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FINAL REPORT ON REFRACTORY METAL COATINGS
BY CHEMICAL VAPOR DEPOSITION, final report.

Gene F. Wakefield

Texas Instruments Incorporated.

TECHNICAL REPORT AFML-TR-66-397

December 1966

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FINAL REPORT ON REFRACTORY METAL COATINGS BY CHEMICAL VAPOR DEPOSITION

Gene F. Wakefield

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FOREWORD

This Final Technical Documentary Report was prepared by the Materials Research and Development Laboratory, Materials and Controls Division of Texas Instruments Incorporated. The work was initiated by the Chemical Processing Branch (MATC) of the Manufacturing Technology Division, Air Force Materials Laboratory, Research and Technology Division, under Contract AF 33 (615)-3046, Project 8-264. The program was accomplished under the technical direction of Mr. G. E. Eichelman and Mr. W. P. Johnson, Chemical Processing Branch, Manufacturing Technology Division, Wright-Patterson Air Force Base, Ohio. The manuscript was released by the authors on 28 November 1966, for publication as an RTD Technical Report.

The work described, to develop the chemical vapor deposition process for application of refractory metal coatings, was performed during the period from 15 May 1965 through 15 September 1966. The contract was managed by Dr. Gene F. Wakefield of Texas Instruments Materials and Controls Division, Dallas, Texas. Contributions to the effort were made by D. R. Williams, C. Cockrum, W. Parker and R. Bracken. Texas Instruments report number is 04-66-36.

Closely related efforts were covered under MMP Projects 8-346, "Refractory Metal Coatings by the Fused Salt Process", 8-184, "Fluidized Bed Techniques for Coating Refractory Metals", and 8-344, "Electrophoretic Deposition of Refractory Metal Coatings".

This project has been accomplished as a part of the Air Force Manufacturing Methods Program, the primary objective of which is to develop, on a timely basis, manufacturing processes, techniques and equipment for use in economical production of USAF materials and components. The program encompasses the following technical areas:

Metallurgy - Rolling, Forging, Extruding, Casting, Fiber, Powder
Chemical - Propellant, Coating, Ceramic, Graphite, Nonmetallics
Electronic - Solid State, Materials & Special Techniques, Thermionics
Fabrication - Forming, Material Removal, Joining, Components

Suggestions concerning additional manufacturing methods development required on this or other subjects will be appreciated.

This technical report has been reviewed and is approved.

Melvin E. Vields

MELVIN E. FIELDS, Colonel, USAF Chief, Manufacturing Technology Division Air Force Materials Laboratory

<u>ABSTRACT</u>

The basic processes necessary to manufacture refractory metal coatings using chemical vapor deposition technology were investigated. The processes developed were demonstrated on a typical oxidation resistant coating of titanium-chromium-silicon but would have general applicability to other similar coatings. The processes utilized both two-stage deposition (metal alloy coating followed by silicon deposition) and a new method developed for the rapid (10 minute) deposition of all three elements in a single step. The processes were used for deposition of metallic coatings and for controlled variation of the composition of titanium and chromium alloys. Substrate property degradation and interaction were minimal for all types of coatings.

Performance of the coatings was evaluated by oxidation testing in air and was found to be nominally comparable to coatings by other processes. Reliability and reproducibility, indicated by statistical analysis of results, were high for this initial effort program.

The processes have potentials for utilization in rapid and economical manufacturing of a variety of coatings on parts with complex shapes.

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SECTION I

INTRODUCTION

The general purpose of this program was to investigate a pure chemical vapor deposition process adaptable to any common refractory metal and designed for improved control of process variables, with the ultimate goal of providing uniform and impervious coatings with a high degree of reproducibility. A process was designed and developed to apply a coating of titanium, chromium, and silicon to the refractory alloy B-66. The effects of process variables on coating performance and coating-substrate system properties were related and the potential for coating more complex shapes evaluated. The potential of the process for scale-up to coating larger and more complex shapes was indicated.

A. Background

The aerospace and aircraft industry's need for materials having high strength at elevated temperatures has stimulated widespread interest in the application of the refractory metal alloys. This is because one of the biggest problems facing designers of aircraft and aircraft engine components is the selection of proper materials to withstand increasingly severe demands of temperature and oxidizing environment to which they will be subjected. Unfortunately, the number of materials from which to choose has been quite limited. Many times a compromise choice or modified performance must be accepted. The refractory metals tantalum, molybdenum, columbium, and tungsten are especially desirable for applications requiring high strength at elevated temperature, particularly where ease of fabrication and ductility are needed. However, before any of these materials can satisfy a wide range of requirements, it must receive an oxidation-resistant coating.

Considerable effort has been expended in developing such coatings. The relatively few coatings now available indicate that this development work has proceeded slowly and in several directions, with major success in a coating of titanium, chromium, and silicon.

Much work has been done in developing methods of applying coatings and in deposition process studies. In programs at the University of Dayton* and at Solar, a division of International Harvester Company,† coating performance was found to be less predictable than desired. In some cases, the nonreproducible results tended to obscure or to limit the observation of interaction of the variables under investigation. Though many techniques -- e.g., pack cementation, fluidized bed, plasma spray, and fused salt -- give coatings on large, complex shapes, a greater degree of control would aid in achieving higher reproducibility.

A key to satisfying this requirement is to develop a technique by which process variables can be rigidly and independently controlled. One technique that offers this advantage is pure "chemical vapor deposition" (CVD). This concept is certainly not new in the field of refractory metals coatings, since pack cementation, the vacuum pack method, and the fluidized bed process are all forms of CVD. However, in all these techniques the vapor phase is, by design, interrupted by a solid phase of material normally acting as the source of vapor contacting the surface of the substrate to be coated. Because the reaction zone is inhomogeneous, control of critical process parameters is often limited. In the pack cementation or vacuum pack method, system response to a controlled change of furnace temperature is very slow, which means that this parameter's effect on the reaction can be only coarsely controlled. Also, since the active material is contained within the pack, any controlled temperature change will necessarily influence the other process variables, especially the composition or concentration of the vapor phase. On the other hand, if the source of the chemical vapor contacting the substrate surface were retained outside the reaction zone, any temperature change in the system could be counterbalanced by adjusting or controlling the outside vapor source. On the same basis, vapor composition could be changed independently of any other variable, including temperature. The fused sait deposition method is, like CVD, a homogeneous reaction medium and might also result in improved coating uniformity and reliability in performance. To investigate the fused salt approach a program, also with the

^{*} Air Force Contract No. AF 33(615)-1312.

 $^{^{\}dagger}$ Air Force Contract No. AF 33(657)-9443.

goal of improved reliability, is being conducted by MATC at Solar, a division of International Harvester*.

