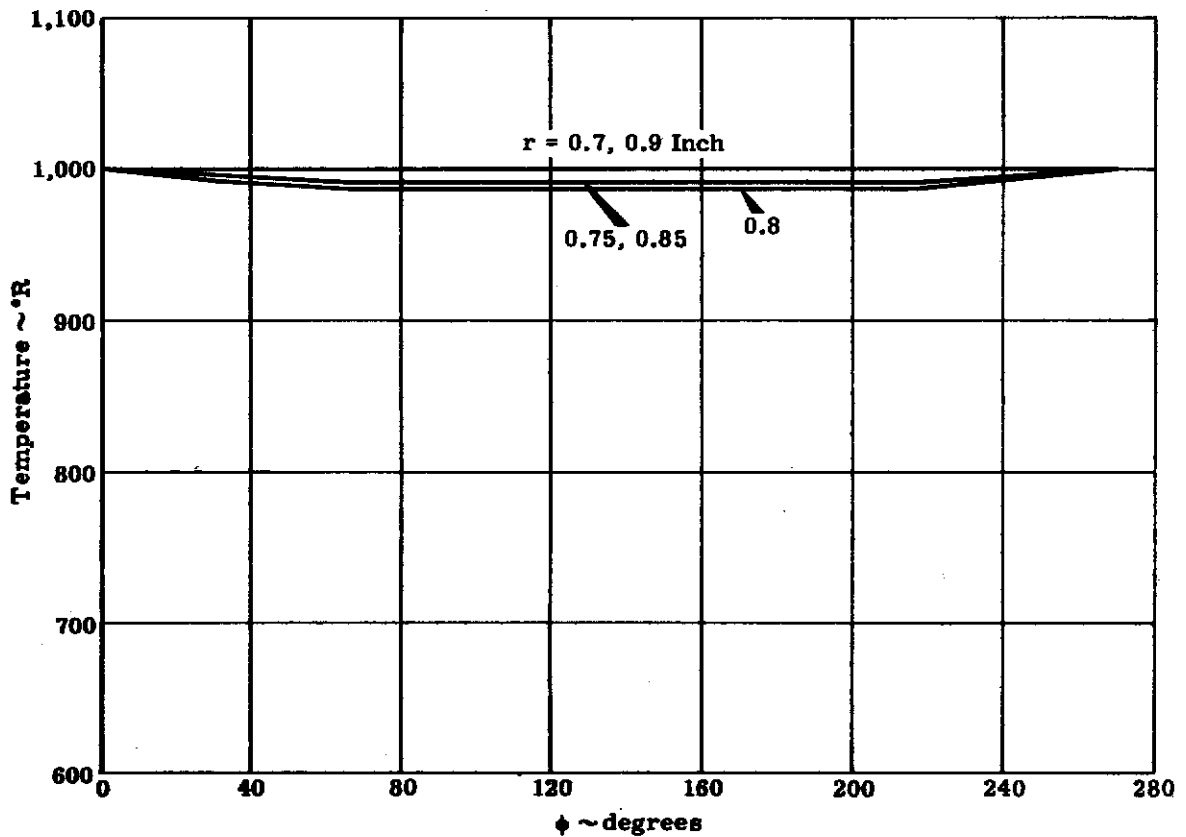


Figure 7. Temperature Distribution, Sample Problem No. 1,
 $\theta = 120$ Seconds

TABLE VI

Program Printout, Sample Problem Number 2

SAMPLE PROBLEM 2

INPUT

GEOMETRY

R1= 0.20000000E-00 R2= 0.80000000E 00
P1= 0. P2= 0.15707961E 01 (RADIANS)

THERMO PHYSICAL PROPERTIES

TRANSFORM ACCORDING TO CASE 4, B= 0.12500000E-02 C= 0.
ALPHA= 0.12200000E-03 KZERO= 0.92500000E-06

INITIAL TEMPERATURE DISTRIBUTION

POLYNOMIAL DATA, 1 TERMS

(0.60000000E 03)(R** 0.) (P** 0.)

BOUNDARY CONDITIONS

AT R1, TEMP L11= 0. L12= 1.00

POLYNOMIAL DATA, 2 TERMS, X=DUMMY VARIABLE
(0.72000000E 03)(X** 0.)
(-0.25400000E 02)(X** 1.0000)

AT R2, FLUX L21= 1.00 L22= 0.

POLYNOMIAL DATA, 1 TERMS, X=DUMMY VARIABLE
(0.) (X** 0.)

AT P1, TEMP K11= 0. K12= 1.00

POLYNOMIAL DATA, 2 TERMS, X=DUMMY VARIABLE
(0.70000000E 03)(X** 0.)
(0.10000000E 03)(X** 1.0000)

AT P2, TEMP K21= 0. K22= 1.00

POLYNOMIAL DATA, 2 TERMS, X=DUMMY VARIABLE
(0.70000000E 03)(X** 0.)
(-0.10000000E 03)(X** 1.0000)

TIME FUNCTION

POLYNOMIAL DATA, 2 TERMS, X=DUMMY VARIABLE
(0.80000000E 00)(X** 0.)
(0.33300000E-02)(X** 1.0000)

EVALUATE POLYNOMIAL FROM 0. TO 0.12000000E 03

Table with 7 columns: T1, T2, T3, T4, STEADY (R, P), IR, TP. It contains numerical data for various time points and parameters.

Contrails

-0.17436476E 02 -0.	0.77290743E-24	0.44223133E-28	0.12074655E 04	1	5
-0.10850725E-08 -0.	0.34593667E-15	0.19793319E-19	0.10141319E 04	2	5
-0.13192373E-15 -0.	-0.86240039E-16	-0.49343616E-20	0.10168617E 04	3	5
-0.76481266E-21 -0.	0.24399023E-15	0.13960291E-19	0.10188134E 04	4	5
-0.46506145E-25 -0.	-0.38802511E-15	-0.22201477E-19	0.10199094E 04	5	5
-0.27509693E-28 -0.	0.42154727E-15	0.24119501E-19	0.10202305E 04	6	5
-0.17436610E 02 -0.	0.47827567E-28	0.71465873E-24	0.99694684E 03	1	6
-0.10850809E-08 -0.	0.21406585E-19	0.31986580E-15	0.99993795E 03	2	6
-0.13192475E-15 -0.	-0.53365398E-20	-0.79740722E-15	0.99810152E 03	3	6
-0.76481858E-21 -0.	0.15098133E-19	0.22560237E-15	0.99673614E 03	4	6
-0.46506504E-25 -0.	-0.24011022E-19	-0.35878234E-15	0.99595206E 03	5	6
-0.27509905E-28 -0.	0.26085375E-19	0.38977817E-15	0.99572010E 03	6	6
0.44150918E 02 0.	0.29595746E-32	0.11549085E-19	0.10081296E 04	1	7
0.27475132E-08 0.	0.13246417E-23	0.51691210E-11	0.98583507E 03	2	7
0.33404420E-15 0.	-0.33022562E-24	-0.12886324E-11	0.97933614E 03	3	7
0.19365829E-20 0.	0.93427398E-24	0.36457975E-11	0.97462137E 03	4	7
0.11775826E-24 0.	-0.14858044E-23	-0.57980231E-11	0.97151839E 03	5	7
0.69657322E-28 0.	0.16141657E-23	0.62989244E-11	0.97064191E 03	6	7
-0.50206479E 02 -0.	0.18313869E-36	0.18663668E-15	0.95803342E 03	1	8
-0.31243510E-08 -0.	0.81968920E-28	0.83534543E-07	0.97204189E 03	2	8
-0.37986036E-15 -0.	-0.20434385E-28	-0.23824686E-07	0.96059722E 03	3	8
-0.22021968E-20 -0.	0.57812939E-28	0.58917179E-07	0.95132963E 03	4	8
-0.13390950E-24 -0.	-0.91941684E-28	-0.93697789E-07	0.94533459E 03	5	8
-0.79211238E-28 -0.	0.99884685E-28	0.13179250E-06	0.94339016E 03	6	8
0.32769869E 02 0.	0.	0.33161342E-11	0.10383348E 04	1	9
0.20392702E-08 0.	0.50722378E-32	0.13499431E-02	0.95889435E 03	2	9
0.24793562E-15 0.	-0.12644800E-32	-0.33653312E-03	0.94220467E 03	3	9
0.14373783E-20 0.	0.35774654E-32	0.95211915E-03	0.92726015E 03	4	9
0.87403001E-25 0.	-0.56893526E-32	-0.15141842E-02	0.91570383E 03	5	9
0.51701334E-28 0.	0.61808656E-32	0.16449972E-02	0.91105254E 03	6	9
0.11926919E-03 0.	-0.	0.48741141E-07	0.18571208E-02	1	10
0.74221262E-14 0.	-0.	0.21815482E 02	0.94421585E 03	2	10
0.90238628E-21 0.	0.	-0.54384753E 01	0.91165409E 03	3	10
0.52314809E-26 0.	-0.	0.15386529E 02	0.90590479E 03	4	10
0.31811190E-30 0.	0.	-0.74469666E 02	0.86845119E 03	5	10
0.18817214E-33 0.	-0.	0.26581643E 02	0.87382974E 03	6	10

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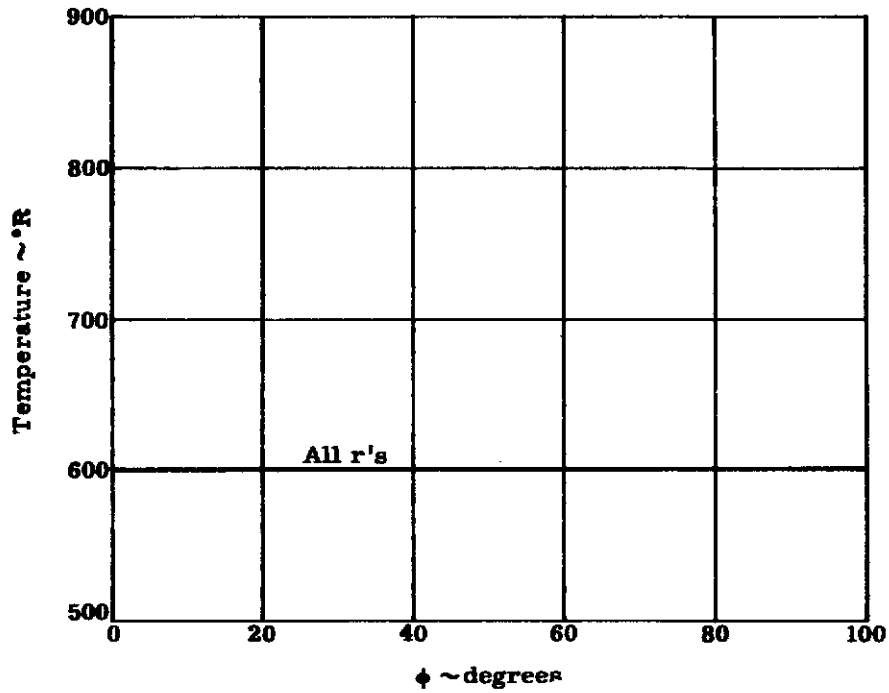


Figure 8. Temperature Distribution, Sample Problem No. 2, $\theta = 0$ Second

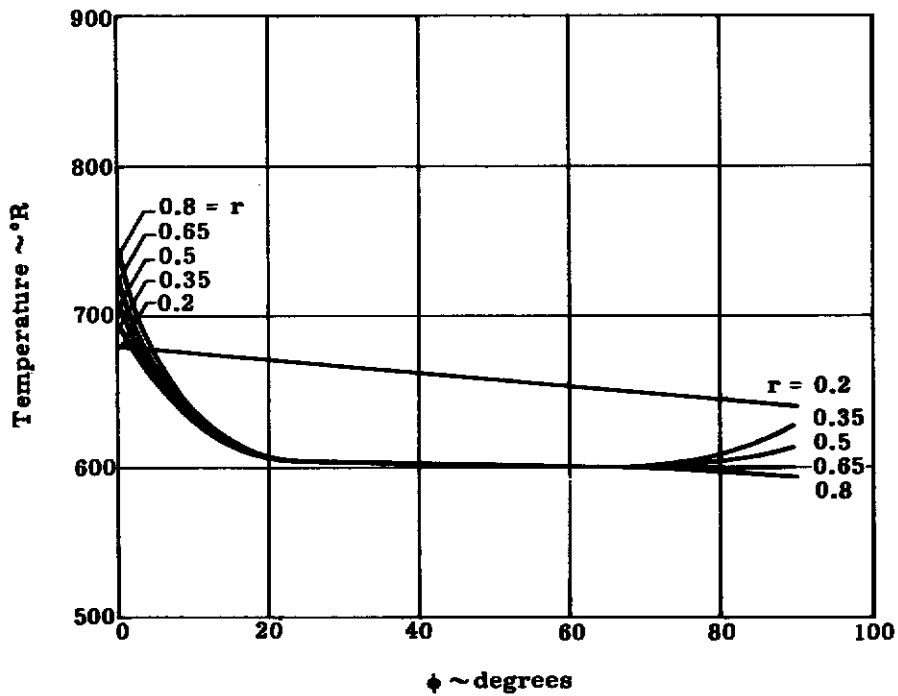


Figure 9. Temperature Distribution, Sample Problem No. 2, $\theta = 40$ Seconds

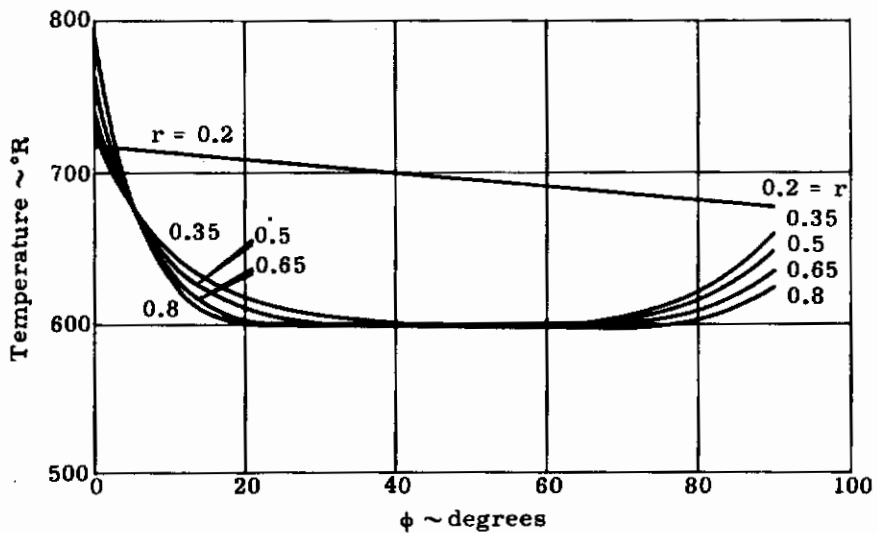


Figure 10. Temperature Distribution, Sample Problem No. 2,
 $\theta = 60$ Seconds

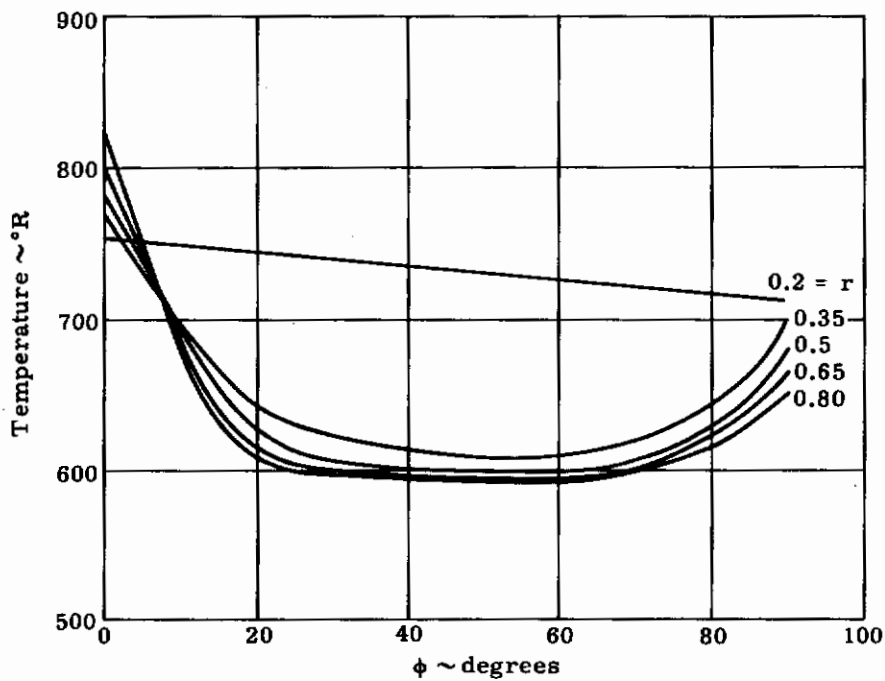


Figure 11. Temperature Distribution, Sample Problem No. 2,
 $\theta = 80$ Seconds

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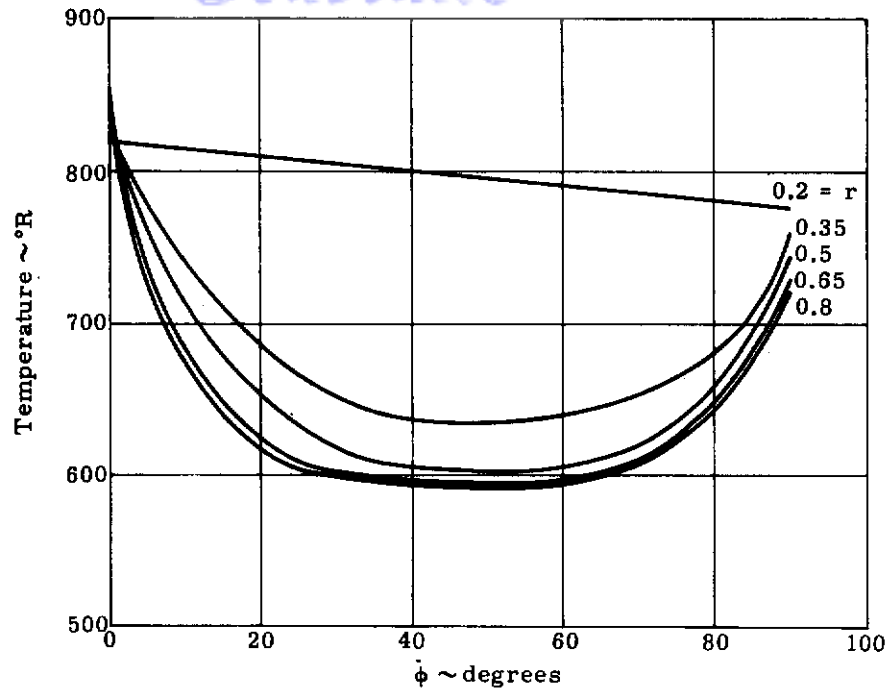


Figure 12. Temperature Distribution, Sample Problem No. 2,
 $\theta = 120$ Seconds

APPENDIX A
STEADY STATE SOLUTION

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APPENDIX A

STEADY STATE SOLUTION

The steady state solution resulting from the coordinate varying boundary condition functions in Equations (75) through (78) is required to obtain $f(r, \phi)$ as discussed in Section II. This steady state solution can be written

$$f(r, \phi) = T_{r_1}(r, \phi) + T_{r_2}(r, \phi) + T_{\phi_1}(r, \phi) + T_{\phi_2}(r, \phi) \quad (99)$$

where the individual functions are the result of the prescribed boundary condition at the particular surface noted by the subscript. A solution similar to that used in Reference 4 is presented in this Appendix. The solution primarily differs from that used in Reference 4 in that basic exponentials are used rather than hyperbolic functions.

The use of hyperbolic functions in the solution of problems in heat transfer is so universal that this departure should be explained. The use of basic exponentials offers several advantages over hyperbolic functions in preparing mathematical solutions applicable to digital computers. A prime advantage is that the solution in basic exponentials can be manipulated to avoid machine overflow by keeping arguments raised to positive or negative powers less than or greater than unity, respectively. Hyperbolic functions, as calculated by machine, necessarily contain both a positive and negative exponent of the argument and cannot be so manipulated. Secondly, the basic exponential form is easier to evaluate in certain indeterminate forms and special cases. The solution used in the digital program is as follows*:

1. $T_{r_1}(r, \phi)$

$$T_{r_1}(r, \phi) = \sum_{\gamma} P_{\gamma} \left[B_{\gamma} \sin \gamma \phi + F_{\gamma} \cos \gamma \phi \right] \xi_{r_1} \quad (100)$$

where in general ($\gamma \neq 0$)

$$\xi_{r_1} = \frac{(-1)^{\ell_{22}} \left(\frac{r}{r_1} \right)^{-\gamma} + \left(\frac{r_1}{r_2} \right)^{\gamma} \left(\frac{r}{r_2} \right)^{\gamma}}{\left(\frac{\gamma}{r_1} \right)^{-\ell_{11}} \left[(-1)^{\ell_{22}} (-1)^{-\ell_{11}} + \left(\frac{r_1}{r_2} \right)^{2\gamma} \right]} \quad \text{and} \quad (101)$$

*Valid for $\phi_1 = 0$.