Inherent in the CVD process is the ability to form an overlay coating. This is done, as illustrated in Figure 1, by applying a coating to a substrate and achieving bonding with only minimum interaction or interdiffusion between coating and substrate. In many cases an interlayer might be necessary between coating and substrate to serve as a diffusion barrier or as a bonding layer compatible with both substances, or to aid in matching thermal expansion. However, use of such a layer does not alter the basic concept of an overlay coating with minimum interaction with the substrate. This overlay principle is followed routinely in the semiconductor industry, and an example of its possibilities is shown in Figure 2. In this case, multiple layers of silicon containing a selected dopant, namely boron or antimony, have been deposited on a silicon substrate.

Extending this overlay concept to protective coatings is considered especially significant for several reasons. First, it helps make choice of coating independent of choice of substrate and vice versa. This means that development of each new alloy may not require development of a new coating. Minimizing the interaction between coating and substrate increases the possibility that desirable substrate properties will be retained in the coated part. This is in contrast to diffusion coatings, which diminish a foil substrate thickness by a significant amount during coating. The composition of the coating does not depend on the composition of the substrate, as does the composition of a coating formed by diffusing chemical elements into a substrate. For diffused coatings which incorporate a portion of the substrate, the role of the minor alloy constituents which become involved in the coating may not be understood and may prove quite detrimental.

^{*} Air Force Contract AF 33(615)-3173

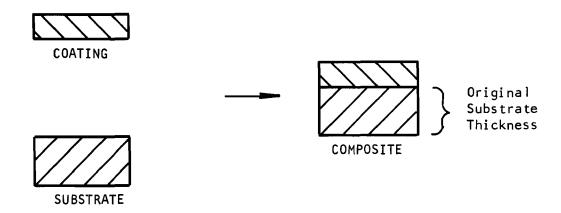
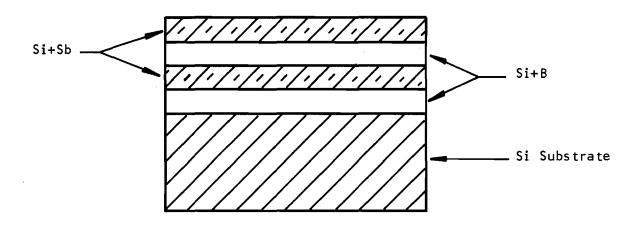


Figure 1 Overlay concept. A coating is bonded to a substrate with little interaction. The coating itself is independent of substrate composition.



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Figure 2 Illustration of a specific overlay deposition which has been made. Multiple layers of silicon containing alternate additions were deposited on a silicon substrate.

A second advantage of the overlay concept is the versatility which can be gained, not only in applying layers of varying composition, but also with respect to the structure and morphology of any specific coating. In using fairly brittle coatings on substrates which are in general more ductile, grain size and orientation assume considerable importance in mechanical compatibility. As the overlay concept is put into more practical application, the ability to develop tailormade coatings will permit better utilization of their inherent properties.

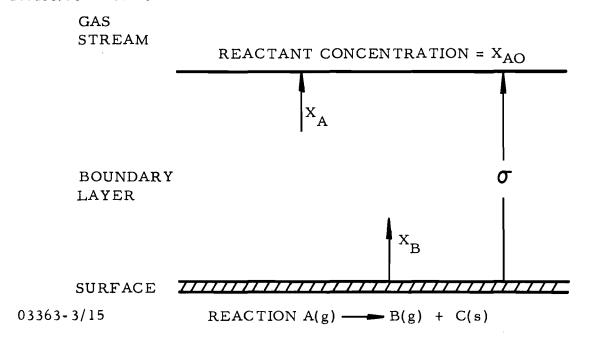
Another advantage inherent in this technique is in multicycle processes for depositing the coating, where the component need not be removed from the system between cycles. Coating refractory metal hardware by most other techniques requires a separate retort, vat, or container for each coating cycle, or the active coating materials must be removed from the equipment and replaced with different materials. However, by injecting the active materials into the reactor from an outside source, the only major operation required between cycles would be purging and cooling, if necessary, without removing the substrate. The savings in time and expense would be especially welcome in a manufacturing operation.

B. Principles Influencing Solid Formation from the Gas Phase

Deposition process mechanisms are investigated primarily to learn what parameters control coating characteristics and thus, coating performance. Such data determine how parameters in the coating process must be altered to obtain desired performance. It is this aspect of control that makes CVD an attractive and promising tool for coatings investigation and production.

In a gas-solid reaction several mechanisms may control the rate of reaction. These mechanisms affect deposit thickness and rate of formation. They also influence deposit morphology, which includes integrity, grain size, density, and whether the material deposits as dendrites or as an adherent layer.

Ideally, a gas-solid reaction can be divided into three steps, any one of which may be rate-limiting: (1) diffusion of the reactants to the surface and diffusion of products from the surface, (2) adsorption of reactants and desorption of products, and (3) reaction on the surface. The step which occurs at the slowest rate in a given series establishes the overall rate. A reaction is illustrated below.



In this instance gas phase concentration of A before reaction is given by X_{A0} . This material must diffuse toward the surface through a layer of counterdiffusing B. The thickness of this layer is δ . Once in the neighborhood of the surface the material is adsorbed. On the surface it reacts, and finally the product B is desorbed and diffuses back into the gas stream.

The most straightforward example of rate limitation is reaction rate limitation. In this case the surface is saturated with reactants. Changes in gas phase reactant concentrations and gas flow rates do not affect the reaction rate. Temperature alone can increase or retard the rate. However, in most cases of CVD, temperatures are high enough that chemical reaction rate is not the limiting step.

When adsorption is the rate-limiting step, the surface is not saturated with reactants. The adsorption equilibrium defines the rates of reaction, and the effects of reactant pressure changes are directly apparent in the rate of the reaction.

For a reaction

$$A(g) \rightarrow B(g) + C(s)$$
,

the rate of reaction, $dX_{\mbox{$\Lambda$}}/dt,$ can be expressed as a function of surface coverage [S $_{\mbox{$\Lambda$}}]$ of A:

$$-\frac{dX_A}{dt} = k[S_A].$$

Assuming a Langmuir isotherm, the surface coverage can be expressed as

$$[S_A] = k' \frac{b_A P_A}{1 + b_A P_A},$$

where $\boldsymbol{b}_{\boldsymbol{A}}$ is related to adsorption energy

 \boldsymbol{P}_{Δ} is partial pressure of A.

The rate of reaction as a function of the gas pressure of the reactant is obtained thus:

$$-\frac{dX_A}{dt} = k^{\prime\prime} \frac{b_A^P A}{1 + b_A^P A}.$$

For the special case of weakly adsorbed material (b_A small) or for low partial pressures of A (P_A small), the rate expression becomes

$$-\frac{dX_A}{dt} = k'' b_A P_A = k''' P_A.$$

That is, the reaction becomes first order in A for a sparsely covered surface.

In many cases diffusion or gas phase transport is made the rate-controlling step by the character of the adsorption isotherm, the activation energy of the reaction, or simply the necessary geometry of the system. Where forced convection has not been obtained, natural convection is the main vehicle for delivering reactant to and removing products from the surface. This usually produces nonuniform coatings with a small amount of material deposited on the work piece, collected mostly in the center. Forced convection of gases directly onto the work piece produces even deposits.

Besides the chemical reaction which yields the deposit, other factors influence the integrity of the deposit and its structural form. Obtaining integrity in deposited materials requires preventing gross flaws in the crystal structure and controlling deposit composition. Some defects can be prevented by using pure starting materials and avoiding gaseous and solid impurities. This is particularly noticeable in the deposition of titanium, where the least amount of oxygen or nitrogen in the atmosphere results in deposition of compounds rather than elemental metal.

The relative pressure (P/P_e) of the deposit also influences the rate of deposit formation and deposit morphology. This relative pressure, called the supersaturation ratio, gives a measure of the driving force that causes condensation or deposit growth on the substrate surface. As this ratio changes, the morphology changes as shown below.

Relation of Supersaturation to Deposit Morphology

P/Pe Supersaturation Homogeneous Nucleation (Fog)

Heterogeneous Nucleation (Fine-Grained Deposit)

Coarse-Grained Deposit

Oriented Deposit

[†]P is actual pressure over the deposit.

P is equilibrium vapor pressure of the deposit.

Such changes in morphology can greatly influence deposit strength and density. Low rates of deposition associated with low values of P/P_e may have to be accepted at times to obtain desired morphologies.

All these factors -- gas phase diffusion, forced convection, super-saturation, and surface saturation -- are important parameters in the gas-solid reaction. The understanding of their influence on the process and deposit and their control are essential to a reproducible, predictable coating process.

C. Method of Approach

The program to investigate and evaluate chemical vapor deposition for the general and specific goals has been carried out in three phases. Emphasis throughout the program was on developing process information useful in scale-up and in producing coatings of high reproducibility. To maximize the amount of information obtained, the series of experiments was designed, ordered, and results evaluated with the aid of statistical techniques.

1. Phase One: Coating Deposition Studies

Process parameters controlling formation of the individual alloy constituents chromium and titanium were studied. The elements were formed by the hydrogen reduction of their halides. Experiments determined the effect of processing variables on the rate and quality of the coating deposition. Information was generated on effects of the physical process variables, e.g., the relationship of substrate-jet geometry to coating uniformity, effect of flow rate on the uniformity of the deposit, and necessity of cooling samples in inert ambient. Physical, metallurgical, and chemical property analyses were used to evaluate coating composition and condition of the deposit.

Parameters controlling codeposition of the chromium and titanium alloys were investigated by determining the effects of process variables on the coating deposition rate and the quality of the deposit obtained. From evaluation of

these deposits, the most suitable process parameters were selected for codeposition of the chromium-titanium layers for subsequent use in the siliconizing studies.

The chromium-titanium layer was siliconized by hydrogen reduction of the silicon halide SiCl₄. These experiments used the most suitable chromium-titanium coated columbium alloy substrates developed in the earlier part of Phase One. Various chromium-to-titanium ratios and thicknesses were siliconized and the properties of the coating substrate system determined.

2. Phase Two: Coating Protectiveness Studies

Optimum process parameters were chosen on the basis of Phase One work, and small samples were coated and evaluated by oxidation testing. Multiple samples were tested to evaluate the reproducibility of the process and thus its value in achieving the objectives of this program. Thirty experiments, involving six samples each, were performed to determine what effects varying process parameters has on coating protectiveness and brittleness. The distribution of the samples among the specific tests is given in Section IV. After satisfactory process conditions were established for coating these specimens, test results were compared with results in the literature on previous tests of the Cr-Ti-Si coating system as produced by other techniques, especially the vacuum pack process.

3. Phase Three: Complex Shapes

After the potential of the CVD process for coating flat test specimens was established, the usefulness of the process for coating more complex shapes was investigated. A curved shape similar to a vane or leading edge segment, and lap-welded samples having hidden or faying surfaces were coated to demonstrate the "throwing power" of the process and its general utility for coating pieces of hardware of typical shapes. Coated parts were subjected to cyclic oxidation tests.

SECTION II

EXPERIMENTAL DESCRIPTION

A. <u>Materials</u>

<u>Substrates</u>: All coatings were deposited on columbium alloy B-66, obtained from the Westinghouse Astronuclear Laboratory. The nominal composition (vendor sanalysis) is

Columbium	89%	0xygen	0.015%
Molybdenum	5%	Nitrogen	0.012%
Vanadium	5%	Carbon	0.014%
Zirconium	1%	Hydrogen	< 0.0005%

The material was received in the annealed condition, had a hardness of 249 Knoop, a surface as shown in Figure 3, and a cross section as shown in Figure 4. The samples were sheared to size, corners and edges rounded, etched in ${\rm HF:H_2SO_4:HNO_3}$, rinsed, and stored under clean conditions until use.

Chemicals: $TiCl_4$ - Purified Grade $TiCl_4$ obtained from the J. T. Baker Chemical Company was used. Gas chromatographic analysis showed only trace quantities (< 0.1%) HCl as impurity. All transfers of the chlorides were carried out so that the chemicals were contained in an inert atmosphere.

CrCl₃ - Anhydrous CrCl₃ was not readily available, and initial attempts to dehydrate the hexahydrate were only moderately successful. The CrCl₃ was finally prepared by passing chlorine over the carbon-free chromium metal at 800°C. A mixture of the di- and trichlorides was formed. The trichloride subsequently distilled to the cooler region of the reactor. Once obtained as anhydrous crystals, this material is not difficult to handle, as it does not absorb water.

 ${
m SiCl}_4$ - Purified Grade ${
m SiCl}_4$ obtained from the J. T. Baker Chemical Company was used. The material was not exposed to the atmosphere.

Hydrogen - The hydrogen was purified just before use by palladium diffuser (Englehard Industries, Model 20-120-C, or Serfass Model 100-D).