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$$P_{\gamma} = \frac{\int_{\phi_1}^{\phi_2} F_{r1}(\phi) [B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi] d\phi}{\int_{\phi_1}^{\phi_2} [B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi]^2 d\phi} \quad (102)$$

If B_{γ} is infinite, that particular term in the series of Equation (100) is indeterminate and must be replaced by

$$P' \sin \gamma\phi \xi_{r1} \quad (103)$$

where

$$P' = \frac{\int_{\phi_1}^{\phi_2} F_{r1}(\phi) \sin \gamma\phi d\phi}{\left[\frac{\phi}{2} - \frac{1}{4\gamma} \sin 2\gamma\phi \right]_{\phi_1}^{\phi_2}} \quad (104)$$

The arbitrary function in Equation (104) is defined in relation to the boundary condition at the surface as follows:

1 If ℓ_{12} equals unity

$$F_{r1}(\phi) = \tau [f_{r1}(\phi)] \quad (\text{transform}) \quad (105)$$

2 If ℓ_{12} equals zero

$$F_{r1}(\phi) = \frac{1}{k_o} f_{r1}(\phi) \quad (106)$$

If γ equals zero, i.e., both k_{11} and k_{21} equal zero, an additional term must be added to Equation (110) as follows:

1 If (ℓ_{12}, ℓ_{22}) equals $(1, 0)$, the term is

$$\frac{1}{\phi_2 - \phi_1} \int_{\phi_1}^{\phi_2} F_{r1}(\phi) d\phi \quad (107)$$

2 If (ℓ_{12}, ℓ_{22}) equals $(1, 1)$ the term is

$$\frac{\ell n \frac{r}{r_2}}{(\phi_2 - \phi_1) \ell n \frac{r_1}{r_2}} \int_{\phi_1}^{\phi_2} F_{r1}(\phi) d\phi. \quad (108)$$

3 If (ℓ_{12}, ℓ_{22}) equals $(0, 1)$ the term is

$$\frac{r_1 \ell n \frac{r}{r_2}}{\phi_2 - \phi_1} \int_{\phi_1}^{\phi_2} F_{r1}(\phi) d\phi. \quad (109)$$

4 If (ℓ_{12}, ℓ_{22}) equals $(0, 0)$ the term is infinite.

2. $T_{r2}(r, \phi)$

$$T_{r2}(r, \phi) = \sum_{\gamma} L_{\gamma} \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi \right] \xi_{r2} \quad (110)$$

where, in general $(\gamma \neq 0)$

$$\xi_{r2} = \frac{(-1)^{\ell_{12}} \left(\frac{r_1}{r_2}\right)^{\gamma} \left(\frac{r}{r_1}\right)^{-\gamma} + \left(\frac{r}{r_2}\right)^{\gamma}}{\frac{\gamma}{r_2} \cdot \ell_{21} \left[(-1)^{\ell_{21}} (-1)^{\ell_{12}} \frac{r_1}{r_2} + 1 \right]} \quad (111)$$

$$L_{\gamma} = \frac{\int_{\phi_1}^{\phi_2} F_{r2}(\phi) \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi \right] d\phi}{\int_{\phi_1}^{\phi_2} \left[B_{\gamma} \sin \gamma\phi + F_{\gamma} \cos \gamma\phi \right]^2 d\phi}. \quad (112)$$

If B_{γ} is infinite, that particular term in the series (Equation (109)) is replaced by

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$$L'_{\gamma} \sin \gamma \phi \xi_{r2} \quad (113)$$

where

$$L'_{\gamma} = \frac{\int_{\phi_1}^{\phi_2} F_{r2}(\phi) \sin \gamma \phi \, d\phi}{\left[\frac{\phi}{2} - \frac{1}{4\gamma} \sin 2\gamma\phi \right]_{\phi_1}^{\phi_2}} \quad (114)$$

The arbitrary function in Equation (114) is defined in relation to the boundary condition at the surface as follows:

1 If l_{22} equals unity

$$F_{r2}(\phi) = \tau \left[f_{r2}(\phi) \right] \quad (\text{transform}). \quad (115)$$

2 If l_{22} equals zero

$$F_{r2}(\phi) = \frac{1}{K_0} f_{r2}(\phi). \quad (116)$$

If γ is equal to zero, an additional term must be added to Equation (109), as follows:

1 If (l_{12}, l_{22}) equals (0, 1) the term is

$$\frac{1}{\phi_2 - \phi_1} \int_{\phi_1}^{\phi_2} F_{r2}(\phi) \, d\phi \quad (117)$$

2 If (l_{12}, l_{22}) equals (1, 1) the term is

$$\frac{l_n \frac{r}{r_2}}{(\phi_2 - \phi_1) l_n \frac{r}{r_1}} \int_{\phi_1}^{\phi_2} F_{r2}(\phi) \, d\phi. \quad (118)$$

3 If (l_{12}, l_{22}) equals $(1, 0)$ the term is

$$\frac{r_2 \ln \frac{r}{r_2}}{(\phi_2 - \phi_1)} \int_{\phi_1}^{\phi_2} F_{r2}(\phi) d\phi. \quad (119)$$

4 If (l_{12}, l_{22}) equals $(0, 0)$ the term is infinite.

3. $T_{\phi_1}(r, \phi)$

$$T_{\phi_1}(r, \phi) = \sum_{\epsilon} Q_{\epsilon} \left[M_{\epsilon} \sin(\epsilon \ln r) + F_{\epsilon} \cos(\epsilon \ln r_1) \right] \xi_{\phi_1} \quad (120)$$

where ϵ are the real positive roots of

$$\begin{vmatrix} \frac{l_{11}\epsilon}{r_1} \cos(\epsilon \ln r_1) + l_{12} \sin(\epsilon \ln r_1) & l_{12} \cos(\epsilon \ln r_1) - \frac{l_{11}\epsilon}{r_1} \sin(\epsilon \ln r_1) \\ \frac{l_{21}\epsilon}{r_2} \cos(\epsilon \ln r_2) + l_{22} \sin(\epsilon \ln r_2) & l_{22} \cos(\epsilon \ln r_2) - \frac{l_{21}\epsilon}{r_1} \sin(\epsilon \ln r_2) \end{vmatrix} = 0 \quad (121)$$

Equation (121) can be solved explicitly similar to Equation (45).

In general ($\epsilon \neq 0$) the terms in Equation (120) are defined as:

$$\xi_{\phi_1} = \frac{(-1)^{k_{22}} e^{-\epsilon\phi} + e^{-\epsilon(2\phi_2 - \phi)}}{\left[(-1)^{k_{11}} (-1)^{k_{22}} + e^{-2\epsilon\phi_2} \right] \epsilon^{-k_{11}}} \quad (122)$$

$$Q_{\epsilon} = \frac{\int_{r_1}^{r_2} F_{\phi_1}(r) \left[M_{\epsilon} \sin(\epsilon \ln r) + F_{\epsilon} \cos(\epsilon \ln r) \right] \frac{dr}{r}}{\int_{r_1}^{r_2} \left[M_{\epsilon} \sin(\epsilon \ln r) + F_{\epsilon} \cos(\epsilon \ln r) \right]^2 \frac{dr}{r}} \quad (123)$$

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$$M_{\epsilon} = \frac{\frac{l_{11}\epsilon}{r_1} \sin(\epsilon \ln r_1) - l_{12} \cos(\epsilon \ln r_1)}{\frac{l_{11}\epsilon}{r_1} \cos(\epsilon \ln r_1) + l_{12} \sin(\epsilon \ln r_1)} \quad (124)$$

If M_{ϵ} is infinite, that particular term in the series from Equation (120) is replaced by

$$Q' \sin(\epsilon \ln r) \xi_{\phi 1} \quad (125)$$

where

$$Q' = \frac{\int_{r_1}^{r_2} F_{\phi 1}(r) \sin(\epsilon \ln r) \frac{dr}{r}}{\left[\frac{\ln r}{2} - \frac{1}{4\epsilon} \sin(2\epsilon \ln r) \right]_{r_1}^{r_2}} \quad (126)$$

The arbitrary function in Equation (126) is defined in relation to the boundary condition at the surface as follows:

1 If k_{12} is equal to unity

$$F_{\phi 1}(r) = T \left[f_{\phi 1}(r) \right] \quad (127)$$

2 If k_{12} is equal to zero

$$F_{\phi 1}(r) = \frac{r}{K_0} f_{\phi 1}(r) \quad (128)$$

If ϵ is equal to zero, i.e., both l_{12} and l_{22} equal zero, an additional term must be added to Equation (120) as follows

1 If (k_{12}, k_{22}) equals $(1,0)$ the term is

$$\frac{1}{\ln \frac{r_2}{r_1}} \int_{r_1}^{r_2} F_{\phi 1}(r) \frac{dr}{r} \quad (129)$$

2 If (k_{12}, k_{22}) equals $(1,1)$ the term is

$$\frac{\phi_2 - \phi}{r_2 \ln \frac{r_2}{r_1}} \int_{r_1}^{r_2} F_{\phi_1}(r) \frac{dr}{r} \quad (130)$$

3 If (k_{12}, k_{22}) equals $(0,1)$ the term is

$$\frac{\phi - \phi_2}{r_2 \ln \frac{r_2}{r_1}} \int_{r_1}^{r_2} F_{\phi_1}(r) \frac{dr}{r} \quad (131)$$

4 If (k_{12}, k_{22}) equals $(0,0)$ the term is infinite.

4. $T_{\phi_2}(r, \phi)$

$$T_{\phi_2}(r, \phi) = \sum_{\epsilon} \left[M_{\epsilon} \sin(\epsilon \ln r) + F_{\epsilon} \cos(\epsilon \ln r) \right] \xi_{\phi_2} \quad (132)$$

where in general

$$\xi_{\phi_2} = \frac{(-1)^{k_{12}} e^{-\epsilon(\phi_2 + \phi)} + e^{\epsilon(\phi - \phi_2)}}{\left[(-1)^{k_{21}} (-1)^{k_{12}} e^{-2\epsilon\phi_2} + 1 \right] (\epsilon)^{k_{21}}} \quad (133)$$

$$W_{\epsilon} = \frac{\int_{r_1}^{r_2} F_{\phi_2}(r) \left[M_{\epsilon} \sin(\epsilon \ln r) + F_{\epsilon} \cos(\epsilon \ln r) \right] \frac{dr}{r}}{\int_{r_1}^{r_2} \left[M_{\epsilon} \sin(\epsilon \ln r) + F_{\epsilon} \cos(\epsilon \ln r) \right] \frac{dr}{r}} \quad (134)$$

If M_{ϵ} is infinite, that particular term in the series from Equation (132) is replaced by

$$W' \sin(\epsilon \ln r) \xi_{\phi_2} \quad (135)$$

where

$$W' = \frac{\int_{r_1}^{r_2} F_{\phi_2}(r) \sin(\epsilon \ln r) \frac{dr}{r}}{\left[\frac{\ln r}{2} - \frac{1}{4\epsilon} \sin(2\epsilon \ln r) \right]_{r_1}^{r_2}} \quad (136)$$

The arbitrary function in Equation (136) is defined in relation to the boundary condition at the surface as follows:

1 If k_{22} is equal to unity

$$F_{\phi_2}(r) = \tau [f_{\phi_2}(r)] \quad (\text{transform}) \quad (137)$$

2 If k_{22} is equal to zero

$$F_{\phi_2}(r) = \frac{r}{K_0} f_{\phi_2}(r) \quad (138)$$

If ϵ is equal to zero an additional term must be added to Equation (132), as follows

1 If (k_{12}, k_{22}) equals $(0, 1)$ the term is

$$\frac{1}{\ln \frac{r_2}{r_1}} \int_{r_1}^{r_2} F_{\phi_2}(r) \frac{dr}{r} \quad (139)$$

2 If (k_{12}, k_{22}) equals $(1, 1)$ the term is

$$\frac{\phi}{\phi_2 \ln \frac{r_2}{r_1}} \int_{r_1}^{r_2} F_{\phi_2}(r) \frac{dr}{r} \quad (140)$$

3 If (k_{12}, k_{22}) equals $(1, 0)$ the term is

$$\frac{\phi}{\ln \frac{r_2}{r_1}} \int_{r_1}^{r_2} F_{\phi 2}(r) \frac{dr}{r} \quad (141)$$

4 If (k_{12}, k_{22}) equals $(0, 0)$ the term is infinite.

The preceding solution characteristically converges to zero at a corner if both adjacent faces have prescribed temperature as the boundary condition. Consequently, a dummy term equal to the sum of one-half of the adjacent transform temperatures prescribed at that corner must be added before $f(r, \phi)$ is used in Equation (80). The average of the prescribed temperatures, rather than either of the temperatures, is used to null out possible mismatch in defining arbitrary functions at a corner.

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APPENDIX B

FORTRAN LANGUAGE OF THE PROGRAM

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```
$IBFTC MAIN                                000010
COMMON/CKOUT/NANS(10)                      000020
COMMON/GCAB/ARE(2),AREARE(200)/HCAB/PHI(2),PHI(200) 000030
COMMON/SPACE1/RR(400),PP(400)             000040
COMMON/TORIAN/STOOD,ONLY,SNOW             000050
COMMON/ZOL/JPRINT(35)                     000060
1 FORMAT(3F10.0,A2,8X,2I10,A6)            000070
2 FORMAT(1H0, 5HTIME=E16.8,5X,2HR=E16.8,5X,4HPHI=E16.8,1X,
1 9H(RADIANS))                             000080
3 FORMAT(F10.0,50X,A6)                     000090
5 FORMAT(70I1)                              000100
10 FORMAT(1H0I5,1X,29HLATTICE POINT ANSWERS IN -R- /(8E16.8)) 000110
11 FORMAT(1H0I5,1X,31HLATTICE POINT ANSWERS IN -PHI- /(8E16.8)) 000120
20 FORMAT(16I5)                             000130
30 FORMAT( 44H0ERROR IN INPUT, PROBLEM UNABLE TO CONTINUE /
1 52H0ALTERATION OF FIXED INPUT SHOULD BE DONE WITH CARE ) 000140
C                                           000150
C                                           000160
C                                           000170
CALL ATHRUZ(ONN,6HON )                     000180
CALL ATHRUZ(ALASS,6HLAST T)                 000190
100 CONTINUE                                000200
OVER=0.0                                    000210
READ(5,20) NANS                             000220
C NANS(1)= THE NUMBER OF GAMMAS USED IN MAIN 000230
C NANS(2)= THE NUMBER OF BETAS USED IN MAIN 000240
C NANS(3)= THE NUMBER OF INTERVALS FOR PHI INTEGRATION IN STATE 000250
C NANS(4)= THE NUMBER OF INTERVALS FOR R INTEGRATION IN STATE 000260
C NANS(5)= THE NUMBER OF LATTICE POINTS IN THE R DIRECTION IN STATE 000270
C NANS(6)= THE NUMBER OF LATTICE POINTS IN THE PHI DIRECTION IN STATE 000280
C NANS(7)= THE NUMBER OF GAMMAS / EPSILONS USED IN STATE 000290
C NANS(8)= THE NUMBER OF INTERVALS FOR PHI INTEGRATION IN MAIN 000300
C NANS(9)= THE NUMBER OF INTERVALS FOR R INTEGRATION IN MAIN 000310
C                                           000320
IF(NANS(1).GT.30.OR.NANS(2).GT.30.OR.NANS(3).GT.200.OR.
1 NANS(4).GT.200.OR.NANS(5).GT.7.OR.NANS(6).GT.25.OR.
2 NANS(7).GT.30.OR.NANS(8).GT.200.OR.NANS(9).GT.200.OR.
3 NANS(10).GT.3) GO TO 9000                000330
READ(5,5) JPRINT                            000340
C JPRINT IS THE INTERMEDIATE PRINT INDICATOR FOR INDIVIDUAL SUBROUTINES 000350
IF(JPRINT(1).EQ.1) WRITE(6,20) NANS        000360
CALL INPUT                                  000370
IF(STOOD.NE.SNOW) CALL STATE                000380
IF(STOOD.EQ.ONLY) GO TO 100                 000390
CALL ARBCOE                                000400
200 CONTINUE                                000410
C READ(5,1) TIME,R,P,OPTION,N,M            000420
READ(5,1) TIME,R,P,OPTION,N,M,OVER         000430
IF(OPTION.EQ.ONN) GO TO 300                 000440
P=P*.01745329                               000450
WRITE(6,2) TIME,R,P                         000460
CALL SERIES(TIME,R,P,NG,NB)                 000470
IF(OVER.EQ.ALASS) GO TO 100                 000480
GO TO 200                                  000490
300 CALL SIZE(RR,N,ARE(1),ARE(2))           000500
CALL SIZE(PP,M,PHI(1),PHI(2))              000510
IF(JPRINT(1).NE.1) GO TO 400                000520
WRITE(6,10) N,(RR(I),I=1,N)                000530
WRITE(6,11) M,(PP(I),I=1,M)                000540
400 DO 500 IP=1,M                           000550
DO 500 IR=1,N                               000560
WRITE(6,2) TIME,RR(IR),PP(IP)              000570
500 CALL SERIES(TIME,RR(IR),PP(IP))         000580
IF(OVER.EQ.ALASS) GO TO 100                 000590
READ(5,3) TIME,OVER                         000600
GO TO 400                                  000610
9000 WRITE(6,30)                            000620
STOP                                        000630
END                                          000640
                                           000650
                                           000660
                                           000670
```

Contrails

```
%IBFTC BES
SUBROUTINE BES(A,X,Y,B,BP)
2  FORMAT(1H ,4E20.8)
DOUBLE PRECISION P      , BJO      , BJI      , ARG      , BR
P=0.
BJO=0.
BJI=0.
ARG=0.
PI=3.1415927
INEG=0
P=Y
ARG =.5*A*X
IF(ARG-.5)90,91,91
90 IF(P)91,11,11
91 CONTINUE
   IF(P)5,11,4
5  INEG=-1
   I=P
   PINT=I
   NIT=I-1
   PF=1.+P-PINT
   P=PF
   GO TO 11
4  IF(ABS (P-1.)-1.)11,11,6
6  INEG=1
   I=P
   PINT=I
   NIT=I-1
   PF=P-PINT
   P=PF
11 CONTINUE
   P=P
   ARG=.5*A*X
   IF(ARG+P)15,10,15
10 B=1.
   BP=0.
   GO TO 50
15 IF(ARG)17,16,17
16 ER=0.
   GO TO 25
17 IF(ARG-4.) 20,30,30
20 ER=EXP (P*ALOG(ARG))
25 AD=P+2.
   AP=P+1.
   A1=ARG**2
   AK=1.
   CALL GAMMA(AP,G)
   T1=ER/G
   B=T1
   BP=T1*P
   DO 28 I=1,100
   TN=-T1/AK*A1/AP
   B=B+TN
   TNP=TN*AD
   BP=BP+TNP
26 IF(ABS (TN)-.1E-07)26,26,27
27 IF(ABS (TNP)-.1E-07)29,29,27
27 T1=TN
   AK=AK+1.
   AD=AD+2.
28 AP=AP+1.
29 BP=BP/X
   IF(INEG)60,50,60
60 BJO=B
   BJI=(B*P-X*BP)/(A*X)
   BJI=BJI
   GO TO 50
```

```
000680
000690
000700
000710
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```
30 F1=SQRT (PI*ARG) 001360
   IDER=0 001370
   A1=-(A*X)**2 001380
   R=4.*P**2 001390
   F2=2.*ARG-(2.*P+1.)*PI/4. 001400
31 CF2=COS (F2) 001410
   SF2=SIN (F2) 001420
   T1= (R-1.)*(R-9.)/(128.*A1) 001430
   T1M=T1 001440
   T2=(1.-R)/(8.*A*X) 001450
   T2M=T2 001460
   S1=1.+T1 001470
   S2=T2 001480
   Z=2. 001490
   J=1 001500
   DO 36 I=1,100 001510
   ICNT=1 001520
   GO TO(40,45,40),J 001530
40 TM1=(R-(4.*Z-3.))**2*(R-(4.*Z-1.))**2/((2.*Z-1.)*Z*128.*A1) 001540
   IF(ABS (TM1)-ABS (T1M))43,51,51 001550
51 IF(ABS (TM1)-1.)43,41,41 001560
41 GO TO(42,42,37),J 001570
42 J=2 001580
   GO TO 44 001590
43 T1=T1*TM1 001600
   T1M=TM1 001610
   S1=S1+T1 001620
   IF(ABS (T1)-.1E-07)41,41,44 001630
44 GO TO(45,45,36),J 001640
45 TM2=(R-(4.*Z-3.))**2*(R-(4.*Z-5.))**2/((2.*Z-1.)*(Z-1.)*128.*A1) 001650
   IF(ABS (TM2)-ABS (T2M))48,49,49 001660
49 IF(ABS (TM2)-1.)48,46,46 001670
46 GO TO(47,37,47),J 001680
47 J=3 001690
   GO TO 36 001700
48 T2=T2*TM2 001710
   T2M=TM2 001720
   S2=S2+T2 001730
   IF(ABS (T2)-.1E-07)46,46,36 001740
36 Z=Z+1. 001750
37 IF(IDER)39,38,39 001760
38 B=(CF2*S1+SF2*S2)/F1 001770
   F2=F2-PI/2. 001780
   R=4.*(P+1.))**2 001790
   IDER=1 001800
   GO TO 31 001810
39 BD=(CF2*S1+SF2*S2)/F1 001820
   BP=P*B/X-A*BD 001830
   BJO=B 001840
   BJO=BJO 001850
   BJ1=BD 001860
   BJ1=BJ1 001870
50 IF(INEG)65,70,80 001880
65 DO 61 I=1,NIT 001890
   BR=P*BJO/ARG-BJ1 001900
   BJO=BJO 001910
   BJO=BR 001920
61 P=P-1. 001930
   B=BJO 001940
   BP=P*B/X-A*BJ1 001950
   GO TO 70 001960
80 P=P+1. 001970
   DO 81 I=1,NIT 001980
   BR=(P*BJ1)/ARG-BJO 001990
   BJO=BJ1 002000
   BJ1=BR 002010
81 P=P+1. 002020
   B=BJ1 002030
```

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BP=A*BJO-P*B/X	002040
70 RETURN	002050
END	002060
\$IBFTC POLYH	002070
SUBROUTINE POLYH(Z,X,Y,NX,NY,NTERMS)	002080
C	002090
C ACCEPTS NO MORE THAN 10 TERMS	002100
C	002110
COMMON/ZO1/JPRINT(35)	002120
DIMENSION A(2),V(2),E(20),COE(10)	002130
DIMENSION X(1),Y(1),Z(7,1)	002140
1 FORMAT(3F10.0)	002150
2 FORMAT(35HOPOLYNOMIAL EXCEEDS PROGRAM DESIGNS)	002160
3 FORMAT(3H X=E16.8,5X,2HY=E16.8,5X, 2HZ(1,1H,1Z,2H)=E16.8)	002170
4 FORMAT(15X,1H(E16.8,5H)(R**F10.4,5H)(P**F10.4,1H))	002180
2000 FORMAT(1X,17HPOLYNOMIAL DATA, ,1Z,1X,5HTERMS)	002190
WRITE(6,2000) NTERMS	002200
IF(NTERMS.LE.10) GO TO 8	002210
WRITE(6,2)	002220
STOP	002230
8 CONTINUE	002240
READ(5,1)(COE(I),E(2*I-1),E(2*I),I=1,NTERMS)	002250
WRITE(6,4) (COE(I),E(2*I-1),E(2*I),I=1,NTERMS)	002260
DO 200 IY=1,NY	002270
V(2)=Y(IY)	002280
DO 200 IX=1,NX	002290
V(1)=X(IX)	002300
Z(IX,IY)=0.	002310
DO 60 NT=1,NTERMS	002320
IF(COE(NT).EQ.0.) GO TO 60	002330
DO 40 L=1,2	002340
K=NT+L-1	002350
IF(E(K).EQ.0.) GO TO 35	002360
IF(V(L).EQ.0.) GO TO 60	002370
C INDETERMINANT FORM 0**0 IS SET TO UNITY	002380
A(L)=V(L)**E(K)	002390
GO TO 40	002400
35 A(L)=1.0	002410
40 CONTINUE	002420
Z(IX,IY)=Z(IX,IY)+COE(NT)*A(1)*A(2)	002430
60 CONTINUE	002440
IF(JPRINT(3).EQ.1) WRITE(6,3) V,IX,IY,Z(IX,IY)	002450
200 CONTINUE	002460
RETURN	002470
END	002480
\$IBFTC VLINE	002490
SUBROUTINE VLINE(NX,NY,XIN,YIN,XARRAY,YARRAY,ZARRAY,L,M,ZOUT)	002500
DIMENSION XARRAY(1),YARRAY(1),ZARRAY(L,M),A(2)	002510
C	002520
DO 100 I=2,NX	002530
K=1	002540
100 IF(XIN.LE.XARRAY(I)) GO TO 110	002550
110 DO 200 I=2,NY	002560
J=I	002570
200 IF(YIN.LE.YARRAY(I)) GO TO 210	002580
210 REG=(YIN-YARRAY(J-1))/(YARRAY(J)-YARRAY(J-1))	002590
DO 220 I=1,2	002600
A(I)=ZARRAY(K-1,J-1)+(ZARRAY(K-1,J)-ZARRAY(K-1,J-1))*REG	002610
220 K=K+1	002620
ZOUT=A(1)+(A(2)-A(1))*(XIN-XARRAY(K-3))/(XARRAY(K-2)-XARRAY(K-3))	002630
RETURN	002640
END	002650

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```
$IBFTC LEFT                                002660
SUBROUTINE LEFT(A,B,C,X,Y,Z)              002670
A=X                                        002680
B=Y                                        002690
C=Z                                        002700
RETURN                                    002710
END                                        002720

$IBFTC SIZE                                002730
SUBROUTINE SIZE(A,MAX,AMIN,AMAX)          002740
DIMENSION A(1)                            002750
A(1)=AMIN                                  002760
A(MAX)=AMAX                                002770
L=MAX-1                                    002780
AL=L                                        002790
DX=(AMAX-AMIN)/AL                          002800
DO 10 I=2,L                                002810
AI=I-1                                     002820
10 A(I)=AI*DX+A(1)                          002830
RETURN                                    002840
END                                        002850

$IBFTC WOOFER                               002860
SUBROUTINE WOOFER                          002870
COMMON/SPACE/YDUM(200),Y(200)             002880
COMMON/XCAB/XTRAN(50),YTRAN(50),MAXMAX    002890
COMMON/ZCCAB/AZERO                         002900
COMMON/ZD1/JPRINT(35)                     002910
1 FORMAT(40HOTRANSFORMED THERMO PHYSICAL PROPERTIES ) 002920
2 FORMAT(10X, 9HXTRAN(I)=E16.8,5X, 9HYTRAN(I)=E16.8,5X,3HI= ,I5) 002930
IF(JPRINT(7).EQ.1) WRITE(6,1)              002940
YTRAN(1)=0.                                002950
DO 40 I=2,MAXMAX                            002960
Y(1)=YDUM(I-1)                              002970
Y(50)=YDUM(I)                                002980
XDIFX=XTRAN(I)-XTRAN(I-1)                   002990
YDIFF=YDUM(I)-YDUM(I-1)                     003000
DX=XDIFF/49.                                003010
DO 30 J=2,49                                003020
AJ=J-1                                       003030
X=XTRAN(I-1)+AJ*DX                           003040
Y(J)=YDUM(I-1)+(YDIFF*(X-XTRAN(I-1)))/XDIFX 003050
30 CONTINUE                                  003060
CALL SIMP(50,DX,Y,YTRAN(I))                 003070
YTRAN(I)=YTRAN(I)/AZERO +YTRAN(I-1)         003080
40 CONTINUE                                  003090
IF(JPRINT(7).EQ.1) WRITE(6,2) (XTRAN(I),YTRAN(I),I,I=1,MAXMAX) 003100
RETURN                                       003110
END                                           003120

$IBFTC POLYGN                               003130
SUBROUTINE POLYGN(X,Y,NY,NTERMS)          003140
C                                           003150
C ACCEPTS NO MORE THAN 10 TERMS           003160
COMMON/ZD1/JPRINT(35)                       003170
DIMENSION X(1),Y(1),COE(10),ORDER(10)     003180
1 FORMAT(3F10.0)                             003190
2 FORMAT(35HOPOLYNOMIAL EXCEEDS PROGRAM DESIGNS) 003200
3 FORMAT(15X,1H(E16.8,5H)(X**F10.4,1H))     003210
4 FORMAT(18H POLYNOMIAL DATA, ,I2,1X,23HTERMS, X=DUMMY VARIABLE) 003220
5 FORMAT(3H X=E16.8,5X,2HY(I3,2H)=E16.8)    003230
```