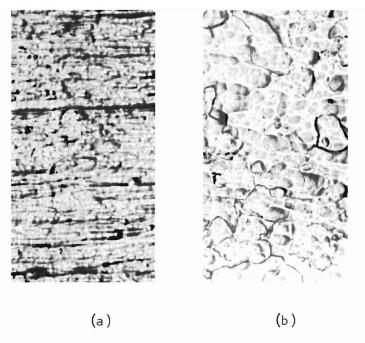


Figure 3 Surface of B-66 Alloy, (a) as-received, (b) etched for deposition (x 500)

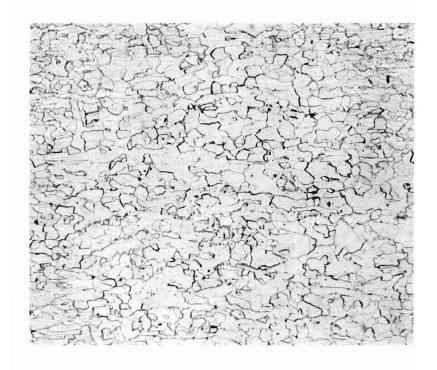


Figure 4 Cross section of B-66 Alloy as-received (x 500, etched)

Helium and Argon - These gases were purified by passing over titanium chips at 750°C. Purity of the gases (estimated < 2 ppm impurity) was checked periodically by heating a blank sample of the substrate in the gases and examining the surface for oxides or nitrides.

B. Equipment

The control unit for the coating system contained in one convenient location temperature and pressure regulators and devices for controlling and measuring all the gases except chromium chlorides. Except for system sample-jet geometry, all the main process variables were controlled from this unit. Figure 5 is a photograph of the control unit and Figure 6 the flow diagram of the coating system.

Temperature controllers for the system included Honeywell Versatronik temperature controllers (accuracy \pm 5°C) on the CrCl $_3$ and helium cleaning furnaces; a Honeywell Pyro-Volt proportional controller (accuracy \pm 2°C) for reactor temperature; and a mercurial thermoregulator (accuracy 0.01°C) on the water bath surrounding the TiCl $_L$ and SiCl $_L$ sources.

The gas flows were measured with Brooks Sho-Rate flow meters having a claimed accuracy of 5%, with the actual flow calibrated with a Precision Scientific Company wet test meter having a claimed accuracy of 0.5%. The results of the calibrations showed a deviation of less than 5% from the smoothed curve values, which was within the desired limits.

A photograph of the coating reactor is shown in Figure 7. Figure 8 is a schematic diagram of the reactor. The reactive gases were impinged directly on the sample so that gas velocity was high at the deposition surface. The temperature was controlled by a saturable-core reactor through the sensing thermocouple. Samples were inserted and removed through the gas lock chamber without introducing impurity gases into the system. This was a major aid in obtaining uniformity among samples and permitted a much faster operation; in addition, the reactor did not have to be cleaned frequently. The exact design of the delivery jet depended on the shape and size of the sample and the information desired from the experiment.

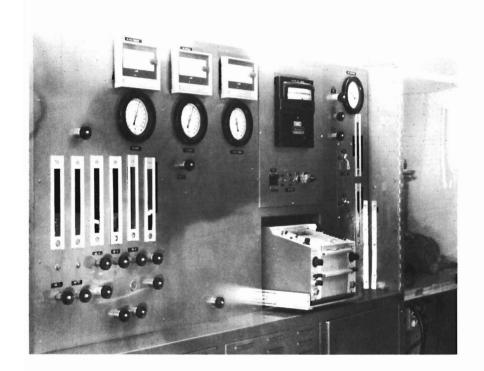


Figure 5 Coating System Control Unit

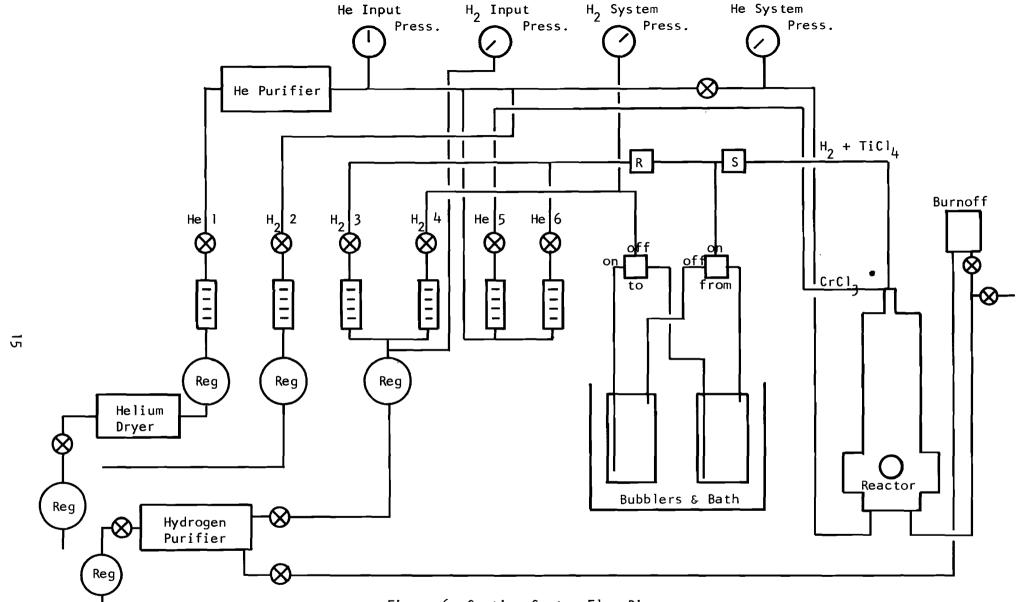


Figure 6 Coating System Flow Diagram

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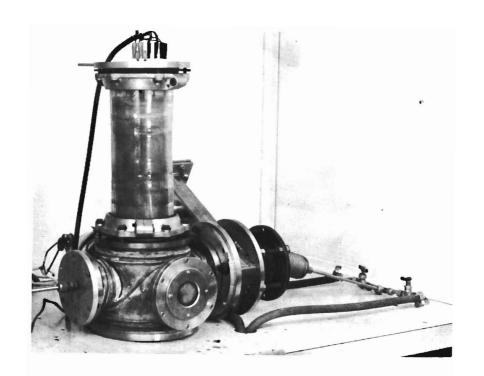


Figure 7. Photograph of Coating Reactor

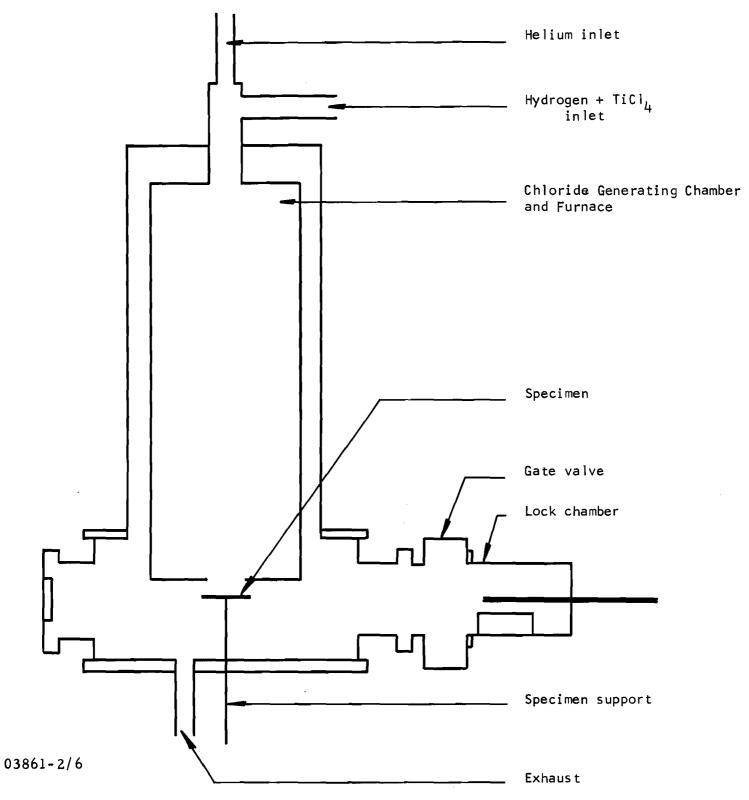


Figure 8 Schematic diagram of coating reactor

SECTION III

PROCESS DEVELOPMENT FOR METAL AND ALLOY DEPOSITION

A. <u>Titanium Deposition</u>

In work carried out at Texas Instruments prior to this study, hydrogen reduction of TiCl₄ at 1350°C under various reactant flow rates and concentrations had been found to deposit approximately 0.2 mg/min/cm² of titanium. At this rate, preparation of samples would not be rapid, and the rate would be less at lower temperature. Consideration of the total number of samples to be prepared suggested that a deposition rate of 0.05 to 0.1 mil/min (1 to 2 mg/min/cm²) at 1100 to 1350°C was desired; therefore, a way of obtaining more rapid deposition of titanium was sought.

Thermodynamic factors of the reaction are such that deposition rates might be increased (at least for reactions proceeding under equilibrium conditions) by altering reaction parameters in various ways. Calculations indicated that this would only slightly increase the deposition rate, however. The most promising changes indicated were to alter flow rate, gas concentration, and geometry, and to add an inert carrier gas, preheat the reactive gases, and partially prereduce the TiCl₁.

1. Pressure of Products

The free energy of the reduction reaction is increased by increasing the pressure of the HCl produced by the reaction. The calculations of the assumed set of conditions in Example 1 below show the effect of HCl pressure and demonstrate the desirability of keeping it low. One way this may be done is to use high flow rates so that product gases are rapidly removed from the reaction zone. Experiments 46 through 50 (in the Appendix) were run at various flow rates to test this approach. Although HCl pressure was reduced, deposition rates increased only slightly.

Example 1 - Hydrogen reduction of
$${\rm TiCl}_4$$

$$\Delta F^{\circ} = 87,700 - 35.8T *$$

at
$$1500^{\circ}$$
K $\Delta F^{\circ} = + 34,000$ cal/mole

$$\Delta F_{1500} = \Delta F^{\circ}_{1500} + RT \ln K$$

$$= 34,000 + 1.98 \times 2.303 \times 1500 \log K$$

$$= 34,000 + 6.86 \times 10^{3} \log \frac{(P_{HC1})^{4} a_{Ti}}{(P_{H_2})^{2} P_{TiCl_{4}}}$$

$$a_{Ti} = 1$$
, $P_{H_2} \cong 1$, $\Delta F_{1500} = 34,000 + 6.86 \times 10^3 \left[\log (P_{HC1})^4 - \log (P_{TiCl_L}) \right]$

where ΔF° is standard free energy change

R is gas constant

T is temperature in °K

a is activity

P is pressure in atmospheres.

Assume sets of conditions:

(1)
$$P_{HC1} = 0.01$$
, $P_{TiCl_4} = 0.01$
 $\Delta F_{1500} = 34,000 + 6.86 \times 10^3 [\log 10^{-8} - \log 10^{-2}] = -7,100 \text{ cal/mole}$
Reaction is slightly favorable.

^{*}McQuillan and McQuillan, <u>Titanium</u> (London, Butterworth's Scientific Publications, 1956), p. 11.

(2)
$$P_{HC1} = 0.1, P_{TiCl_{L}} = 0.01$$

 $\Delta F_{1500} = 34,000 + 6.86 \times 10^3 [log 10^{-4} - log 10^{-2}] = +21,000 cal/mol Reaction is unfavorable.$

Therefore, the reduction of $TiCl_4$ requires that the pressure of HCl be limited to low values (less than 10^{-2} atmosphere).

HCl pressure can also be lowered by diluting the gas stream with an inert gas. This lowers reactant concentrations, but these concentrations have less effect on reaction thermodynamics than does HCl concentration. ($P_{TiCl_{L}}$ to the first power and P_{H_2} squared, but P_{HCl} to the <u>fourth</u> power.) Experiments 43 and 45, listed in Appendix A, were performed to test this, using argon as the diluent. Deposition rates were not significantly increased.

Still a third way to reduce HCl pressure is to prereduce ${\rm TiCl}_4$ to the trivalent state before the deposition reaction. This can be accomplished with the reaction:

for which

$$\Delta F^{\circ}_{1200} = -16.55$$
 cal/mole.

Placing titanium chips in a tube heated by the $CrCl_3$ furnace results in the above reaction to yield $TiCl_3$ vapor, which is carried in the gas stream to the sample.

For the reduction of TiCl₃
$$TiCl_3 + 3/2H \longrightarrow Ti + 3HCl$$

$$\Delta F^{\circ}_{1500} = +23,000 \text{ cal/mole}$$

$$\Delta F_{1500} = \Delta F^{\circ} + RT \ln \frac{P_{HCl}^{3}}{(P_{H_2})^{-(P_{TiCl}_3)}}$$

$$P_{H_2} \cong 1$$

$$= 23,000 + 6.86 \times 10^{3} [\log{(P_{HCl})^{3}} - \log{(P_{TiCl}_3)^{3}}]$$

Assume conditions

(3)
$$P_{\text{TiCl}_3} = 0.01, P_{\text{HCl}} = 0.002$$

 $\Delta F_{1500} = 23,000 + 6.86 \times 10^3 [2 \log 2 \times 10^{-3} - \log 10^{-2}]$

= - 300 cal/mole

Reaction slightly favorable.

(4)
$$P_{TiCl_3} = 0.1$$
; $P_{HCl} = 0.002$

$$\Delta F_{1500} = 23,000 + 6.86 \times 10^3 [2 \log 2 \times 10^{-3} - \log 10^{-1}]$$

$$= -5,300 \text{ cal/mole}$$

Reaction favorable.

Experiments 67 through 98 (Appendix A) were carried out in this fashion and yielded deposition rates in the desired range (1 to 2 mg/minute).