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```

COMMON/ZO1/JPRINT(35)
1 FORMAT(1X,32H*GAMEPS* ILLEGAL SELECTION ERROR,5X,7HISEEK= I5)
500 FORMAT(1X,32H*GAMEPS* ERROR IN TAKING THE LOG)
950 FORMAT(1X,17H*GAMEPS* ISEEK= I3,3X, 5HI12= I5,3X, 5HI22= I5,3X,
1 6HMANY= I5,5X,5HREG1=E16.8,5X,7HARE(1)=E16.8/1X,7HARE(2)=E16.8,
2 5X, 7HPHI(1)=E16.8,5X, 7HPHI(2)=E16.8,5X, 8HAL(1,2)=E16.8/ 1X,
3 8HAL(2,2)=E16.8,5X, 8HAK(1,2)=E16.8,5X, 8HAK(2,2)=E16.8/ 1X,
4 7HVALUES=/ (1X,6E20.8))
PI=3.1415926
IF(ISEEK.EQ.1.OR.ISEEK.EQ.2) GO TO 100
WRITE(6,1) ISEEK
STOP
100 IF(ISEEK.NE.1) GO TO 2
C GAMMAS OR SIGMAS
REG1=PHI(2)-PHI(1)
REG2=AK(1,2)
REG3=AK(2,2)
GO TO 3
C EPSILONS
2 REG1=ARE(2)/ARE(1)
IF(REG1.LE.0.) GO TO 9997
REG1=ALOG(REG1)
REG2=AL(1,2)
REG3=AL(2,2)
3 I12=SIGN(ABS(REG2)+.005,REG2)
I22=SIGN(ABS(REG3)+.005,REG3)
IF(I12.NE.I22) GO TO 20
C SPECIAL CASE OF BOUNDARY CONDITIONS
C REG2=REG3=0.
IF(I12.NE.0) GO TO 8
VALUES(1)=0.
DO 5 M=2,MANY
AM=M-1
5 VALUES(M)=AM*PI/REG1
GO TO 900
C REG2=REG3
8 DO 10 M=1,MANY
AM=M
10 VALUES(M)=AM*PI/REG1
GO TO 900
20 PIO2=1.5707963
C SPECIAL CASE OF BOUNDARY CONDITIONS
DO 15 M=1,MANY
AM=2*M-1
15 VALUES(M)=PIO2*AM/REG1
GO TO 900
9997 WRITE(6,500)
STOP
900 IF(JPRINT(10).EQ.1) WRITE(6,950) ISEEK,I12,I22,
1 MANY,REG1,ARE(1),ARE(2),PHI(1),PHI(2),AL(1,2),AL(2,2),AK(1,2),
2 AK(2,2), (VALUES(I),I=1,MANY)
RETURN
END
003860
003870
003880
003890
003900
003910
003920
003930
003940
003950
003960
003970
003980
003990
004000
004010
004020
004030
004040
004050
004060
004070
004080
004090
004100
004110
004120
004130
004140
004150
004160
004170
004180
004190
004200
004210
004220
004230
004240
004250
004260
004270
004280
004290
004300
004310
004320
004330
004340
004350
004360
004370
004380

$IRFTC GTIME
SUBROUTINE GTIME(TIME,BETA,GS(MP))
COMMON/COMTIM/XLAMDA(50),GLAMDA(50),NLAMDA
COMMON/JOHN/GTHE TA,SAMETM
COMMON/SPACE/X(100),Y(100),SX(20),SY(20),DUM(160)
COMMON/VCAB/ALPHO
1 FORMAT(1H0,40H*GTIME* ERROR, TIME OUT OF BOUNDS, MIN=E16.8,
15X,4HMAX=E16.8,5X,5HTIME=E16.8)
C
IF(TIME.LT.XLAMDA(1).OR.TIME.GT.XLAMDA(NLAMDA))GO TO 55
ABQ=-ALPHO*BETA**2
IF(TIME.EQ.SAMETM) GO TO 25
004390
004400
004410
004420
004430
004440
004450
004460
004470
004480
004490
004500

```

Contrails

SAMETM=TIME	004510
DX=TIME/19.0	004520
DO 18 I=1,20	004530
AI=I-1	004540
SX(I)=DX*AI	004550
18 CALL CLINE(NLAMDA,SX(I),XLAMDA,GLAMDA,SY(L))	004560
GTHETA=SY(20)	004570
C	004580
25 DO 30 I=2,20	004590
A=SY(I)*EXP(ABQ*(TIME-SX(I)))	004600
30 IF(ABS(A).GE..0001) GO TO 35	004610
I=20	004620
35 BEGIN=SX(I-1)	004630
DX=(TIME-BEGIN)/99.0	004640
DO 45 I=1,100	004650
AI=I-1	004660
X(I)=AI*DX+BEGIN	004670
CALL CLINE(NLAMDA,X(I),XLAMDA,GLAMDA,Y(I))	004680
45 Y(I)=Y(I)*EXP(ABQ*(TIME-X(I)))	004690
C	004700
CALL SIMP(100,DX,Y,REG)	004710
GSIMP=-ABQ*REG	004720
GO TO 60	004730
C	004740
55 WRITE(6,1) XLAMDA(1),XLAMDA(NLAMDA),TIME	004750
STOP	004760
C	004770
60 RETURN	004780
END	004790
\$IBFTC WURZEL	004800
SUBROUTINE WURZEL	004810
COMMON/CKOUT/NANS(10)	004820
COMMON/ECAB/GAMMOS(30),BAYTOS(30,30)	004830
COMMON/GCAB/ARE(2),AREARE(200)/HCAB/PHI(2),PHIPHI(200)	004840
COMMON/ZO1/JPRINT(35)	004850
3 FORMAT(14HOFIRST GUESS= E16.8,5X, 10HINTERVAL= E16.8)	004860
5 FORMAT(1H1)	004870
NB=NANS(2)	004880
NG=NANS(1)	004890
C	004900
DO 4000 IG=1,NG	004910
C INITIAL GUESS MUST NOT EXCEED FIRST MAX OR MIN	004920
G=GAMMOS(IG)	004930
IF(G.EQ.0.0) GO TO 350	004940
IF(IG.LT.3) GO TO 100	004950
X= 2.* BAYTOS(IG-1,1) - BAYTOS(IG-2,1)	004960
GO TO 370	004970
100 X= (.25*G) / ARE(1)	004980
GO TO 370	004990
350 X=1.0/(ARE(2)-ARE(1))	005000
370 DX=0.1/(ARE(2)-ARE(1))	005010
C	005020
C	005030
IF(JPRINT(12).GE.1) WRITE(6,3) X,DX	005040
400 CALL FFP(IG,X,Y,YP)	005050
500 IF(Y.EQ.0.0) GO TO 2100	005060
IF(SIGN(1.0,YP).NE.SIGN(1.0,Y)) GO TO 650	005070
C	005080
C THE FIRST ROOT IS BEHIND YOUR GUESS	005090
610 CALL LEFT(XH,YH,SH,X,Y,YP)	005100
X=XH-.1*DX	005110
IF(X.LE.0.0) GO TO 640	005120
CALL FFP(IG,X,Y,YP)	005130
IF(SIGN(1.0,YH).NE.SIGN(1.0,Y)) GO TO 630	005140
IF(ABS(Y).GE.ABS(YH)) GO TO 640	005150

Contrails

GO TO 610	005160
630 CALL LEFT(XL,YL,SL,X,Y,YP)	005170
GO TO 1000	005180
640 CALL LEFT(XL,YL,SL,XH,YH,SH)	005190
DX=2.0*DX	005200
GO TO 660	005210
C	005220
C THE FIRST ROOT IS FORWARD OF YOUR GUESS,(OR YOU ARE PAST THE FIRST	005230
C MAX OR MIN)	005240
650 CALL LEFT(XL,YL,SL,X,Y,YP)	005250
660 X=XL+DX	005260
CALL FFP(IG,X,Y,YP)	005270
IF(Y.EQ.0.) GO TO 2100	005280
IF(SIGN(1.0,YL).NE.SIGN(1.0,Y)) GO TO 800	005290
IF(ABS(Y).GE.ABS(YL)) DX=2.0*DX	005300
GO TO 650	005310
800 CALL LEFT(XH,YH,SH,X,Y,YP)	005320
C	005330
1000 IB=1	005340
CALL BRIDGE(IB,XH,YH,SH,XL,YL,SL,IG)	005350
IF(NB.LE.1) GO TO 4000	005360
DX=2.5*DX	005370
GO TO 2120	005380
C	005390
C Y=0 TRAP	005400
2100 BAYTOS(IG,1)=X	005410
IF(NB.LE.1) GO TO 4000	005420
XH=X+DX	005430
CALL FFP(IG,XH,YH,SH)	005440
C	005450
C SECOND AND REMAINING ROOTS	005460
2120 DO 3980 IB=2,NB	005470
CALL LEFT(XL,YL,SL,XH,YH,SH)	005480
IF(IB.LT.3) GO TO 2200	005490
DX=BAYTOS(IG,IB-1)-BAYTOS(IG,IB-2)	005500
IF(DX.LE.0.) DX=.25	005510
2200 X=XL+DX	005520
CALL FFP(IG,X,Y,YP)	005530
IF(Y.EQ.0.) GO TO 3970	005540
IF(SIGN(1.0,YL).NE.SIGN(1.0,Y)) GO TO 2500	005550
CALL LEFT(XL,YL,SL,X,Y,YP)	005560
GO TO 2200	005570
2500 CONTINUE	005580
CALL LEFT(XH,YH,SH,X,Y,YP)	005590
CALL BRIDGE(IB,XH,YH,SH,XL,YL,SL,IG)	005600
GO TO 3980	005610
C	005620
C Y=0 TRAP	005630
3970 BAYTOS(IG,IB)=X	005640
IF(IB.GE.NB) GO TO 3980	005650
XH=X+DX	005660
CALL FFP(IG,XH,YH,SH)	005670
3980 CONTINUE	005680
IF(JPRINT(12).GE.1) WRITE(6,5)	005690
4000 CONTINUE	005700
RETURN	005710
END	005720
\$IBFTC BRIDGE	005730
SUBROUTINE BRIDGE(IB,XH,YH,SH,XL,YL,SL,IG)	005740
COMMON/ECAB/GAMMOS(30),BAYTOS(30,30)	005750
COMMON/ZO1/JPRINT(35)	005760
DOUBLE PRECISION D1,D2,D3	005770
1 FORMAT(34H0ATTEMPT AT ROOT WAS NOT JUSTIFIED /20H XH,YH,SH,XL,YL,S	005780
1L, 6E16.8)	005790
3 FORMAT(6H0BETA(,I2,1H,,I2,5H) = E16.8/)	005800
4 FORMAT(23H0FINE SEARCH PHASE IG= I2,3X,4HIB= I2)	005810

Contrails

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C                                                    005820
C                                                    005830
C (XH,YH) AND (XL,YL) BOARDER A ROOT                005840
C                                                    005850
C      J=1                                           005860
C      100 CONTINUE                                  005870
C            IF(SIGN(1.0,YH).EQ.SIGN(1.0,YL)) GO TO 9000 005880
C                                                    005890
C C COURSE SEARCH PHASE                              005900
C      200 SLINE=(YH-YL)/(XH-XL)                     005910
C            A=XH-YH/SLINE                           005920
C            CALL FFP(IG,A,B,BP)                     005930
C                                                    005940
C C NEIGHBORHOOD CRITERION                           005950
C      IF(SIGN(1.0,SL).EQ.SIGN(1.0,SH).AND.SIGN(1.0,SL).EQ.SIGN(1.0,BP))
C        1                                           J=J+1
C          IF(J.GE.3) GO TO 7500                      005970
C                                                    005980
C                                                    005990
C C ITERATIVE CONTRACTION                            006000
C      DEL=(XH-XL)/3.0                               006010
C      DO 220 I=1,3                                  006020
C        X=XL+DEL                                    006030
C        CALL FFP(IG,X,Y,YP)                         006040
C        IF(Y.EQ.0.) GO TO 8050                      006050
C        IF(SIGN(1.0,YL).NE.SIGN(1.0,Y)) GO TO 225   006060
C        CALL LEFT(XL,YL,SL,X,Y,YP)                 006070
C      220 CONTINUE                                  006080
C            GO TO 100                                006090
C      225 CONTINUE                                  006100
C            CALL LEFT(XH,YH,SH,X,Y,YP)              006110
C            GO TO 100                                006120
C                                                    006130
C C FINE SEARCH PHASE                                006140
C      7500 CONTINUE                                  006150
C            DO 8000 I=1,10                           006160
C              IF(JPRINT(13).GE.2) WRITE(6,4) IG,IB  006170
C              IF(ABS(B).LE.1.E-6) GO TO 8040         006180
C              D1=A                                    006190
C              D2=B                                    006200
C              D3=BP                                   006210
C              D1=D1-D2/D3                             006220
C              A=D1                                    006230
C              CALL FFP(IG,A,B,BP)                   006240
C            8000 CONTINUE                             006250
C              WRITE(6,4) IG,IB                      006260
C                                                    006270
C C      8040 BAYTOS(IG,IB)=A                         006280
C            GO TO 8080                               006290
C                                                    006300
C C      8050 BAYTOS(IG,IB)=X                         006310
C                                                    006320
C C      8080 IF(JPRINT(13).GE.1) WRITE(6,3) IG,IB,BAYTOS(IG,IB) 006330
C            RETURN                                   006340
C      9000 WRITE(6,1) XH,YH,SH,XL,YL,SL           006350
C            STOP                                     006360
C            END                                     006370

SIBFTC ABES                                          006380
SUBROUTINE ABES(B6,R6,G6,A6,AP6,Y6,YP6)            006390
C  WRITE(6,10001) B6,R6,G6                        006400
C                                                    006410
C C INPUT BETA,R,GAMMA, OUTPUT J-GAMMA,J-PRIME-GAMMA,Y-GAMMA,Y-PRIME GAMM 006420
C                                                    006430
C      DIMENSION Y(2)                               006440
C      DOUBLE PRECISION Y                           006450
C      DOUBLE PRECISION D1,D2,D3,D4,D5             006460

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Contrails

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DOUBLE PRECISION XO2S,P1,ONEOPI          006470
DOUBLE PRECISION PIO2                     006480
DOUBLE PRECISION PIO4                     006490
DOUBLE PRECISION WOOD                     006500
DOUBLE PRECISION BRO2                     006510
DOUBLE PRECISION XSQRD                    006520
DOUBLE PRECISION BLOCKA                   006530
DOUBLE PRECISION BLOCKB                   006540
DOUBLE PRECISION BLOCKC                   006550
DOUBLE PRECISION BLOCKD                   006560
DOUBLE PRECISION BLOCKE                   006570
DOUBLE PRECISION SUM1                     006580
DOUBLE PRECISION SUM2                     006590
DOUBLE PRECISION SUM3                     006600
DOUBLE PRECISION BFACT                     006610
DOUBLE PRECISION BFACT2                   006620
DOUBLE PRECISION ADENOM                   006630
DOUBLE PRECISION BOENOM                   006640
DOUBLE PRECISION HANK1                    006650
DOUBLE PRECISION HANK2                    006660
DOUBLE PRECISION HANK3                    006670
DOUBLE PRECISION HANK4                    006680
DOUBLE PRECISION TERM                     006690
11 FORMAT(1X, 6HGAMMA=1PE16.8,5X,28HNEGATIVE ORDERS ARE ILLEGAL , 006700
   1 7HIN ABES) 006710
12 FORMAT(1X,48HNEGATIVE AND ZERO ARGUMENTS ARE ILLEGAL IN ABES., 5X, 006720
   1 4HARG= 3E16.8) 006730
10000 FORMAT(4H N= 15,5X,5HY(N)=E16.8,5X,E16.8) 006740
10001 FORMAT(17H B,R,G,J,JP,Y,YP, 7E16.8) 006750
C 006760
C REF.A = APPLIED MATH FOR ENGR AND PHY *L.A.PIPES* MCGRAM-HILL 1958 006770
C (NOTE DISCREPANCY IN PIPES TEXT, LOG SHOULD BE LN) 006780
C REF.B = THEORY OF BESSEL FUNCTIONS *G.N.WATSON* CAMBRIDGE UNIV. 1958 006790
C 006800
BR=B6*R6 006810
IF(BR.LE.0.) GO TO 9001 006820
IF(G6.LT.0.) GO TO 9002 006830
C 006840
PI=3.1415926500 006850
ONEOPI=.3183098900 006860
PIO2=1.5707963300 006870
PIO4=.7853981600 006880
BRO2=BR/2.0 006890
XO2S=- (BRO2*BRO2) 006900
TWOBR=2.*BR 006910
XSQRD=- (TWOBR*TWOBR) 006920
AAA=8. 006930
TOL=1.E-09 006940
C 006950
CALL BES(B6,R6,G6,A6,AP6) 006960
NG=G6+.00001 006970
GMIX=NG 006980
FRACT=G6-GMIX 006990
IF(FRACT.LE..0001) GO TO 50 007000
C 007010
C GAMMA IS NOT AN INTEGER 007020
CALL BES(B6,R6,-G6,A5,AP5) 007030
D1=G6 007040
D2=A6 007050
D3=A5 007060
D4=DCOS(PI*D1) 007070
D5=DSIN(PI*D1) 007080
C (4.2) P.350 REF.A 007090
Y6=(D4*D2-D3)/D5 007100
D2=AP6 007110
D3=AP5 007120
YP6=(D4*D2-D3)/D5 007130
C WRITE(6,10001) B6,R6,G6,A6,AP6,Y6,YP6 007140
RETURN 007150

```


Contrails

```
50 IF(BR.GE.AAA)GO TO 100                                007160
C                                                         007170
C GAMMA IS AN INTEGER, ARGUMENT IS LESS THAN THE CROSS OVER POINT 007180
  IF(NG.NE.1) GO TO 55                                    007190
  AONE=A6                                                 007200
  APONE=AP6                                               007210
  GO TO 56                                                007220
55 CONTINUE                                              007230
  CALL BES(B6,R6,1.,AONE,APONE)                          007240
56 CONTINUE                                              007250
  WOOD=2.000*(DLOG(BRO2)+.577215700)                    007260
  D1=AONE                                                 007270
  DO 85 N=1,2                                             007280
  AN=N                                                    007290
  BLOCKA=D1*WOOD                                         007300
  BLOCKB=0.000                                           007310
  SUM1=BRO2                                              007320
  IF(N.EQ.2) SUM1=BRO2*BRO2/2.000                       007330
  SUM2=1.000                                             007340
  IF(N.EQ.2) SUM2=1.500                                  007350
  DO 75 K=1,15                                           007360
  NR=K-1                                                 007370
  R=NR                                                    007380
  IF(NR.LT.1)GO TO 60                                    007390
  D2=R*(AN+R)                                            007400
  SUM1=SUM1*(XD2S/D2)                                    007410
  D3=1.0/(AN+R)+1.0/R                                   007420
  SUM2=SUM2+D3                                           007430
60 SUM3=SUM1+SUM2                                        007440
  BLOCKB=BLOCKB+SUM3                                     007450
  XYZ=SUM3                                               007460
  IF(ABS(XYZ).LE.TOL) GO TO 78                          007470
75 CONTINUE                                              007480
78 BLOCKD=1.000/BRO2                                     007490
  IF(N.EQ.2) BLOCKD=1.000/(BRO2*BRO2)+1.000           007500
C (4.7) P.351 REF.A                                     007510
  Y(N)=ONEDPI*(BLOCKA-BLOCKB-BLOCKD)                   007520
  D2=BR                                                  007530
  D3=APONE                                               007540
  D4=B6                                                  007550
  D1=D1/D2-D3/D4                                         007560
85 CONTINUE                                              007570
  GO TO 200                                              007580
C                                                         007590
C GAMMA IS AN INTEGER, ARGUMENT IS GREATER THAN THE CROSS OVER POINT 007600
100 CONTINUE                                             007610
  D5=BR                                                  007620
  BLOCKA=DSQRT((2.000*PI)/D5)/PI                        007630
  DO 115 N=1,2                                           007640
  AN=N                                                    007650
  D2=AN                                                  007660
  D1=D5-PI02*D2-PI04                                    007670
  BLOCKB=D5 IN(D1)                                       007680
  BLOCKC=DCOS(D1)                                       007690
  BLOCKD=0.000                                          007700
  BLOCKE=0.000                                          007710
  BFACT=1.000                                           007720
  ADENOM=1.000                                          007730
  J=1                                                    007740
C J=CONVERGENCE INDICATOR                               007750
  BFACT2=1.000                                          007760
  BDENOM=TWOBR                                          007770
  IF(N.EQ.2)GO TO 101                                    007780
  HANK1=.88622692500                                    007790
  HANK3=1.32934039000                                  007800
  HANK4=1.77245385000                                  007810
  GO TO 102                                              007820
101 HANK1=1.32934039000                                  007830
  HANK4=.88622692500                                    007840
  HANK3=2.5*HANK1                                       007850
```