2. Concentration of Reactants

Increasing $TiCl_{4}$ pressure provides a more favorable energy for the reduction reaction (assumed sets of conditions, Example 2). In the original experimental setup, the partial pressure or concentration of $TiCl_{4}$ obtainable was limited to saturation of the carrier gas stream with liquid $TiCl_{4}$ at $20^{\circ}C$; i.e., since the vapor pressure of $TiCl_{4}$ at $20^{\circ}C$ is 10 mm of mercury, the maximum concentration possible is 10/760, or 1.35% $TiCl_{4}$. The concentration could be increased by heating the liquid $TiCl_{4}$ above $20^{\circ}C$, but this would require all transfer lines to be kept at least this warm. A more practical technique is to meter the liquid $TiCl_{4}$ directly into the heated furnace. This permits $TiCl_{4}$ concentrations up to 100%.

Example 2 - Increase TiCl₄ pressure Assume conditions

(5)
$$P_{TiCl_4} = 0.1$$
, $P_{HCl} = 0.01$

$$\Delta F_{1500} = 34,000 + 6.86 \times 10^3 [\log 10^{-8} - \log 10^{-1}]$$

$$= -14,000 \text{ cal/mole}$$

Reaction is favorable.

(6)
$$P_{\text{TiCl}_4} = 0.5$$
, $P_{\text{HCl}} = 0.01$

$$\Delta F_{1500} = 34,000 + 6.86 \times 10^3 [\log 10^{-8} - \log 0.5]$$

$$= -18,500 \text{ cal/mole}$$

Reaction is more favorable.

A method for controlling high concentrations of ${\rm TiCl_4}$ was set up (Figure 9) and the role of ${\rm TiCl_4}$ investigated in experiments 53 through 66 (Appendix A). These showed that in some instances the deposition rate can be increased. When ${\rm TiCl_4}$ concentration was greater than 30%, the substrate lost weight (see Figure 10), probably through one of the reactions:

$$nTiCl_4 + Cb \longrightarrow nTiCl_4 + CbCl_n \uparrow$$
or
$$nHCl + Cb \longrightarrow \frac{n}{2} H_2 + CbCl_n \uparrow$$

(A similar effect has been widely noted in the production of silicon by hydrogen reduction of the chlorides.) In the case of either reaction, attack on the substrate can be eliminated by using a lower TiCl₄ and/or a higher hydrogen pressure.

3. Change in Geometry

Though not apparent from the thermodynamic analysis, the geometry relating the gas delivery jet and the sample is quite important in forming TiCl₃ and in gas-sample contact time. The jet arrangement was modified to control gas

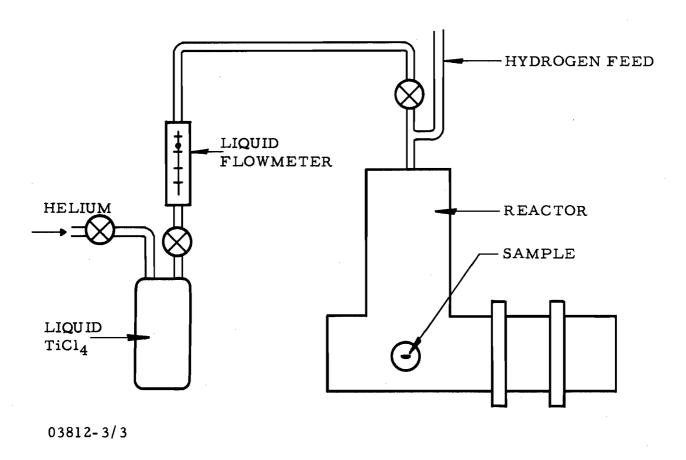


Figure 9 Liquid TiCl₄ Metering Setup

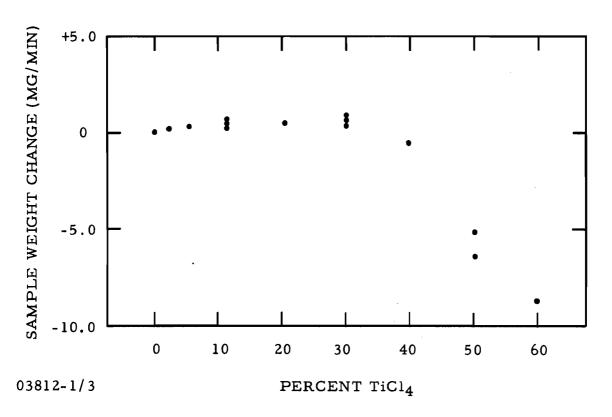


Figure 10 Sample Weight Change versus Percent TiCl₄

velocity over the sample and to increase contact time. Deposit uniformity also benefited by this change.

4. Preheating Reactive Gases

Calculations of the residence time of the gases in the sample region show that for average flows, contact time is extremely short. Assuming a reactive volume of gas around the sample to be about 2 cm³, a flow rate of 2000 cm³/min would yield a residence time of 0.06 second. (Such a condition may be demanded by the requirement of maintaining a low HCl pressure.) It is not likely that all the gas would approach thermal equilibrium in this time. By running the CrCl₃ furnace at various temperatures, the gas stream was preheated to the levels indicated in experiments 50 through 52 in Appendix A. However, this preheating had little effect.

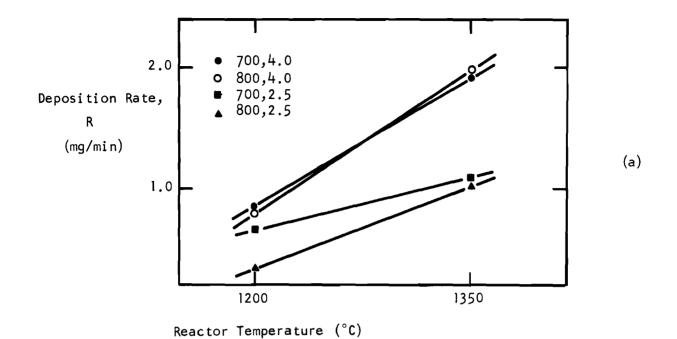
After having outlined the general range of interest of the variables, a statistical cube design series of experiments was set up and carried out. Preliminary experiments had indicated that the major variables would be the temperature of the reactor, flow rate of gases, and temperature of the titanium chips used in prereducing the TiCl₄. The series listed in Table I consists in running the corners of a cube with a central point replication four times. The results of this type of experiment were analyzed by a multiple regression and correlation analysis and a factorial analysis of variance. For the statistical analysis, the rate of weight gain was used as the dependent variable. The results of this series are shown in the plots of Figure 11, where the dependent variable is plotted versus each independent variable.

The titanium deposition results can be portrayed on a three-dimensional scale as shown in Figure 12 in addition to the two-dimensional plots of Figure 11. Since there are two independent variables and one dependent variable, the deposition rate is uniquely defined for specific temperatures and flow rates. Thus, the deposition rate can be represented as a plane in the temperature, flow rate, deposition rate, three-dimensional plot. That is, the deposition rate that will occur for any particular value of flow and reactor temperature

TABLE I

Plan for Titanium Deposition

Run No.	×ı	× ₂	×3	У
86	-1	1	1	1.91
87	-1	-1	-1	0.70
88	0	0	0	1.05
89	1	-1	-1	0.30
90	0	0	0	0.77
92	-1	-1	1	1.04
93	0	0	0	0.88
94	1	1	1	1.92
95	1	1	-1	0.78
96	1	-1	1	1.16
97	-1	1	-1	0.83
98	0	0	0	1.28
Variable	Level	-1	0	1
×ı	Prereduction furnace temper-ature (°C)	700	750	800
× ₂	× ₂ Flow rate (liters/min)		3.3	4.0
× ₃	Reactor temper- ature (°C)	1200	1275	1350
У	Weight gain (mg/min)			



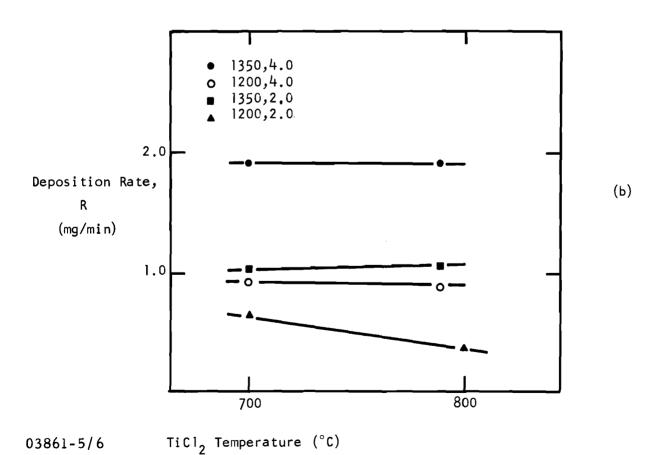
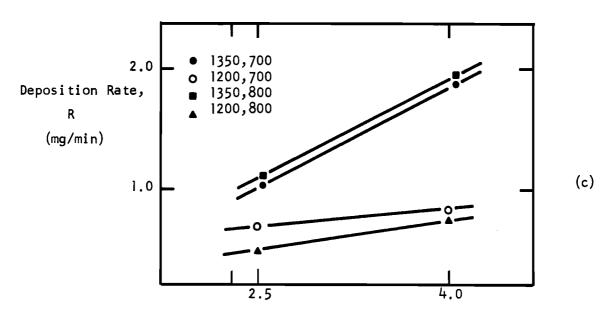


Figure || Plots of Deposition Rate of Titanium from Statistical Plan



Flow Rate (liters/min)

03861-6/6

Figure || (Continued) Plots of Deposition Rate of Titanium from Statistical Plan

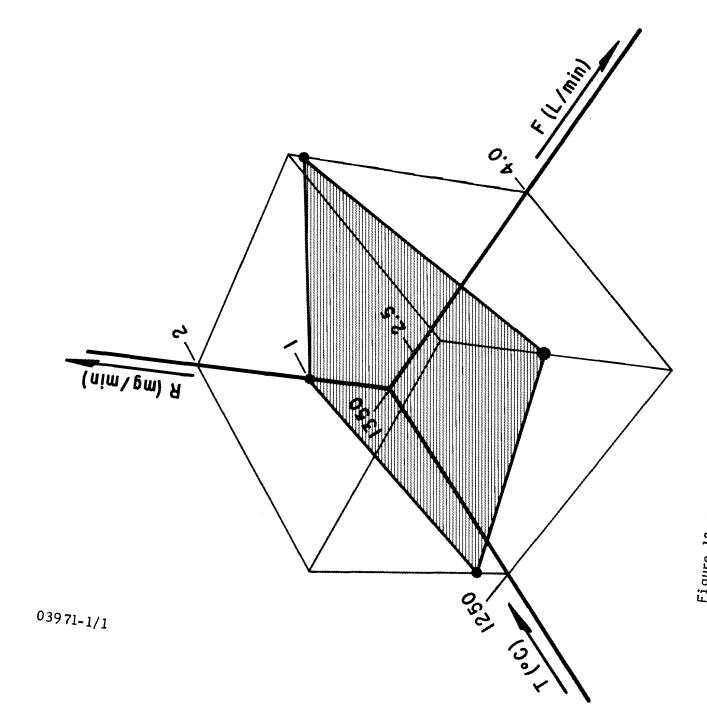


Figure 12 Three-Axis Plot of Titanium Deposition Data

29

is the point intersecting the deposition plane. This then represents the response surface of the deposition rate as influenced by the flow and the reactor temperature.

Results of the multiple regression analysis are shown in Table II. The last column lists the value of F required to establish a level of confidence for the parameters. The 95% level was selected as the goal for this study. The variable \mathbf{x}_1 , which is the titanium tetrachloride prereduction temperature, is not a significant factor. The variable \mathbf{x}_2 , the total flow-rate of gases, is significant a levels greater than 95% for this study. The variable \mathbf{x}_3 , the reactor temperature, is likewise significant considerably beyond the 95% confidence level.

Table III shows the factorial analysis of variance of the data for titanium depositions. Column 5 shows the F values required for the 95% confidence level. These figures agree with the multiple regression analysis. In addition, this table shows that these are the only variables significant in this model at this level and that no first-order interaction between the variables approaches the 95% level of confidence.

A photomicrograph of a typical titanium coating is shown in Figure 13. The grain structure of the coating is a continuation of the original substrate. This same structure is evident in the photograph of the titanium surface shown in Figure 14. Figure 15 illustrates the degree of edge coverage normally obtained in these depositions.

A possible aid in interpreting and understanding titanium deposition rate data is to plot the logarithm of the deposition rate versus 1/T in a typical Arrhenius plot, as shown in Figure 16. The data appear to group on two general lines, the upper one including the deposition at high flow rate, the lower one including the depositions at the lower flow rate. The slopes of these curves, which are related to the overall ΔH , heats of reaction of the deposition, differ only slightly. While the deviation is not large enough to be interpreted

TABLE II

Multiple Regression and Correlation Analysis of

Titanium Depositions

Variable	Regression <u>C</u> oefficient	Observed T Value	T Value [*] for 95% Level
×ı	-8.00×10^{-4}	-0.48	2.36
× ₂	0.37	3.36	2.36
× ₃	5.70×10^{-3}	5.17	2.36

Intercept: -6.82

Standard Error of Estimate:

0.23

Multiple Correlation Coefficient:

0.9195

^{*} Reference: "Introduction to Statistical Method", Ehrenfeld, S., and Littauer, S., (McGraw-Hill Book Co., New York, 1965).