Contrails

```
102 HANK2=HANK1                                007860
    DO 110 K=1,7                                007870
    M=K-1                                        007880
    AM=M                                         007890
    D1=2*M                                       007900
    GO TO (104,105,104),J                       007910
104 IF(M.EQ.0)GO TO 103                        007920
    HANK1=HANK1*(D2+D1-.5D0)*(D2+D1-1.5D0)    007930
    BFACT=BFACT*D1*(D1-1.0D0)                  007940
    HANK2=HANK2/((D2-D1+.5D0)*(D2-D1+1.5D0))  007950
    ADENOM=ADENOM*XSQRD                        007960
103 TERM=(HANK1/(BFACT*HANK2))/ADENOM          007970
    BLOCKD=BLOCKD+TERM                         007980
    XYZ=TERM                                    007990
    IF(ABS(XYZ).LE.TOL) J=J+1                  008000
105 GO TO (109,109,110,110),J                  008010
109 IF(M.EQ.0)GO TO 108                        008020
    HANK3=HANK3*(D2+D1+.5D0)*(D2+D1-.5D0)    008030
    BFACT2=BFACT2*(D1+1.0D0)*D1               008040
    HANK4=HANK4/((D2-D1-.5D0)*(D2-D1+.5D0))  008050
    BDENOM=BDENOM*XSQRD                        008060
108 TERMA=(HANK3/(BFACT2*HANK4))/BDENOM       008070
    BLOCKE=BLOCKE+TERMA                        008080
    XYZ=TERMA                                   008090
    IF(ABS(XYZ).LE.TOL) J=J+2                  008100
110 IF(J.GT.3) GO TO 114                       008110
C (5) P.199 REF.B                               008120
114 Y(N)=BLOCKA*(BLOCKB*BLOCKD+BLOCKC*BLOCKE)  008130
C WRITE(6,10000) N,Y(N)                         008140
115 CONTINUE                                    008150
C                                                 008160
C RECURRENCE FORMULAS                           008170
200 IF(NG.NE.0)GO TO 210                       008180
C                                                 008190
C GAMMA ZERO                                    008200
    D1=B6                                       008210
    YP6=-Y(1)*D1                                008220
    D1=BR                                         008230
    Y6=(2.0D0*Y(1))/D1-Y(2)                    008240
    RETURN                                       008250
C                                                 008260
210 IF(NG.NE.1)GO TO 220                       008270
C                                                 008280
C GAMMA ONE                                    008290
    Y6=Y(1)                                       008300
    D1=GMIX                                       008310
    D2=R6                                         008320
    D3=B6                                         008330
    YP6=(D1*Y(1))/D2-D3*Y(2)                   008340
    RETURN                                       008350
C                                                 008360
220 IF(NG.NE.2)GO TO 230                       008370
C                                                 008380
C GAMMA TWO                                    008390
    D3=Y(1)                                       008400
    D4=Y(2)                                       008410
    Y6=Y(2)                                       008420
    GO TO 270                                    008430
C                                                 008440
230 D4=Y(2)                                       008450
    D2=Y(1)                                       008460
    DO 240 N=3,NG                                008470
    AN=N-1                                       008480
    D5= BR                                       008490
    D1=AN                                         008500
    D1=(2.0D0*D1)/D5                             008510
    D3=D4                                         008520
    D4=D1*D3-D2                                  008530
C WRITE(6,10000) N,D4,D3                       008540
```


Contrails

	D2=D3	008550
240	CONTINUE	008560
	Y6=D4	008570
C		008580
270	CONTINUE	008590
	D1=B6	008600
	D2=GMIX	008610
	D5=R6	008620
	YP6=D1*D3-(D2*D4)/D5	008630
C	WRITE(6,10001) B6,R6,G6,A6,AP6,Y6,YP6	008640
	RETURN	008650
C		008660
9001	CONTINUE	008670
	WRITE(6,12) BR,B6,R6	008680
	STOP	008690
9002	CONTINUE	008700
	WRITE(6,11) G6	008710
	STOP	008720
	END	008730
\$IBFTC	GAMMA	008740
	SUBROUTINE GAMMA(X,Y)	008750
	DIMENSION A(10)	008760
	A(1) = .83333333E-01	008770
	A(2) = -.27777778E-02	008780
	A(3) = .79365079E-03	008790
	A(4) = -.59523810E-03	008800
	A(5) = .84175084E-03	008810
	A(6) = -.19175269E-02	008820
	A(7) = .64102564E-02	008830
	A(8) = -.29550654E-01	008840
	A(9) = .17964437E 00	008850
	A(10) = -.13924322E 01	008860
	IF(X)10,13,30	008870
10	K000FX=X	008880
	XINT=K000FX	008890
	XF=XINT-X	008900
	AX=X	008910
	XM=1.0	008920
	J=3-K000FX	008930
	DO 15 I=1,J	008940
	XM=XM*AX	008950
15	AX=AX+1.0	008960
	IF(ABS(XM)-1.0E-30)11,11,29	008970
11	IE=K000FX/2	008980
	IF(K000FX-2*IE)13,12,13	008990
12	Y=-1.0E 30	009000
	GO TO 50	009010
13	Y=1.0E 30	009020
	GO TO 50	009030
29	AX=3.0-XF	009040
	GO TO 38	009050
30	IF(X-1.0)34,37,36	009060
34	IF(X-1.0E-30)13,13,37	009070
37	AX=X+2.0	009080
	XM=X*(X+1.0)	009090
	GO TO 38	009100
36	IF(X-2.0)39,39,14	009110
39	AX=X+1.0	009120
	XM=X	009130
	GO TO 38	009140
14	AX=X	009150
	XM=1.0	009160
	IF(X-30.0)38,13,13	009170
38	C=1.0/AX**2	009180
	T1=A(1)/AX	009190

Contrails

F=T1	009200
DO 35 I=2,10	009210
TN=T1*A(I)/A(I-1)*C	009220
IF(ABS(TN)-ABS(T1))32,32,33	009230
32 F=F+TN	009240
T1=TN	009250
IF(ABS(TN)-.2E-08)33,33,35	009260
35 CONTINUE	009270
33 EX=(AX-.5)*ALOG(AX)-AX+F+.91893853	009280
Y=EXP(EX)	009290
Y=Y/XM	009300
50 RETURN	009310
END	009320
\$IBFTC AMSAFC	009330
FUNCTION AMSAFC(IG,V)	009340
COMMON/ECAB/DUM(30),EPILON(30),DUMY(870)	009350
COMMON/WCAB/AM(30),AF(30)	009360
COMMON/Z01/JPRINT(35)	009370
1 FORMAT(1X,7HMSAFC=E16.8)	009380
C	009390
A=ALOG(V)	009400
C	009410
A=A*EPILON(IG)	009420
IF(AM(IG).EQ.0.) GO TO 110	009430
B=AM(IG)*SIN(A)	009440
GO TO 120	009450
110 B=0.0	009460
120 IF(AF(IG).EQ.0.) GO TO 130	009470
C=COS(A)	009480
GO TO 140	009490
130 C=0.0	009500
140 AMSAFC=B+C	009510
IF(JPRINT(16).EQ.1) WRITE(6,1) AMSAFC	009520
8800 RETURN	009530
END	009540
\$IBFTC SERIES	009550
SUBROUTINE SERIES(TIME,R,P)	009560
C	009570
DOUBLE PRECISION SERIA,SERIB,TERMA,TERMB	009580
COMMON/ACAB/ARBA(30,30),ARBG(30,30)	009590
COMMON/CCAB/CSUB(30,30)	009600
COMMON/CKOUT/NANS(10)	009610
COMMON/ECAB/GAMMOS(30),BAYTOS(30,30)	009620
COMMON/JOHN/GTHETA,SAMETH	009630
COMMON/OCAB/STEADY(7,25),XR(7),YP(25),KR,KP	009640
COMMON/V CAB/ALPHO	009650
COMMON/Z01/JPRINT(35)	009660
3 FORMAT(3X,6HTERM A,10X,6HTERM B,10X,8HSERIES A,8X,8HSERIES B,	009670
1 8X,5HGAMMA,11X,4HBETA,12X,6HGTHETA,10X,5HGSIMP)	009680
4 FORMAT(1X,8E16.8)	009690
5 FORMAT(9X,12HTEMPERATURE=E16.8,10X,E16.8)	009700
C	009710
NG=NANS(1)	009720
NB=NANS(2)	009730
C	009740
IF(JPRINT(17).GE.2) WRITE(6,3)	009750
C	009760
SERIA=0.000	009770
SERIB=0.000	009780
DO 400 IG=1,NG	009790
DO 400 IB=1,NB	009800
CALL ABES(BAYTOS(IG,IB),R,GAMMOS(IG),U,UP,W,WP)	009810
B2=CSUB(IG,IB)*U+W	009820

Contrails

	REG1=B2*BSNFCS(IG,P)	009830
C	TERMA=ARBA(IG,IB)*REG1*EXP(-ALPHO*BAYTOS(IG,IB)**2*TIME)	009840
	CALL GTIME(TIME,BAYTOS(IG,IB),GSIMP)	009850
	TERMB= ARBG(IG,IB)*REG1*(-GTHETA+GSIMP)	009860
C	SERIA=SERIA+TERMA	009880
	SERIB=SERIB+TERMB	009890
C	IF(JPRINT(17).GE.2)	009900
	1 WRITE(6,4) TERMA,TERMB,SERIA,SERIB,GAMMOS(IG),BAYTOS(IG,IB),	009910
	2 GTHETA,GSIMP	009920
400	CONTINUE	009930
C	CALL VLINE(KR,KP,R,P,XR,YP,STEADY,7,25,S)	009940
	HEAT=SERIA+SERIB +GTHETA*S	009950
	IF(JPRINT(17).GE.1) WRITE(6,5) HEAT,S	009960
	HEAT=ATRA(HEAT)	009970
500	WRITE(6,5) HEAT	009980
	RETURN	009990
	END	010000
		010010
		010020
		010030
		010040
\$IBFTC	STATE	010050
	SUBROUTINE STATE	010060
	DOUBLE PRECISION T1,T2,T3,T4	010070
	COMMON/ACAB/FARE1(200),FARE2(200),FPHI1(200),FPHI2(200),DUME(1000)	010080
	COMMON/CCAB/STO1(25),STO2(7),STO3(7),STO4(7),STO5(25),STO6(25),	010090
	1 DUMY(804)	010100
	COMMON/CKOUT/NANS(10)	010110
	COMMON/ECAB/GAMMOS(30),EPILON(30),DUM(870)	010120
	COMMON/GCAB/ARE(2),AREARE(200)/HCAB/PHI(2),PHIPHI(200)	010130
	COMMON/DCAB/STEADY(7,25),XR(7),YP(25),KR,KP	010140
	COMMON/RCAB/AL(2,2),AK(2,2)	010150
	COMMON/ZO1/JPRINT(35)	010160
	EQUIVALENCE (KR,NR)	010170
	EQUIVALENCE (KP,NP)	010180
13	FORMAT(6H0 T1,13X,2HT2,14X,2HT3,14X,2HT4,14X,13HSTEADY(IR,IP),	010190
1	10X,2HIR,8X,2HIP)	010200
14	FORMAT(1X,5E16.8,2I10)	010210
C	NG=NANS(7)	010220
	NP=NANS(6)	010230
	NR=NANS(5)	010240
	DO 100 IP=1,NP	010250
	DO 100 IR=1,NR	010260
	STEADY(IR,IP)=0.0	010270
100	CONTINUE	010280
C		010290
	N=1	010300
	CALL GAMEPS(N,NANS(7),GAMMOS)	010310
	N=2	010320
	CALL GAMEPS(N,NANS(7),EPILON)	010330
C		010340
	DO 400 IG=1,NG	010350
	CALL COE(IG,CP,CL,CQ,CW)	010360
	DO 300 IP=1,NP	010370
	STO1(IP)=BSNFCS(IG,YP(IP))	010380
	N=3	010390
	STO5(IP)=XI(IG,YP(IP),N)	010400
	N=4	010410
	STO6(IP)=XI(IG,YP(IP),N)	010420
300	CONTINUE	010430
C		010440
	DO 400 IR=1,NR	010450
	N=1	010460
	STO2(IR)=XI(IG,XR(IR),N)	010470
	N=2	010480

Contrails

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      STO3(IR)=X(IG,XR(IR),N)
      STO4(IR)=AMSAFC(IG,XR(IR))
400 CONTINUE
C
      IF(IG.EQ.NG.OR.JPRINT(18).GE.1) WRITE(6,13)
      DO 3000 IP=1,NP
      DO 3000 IR=1,NR
      T1=CP*STO1(IP)*STO2(IR)
      T2=CL*STO1(IP)*STO3(IR)
      T3=CQ*STO4(IR)*STO5(IP)
      T4=CW*STO4(IR)*STO6(IP)
      STEADY(IR,IP)=STEADY(IR,IP)+T1+T2+T3+T4
      IF(IG.EQ.NG.OR.JPRINT(18).GE.1) WRITE(6,14) T1,T2,T3,T4,
1    STEADY(IR,IP),IR,IP
3000 CONTINUE
4000 CONTINUE
C
      NP1=NANS(3)
      NR1=NANS(4)
      IF(AL(1,2)+AK(1,2).EQ.2.0) STEADY(1,1)=STEADY(1,1)+.5*FARE1(1)+
1                                     .5*FPHI1(1)
      IF(AL(1,2)+AK(2,2).EQ.2.0) STEADY(1,NP)=STEADY(1,NP)+.5*FARE1(NP1)
1                                     +.5*FPHI2(1)
      IF(AL(2,2)+AK(1,2).EQ.2.0) STEADY(NR,1)=STEADY(NR,1)+.5*FARE2(1)
1                                     +.5*FPHI1(NR1)
      IF(AL(2,2)+AK(2,2).EQ.2.0) STEADY(NR,NP)=STEADY(NR,NP)+
1                                     .5*FARE2(NP1)+.5*FPHI2(NR1)
C
      RETURN
      END

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$IRFTC XI
      FUNCTION XI(IG,N,NBOUND)
      COMMON/ECAB/GAMMOS(50),EPILON(30),DUM(870)
      COMMON/GCAB/ARE(2),AREARE(200)/HCAB/PHI(2),PHIPHI(200)
      COMMON/RCAB/AL(2,2),AK(2,2)
      COMMON/UTILTY/R1OR2,R2OR1,AR1OR2,AR2OR1,AR1,AR2
      COMMON/ZD1/JPRINT(35)
      DOUBLE PRECISION D1,D2,D3,D4
1  FORMAT(49H1JOB TERMINATED, DO NOT PUT FLUX AT ALL BOUNDRIES
1  (8E16.8))
2  FORMAT(1X,3HXI=E16.8,5X, 8HBOUNDARY,2X,I1)
C
      GO TO(100,2000,3000,4000),NBOUND
C
C
C AT BOUNDARY R1
C
100 CONTINUE
      IF(GAMMOS(IG).NE.0.) GO TO 1000
      IF(AL(1,2).EQ.0.) GO TO 200
      IF(AL(2,2).EQ.0.) GO TO 800
      XI=ALOG(V/ARE(2))/AR1OR2
      GO TO 8900
200 IF(AL(2,2).EQ.0.) GO TO 9000
      XI=ARE(1)*ALOG(V/ARE(2))
      GO TO 8900
C
1000 CONTINUE
      D1=(V/ARE(1))*(-GAMMOS(IG))
      D2=(R1OR2**GAMMOS(IG))*((V/ARE(2))*GAMMOS(IG))
      D4= R1OR2**(2.0*GAMMOS(IG))
      IF(AL(1,1).EQ.0.) GO TO 1050
      D3=-GAMMOS(IG)/ARE(1)
      D4=D4*(-D3)
      GO TO 1100
1050 CONTINUE

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010990
011000
011010
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011080
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011100
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011120
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Contrails

<pre> D3=1.000 1100 CONTINUE IF(AL(2,2).EQ.0.0) GO TO 8700 D3=-D3 D1=-D1 GO TO 8700 C C AT BOUNDARY R2 C 2000 CONTINUE IF(GAMMOS(IG).NE.0.) GO TO 2800 IF(AL(1,2).EQ.0.) GO TO 2030 IF(AL(2,2).EQ.0.) GO TO 2020 XI=ALOG(V/ARE(1))/AR2OR1 GO TO 8900 2020 CONTINUE XI=ARE(2)*ALOG(V/ARE(1)) GO TO 8900 2030 CONTINUE IF(AL(2,2).EQ.0.) GO TO 9000 GO TO 8800 C 2800 CONTINUE D1=(R1OR2**GAMMOS(IG))*(V/ARE(1))*(-GAMMOS(IG)) D2=(V/ARE(2))*GAMMOS(IG) D3=R1OR2**(2.*GAMMOS(IG)) IF(AL(2,1).EQ.0.) GO TO 2835 D3=-(GAMMOS(IG)/ARE(2))*D3 D4=GAMMOS(IG)/ARE(2) GO TO 2880 2835 CONTINUE D4=1.000 2880 CONTINUE IF(AL(1,2).EQ.0.) GO TO 8700 D1=-D1 D3=-D3 GO TO 8700 C C AT BOUNDARY P1 C 3000 CONTINUE IF(EPIILON(IG).NE.0.) GO TO 3705 IF(AK(1,2).EQ.0.) GO TO 3010 IF(AK(2,2).EQ.0.) GO TO 8800 XI=PHI(2)-V GO TO 8900 3010 CONTINUE IF(AL(2,2).EQ.0.) GO TO 9000 XI=V-PHI(2) GO TO 8900 C 3705 CONTINUE D1=EXP(-EPIILON(IG)*V) D2=EXP(-EPIILON(IG)*(2.*PHI(2)-V)) D4=EXP(-2.*EPIILON(IG)*PHI(2)) IF(AK(1,1).NE.0.) GO TO 3815 D3=1.000 GO TO 3860 3815 CONTINUE D3=-EPIILON(IG) D4=D4*EPIILON(IG) 3860 CONTINUE IF(AK(2,2).EQ.0.) GO TO 8700 D1=-D1 D3=-D3 GO TO 8700 C C AT BOUNDARY P2 C </pre>	<pre> 011150 011160 011170 011180 011190 011200 011210 011220 011230 011240 011250 011260 011270 011280 011290 011300 011310 011320 011330 011340 011350 011360 011370 011380 011390 011400 011410 011420 011430 011440 011450 011460 011470 011480 011490 011500 011510 011520 011530 011540 011550 011560 011570 011580 011590 011600 011610 011620 011630 011640 011650 011660 011670 011680 011690 011700 011710 011720 011730 011740 011750 011760 011770 011780 011790 011800 011810 011820 011830 </pre>
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Contrails

4000 CONTINUE	011840
IF(EPIILON(IG).NE.0.) GO TO 5000	011850
IF(AK(1,2).EQ.0.) GO TO 4025	011860
IF(AK(2,2).EQ.0.) GO TO 4016	011870
XI=V/PHI(2)	011880
GO TO 8900	011890
4016 CONTINUE	011900
XI=V	011910
GO TO 8900	011920
4025 IF(AK(2,2).EQ.0.) GO TO 9000	011930
GO TO 8800	011940
C	011950
5000 CONTINUE	011960
D1=EXP(-EPIILON(IG)*(PHI(2)+V))	011970
D2=EXP(EPIILON(IG)*(V-PHI(2)))	011980
D3= EXP(-2.*EPIILON(IG)*PHI(2))	011990
IF(AK(2,1).EQ.0.) GO TO 5046	012000
D3=-D3*EPIILON(IG)	012010
D4=EPIILON(IG)	012020
GO TO 6000	012030
5046 CONTINUE	012040
D4=1.000	012050
6000 CONTINUE	012060
IF(AK(1,2).EQ.0.) GO TO 8700	012070
D1=-D1	012080
D3=-D3	012090
C	012100
8700 XI=(D1+D2)/(D3+D4)	012110
GO TO 8900	012120
C	012130
8800 XI=1.0	012140
C	012150
8900 CONTINUE	012160
IF(JPRINT(19).EQ.1) WRITE(6,2) XI,NBOUND	012170
RETURN	012180
C	012190
C	012200
9000 CONTINUE	012210
WRITE(6,1) AL,AK	012220
STOP	012230
END	012240
\$IBFTC FFP	012250
SUBROUTINE FFP(IG,B,F,FP)	012260
DOUBLE PRECISION D1,D2	012270
DOUBLE PRECISION T,TP	012280
DIMENSION T(4),TP(4),C(4),P(4)	012290
COMMON/ECAB/GAMMOS(30),BAYTDS(30,30)	012300
COMMON/GCAB/ARE(2),AREARE(200)/HCAB/PHI(2),PHIPHI(200)	012310
COMMON/RCAB/AL(2,2),AK(2,2)	012320
COMMON/ZD1/JPRINT(35)	012330
14 FORMAT(1X,2HK=I2,5X,5HC(K)=E16.8,5X,5HP(K)=E16.8,5X,5HT(K)=E16.8,	012340
1 5X,6HTP(K)=E16.8,E16.8)	012350
20 FORMAT(3H X=E16.8,3X,2HY=E16.8,3X,3HYP=E16.8,3X,6HGAMMA I2)	012360
BR=B*ARE(1)	012370
REG1= ((GAMMOS(IG)**2)/BR) -BK	012380
A=AL(1,1)	012390
D=AL(1,2)	012400
R=ARE(1)	012410
CALL ABES(B,ARE(1),GAMMOS(IG),C(1),P(1),C(2),P(2))	012420
CALL ABES(B,ARE(2),GAMMOS(IG),C(3),P(3),C(4),P(4))	012430
DO 13 K=1,4	012440
D1=A*P(K)	012450
D2=D*C(K)	012460
T(K)=D1+D2	012470
D1=A*C(K)*REG1	012480
D2=R*D*P(K)/B	012490

Contrails

TP(K)=D1+D2	012500
IF(K.EQ.1) GO TO 13	012510
A=AL(2,1)	012520
D=AL(2,2)	012530
BR=B*ARE(2)	012540
REG1= ((GAMMOS(IG)**2)/BR) -BR	012550
R=ARE(2)	012560
13 IF(JPRINT(20).GE.2) WRITE(6,14) K,C(K),P(K),T(K),TP(K),REG1	012570
F=T(1)*T(4)-T(2)*T(3)	012580
FP=T(1)*TP(4)-T(2)*TP(3)+TP(1)*T(4)-TP(2)*T(3)	012590
IF(JPRINT(20).GE.1) WRITE(6,20) B,F,FP,IG	012600
RETURN	012610
END	012620
\$IBFTC ARBCOE	012630
SUBROUTINE ARBCOE	012640
COMMON/ACAB/ARBA(30,30),ARBG(30,30)	012650
COMMON/CKOUT/NANS(10)	012660
COMMON/ECAB/GAMMOS(30),BAYTOS(30,30)	012670
COMMON/GCAB/ARE(2),AREARE(200)/HCAB/PHI(2),PHIPHI(200)	012680
COMMON/OCAB/STEADY(7,25),XR(7),YP(25),KR,KP	012690
COMMON/PCAB/DZERO(7,25),XRD(7),YPD(25),LR,LP	012700
COMMON/SPACE/STO1(200),STO2(200)	012710
COMMON/SPACE1/A1(200),A2(200),A3(200),A4(200)	012720
COMMON/ZO1/JPRINT(35)	012730
1 FORMAT(8H A(I,J)=E16.8,5X, 7HG(I,J)=E16.8,5X,3HI= I3,3X,3HJ= I3)	012740
C	012750
NG=NANS(1)	012760
NB=NANS(2)	012770
NR=NANS(9)	012780
NP=NANS(8)	012790
C	012800
CALL SIZE(PHIPHI,NP,PHI(1),PHI(2))	012810
CALL SIZE(AREARE,NR,ARE(1),ARE(2))	012820
C	012830
DXR=AREARE(2)-AREARE(1)	012840
DXP=PHIPHI(2)-PHIPHI(1)	012850
C	012860
CALL GAMEPS(1,NANS(1),GAMMOS)	012870
CALL WURZEL	012880
C	012890
DO 500 IG=1,NG	012900
CALL DENOM(IG,DEHOM)	012910
C	012920
DO 91 I=1,NP	012930
STO1(I)=BSNFCS(IG,PHIPHI(I))	012940
91 CONTINUE	012950
C	012960
DO 500 IB=1,NB	012970
CALL ENOM(IG,IB,EHOM)	012980
C	012990
STO2 IS FILLED IN ENOM	012990
DO 300 IP=1,NP	013000
DO 200 IR=1,NR	013010
CALL VLINE(LR,LP,AREARE(IR),PHIPHI(IP),XRD,YPD,DZERO,7,25,D)	013020
D=TRAN(D)	013030
CALL VLINE(KR,KP,AREARE(IR),PHIPHI(IP),XR,YP,STEADY,7,25,S)	013040
REG=STO1(IP)*STO2(IR)	013050
A1(IR)=D*REG	013060
A2(IR)=S*REG	013070
200 CONTINUE	013080
CALL SIMP(NR,DXR,A1,A)	013090
A3(IP)=A	013100
CALL SIMP(NR,DXR,A2,B)	013110
A4(IP)=B	013120
300 CONTINUE	013130
CALL SIMP(NP,DXP,A3,V1)	013140
CALL SIMP(NP,DXP,A4,V2)	013150

Contrails

C	REG=EHOM*DEHOM	013160
	ARBA(IG,IB)=V1/REG	013170
	ARBG(IG,IB)=V2/REG	013180
C		013190
	IF(JPRINT(21).EQ.1) WRITE(6,1) ARBA(IG,IB),ARBG(IG,IB),IG,IB	013200
500	CONTINUE	013210
	RETURN	013220
	END	013230
		013240
\$IBFTC	TRAN	013250
	FUNCTION TRAN(V)	013260
	COMMON/XCAB/XTRAN(50),YTRAN(50),MAXMAX	013270
	COMMON/YCAB/BEE,SEE/ZBCAB/ICASE	013280
	COMMON/ZO1/JPRINT(35)	013290
	50 FORMAT(1X,17H*TRANSFORM ERROR=)	013300
	51 FORMAT(1X,14H*TRAN= CASE= 13,3X,4HBEE=E16.8,5X,4HSEE=E16.8,5X,	013310
	1 2HV=E16.8 / 1X,2HQ=E16.8,5X,5HREG1=E16.8,5X,5HREG2=E16.8,5X,	013320
	2 5HREG3=E16.8 / 1X,5HREG4=E16.8,5X,5HREG5=E16.8,5X,5HTRAN=E16.8)	013330
C	SET ICASE EQUAL TO THE CASE NUMBER OF THE DESIRED TRANSFORM	013340
	REG1=0.	013350
	REG2=0.	013360
	REG3=0.	013370
	REG4=0.	013380
	REG5=0.	013390
	Q=0.	013400
	GO TO (1,2,3,4,5,6,7),ICASE	013410
	1 TRAN=1.0/BEE-EXP(-BEE*V)/BEE	013420
C	CASE 1	013430
	GO TO 100	013440
	2 IF(SEE.LE.0.) GO TO 9997	013450
C	CASE 2	013460
	REG1=4.*SEE-BEE*BEE	013470
	Q=REG1	013480
	REG2=2.*SEE*V	013490
	IF(REG1)11,12,13	013500
	11 REG1=SQRT(-REG1)	013510
	REG3=REG2/(BEE-REG1)	013520
	REG4=REG2/(BEE+REG1)	013530
	REG5=(REG3+1.)/(REG4+1.)	013540
	IF(REG5.LE.0.) GO TO 9997	013550
	TRAN=ALOG(REG5)/REG1	013560
	GO TO 100	013570
	12 REG3=REG2+BEE	013580
	TRAN=2./BEE-2./REG3	013590
	GO TO 100	013600
	13 REG1=SQRT(REG1)	013610
	REG3=REG2+BEE	013620
	REG4=ATAN2(REG3,REG1)	013630
	REG5=ATAN2(BEE,REG1)	013640
	TRAN=2.*(REG4-REG5)/REG1	013650
	GO TO 100	013660
	3 REG1=1.+BEE*V	013670
C	CASE 3	013680
	IF(REG1.LE.0..OR.BEE.EQ.0.) GO TO 9997	013690
	TRAN=ALOG(REG1)/BEE	013700
	GO TO 100	013710
	4 TRAN=V+BEE*V*V/2.	013720
C	CASE 4	013730
	GO TO 100	013740
	5 CONTINUE	013750
C	CASE 5	013760
	6 TRAN=V	013770
C	CASE 6	013780
	GO TO 100	013790
	7 CALL CLINE(MAXMAX,V,XTRAN(1),YTRAN(1),TRAN)	013800

Contrails

C CASE 7	013810
GO TO 100	013820
9997 WRITE(6,50)	013830
STOP	013840
100 IF(JPRINT(22).EQ.1) WRITE(6,51) ICASE,BEE,SEE,V,	013850
1 Q,REG1,REG2,REG3,REG4,REG5,TRAN	013860
RETURN	013870
END	013880
\$IBFTC BSNFCS	013890
FUNCTION BSNFCS(IG,P)	013900
COMMON/ECAB/GAMMOS(30),DUM(900)	013910
COMMON/ZECAB/BSUB(30),FSUB(30)	013920
COMMON/ZO1/JPRINT(35)	013930
1 FORMAT(1X,7HBSNFCS=E16.8)	013940
A=GAMMOS(IG)*P	013950
IF(BSUB(IG).EQ.0.) GO TO 100	013960
R1=BSUB(IG)*SIN(A)	013970
GO TO 200	013980
100 R1=0.	013990
200 IF(FSUB(IG).EQ.0.) GO TO 300	014000
R2=FSUB(IG)*COS(A)	014010
GO TO 400	014020
300 R2=0.	014030
400 BSNFCS=R1+R2	014040
IF(JPRINT(23).EQ.1) WRITE(6,1) BSNFCS	014050
RETURN	014060
END	014070
\$IBFTC COECOE	014080
SUBROUTINE COECOE(IG,CP,CL,CQ,CW)	014090
COMMON/ACAB/FARE1(200),FARE2(200),FPHI1(200),FPHI2(200),A1(200),	014100
1 A2(200),DUMY(600)	014110
COMMON/CKOUT/NANS(10)	014120
COMMON/ECAB/GAMMOS(30),EPILON(30),DUM(870)	014130
COMMON/GCAB/ARE(2),AREARE(200)/HCAB/PHI(2),PHIPHI(200)	014140
COMMON/ZO1/JPRINT(35)	014150
1 FORMAT(1X,6HCP(1)=E16.8,5X,6HCL(1)=E16.8,5X,6HCQ(1)=E16.8,5X,	014160
1 6HCW(1)=E16.8,5X,3HI= 15)	014170
C	014180
CALL DENOM(IG,DEHOM)	014190
NTERVL=NANS(3)	014200
DO 500 INN=1,NTERVL	014210
B=BSNFCS(IG,PHIPHI(INN))	014220
A1(INN)=FARE1(INN)*B	014230
A2(INN)=FARE2(INN)*B	014240
500 CONTINUE	014250
CALL SIMP(NTERVL,PHIPHI(2)-PHIPHI(1),A2,AREA2)	014260
CALL SIMP(NTERVL,PHIPHI(2)-PHIPHI(1),A1,AREA1)	014270
CP=AREA1/DEHOM	014280
CL=AREA2/DEHOM	014290
C	014300
C	014310
NTERVL=NANS(4)	014320
CALL CENOM(IG,CEHOM)	014330
DO 1000 INN=1,NTERVL	014340
B=AMSAFC(IG,AREARE(INN))/AREARE(INN)	014350
A1(INN)=FPHI1(INN)*B	014360
A2(INN)=FPHI2(INN)*B	014370
1000 CONTINUE	014380
CALL SIMP(NTERVL,AREARE(2)-AREARE(1),A1,AREA1)	014390
CALL SIMP(NTERVL,AREARE(2)-AREARE(1),A2,AREA2)	014400
CQ=AREA1/CEHOM	014410
CW=AREA2/CEHOM	014420

Contrails

C	IF(JPRINT(24).EQ.1) WRITE(6,1) CP,CL,CQ,CW,IG RETURN END	014430 014440 014450 014460
\$IBFTC CENOM	SUBROUTINE CENOM(IG,CEHOM) DIMENSION A(2),X(2) COMMON/ECAB/DUM(30),EPILON(30),DUMY(870) COMMON/GCAB/ARE(2),AREARE(200) COMMON/RCAB/AL(2,2),AK(2,2) COMMON/UTILTY/R1OR2,R2DR1,AR1OR2,AR2OR1,AR1,AR2 COMMON/WCAB/AM(30),AF(30) COMMON/ZO1/JPRINT(35) 1 FORMAT(7H CEHOM=E16.8,5X,7HAM(IG)=E16.8,5X,4MIG= 15)	014470 014480 014490 014500 014510 014520 014530 014540 014550 014560
C	T1=0.0 T2=0.0 T3=0.0	014570 014580 014590 014600
C	IF(EPILON(IG).EQ.0.) GO TO 7600 REG1=SIN(EPILON(IG)*AR1) REG2=COS(EPILON(IG)*AR1) UNDER=AL(1,1)*EPILON(IG)*REG2/ARE(1)+AL(1,2)*REG1 IF(UNDER.EQ.0.) GO TO 4000 AM(IG)=(AL(1,1)*EPILON(IG)*REG1/ARE(1) -AL(1,2)*REG2)/UNDER AF(IG)=1.0 GO TO 4065	014610 014620 014630 014640 014650 014660 014670 014680 014690
4000	CONTINUE AM(IG)=1.0 AF(IG)=0.0	014700 014710 014720
C	4065 CONTINUE X(1)=EPILON(IG)*AR1 X(2)=EPILON(IG)*AR2	014730 014740 014750 014760
C	IF(AM(IG).EQ.0.) GO TO 4800 IF(AF(IG).EQ.0.) GO TO 5200	014770 014780 014790
C	DO 4300 I=1,2 A(I)=-.5*SIN(X(I))*2	014800 014810
4300	CONTINUE T3=(2.0*AM(IG)/EPILON(IG))*(A(2)-A(1))	014820 014830 014840
C	4800 CONTINUE DO 4850 I=1,2 A(I)=X(I)/2.0+SIN(2.0*X(I))/4.0	014850 014860 014870 014880
4850	CONTINUE T2=(A(2)-A(1))/EPILON(IG)	014890 014900
C	IF(AM(IG).EQ.0.) GO TO 6123	014910
5200	CONTINUE DO 5256 I=1,2 A(I)=X(I)/2.0-SIN(2.0*X(I))/4.0	014920 014930 014940 014950
5256	CONTINUE T1=(AM(IG)*2)*(A(2)-A(1))/EPILON(IG)	014960 014970
C	6123 CONTINUE CEHOM=T1+T2+T3 GO TO 8653	014980 014990 015000 015010
C	7600 CONTINUE AM(IG)=0.0 AF(IG)=1.0 CEHOM=AR2OR1	015020 015030 015040 015050 015060
C	8653 CONTINUE IF(JPRINT(25).EQ.1) WRITE(6,1) CEHOM,AM(IG),IG RETURN END	015070 015080 015090 015100 015110

Contrails

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$IBFTC ATRAN                                015120
      FUNCTION ATRAN(T)                       015130
      COMMON/XCAB/XTRAN(50),YTRAN(50),MAXMAX  015140
      COMMON/YCAB/BEE,SEE/ZBCAB/ICASE        015150
      COMMON/ZD1/JPRINT(35)                  015160
      50 FORMAT(1X,17H*INVERSION ERROR=)     015170
      51 FORMAT(1X,14H*ATLAN* CASE= I3,3X,4HBEE=E16.8,5X,4HSEE=E16.8,5X,  015180
      1 2HT=E16.8 / 1X,2HQ=E16.8,5X,5HREG1=E16.8,5X,5HREG2=E16.8,5X,  015190
      2 5HREG3=E16.8 / 1X,5HREG4=E16.8,30X,6HATLAN=E16.8) 015200
      C SET ICASE EQUAL TO THE CASE NUMBER OF THE DESIRED INVERSION 015210
      REG1=0.                                  015220
      REG2=0.                                  015230
      REG3=0.                                  015240
      REG4=0.                                  015250
      Q=0.                                      015260
      GO TO (1,2,3,4,5,6,7),ICASE            015270
      1 REG1=1.-T*BEE                          015280
      C CASE 1                                  015290
      IF(REG1.LE.0.) GO TO 9997                015300
      ATRAN=-ALOG(REG1)/BEE                    015310
      GO TO 100                                015320
      2 IF(SEE.EQ.0.) GO TO 9997              015330
      C CASE 2                                  015340
      REG1=4.*SEE-BEE*BEE                      015350
      Q=REG1                                    015360
      IF(REG1)11,12,13                         015370
      11 REG1=SQRT(-REG1)                       015380
      REG2=EXP(REG1*T)*(BEE-REG1)/(BEE+REG1)  015390
      ATRAN=((BEE+REG1)*REG2-BEE+REG1)/(2.*SEE*(1.-REG2)) 015400
      GO TO 100                                015410
      12 ATRAN=BEE*T/(4.*SEE/BEE-2.*SEE*T)     015420
      GO TO 100                                015430
      13 REG1=SQRT(REG1)                       015440
      REG2=BEE/REG1                             015450
      REG3=REG1*T/2.                            015460
      REG4=SIN(REG3)/COS(REG3)                 015470
      REG4=REG3*REG2                            015480
      TWOSEE=2.*SEE                             015490
      ATRAN=REG1*(REG2+REG3)/(TWOSEE-TWOSEE*REG4)-BEE/TWOSEE 015500
      GO TO 100                                015510
      3 IF(BEE.EQ.0.) GO TO 9997              015520
      C CASE 3                                  015530
      ATRAN=(EXP(BEE*T)-1.)/BEE                 015540
      GO TO 100                                015550
      4 REG1=1.+2.*BEE*T                       015560
      C CASE 4                                  015570
      IF(REG1.LT.0.) GO TO 9997                015580
      ATRAN=(-1.+SQRT(REG1))/BEE                015590
      GO TO 100                                015600
      5 CONTINUE                               015610
      C CASE 5                                  015620
      6 ATRAN=T                                 015630
      C CASE 6                                  015640
      GO TO 100                                015650
      7 CALL CLINE(MAXMAX,T,YTRAN(1),XTRAN(1),ATLAN) 015660
      C CASE 7                                  015670
      IF(ATLAN.GT.XTRAN(MAXMAX)) GO TO 9997    015680
      GO TO 100                                015690
      9997 WRITE(6,50)                          015700
      STOP                                      015710
      100 IF(JPRINT(26).EQ.1) WRITE(6,51) ICASE,BEE,SEE,T, 015720
      1 Q,REG1,REG2,REG3,REG4,ATLAN            015730
      RETURN                                    015740
      END                                       015750
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Contrails

	\$IBFTC DENOM	015760
	SUBROUTINE DENOM(IG,DEHOM)	015770
C		015780
	DIMENSION A(2),X(2)	015790
	COMMON/ECAB/GAMMOS(30),DUM(900)	015800
	COMMON/MCAB/PHI(2),PHI(200)	015810
	COMMON/RCAB/AL(2,2),AK(2,2)	015820
	COMMON/ZECAB/BSUB(30),FSUB(30)	015830
	COMMON/ZO1/JPRINT(35)	015840
	1 FORMAT(7H DEHOM=E16.8,5X,9HBSUB(IG)=E16.8,5X,4HIG= I5)	015850
C		015860
	IF(GAMMOS(IG).EQ.0.) GO TO 8011	015870
C		015880
	T3=0.0	015890
	T2=0.0	015900
	T1=0.0	015910
C		015920
	X(1)=GAMMOS(IG)*PHI(1)	015930
	X(2)=GAMMOS(IG)*PHI(2)	015940
	REG1=COS(X(1))	015950
	REG2=SIN(X(1))	015960
	UNDER=AK(1,1)*GAMMOS(IG)*REG1+AK(1,2)*REG2	015970
	IF(UNDER.EQ.0.) GO TO 106	015980
	BSUB(IG)=(-AK(1,2)*REG1+AK(1,1)*GAMMOS(IG)*REG2)/UNDER	015990
	FSUB(IG)=1.0	016000
	GO TO 150	016010
106	CONTINUE	016020
	BSUB(IG)=1.0	016030
	FSUB(IG)=0.0	016040
150	CONTINUE	016050
	IF(BSUB(IG).EQ.0.) GO TO 3000	016060
	IF(FSUB(IG).EQ.0.) GO TO 1000	016070
	DO 400 I=1,2	016080
	A(I)=BSUB(IG)*(SIN(X(I)))**2/GAMMOS(IG)	016090
400	CONTINUE	016100
	T2=A(2)-A(1)	016110
1000	CONTINUE	016120
	B=BSUB(IG)**2/2.0	016130
	G=2.0*GAMMOS(IG)	016140
	DO 1002 I=1,2	016150
	A(I)=B*PHI(I)-B*SIN(2.0*X(I))/G	016160
1002	CONTINUE	016170
	T1=A(2)-A(1)	016180
	IF(FSUB(IG).EQ.0.) GO TO 7000	016190
3000	CONTINUE	016200
	G=4.0*GAMMOS(IG)	016210
	DO 3800 I=1,2	016220
	A(I)=PHI(I)/2.0 +SIN(2.0*X(I))/G	016230
3800	CONTINUE	016240
	T3=A(2)-A(1)	016250
7000	CONTINUE	016260
	DEHOM=T1+T2+T3	016270
	GO TO 8900	016280
C		016290
8011	CONTINUE	016300
	BSUB(IG)=0.0	016310
	FSUB(IG)=1.0	016320
	DEHOM=PHI(2)-PHI(1)	016330
C		016340
8900	CONTINUE	016350
	IF(JPRINT(27).EQ.1) WRITE(6,1) DEHOM,BSUB(IG),IG	016360
	RETURN	016370
	END	016380

Contrails

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$IBF=0 INPUT                                016390
SUBROUTINE INPUT                             016400
C                                             016410
COMMON/ACAB/FARE(1200),FARE2(200),FPHI1(200),FPHI2(200),DUME(1000) 016420
COMMON/CKOUT/NANS(10)                       016430
COMMON/COMTIM/XLAMDA(50),GLAMDA(50),NLAMDA  016440
COMMON/GCAB/ARE(2),AREARE(200)/HCAB/PHI(2),PHIPHI(200) 016450
COMMON/QCAB/STEADY(7,25),XR(7),YP(25),KR,KP 016460
COMMON/PCAB/DZERO(7,25),XRD(7),YPD(25),LR,LP 016470
COMMON/RCAB/AL(2,2),AK(2,2)                016480
COMMON/SPACE/DUM(200),R(200)               016490
COMMON/TORIAN/STOOD,ONLY,SNOW              016500
COMMON/UTILTY/R1OR2,R2OR1,AR1OR2,AR2OR1,AR1,AR2 016510
COMMON/VCAB/ALPHG                           016520
COMMON/XCAB/XTRAN(50),YTRAN(50),MAXMAX     016530
COMMON/YCAB/BEE,SEE                          016540
COMMON/ZBCAB/ICASE                           016550
COMMON/ZCCAB/AZERO                           016560
EQUIVALENCE (R,P)                           016570
C                                             016580
1 FORMAT(6F10.0)                             016590
2 FORMAT(A5,5X,I10,4F10.0)                  016600
3 FORMAT(A5,5X,2I10,A6)                    016610
4 FORMAT(A5,5X,A5,5X,I10,A2)               016620
5 FORMAT(12A6)                              016630
6 FORMAT(1H112A6)                           016640
8 FORMAT( 6HOINPUT/ 9H GEOMETRY/1X,3HR1=E16.8,5X,3HR2=E16.8/ 1X, 016650
1 3HP1=E16.8,5X,3HP2=E16.8, 5X,9H(RADIANS)) 016660
9 FORMAT(27HOTHERMO PHYSICAL PROPERTIES)    016670
10 FORMAT(1X,11HPOINT DATA,3X,I2,12H POINT PAIRS / (4E20.8)) 016680
11 FORMAT(1X, 6HALPHA=E16.8,5X, 6HKZERO=E16.8) 016690
12 FORMAT(1X,26HEVALUATE POLYNOMIAL FROM E16.8,3X,5HTD E16.8) 016700
13 FORMAT(29H TRANSFORM ACCORDING TO CASE 11,3H,B=E16.8,3X,2HC=E16.8) 016710
14 FORMAT(34HOINITIAL TEMPERATURE DISTRIBUTION ) 016720
15 FORMAT(1X,12HPOINT DATA, I2,1HX,I2,18H, VALUES OF F(R,P)) 016730
16 FORMAT(3X,5E20.8)                        016740
1000 FORMAT(44HOERROR IN INPUT, PROBLEM UNABLE TO CONTINUE ,10X,I12) 016750
4000 FORMAT(1H0,20HBOUNDARY CONDITIONS )    016760
4001 FORMAT(1X,7HAT R1, ,A6,3X,4HL11=F10.2,3X,4HL12=F10.2) 016770
4002 FORMAT(1X,7HAT R2, ,A6,3X,4HL21=F10.2,3X,4HL22=F10.2) 016780
4003 FORMAT(1X,7HAT P1, ,A6,3X,4HK11=F10.2,3X,4HK12=F10.2) 016790
4004 FORMAT(1X,7HAT P2, ,A6,3X,4HK21=F10.2,3X,4HK22=F10.2) 016800
4005 FORMAT(1X,11HPOINT DATA,3X,I2,43H EVENLY SPACED ORDINATES BETWEEN 016810
IR1 AND R2 / (6E20.8))                    016820
4006 FORMAT(1X,11HPOINT DATA,3X,I2,43H EVENLY SPACED ORDINATES BETWEEN 016830
IP1 AND P2 / (6E20.8))                    016840
4007 FORMAT(14HOTIME FUNCTION)              016850
4009 FORMAT(44HOSTEADY STATE SOLUTION IS BEING INPUT ) 016860
4010 FORMAT(1H1)                            016870
10000 FORMAT(7F10.0)                       016880
CALL ATHRUZ(PLY,6HPOLY )                   016890
CALL ATHRUZ(CACE,6HCASE )                  016900
CALL ATHRUZ(PONT,6HPDINT )                016910
CALL ATHRUZ(SNOW,6HND )                   016920
CALL ATHRUZ(PLUX,6HFLUX )                 016930
CALL ATHRUZ(FEVER,6HTEMP )                016940
CALL ATHRUZ(ONLY,6HON )                   016950
CALL ATHRUZ(SSOP,6HTRANSF)                016960
J=0                                         016970
C TITLE                                     016980
READ(5,5) (R(I),I=1,12)                   016990
WRITE(6,6) (R(I),I=1,12)                  017000
C                                             017010
C GEOMETRY, AREARE                          017020
READ(5,1) ARE                              017030
R1OR2=ARE(1)/ARE(2)                       017040
R2OR1=ARE(2)/ARE(1)                       017050
AR1OR2=ALOG(R1OR2)                        017060
AR2OR1=ALOG(R2OR1)                        017070

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Contrails

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      AR1=ALOG(ARE(1))
      AR2=ALOG(ARE(2))
C PHIPHI
      READ(5,1) PHI
C PHI IS INPUT IN DEGREES AND FRACTIONS
      DO 18 I=1,2
      18 PHI(I)=PHI(I)*.01745329
      WRITE(6,8) ARE,PHI
      IF(ARE(1).LE.0.0.OR.ARE(2).LE.0.0.OR.PHI(1).NE.0.0.OR.PHI(2).LE.0.
      10) GO TO 900
C
C THERMO PHYSICAL PROPERTIES, TRAN
      WRITE(6,9)
      READ(5,2) WHAT,N,A,B,C,D
      IF(N.GT.50) GO TO 19
      IF(WHAT.EQ.PONT) GO TO 20
      IF(WHAT.EQ.PLY) GO TO 30
      IF(WHAT.EQ.CACE) GO TO 40
      19 J=1
      GO TO 900
      20 ALPHD=A
      AZERO=B
      READ(5,1) (XTRAN(I),DUM(I),I=1,N)
      MAXMAX=N
      WRITE(6,10) MAXMAX,(XTRAN(I),DUM(I),I=1,MAXMAX)
      WRITE(6,11) A,B
      GO TO 35
      30 ALPHD=C
      AZERO=D
      MAXMAX=50
      CALL SIZE(XTRAN,MAXMAX,A,B)
      CALL POLYGN(XTRAN,DUM,MAXMAX,N)
      WRITE(6,12) A,B
      WRITE(6,11) C,D
      35 ICASE=7
C
      CALL WOOFER
C
      GO TO 60
      40 ICASE=N
      BEE=A
      SEE=B
      ALPHD=C
      AZERO=D
      WRITE(6,13) N,BEE,SEE
      IF(ICASE.GT.7) GO TO 19
      WRITE(6,11) C,D
C
C INITIAL TEMPERATURE DISTRIBUTION, DZERO(IR,IP)
      60 WRITE(6,14)
      READ(5,3) WHAT,N,M
      IF(WHAT.EQ.PLY) GO TO 90
      IF(N.GT.7.OR.M.GT.25) GO TO 68
      IF(WHAT.EQ.PONT) GO TO 70
      68 J=2
      GO TO 900
      70 CALL SIZE(XRD,N,ARE(1),ARE(2))
      CALL SIZE(YPD,M,PHI(1),PHI(2))
      WRITE(6,15) N,M
      DO 75 J=1,M
      L=M-J+1
      READ(5,10000) (DZERO(I,L),I=1,N)
      75 WRITE(6,16) (DZERO(I,L),I=1,N)
      LR=N
      LP=M
      GO TO 95
      90 NTERMS=N
      LR=7
      LP=25
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Contrails

CALL SIZE(XRD,LR,ARE(1),ARE(2))	017770
CALL SIZE(YPD,LP,PHI(1),PHI(2))	017780
CALL POLYH(DZERO,XRD,YPD,LR,LP,NTERMS)	017790
C	017800
C BOUNDARY CONDITIONS, FARE1(PHI),AL(1,1),AL(1,2)	017810
95 WRITE(6,4000)	017820
100 READ(5,4) X,WHAT,N,STOOD	017830
IF(X.EQ.FEVER) GO TO 110	017840
IF(X.EQ.PHLUX) GO TO 120	017850
J=3	017860
GO TO 900	017870
110 AL(1,1)=0.	017880
AL(1,2)=1.	017890
GO TO 130	017900
120 AL(1,1)=-1	017910
AL(1,2)=0.	017920
130 WRITE(6,4001) X,AL(1,1),AL(1,2)	017930
IF(STOOD.EQ.SNOW) GO TO 160	017940
IF(N.GT.200) GO TO 135	017950
CALL SIZE(PHIPHI,NANS(3),PHI(1),PHI(2))	017960
CALL SIZE(AREARE,NANS(4),ARE(1),ARE(2))	017970
CALL SIZE(XR,NANS(5),ARE(1),ARE(2))	017980
CALL SIZE(YP,NANS(6),PHI(1),PHI(2))	017990
NP=NANS(3)	018000
NR=NANS(4)	018010
IF(WHAT.EQ.PONT) GO TO 140	018020
IF(WHAT.EQ.PLY) GO TO 150	018030
135 J=4	018040
GO TO 900	018050
140 READ(5,10000) (DUM(I),I=1,N)	018060
WRITE(6,4006) N,(DUM(I),I=1,N)	018070
CALL SIZE(P,N,PHI(1),PHI(2))	018080
DO 145 IP=1,NP	018090
145 CALL CLINEIN,PHIPHI(IP),P,DUM,FARE1(IP))	018100
GO TO 155	018110
150 CALL POLYGN(PHIPHI,FARE1,NANS(3),N)	018120
155 IF(AL(1,2).EQ.1.0) GO TO 157	018130
DO 156 IP=1,NP	018140
FARE1(IP)=FARE1(IP)/AZERO	018150
156 CONTINUE	018160
GO TO 160	018170
157 CONTINUE	018180
DO 158 IP=1,NP	018190
FARE1(IP)=TRAN(FARE1(IP))	018200
158 CONTINUE	018210
C	018220
C FARE2(PHI),AL(2,1),AL(2,2)	018230
160 READ(5,4) X,WHAT,N	018240
IF(X.EQ.FEVER) GO TO 170	018250
IF(X.EQ.PHLUX) GO TO 180	018260
J=5	018270
GO TO 900	018280
170 AL(2,1)=0.	018290
AL(2,2)=1.	018300
GO TO 190	018310
180 AL(2,1)=1.	018320
AL(2,2)=0.	018330
190 WRITE(6,4002) X,AL(2,1),AL(2,2)	018340
IF(STOOD.EQ.SNOW) GO TO 220	018350
IF(N.GT.200) GO TO 195	018360
IF(WHAT.EQ.PONT) GO TO 200	018370
IF(WHAT.EQ.PLY) GO TO 210	018380
195 J=6	018390
GO TO 900	018400
200 READ(5,10000) (DUM(I),I=1,N)	018410
WRITE(6,4006) N,(DUM(I),I=1,N)	018420
CALL SIZE(P,N,PHI(1),PHI(2))	018430
DO 205 IP=1,NP	018440
205 CALL CLINEIN,PHIPHI(IP),P,DUM,FARE2(IP))	018450

Contrails

GO TO 211	018460
210 CALL POLYGN(PHIPHI,FARE2,NP,N)	018470
211 IF(AL(2,2).EQ.1.0) GO TO 213	018480
DO 212 IP=1,NP	018490
FARE2(IP)=FARE2(IP)/AZERO	018500
212 CONTINUE	018510
GO TO 220	018520
213 CONTINUE	018530
DO 214 IP=1,NP	018540
FARE2(IP)=TRAN(FARE2(IP))	018550
214 CONTINUE	018560
C	018570
Q FPHI1(ARE),AK(1,1),AK(1,2)	018580
220 READ(5,4) X,WHAT,N	018590
IF(X.EQ.FEVER) GO TO 230	018600
IF(X.EQ.PHLUX) GO TO 240	018610
J=7	018620
GO TO 900	018630
230 AK(1,1)=0.	018640
AK(1,2)=1.	018650
GO TO 250	018660
240 AK(1,1)=-1	018670
AK(1,2)=0.	018680
250 WRITE(6,4003) X,AK(1,1),AK(1,2)	018690
IF(STOOD.EQ.SNOW) GO TO 280	018700
IF(N.GT.200) GO TO 255	018710
IF(WHAT.EQ.PONT) GO TO 260	018720
IF(WHAT.EQ.PLY) GO TO 270	018730
255 J=8	018740
GO TO 900	018750
260 READ(5,10000) (DUM(I),I=1,N)	018760
WRITE(6,4005) N,(DUM(I),I=1,N)	018770
CALL SIZE(R,N,ARE(1),ARE(2))	018780
DO 265 IR=1,NR	018790
265 CALL CLINE(N,AREARE(IR),R,DUM,FPHI1(IR))	018800
GO TO 271	018810
270 CALL POLYGN(AREARE,FPHI1,NR,N)	018820
271 IF(AK(1,2).EQ.1.0) GO TO 273	018830
DO 272 IR=1,NR	018840
FPHI1(IR)=AREARE(IR)*FPHI1(IR)/AZERO	018850
272 CONTINUE	018860
GO TO 280	018870
273 CONTINUE	018880
DO 274 IR=1,NR	018890
FPHI1(IR)=TRAN(FPHI1(IR))	018900
274 CONTINUE	018910
C	018920
C FPHI2(ARE),AK(2,1),AK(2,2)	018930
280 READ(5,4) X,WHAT,N	018940
IF(X.EQ.FEVER) GO TO 290	018950
IF(X.EQ.PHLUX) GO TO 300	018960
J=9	018970
GO TO 900	018980
290 AK(2,1)=0.	018990
AK(2,2)=1.	019000
GO TO 310	019010
300 AK(2,1)=1.	019020
AK(2,2)=0.	019030
310 WRITE(6,4004) X,AK(2,1),AK(2,2)	019040
IF(STOOD.EQ.SNOW) GO TO 360	019050
IF(N.GT.200) GO TO 315	019060
IF(WHAT.EQ.PONT) GO TO 320	019070
IF(WHAT.EQ.PLY) GO TO 330	019080
315 J=10	019090
GO TO 900	019100
320 READ(5,10000) (DUM(I),I=1,N)	019110
WRITE(6,4005) N,(DUM(I),I=1,N)	019120
CALL SIZE(R,N,ARE(1),ARE(2))	019130
DO 325 IR=1,NR	019140

Contrails

325	CALL CLINE(N,AREARE(IR),R,DUM,FPHI2(IR))	019150
	GO TO 331	019160
330	CALL POLYGN(AREARE,FPHI2,NR,N)	019170
331	IF(AK(2,2).EQ.1.0) GO TO 333	019180
	DO 332 IR=1,NR	019190
	FPHI2(IR)= AREARE(IR)*FPHI2(IR)/AZERO	019200
332	CONTINUE	019210
	GO TO 400	019220
333	CONTINUE	019230
	DO 334 IR=1,NR	019240
	FPHI2(IR)=TRAN(FPHI2(IR))	019250
334	CONTINUE	019260
	GO TO 400	019270
C		019280
C	STEADY STATE INPUT OPTION	019290
360	WRITE(6,4009)	019300
	READ(5,3) WHAT,N,M,X	019310
	IF(WHAT.EQ.PLY) GO TO 380	019320
	IF(N.GT.7.OR.M.GT.25) GO TO 365	019330
	IF(WHAT.EQ.PONT) GO TO 370	019340
365	J=11	019350
	GO TO 900	019360
370	WRITE(6,15) N,M	019370
	DO 375 J=1,M	019380
	L=M-J+1	019390
	READ(5,10000) (STEADY(I,L),I=1,N)	019400
375	WRITE(6,16) (STEADY(I,L),I=1,N)	019410
	CALL SIZE(XR,N,ARE(1),ARE(2))	019420
	CALL SIZE(YP,M,PHI(1),PHI(2))	019430
	KR=N	019440
	KP=M	019450
	GO TO 390	019460
380	KR=7	019470
	KP=25	019480
	CALL SIZE(XR,KR,ARE(1),ARE(2))	019490
	CALL SIZE(YP,KP,PHI(1),PHI(2))	019500
	NTERMS=N	019510
	CALL POLYH(STEADY,XR,YR,KR,KP,NTERMS)	019520
390	IF(X.NE.SSOP) GO TO 400	019530
	DO 395 IR=1,KR	019540
	DO 395 IP=1,KP	019550
395	STEADY(IR,IP)=TRAN(STEADY(IR,IP))	019560
C		019570
C	TIME FUNCTION	019580
400	READ(5,2) WHAT,N,B	019590
	A=0.0	019600
	WRITE(6,4007)	019610
	IF(WHAT.EQ.PONT) GO TO 420	019620
	IF(WHAT.EQ.PLY) GO TO 430	019630
410	J=12	019640
	GO TO 900	019650
420	NLAMDA=N	019660
	IF(NLAMDA.GT.50) GO TO 410	019670
	READ(5,1) (XLAMDA(I),GLAMDA(I),I=1,N)	019680
	WRITE(6,10) N,(XLAMDA(I),GLAMDA(I),I=1,N)	019690
	GO TO 440	019700
430	NLAMDA=50	019710
	CALL SIZE(XLAMDA,NLAMDA,A,B)	019720
	CALL POLYGN(XLAMDA,GLAMDA,NLAMDA,N)	019730
	WRITE(6,12) A,B	019740
440	WRITE(6,4010)	019750
	RETURN	019760
C	ERRORS	019770
900	WRITE(6,1000) J	019780
	STOP	019790
	RETURN	019800
	END	019810

Contrails

```
$IBFTC SIMP                                019820
  SUBROUTINE SIMP(NT,DX,V,AREA)            019830
C   EQUALLY SPACED POINTS                019840
C   EVEN NO. OF INTERVALS---SIMPSONS RULE 019850
C   ODD NO.--- FIRST N-1 INTERVALS-SIMPSONS LAST INTERVAL-TRAPEZOID 019860
C   DIMENSION V(1)                        019870
C   DIMENSION V(1)                        019880
  AREA=0.0                                  019890
  IF(NT-1)47,50,47                          019900
  47 IF((NT/2)*2-NT)28,29,28                019910
  29 NOD=1                                    019920
  N1=NT-1                                    019930
  GO TO 32                                    019940
  28 N1=NT                                    019950
  NOD=0                                       019960
  32 AREA1=0.0                               019970
  IF(N1-1)35,35,445                          019980
  445 J=N1-1                                  019990
  DO 33 I=2,J,2                               020000
  33 AREA=V(I)+AREA                          020010
  IF(N1-3)446,446,7447                       020020
  7447 J=N1-2                                 020030
  DO 34 I=3,J,2                               020040
  34 AREA1=V(I)+AREA1                        020050
  446 AREA =.33333333*(V(1)+V(N1)+2.0*AREA1+4.0*AREA) 020060
  IF(NOD)35,36,35                             020070
  35 AREA= AREA +.5*(V(N1)+V(N1+1))          020080
  36 AREA= DX*AREA                            020090
  50 RETURN                                    020100
  END                                          020110

$IBFTC ATHRUZ                                020120
  SUBROUTINE ATHRUZ(A, B)                   020130
  DIMENSION A(1), B(1)                      020140
  DO 25 I = 1, 22                            020150
  A(I) = B(I)                                 020160
  C = COMPL(B(I+1))                          020170
  IF( C .EQ. 0.) RETURN                      020180
  25 CONTINUE                                 020190
  RETURN                                       020200
  END                                          020210

$IBFTC CLINE                                020220
  SUBROUTINE CLINE(N,X,A,B,Y)               020230
  DIMENSION A(1),B(1)                       020240
C   DIMENSION A(1),B(1)                     020250
  IF (N) 100,100,101                         020260
  101 DO 102 J=2,N                            020270
  I=J                                          020280
  IF (X-A(J)) 103,103,102                    020290
  102 CONTINUE                                 020300
  103 Y=B(I-1) + (X+A(I-1)) *(B(I)-B(I-1))/(A(I)-A(I-1)) 020310
  RETURN                                       020320
  100 Y=0.0                                    020330
  RETURN                                       020340
  END                                          020350
```

Contrails

APPENDIX C

DESCRIPTION OF SUBROUTINES

Contrails

APPENDIX C

DESCRIPTION OF SUBROUTINE

Independent intermediate print is available for many of the subroutines and subprograms. Each routine has been assigned a sequence number. This sequence number corresponds to a column number of card B in the input data (see Section IIIB). Hence, a particular column i of card B controls the intermediate print of the routine or subprogram which has been assigned the sequence number i. A description of this resulting intermediate print for a particular routine is given in the discussion of that routine.

The routines and their associated sequence numbers are as follows:

Routine	Sequence No.	Routine	Sequence No.
MAIN	1	SERIES	17
BES	2	STATE	18
PØLYH	3	XI (function)	19
VLINE	4	FFP	20
LEFT	5	ARBCØE	21
SIZE	6	TRAN (function)	22
WØØFER	7	BSNFCS (function)	23
PØLYGN	8	CØECØE	24
ENØM	9	CENØM	25
GAMEPS	10	ATRAN (function)	26
GTIME	11	DENØM	27
WURZEL	12	INPUT	28
BRIDGE	13		
ABES	14	ATHRUZ)	
GAMMA	15	SIMP)	System routines
AMSAFC (function)	16	CLINE)	

1. MAIN (Sequence Number 1)

MAIN is the control point for the program. It sets up the order in which the program will execute.

MAIN reads in the fixed input and in the times and locations at which answers are requested. The remainder of the input data is handled in sub-routine INPUT.

Contrails

The fixed input in MAIN consists of the two leading cards A and B of the input data. Card A contains the mathematical options for the program which are stored in an array called NANS. Card B contains the intermediate print options for the program which are stored in an array called JPRINT. The following are the mathematical options:

- NANS(1) = the number of gammas used for a solution (30 max)
- NANS(2) = the number of betas used for a solution (30 max)
- NANS(3) = the number of intervals for phi integration in STATE (200 max)
- NANS(4) = the number of intervals for r integration in STATE (200 max)
- NANS(5) = the number of lattice points in the r direction in STATE (7 max)
- NANS(6) = the number of lattice points in the phi direction in STATE (25 max)
- NANS(7) = the number of gammas/epsilons used in STATE (30 max)
- NANS(8) = the number of intervals for phi integration in ARBCØE (200 max)
- NANS(9) = the number of intervals for r integration in ARBCØE (200 max)

Each subscript of NANS is a multiple of a five column span on card A. Thus, the number of β 's, NANS(2), would appear in columns 6 through 10, and it must be right adjusted in its field with no decimal point, as must all of the entries on cards A and B.

Intermediate print for MAIN consists of a printout of the image of card A, and the lattice network in r and ϕ , if applicable.

A detailed flow diagram of MAIN is given in Appendix D.

2. BES (Sequence Number 2)

Subroutine BES is one of the two subroutines used in a previous program (Reference 4). This subroutine calculates Bessel functions of the form $J_\gamma(\beta r)$ and $J'_\gamma(\beta r)$. The arguments for this routine are A, X, Y, B, and BP, which correspond to β , r, γ , $J_\gamma(\beta r)$ and $J'_\gamma(\beta r)$.

This subroutine uses series solution and recursion formulas similar to those used in ABES. (See References 6 and 7).

3. PØLYH (Sequence Number 3)

Subroutine PØLYH is a utility subroutine of subroutine Input. PØLYH reads in and evaluates a polynomial of ten or less terms where each term is of the form:

$$A X^B Y^C.$$

Contrails

The restrictions on A, B, and C are those imposed by the computer in raising a number to a power. In general it should be noted that a computer error usually occurs when

$$|B| \ln X > 88.02$$

or

$$|C| \ln Y > 88.02$$

(A negative X or Y also causes a computer error, but the X's and Y's used in this program correspond to r and ϕ , and these cannot be negative without voiding the solution).

PØLYH automatically prints out the polynomial that was input.

The argument list for PØLYH is Z, X, Y, NX, NY, and NTERMS where:

- Z = the name of the array where the evaluations of the polynomial will be stored
- X = an independent variable array
- Y = an independent variable array
- NX = the number of points in X where the polynomial is to be evaluated
- NY = the number of points in Y where the polynomial is to be evaluated
- NTERMS = the number of terms comprising the polynomial

A 1 placed in column 3 of card B will produce intermediate print of the arrays X, Y, and Z.

4. VLINE (Sequence Number 4)

Subroutine VLINE linearly interpolates on a doubly subscripted array Z(X,Y). The arguments for VLINE are NX, NY, XIN, YIN, XARRAY, YARRAY, L, M, and ZØUT, where:

- NX = the number of points in the X direction
- NY = the number of points in the Y direction
- XIN = the value X at which the value for Z(X,Y) is desired
- YIN = the value Y at which a value for Z(X,Y) is desired
- XARRAY = the name of the singularly subscripted array containing the X independent variable
- YARRAY = the name of the singularly subscripted array containing the Y independent variable
- ZARRAY = the name of the doubly subscripted array containing the Z dependent variable

Contrails

L and M = the dimensions on the Z array, where the Z array is of the form Z(L,M)

Note: L and M must agree with the dimensions given for the Z array in the calling program.

ZØUT = the interpolated value of the dependent variable.

VLINE was especially designed for the lattice networks of the steady state solution and the initial temperature distribution which are stored as double subscripted variables in the program.

5. LEFT (Sequence Number 5)

Subroutine LEFT was created to avoid unnecessary rewriting of a short set of instructions. It simply moves an X, Y, and Z prime into their appropriate storage areas. LEFT is used in subroutines WURZEL and BRIDGE.

6. SIZE (Sequence Number 6)

Subroutine SIZE divides an interval into a given number of constant size increments. The arguments for size are:

A, MAX, AMIN, and AMAX

where

A = the array name where the increments are to be stored
MAX = the number of increments plus one
AMIN = the lower limit of the interval
AMAX = the upper limit of the interval

7. WØØFER (Sequence Number 7)

Subroutine WØØFER cumulatively integrates numerical input data of the thermophysical properties to obtain the required transformation and inversion. This subroutine performs a numerical integration of Equation (11) and stores an array of real temperature and transform temperature.

A 1 placed in column 7 of card 8 will produce intermediate print of the arrays XTRAN (real temperature) and YTRAN (transform temperature). The input thermophysical property data is always printed out.

8. PØLYGN (Sequence Number 8)

Subroutine PØLYGN performs the same basic function as subroutine PØLYH but for one independent variable instead of two; i.e., PØLYGN reads and evaluates a polynomial whose terms are of the form

$$Y = AX^B$$

(See Subroutine PØLYGH).

The arguments for PØLYGN are X, Y, NY, and NTERMS, where:

- X = the independent variable
- Y = the dependent variable array
- NY = the number of points at which the polynomial is to be evaluated,
- NTERMS = the number of terms comprising the polynomial.

The polynomial evaluated by PØLYGN is automatically printed out.

A 1 placed in column 8 of card B will produce intermediate print of the arrays X and Y.

9. ENØM (Sequence Number 9)

Subroutine ENØM solves three related expressions:

$$C_{\gamma\beta} = \frac{-l_{11} Y'_Y(\beta r_1) - l_{12} Y_Y(\beta r_1)}{l_{11} J'_Y(\beta r_1) + l_{12} J_Y(\beta r_1)}$$

$$EHØM = \int_{r_1}^{r_2} r \left[C_{\gamma\beta} J_Y(\beta r) + Y_Y(\beta r) \right]^2 dr$$

$$STØ2 = \left[C_{\gamma\beta} J_Y(\beta r) + Y_Y(\beta r) \right] r.$$

STØ2 is actually an array of expressions. It is evaluated at a number of r's and stored in common for use in subroutine ARBCØE.

The arguments for ENØM are IG, IB, and EHØM where:

Contrails

IG = the index number of an angular eigenvalue (γ)
IB = the index number of a radial eigenvalue (β)
EHØM = the name of the location where the value of the integral
EHØM is stored

A 1 in column 9 of card B produces intermediate print of EHØM and $C_{\gamma\beta}$.

10. GAMEPS (Sequence Number 10)

Subroutine GAMEPS solves for the eigenvalues γ and ϵ . The subroutine evaluates Equations (46) through (51) to obtain γ 's and a similar set to obtain ϵ 's.

The arguments for GAMEPS are ISEEK, MANY, and VALUES, where:

ISEEK = the selection indicator for γ 's or ϵ 's
MANY = the number of eigenvalues desired
VALUES = the array name where the eigenvalues will be stored.

A 1 in column 10 of card B (above) produces intermediate print in GAMEPS of the arguments, the eigenvalues, and some of the indicators and constants used by GAMEPS.

11. GTIME (Sequence Number 11)

Subroutine GTIME solves the expression

$$\text{GSIMP} = \alpha\beta^2 \int_0^{\theta} g(\lambda) e^{-\alpha\beta^2(\theta-\lambda)} d\lambda$$

and also obtains

$$\text{GTHETA} = g(\theta)$$

where $g(\theta)$ is the time function which is input by the program user, and θ is a time at which program answers are being computed.

The arguments for GTIME are: TIME, BETA, and GSIMP, where:

TIME = θ
BETA = β
GSIMP = the name of the location where the evaluation of the integral
GSIMP (above) will be stored.

Intermediate print is not furnished for GTIME since GSIMP and GTHETA are part of the intermediate print of subroutine SERIES.

12. WURZEL (Sequence Number 12)

Subroutine WURZEL is a root searching routine which works in conjunction with subroutines FFP and BRIDGE. Its method, one of trial and error, is based on limiting root spacing of Equation (52)†.

WURZEL makes continuous guesses at an abscissa value and calls subroutine FFP to get a corresponding ordinate value of Equation (53) and slope of Equation (54). The relationship of the signs of Equations (53) and (54) on the first guess, determines whether the direction of the continued search should be forward or backward. This scheme is illustrated in Figure 13. A first guess at point A would initiate a forward search. A first guess at point B would initiate a backward search. The scheme, however, does not provide recovery from a first guess past the first finite maximum or minimum such as point C, and would subsequently miss the first root. Accordingly, an educated first guess is required. As programmed, a first guess of $0.25\alpha/r_1^*$ is used as the first guess for the first two γ 's. The first guesses for the remainder of the given set of γ 's is based on the spacing of the previously determined first roots.

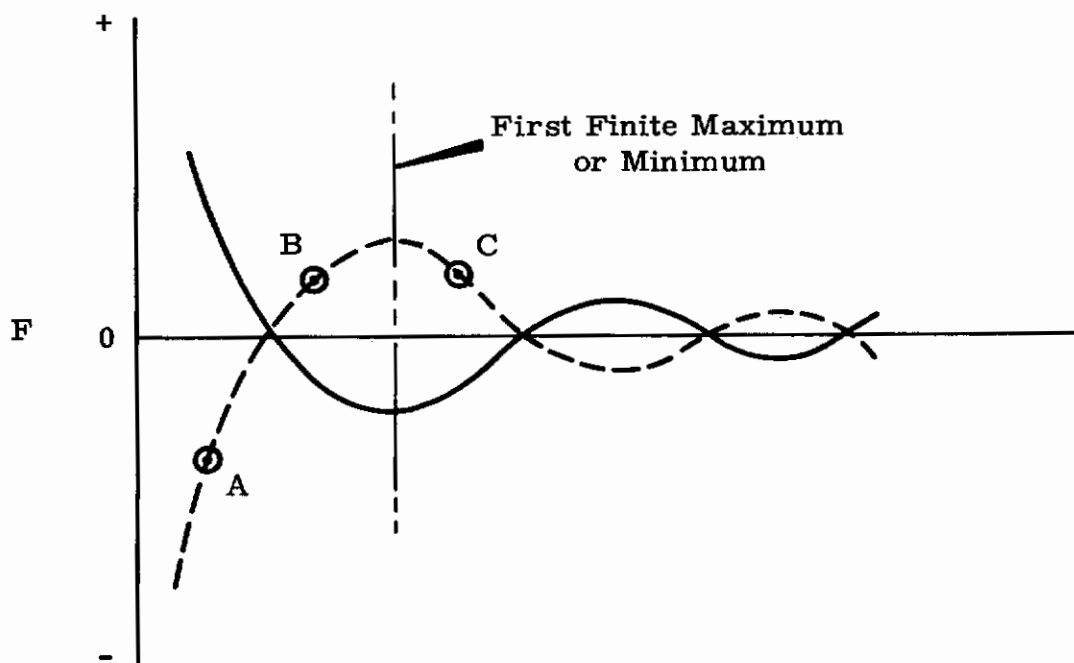


Figure 13. "First Guess" System

* For γ equals zero, the first guess is $1/r_2 - r_1$.

† The root spacing approaches $\pi/r_2 - r_1$ as β approaches infinity.

13. BRIDGE (Sequence Number 13)

Subroutine BRIDGE is a direct application of Newton's method of successive approximations (Reference 11) and the regula falsi method (Reference 2) used in root searching. The regula falsi method is applied in the coarse search phase, until it obtains two points in the immediate vicinity of, and bordering a root. Newton's method of successive approximations is then applied in the fine search phase to obtain the root within a desired accuracy.

The arguments for BRIDGE are:

IB, XH, YH, SH, XL, YL, SL, and IG

where

IB = the index number of the desired radial eigenvalue (β).

(XH, YH and SH) = the coordinates and slope respectively of a point forward of the β .

(XL, YL and SL) = the coordinates and slope respectively of a point behind the β .

IG = the index number of an angular eigenvalue (γ)

14. ABES (Sequence Number 14)

Subroutine ABES solves for Bessel functions of the second kind of the form $Y_\gamma(\beta r)$ and $Y'_\gamma(\beta r)$. It also obtains Bessel functions of the first kind of the form $J_\gamma(\beta r)$ and $J'_\gamma(\beta r)$ by a call to subroutine BES.

The arguments for ABES are B6, R6, C6, A6, AP6, Y6, and YP6, which correspond to

β , r , γ , $J_\gamma(\beta r)$, $J'_\gamma(\beta r)$, $Y_\gamma(\beta r)$, and $Y'_\gamma(\beta r)$.

If the desired functions are of non-integer order, an explicit expression for Bessel functions of the second kind, in terms of Bessel functions of the first kind, is used. This expression is

$$Y_\gamma(\beta r) = \frac{1}{\sin \gamma \pi} \left[\cos(\gamma \pi) J_\gamma(\beta r) - J_{-\gamma}(\beta r) \right]$$

Integer order functions are obtained by calculating Y_1 and Y_2 from the following equations and using recurrence formulas (see Reference 6 or 7)

Contrails

to obtain Y_n and Y_n' . The use of recurrence formulas avoids the need of special expressions for high orders and provides a scheme for obtaining the derivatives. The derivative is required by WURZEL and ENOM.

For small arguments

$$\begin{aligned} \pi Y_{\gamma}(\beta r) &= 2J_{\gamma}(\beta r) \left[\ln \frac{\beta r}{2} + \sigma \right] - \sum_{r=0}^{\infty} (-1)^r \frac{(\beta r/2)^{\gamma+2r}}{r!(\gamma+r)!} \\ &\times \left[\sum_{m=1}^{\gamma+r} m^{-1} + \sum_{m=1}^r m^{-1} \right] \\ &- \sum_{r=0}^{\gamma-1} (\beta r/2)^{-\gamma+2r} \frac{(\gamma-r-1)!}{r!} \end{aligned}$$

where σ is Euler's constants.

For larger arguments (βr)

$$\begin{aligned} \pi Y_{\gamma}(\beta r) &\sim \left(\frac{2\pi}{\beta r}\right)^{\frac{1}{2}} \sin\left(\beta r - \frac{1}{2}\gamma\pi - \frac{1}{4}\pi\right) \sum_{m=0}^{\infty} \frac{(-1)^m f_{\gamma}(\gamma, 2m)}{(2\beta r)^{2m}} \\ &+ \cos\left(\beta r - \frac{1}{2}\gamma\pi - \frac{1}{4}\pi\right) \sum_{m=0}^{\infty} \frac{(-1)^m f_{\gamma}(\gamma, 2m+1)}{(2\beta r)^{2m+1}} \end{aligned}$$

where

$$f_{\gamma}(a, b) = \frac{\Gamma(a+b+\frac{1}{2})}{b! \Gamma(a-b+\frac{1}{2})} .$$

15. GAMMA (Sequence Number 15)

Subroutine GAMMA is the other of the two routines used in Reference 4. GAMMA calculates GAMMA functions for use in Subroutine BES.

The arguments for GAMMA are X and Y where

$$Y = \Gamma(X).$$

16. AMSAFC (Sequence Number 16)

Function subprogram AMSAFC solves the expression

Contrails

$$\text{AMSAFC} = M_{\epsilon} \sin (\epsilon \ln r) + F_{\epsilon} \cos (\epsilon \ln r)$$

The arguments for AMSAFC are IG and V,

where,

IG = the index number of the ϵ

V = r

the values of M_{ϵ} , F_{ϵ} , and ϵ are stored in common. A 1 placed in column 16 of card B will print the expression AMSAFC as intermediate print.

17. SERIES (Sequence Number 17)

Subroutine SERIES solves the equation

$$T(\theta, r, \phi) = \text{TERMA} + \text{TERMB} + g(\theta) \times f(r, \phi)$$

where TERMA is defined by the double summation of Equation (44) and TERMB is defined by the double summation of Equation (83).

The arguments for SERIES are: TIME, R, and P, which correspond to θ , r, and ϕ .

The intermediate print for SERIES is in two forms. A 1 in column 17 of card B prints $T(\theta, r, \phi)$, before it is inverted, and the corresponding steady state solution $f(r, \phi)$.

A 2 placed in the same column prints the same information as for a 1, as well as the terms in the series, the time function, the eigenvalues, and the partial sums.

18. STATE (Sequence Number 18)

Subroutine STATE solves the steady state portion of the problem. In particular, it solves the equation

$$f(r, \phi) = \Sigma T_1 + \Sigma T_2 + \Sigma T_3 + \Sigma T_4$$

where

T_1 = term defined in (100)

T_2 = term defined in (110)

T_3 = term defined in (120)

T_4 = term defined in (132).

The intermediate print in STATE is obtained by placing a 1 in column 18 of card B.

The printout consists of T_1 , T_2 , T_3 , T_4 and the partial sum of the four series for each eigenvalue.

19. XI (Sequence Number 19)

Function subprogram XI solves for the ξ expressions in Equations (101), (111), (122), and (133) in the steady state solution. For example, at boundary r_1 , XI solves Equation (101). However, special cases in Equation (107), (108), and (109) can be obtained by setting

$$\xi_{r1} = 1.0,$$

$$\xi_{r1} = \ln(r/r_2) / \ln r_1/r_2,$$

or

$$\xi_{r1} = r_1 \ln(r/r_2).$$

depending upon the boundary condition indicators.

The arguments for XI are: IG, V, and NBOUND,

where

IG = the index number of an eigenvalue

V = either r or ϕ depending upon which boundary is being requested

NBOUND = the number of the boundary being requested (1, 2, 3, or 4 for boundaries r_1 , r_2 , ϕ_1 , or ϕ_2 , respectively).

A 1 in column 19 of card B will produce intermediate print of the value of ξ and the boundary which has been evaluated.

20. FFP (Sequence Number 20)

Subroutine FFP calculates the functions

$$F = \begin{vmatrix} T_1 & T_2 \\ T_3 & T_4 \end{vmatrix} \quad (53)$$

and

$$\frac{\partial F}{\partial \beta} = FP = \begin{vmatrix} T_1 & T_2 \\ \frac{\partial T_3}{\partial \beta} & \frac{\partial T_4}{\partial \beta} \end{vmatrix} + \begin{vmatrix} \frac{\partial T_1}{\partial \beta} & \frac{\partial T_2}{\partial \beta} \\ T_3 & T_4 \end{vmatrix} \quad (54)$$

for a particular value of the independent variable β for a given γ . Subroutine WURZEL and BRIDGE use this subroutine (FFP) in searching for roots of equation (52).

The arguments for FFP are: IG, B, F, and FP

where:

IG = the index number of the angular eigenvalue (γ).

B = the independent variable in equation (53) and (54).

F ~ defined above in equation (53)

FP ~ defined above in equation (54)

Intermediate print in FFP is of two forms:

- 1 A 1 in column 20 of card B (above) prints out B, F, FP, and IG.
- 2 A 2 in column 20 of card B prints out the elements of the determinant F, and their derivatives and the Bessel functions necessary for these computations as well as B, F, FP, and IG.

21. ARBCØE (Sequence Number 21)

Subroutine ARBCØE solves the equations for $A_{\gamma\beta}$, Equation (62), and $G_{\gamma\beta}$, Equation (80). $A_{\gamma\beta}$ and $G_{\gamma\beta}$ are the arbitrary coefficients of the transient solution. Special cases of the arbitrary coefficients emerge automatically in ARBCØE due to the evaluation of composite expressions in supporting subroutines and subprograms.

A 1 in column 21 of card B produces intermediate print of the arbitrary coefficients.

22. TRAN (Sequence Number 22)

Function subprogram TRAN transforms a variable according to the special cases of Table I.

The single argument of TRAN is V which is the variable to be transformed according to Equation (11).

A 1 in column 22 of card B produces intermediate print of the variable V, the transform variable T, and the constants and intermediate expressions used in the transform calculations.

23. BSNFCS (Sequence Number 23)

Function subprogram BSNFCS solves the expression

$$\text{BSNFCS} = B_{\gamma} \sin (\gamma \phi) + F_{\gamma} \cos (\gamma \phi)$$

The arguments of BSNFCS are IG and P

where

IG = the index number of an eigenvalue (γ)

P = ϕ

B_{γ} and F_{γ} are obtained through common.

A 1 in column 23 of card B produces intermediate print of the expression BSNFCS.

24. CØECØE (Sequence Number 24)

Subroutine CØECØE solves Equations (102), (112), (123) and (134) for the arbitrary coefficients of the steady state solution, P_γ , L_γ , Q_ϵ and W_ϵ , respectively. Special cases of the arbitrary coefficients resulting from B_γ or M_ϵ being infinite, or γ or ϵ being zero, are incorporated in CØECØE through supporting subroutines and subprograms.

The arguments for CØECØE are: IG, CP, CL, CQ, and CW where: IG = the index number of an eigenvalue, and CP, CL, CQ, and CW correspond to P_γ , L_γ , Q_ϵ and W_ϵ respectively.

Intermediate print for CØECØE consists of printout of the arbitrary coefficients calculated in each case, and it is obtained by placing a 1 in column 24 of card B.

25. CENØM (Sequence Number 25)

Subroutine CENØM solves the integral:

$$CEHØM = \int_{r_1}^{r_2} \left[M_\epsilon \sin(\epsilon \ln r) + F_\epsilon \cos(\epsilon \ln r) \right]^2 \frac{dr}{r}$$

which has special cases of

$$CEHØM = \left[\frac{\ln r}{2} - \frac{1}{4\epsilon} \sin(2\epsilon \ln r) \right]_{r_1}^{r_2}$$

for an infinite M_ϵ , and

$$CEHØM = \ln(r_2/r_1)$$

when ϵ equals zero.

Subroutine CENØM also solves Equation (124) for M_ϵ and sets F_ϵ equal to zero for special cases.

The arguments to CENØM are: IG, and CEHØM where

IG = the index number of an eigenvalue and

CEHØM = the name of the location where the value of the integral
CEHØM above will be stored.

The values of M_ϵ and F_ϵ are stored in common.

A 1 in column 25 of card B produces intermediate print of IG, CEHØM,
and M_ϵ .

26. ATRAN (Sequence Number 26)

Function subprogram ATRAN is the inversion routine which corresponds
to the transformation routine TRAN. (See function subprogram TRAN.)

The single argument of ATRAN is T which is the transform variable
to be inverted to real temperature.

A 1 in column 26 of card B produces print similar to the intermediate
print of function subprogram TRAN.

27. DENØM (Sequence Number 27)

Subroutine DENØM solves the integral:

$$DEHØM = \int_{\phi_1}^{\phi_2} [B_\gamma \sin(\gamma\phi) + F_\gamma \cos(\gamma\phi)]^2 d\phi$$

which has special cases of:

$$DEHØM = \left[\frac{\phi}{2} - \frac{1}{4\gamma} \sin(2\gamma\phi) \right]_{\phi_1}^{\phi_2}$$

for B_γ of infinity and

$$DEHØM = \phi_2 - \phi_1$$

for γ equal to zero

DENØM also solves Equation (60) for B_γ and sets F_γ equal to zero for
special cases. F_γ is similar to F_ϵ in subroutine CENØM.

The arguments for DENØM are IG and DEHØM

where:

IG = the index number of an eigenvalue and

DEHØM = the name of the location where the value of the integral
DEHØM above will be stored.

The values of B_γ and F_γ are stored in common. Intermediate print in DENØM consists of the values of DEHØM, IG, and B_γ . To obtain this print a 1 is placed in column 27 of card B.

28. INPUT (Sequence Number 28)

Subroutine INPUT reads in and prints out the problem description and the necessary supporting data for a problem solution. It also serves the needs of the other subroutines by putting raw data in a form usable to the transformation of the thermophysical properties, and each of the boundary condition functions is done by subroutine INPUT. Communication from INPUT to the remainder of the program is accomplished through COMMON.

Subroutine INPUT checks all of the input data for such things as sequence, mode, form, and limits. Together with the computer FORTRAN monitor, this provides a good level of assurance to the program user that computer time will not be expended on an erroneous problem. However, this is by no means complete protection.

**APPENDIX D
FLOW DIAGRAMS**

Contrails

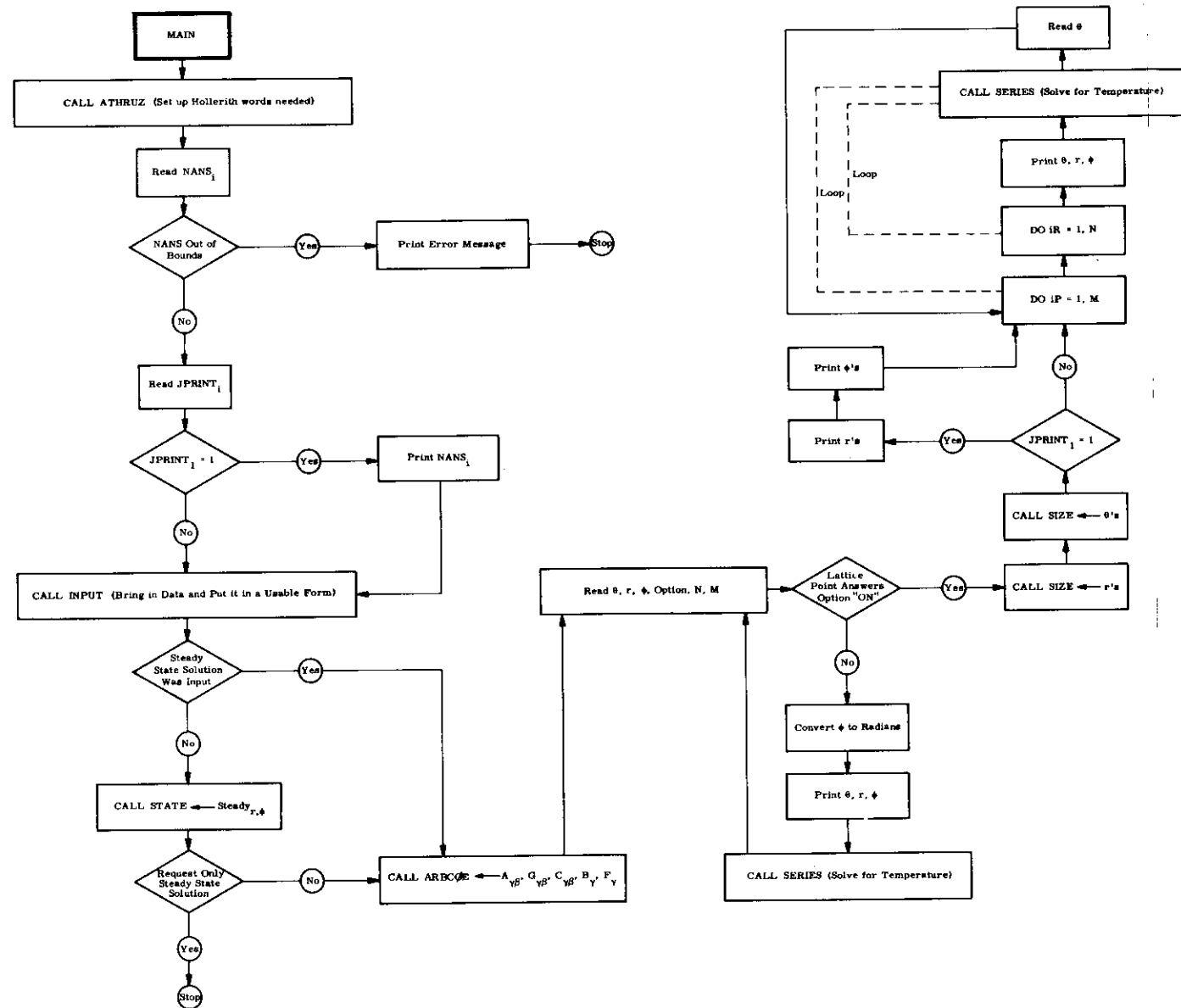


Figure 14. Flow Diagram, MAIN Routine

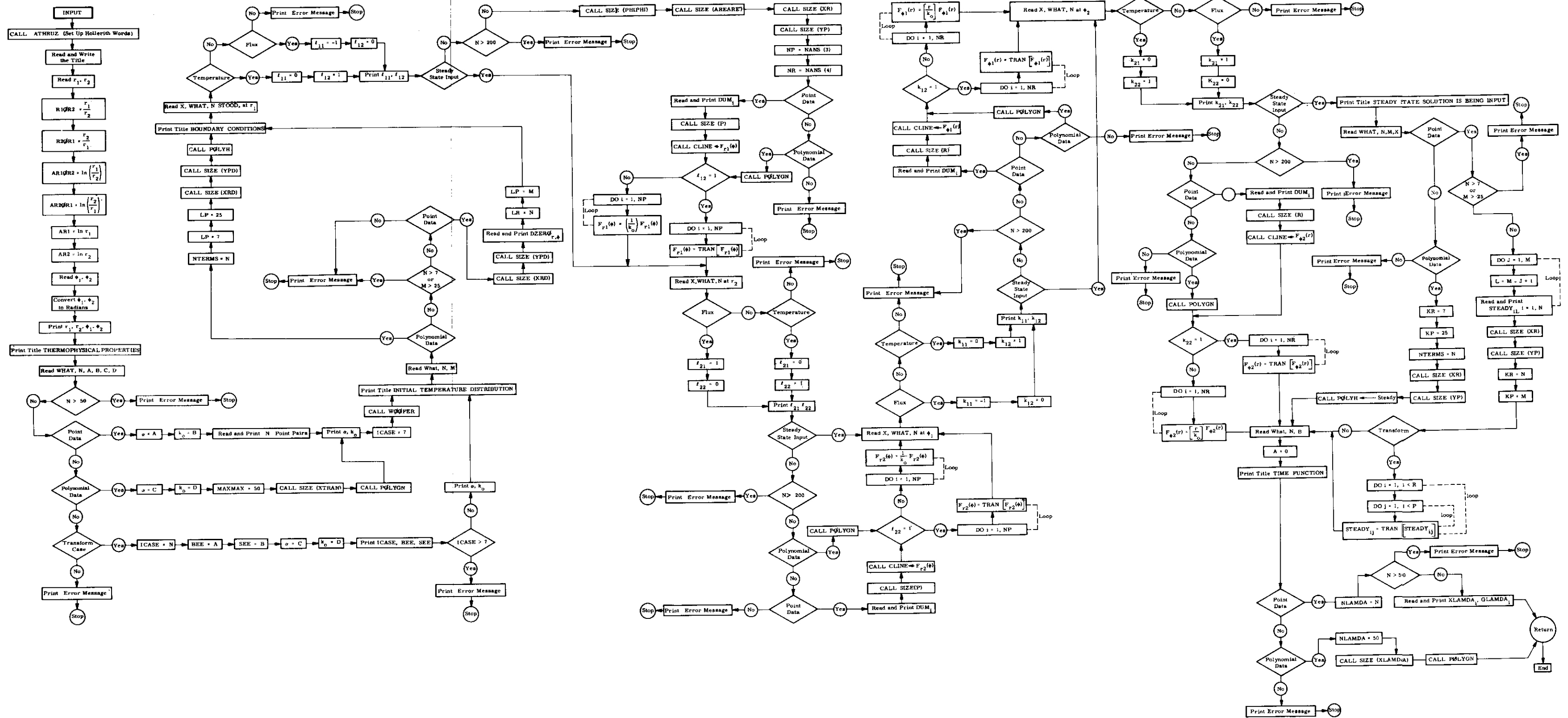


Figure 15. Flow Diagram, Subroutine INPUT

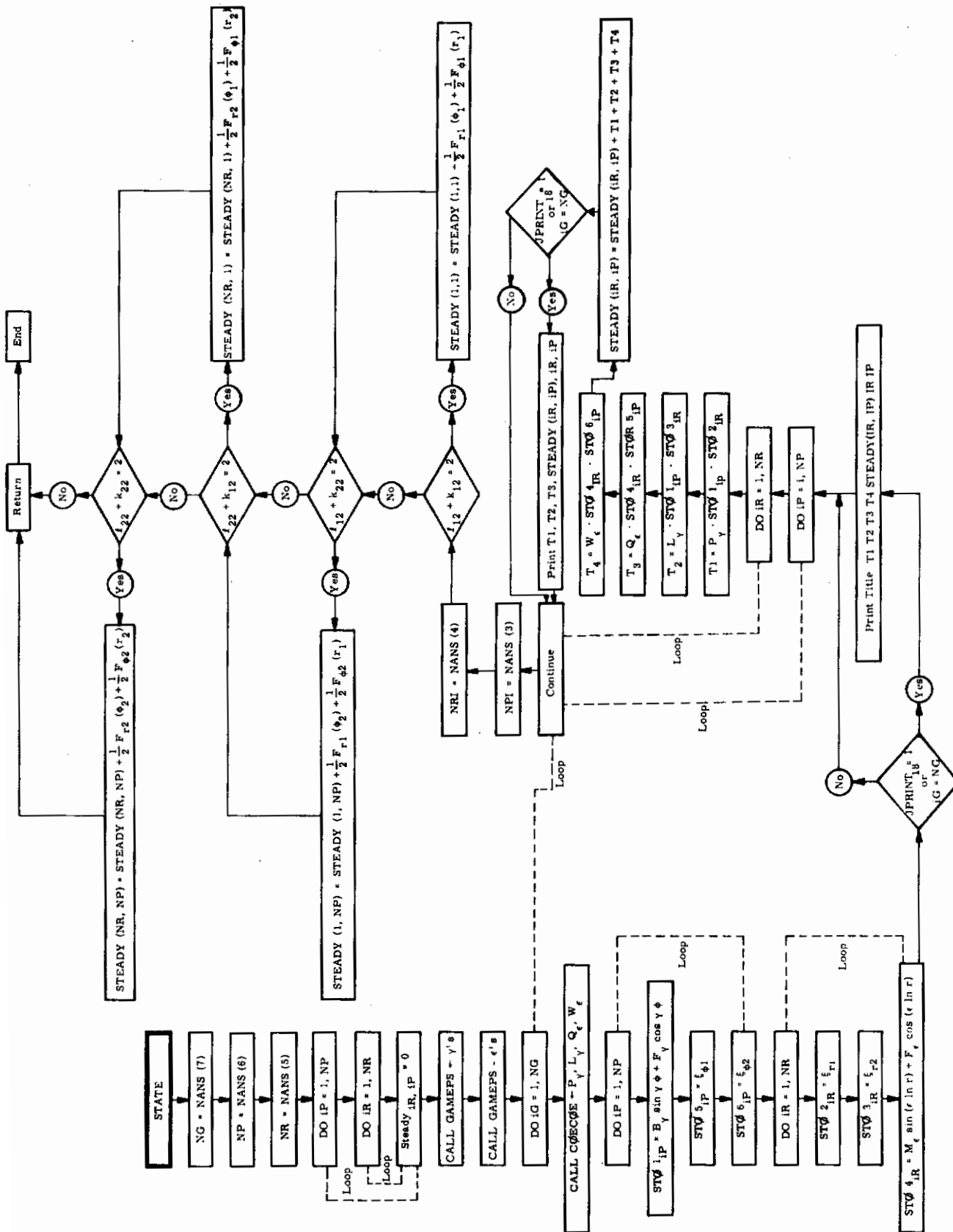


Figure 16. Flow Diagram, Subroutine STATE

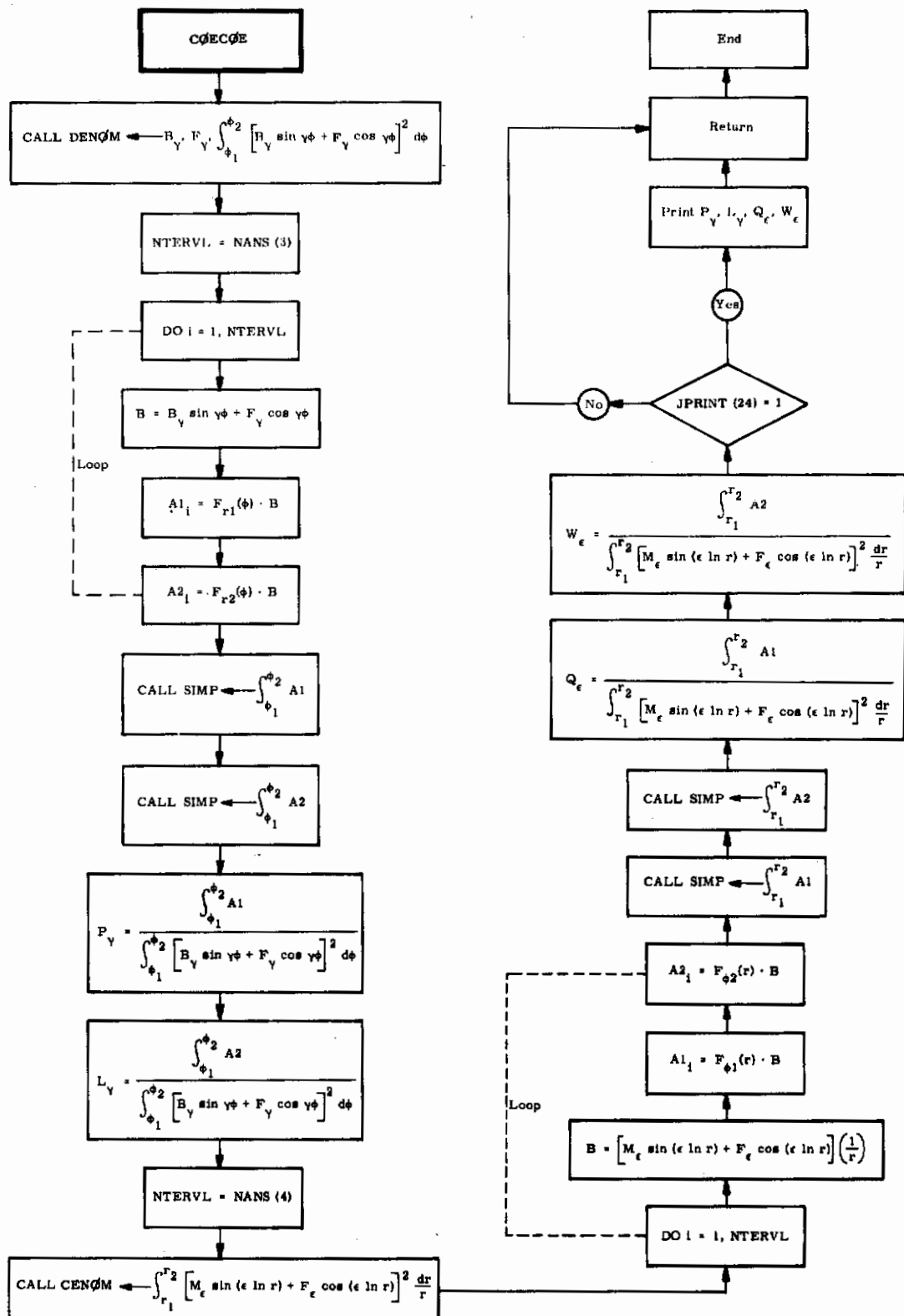


Figure 17. Flow Diagram, Subroutine CØECØE

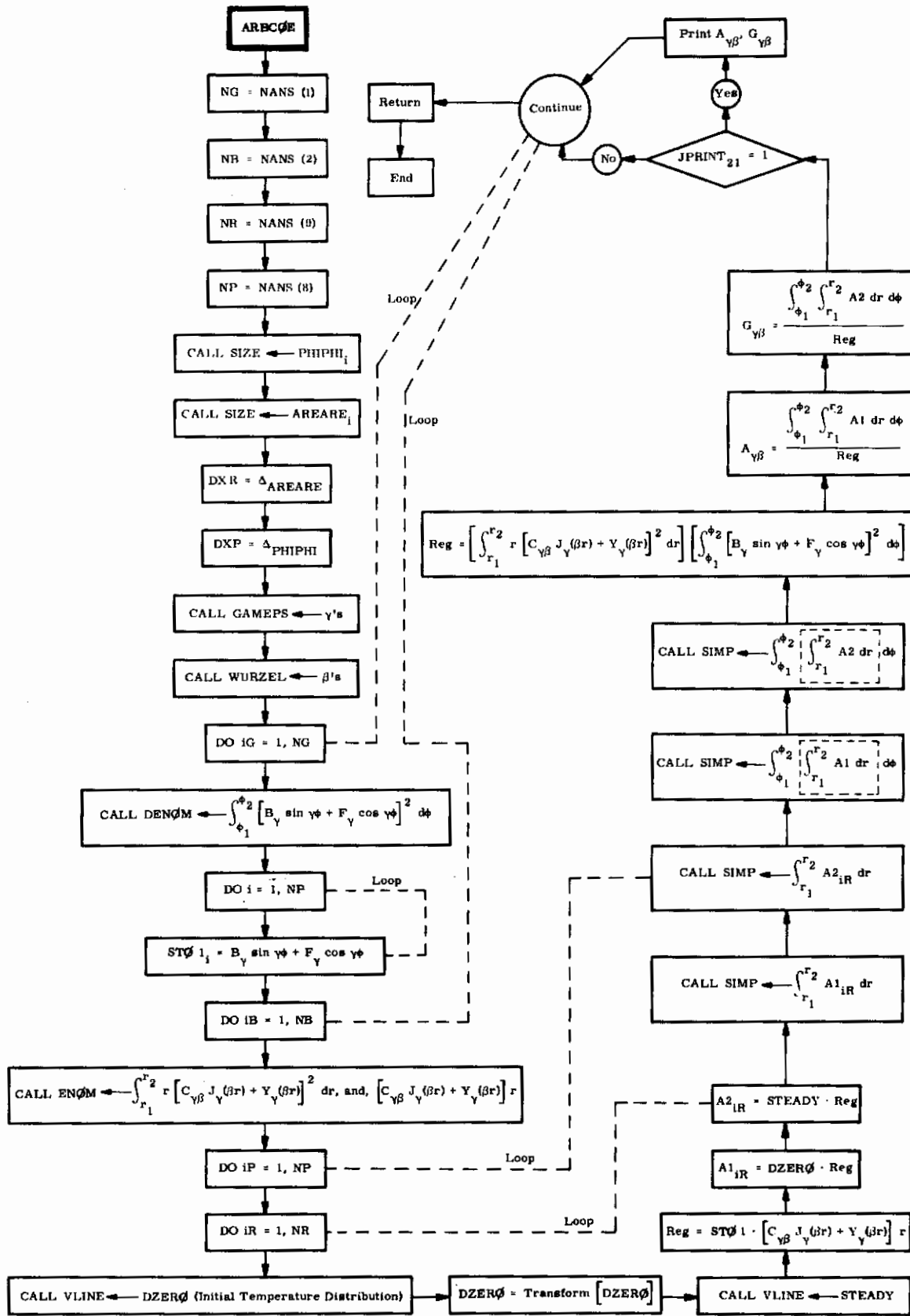


Figure 18. Flow Diagram, Subroutine ARBCOE

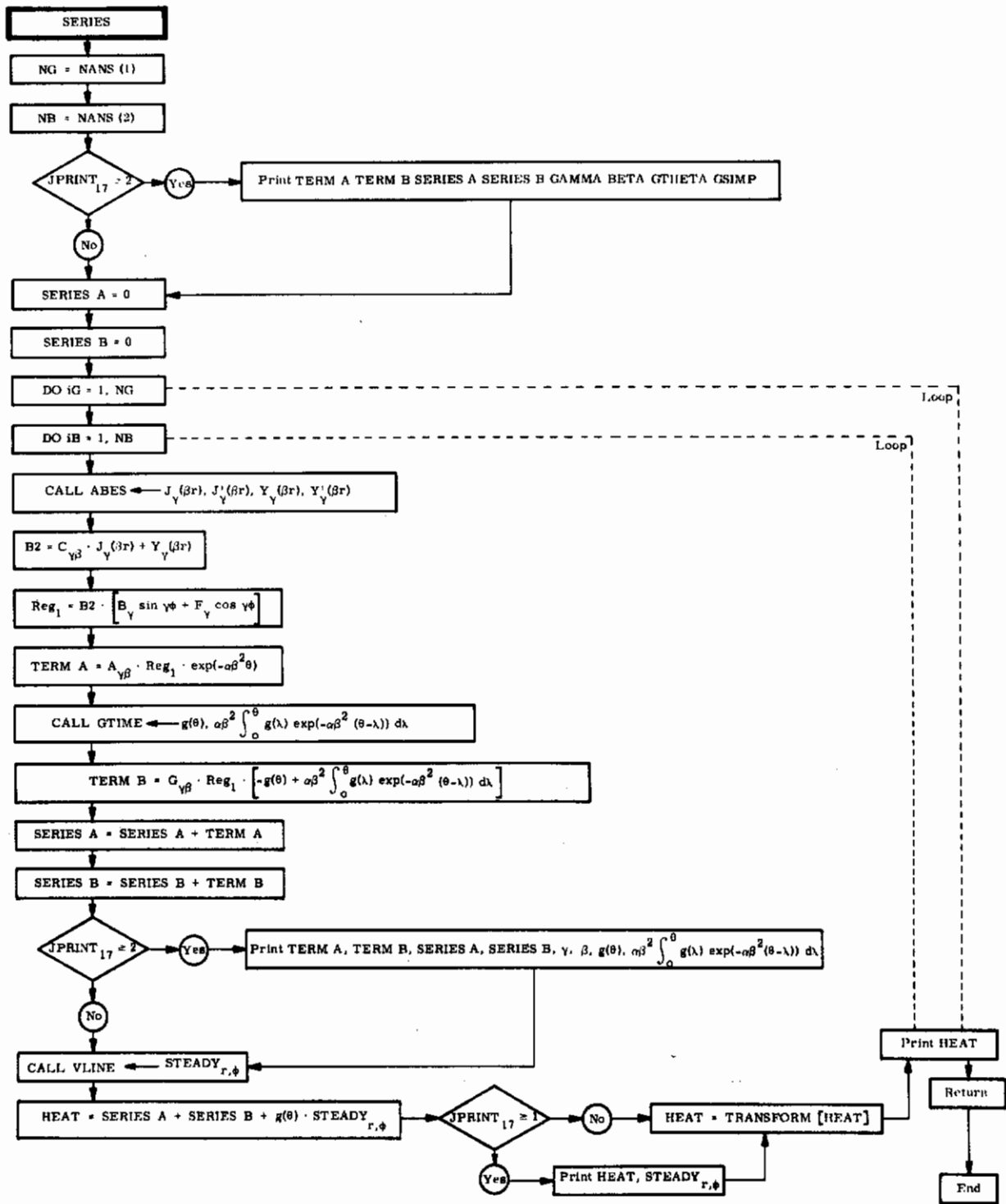


Figure 19. Flow Diagram, Subroutine SERIES

APPENDIX E
SAMPLE TEST PROBLEM

Contrails

APPENDIX E

SAMPLE TEST PROBLEM

In addition to the sample problems presented in preceding sections, a series of 12 test problems were run on the computer at Wright-Patterson Air Force Base. Similar problems were calculated for comparison using the Thermal Analyzer Computer program (Reference 13).

This appendix presents a typical test problem and a comparison between the two programs. The specific problem is the determination of the temperature response of the center radial line in a 60 degree sector with zero initial temperature distribution, a temperature of $100 \sin 6\theta$ at the surface r_2 , and the remaining surfaces insulated. Print out of the input and solution obtained with the computer program developed in this report is shown in Table VII. Data from both programs is shown in Figure 20.

TABLE VII

Program Printout, Sample Test Problem

SP 8

INPUT

GEOMETRY

R1= 0.5000000E 00 R2= 0.8000000E 00
P1= 0. P2= 0.10471974E 01 (RADIAN)

THERMO PHYSICAL PROPERTIES

TRANSFORM ACCORDING TO CASE 4,B= 0.9999999E-02 C= 0.
ALPHA= 0.4000000E-04 KZERO= 0.2310000E-04

INITIAL TEMPERATURE DISTRIBUTION

POLYNOMIAL DATA, 1 TERMS
(0.) (R** 0.) (P** 0.)

BOUNDARY CONDITIONS

AT R1, FLUX L11= -1.00 L12= 0.

POLYNOMIAL DATA, 1 TERMS, X=DUMMY VARIABLE
(0.) (X** 0.)

AT R2, TEMP L21= 0. L22= 1.00

POLYNOMIAL DATA, 5 TERMS, X=DUMMY VARIABLE
(0.3000000E 03)(X** 1.0000)
(-0.4500000E 03)(X** 3.0000)
(0.2025000E 03)(X** 5.0000)
(-0.4349999E 02)(X** 7.0000)
(0.5399999E 01)(X** 9.0000)

AT P1, FLUX K11= -1.00 K12= 0.

POLYNOMIAL DATA, 1 TERMS, X=DUMMY VARIABLE
(0.) (X** 0.)

AT P2, FLUX K21= 1.00 K22= 0.

POLYNOMIAL DATA, 1 TERMS, X=DUMMY VARIABLE
(0.) (X** 0.)

TIME FUNCTION

POLYNOMIAL DATA, 1 TERMS, X=DUMMY VARIABLE
(0.0999999E 01)(X** 0.)

EVALUATE POLYNOMIAL FROM 0. TO 0.3000000E 03

UNDRFLOW AT 32300 IN AC AND MQ.
UNDRFLOW AT 32300 IN AC AND MQ.

Table with 7 columns: T1, T2, T3, T4, STEADY(IR,IP), IR, IP. It contains numerical data for various time points and boundary conditions.

Contrails

-0.	-0.49741732E-00	-0.	0.	0.14672258E 03	6	6
0.	0.17355134E-08	-0.	0.	0.92751843E 02	1	7
0.	0.10129473E-06	0.	-0.	0.93741185E 02	2	7
0.	0.72796263E-05	0.	-0.	0.96709935E 02	3	7
0.	0.35239718E-03	0.	-0.	0.10212471E 03	4	7
0.	0.12285435E-01	-0.	0.	0.11087208E 03	5	7
0.	0.32466700E-00	-0.	0.	0.12407804E 03	6	7
0.	0.32616902E-08	-0.	0.	0.87400795E 02	1	8
0.	0.19037136E-06	0.	-0.	0.87130707E 02	2	8
0.	0.13681188E-04	0.	-0.	0.86409051E 02	3	8
0.	0.66228844E-03	0.	-0.	0.85421223E 02	4	8
0.	0.23089010E-01	-0.	0.	0.84641580E 02	5	8
0.	0.61017287E 00	-0.	0.	0.85295413E 02	6	8
-0.	-0.60274187E-09	-0.	0.	0.82602119E 02	1	9
-0.	-0.35179549E-07	0.	-0.	0.81123461E 02	2	9
-0.	-0.25282060E-05	0.	-0.	0.76720396E 02	3	9
-0.	-0.12238714E-03	0.	-0.	0.68821730E 02	4	9
-0.	-0.42667181E-02	-0.	0.	0.56572578E 02	5	9
-0.	-0.11275648E-00	-0.	0.	0.40928707E 02	6	9
-0.	-0.34710202E-08	0.	0.	0.80681277E 02	1	10
-0.	-0.20258909E-06	0.	-0.	0.78692923E 02	2	10
-0.	-0.14559225E-04	0.	-0.	0.72672613E 02	3	10
-0.	-0.70479304E-03	0.	-0.	0.61380392E 02	4	10
-0.	-0.24570825E-01	-0.	0.	0.41679476E 02	5	10
-0.	-0.64933278E 00	-0.	0.	0.46839292E 01	6	10

TIME=	0.30000000E 02	R=	0.52500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.73015690E-02		
TIME=	0.59999999E 02	R=	0.52500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.31241775E-01		
TIME=	0.12000000E 03	R=	0.52500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.99881590E 00		
TIME=	0.20000000E 03	R=	0.52500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.59490845E 01		
TIME=	0.30000000E 03	R=	0.52500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.14803585E 02		
TIME=	0.30000000E 02	R=	0.57500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		-0.12274235E-00		
TIME=	0.59999999E 02	R=	0.57500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.56388974E-01		
TIME=	0.12000000E 03	R=	0.57500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.32958269E 01		
TIME=	0.20000000E 03	R=	0.57500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.11141579E 02		
TIME=	0.30000000E 03	R=	0.57500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.20894477E 02		
TIME=	0.30000000E 02	R=	0.62500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		-0.39322674E-00		
TIME=	0.59999999E 02	R=	0.62500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.14172167E 01		
TIME=	0.12000000E 03	R=	0.62500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.10520564E 02		
TIME=	0.20000000E 03	R=	0.62500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.22130169E 02		
TIME=	0.30000000E 03	R=	0.62500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.32333116E 02		
TIME=	0.30000000E 02	R=	0.67500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.11884510E 01		
TIME=	0.59999999E 02	R=	0.67500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.10009040E 02		
TIME=	0.12000000E 03	R=	0.67500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.26531497E 02		
TIME=	0.20000000E 03	R=	0.67500000E 00	PHI=	0.52359870E 00 (RADIANS)
	TEMPERATURE=		0.39367791E 02		

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TIME= 0.3000000E 03	R= 0.6750000E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.48302764E 02		
TIME= 0.3000000E 02	R= 0.7249999E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.17429347E 02		
TIME= 0.5999999E 02	R= 0.7249999E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.35588296E 02		
TIME= 0.1200000E 03	R= 0.7249999E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.52197002E 02		
TIME= 0.2000000E 03	R= 0.7249999E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.61719963E 02		
TIME= 0.3000000E 03	R= 0.7249999E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.67554465E 02		
TIME= 0.3000000E 02	R= 0.7750000E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.67873026E 02		
TIME= 0.5999999E 02	R= 0.7750000E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.76943004E 02		
TIME= 0.1200000E 03	R= 0.7750000E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.83214576E 02		
TIME= 0.2000000E 03	R= 0.7750000E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.86429211E 02		
TIME= 0.3000000E 03	R= 0.7750000E 00	PHI= 0.52359870E 00 (RADIAN)
TEMPERATURE= 0.88313639E 02		

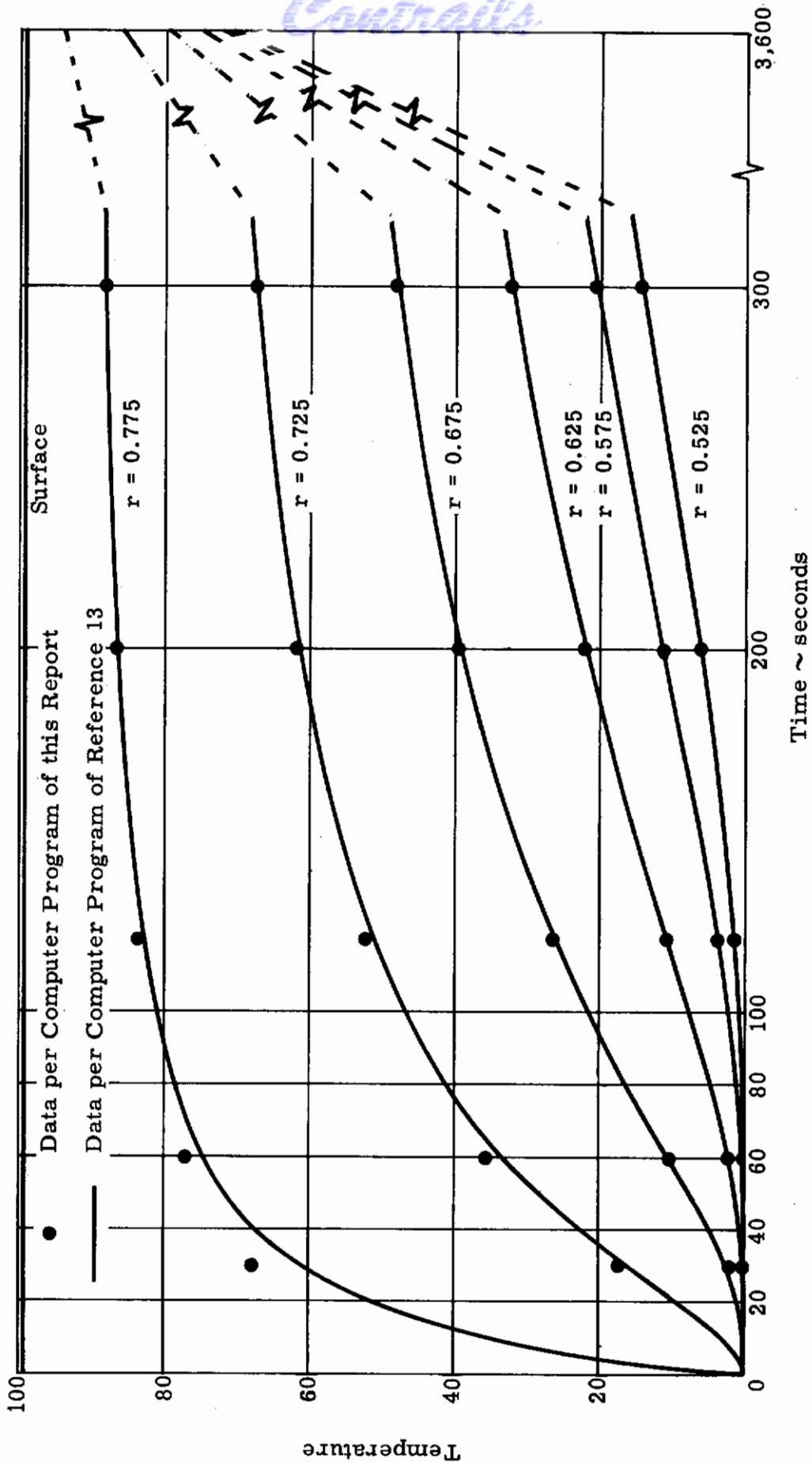


Figure 20. Temperature History, Sample Test Problem

Contrails

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13. ABSTRACT This study was undertaken to develop an analytical method of determining temperature distribution in a sector of a hollow circular cylinder with temperature dependent thermophysical properties based on exact mathematical solution. The subject nonlinear differential equation is reduced to a linear differential equation by an integral transform and with the assumption that only the thermal conductivity is a function of temperature whereas the thermal diffusivity remains constant. The solution has been programmed in FORTRAN IV for application to an IBM 7094. In addition, a successive transform applicable to the solution of similar problems involving temperature dependent thermal diffusivity is illustrated and discussed.			

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