TABLE III

Factorial Analysis of Variance for Titanium Depositions

NUMBER OF VARIABLES 3
NUMBER OF REPLICATES 1

VARIABLE NO. OF LEVELS

1 2
2 2
3 2

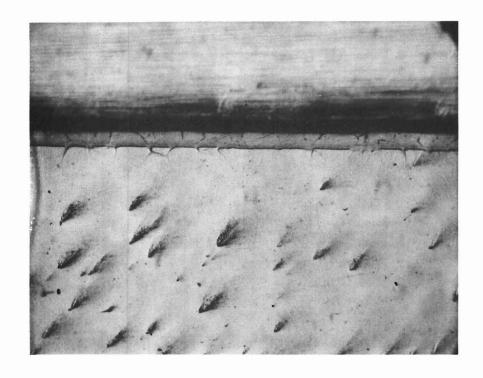
GRAND MEAN

1.08000

SOURCE OF VARIATION	DEGREES OF FREEDOM	SUMS OF SQUARES	F * VALUE	F *** REQUIRED
1	1	0.01280	<1	10.1
2	1	0.62720	8.65	
3	1	1.46205	20.2	
12	1	0.00720	<1	
13	1	0.04205	<1	
23	1	0.13005	1.79	
RESIDUAL	1	0.02645	<1	10.1
TOTAL	7	2.30780		

 $* \overset{\wedge}{\sigma}^2 = 0.0725$

*** For 95% confidence level



mounting titanium substrate

Figure 13 Coating of titanium on B-66 Alloy (x 265, unetched) Run 80, 17-micron titanium thickness

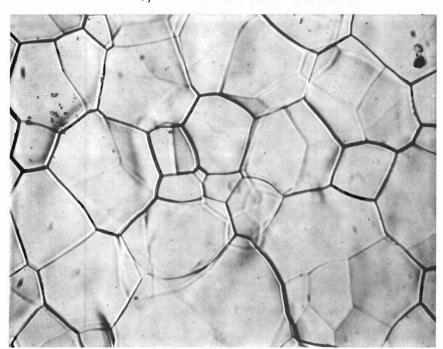


Figure 14 Surface of titanium "as deposited" on B-66 Alloy (x 500)

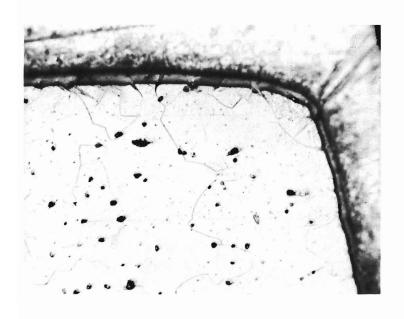


Figure 15 Titanium Deposited on B-66 Alloy Showing Coating Over Corner Section (X 135)

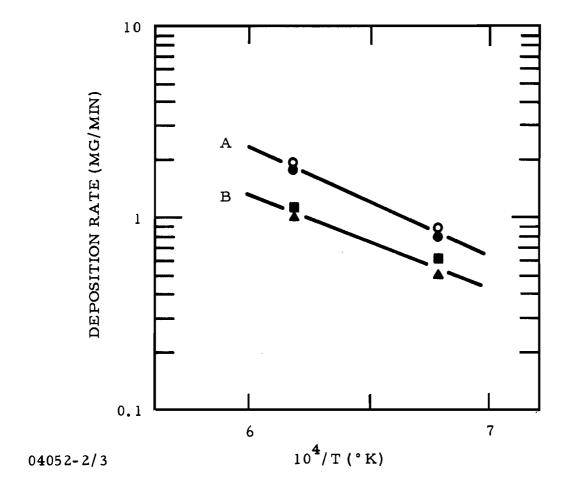


Figure 16 Arrhenius Plot of Titanium Deposition Data. Curve A includes data points for flow = 5.5, Curve B Flow = 2.5.

unequivocally, the lower energy of reaction at the lower flow rate is characteristic of other chemical vapor deposition experiments and is sometimes interpreted as indicating a high degree of diffusion control. The values of the energies obtained are approximately 24 and 22 kcal/mole for the upper and lower curve. In Table IV these two values are compared with known values for possible reactions which may occur during deposition. These data are consistent with previous work at Texas Instruments which included the total reduction at the sample. Present process studies include only the second reaction.

TABLE IV.

Activation Energies of Reactions Occurring During
Titanium Depositions

Reaction	∆H (kcal/mole)	Reference
TiCl ₄ + 2H ₂ → Ti + 4HCl	(41)*	R. B. Byer and D. M. Mason, Ind. Eng. Chem. Process Design and Dev., 2, 78 (1963)
	37	R. Bracken, "Chemical Vapor Deposition of Titanium," Texas Instruments Report
TiCl ₃ + 3/2H ₂ → Ti + 3HCl	22-24	This work

^{*} assuming TiCl $_3$ intermediate is limiting factor

B. <u>Deposition of Chromium Metal</u>

The attempt to obtain chromium chloride in the vapor state in reproducible and controllable fashion was initially based on the saturation of an argon gas stream passing through a bed of solid chromium trichloride (CrCl₃) at an elevated temperature. However, this process has several deficiencies. One is the difficulty in obtaining pure CrCl₃. The reaction of most chlorine-containing chemicals produces only chromium dichloride (CrCl₂) rather than the trichloride.

Only in the reaction of elemental chromium and chlorine is CrCl₃ produced. CrCl₃ is listed in literature as being relatively stable and so should sublime without decomposition. However, when the trichloride was held at elevated temperatures, it seemed to change to the dichloride. This indicates that the trichloride sublimes by decomposing to the tetrachloride (which is known to exist and has a relatively high vapor pressure at these temperatures), leaving a solid residue of the dichloride. Such an occurrence would gradually lower the effective pressure of chromium in the vapor state and make the source difficult to control over long periods of time.

The dichloride was considered as a source of chromium chloride vapor to be reduced at the sample. Problems inherent in using the dichloride are that it tends to absorb water and that it is a liquid at the temperature required for vaporization. A slight equipment modification was therefore necessary so that the dichloride could be used in the liquid form. The CrCl₂ was prepared by reaction of HCl gas and chromium metal. Following preparation, the chloride was purified by sublimation and loaded into the reservoir of the reactor system. This procedure apparently did not successfully eliminate all impurities because deposits obtained with this material were discolored, indicating the presence of oxides and/or nitrides. This process was very difficult to use, and results did not justify continuing with it.

An alternate way to form CrCl₂ is the in-situ reaction of HCl with chromium metal in the furnace used to heat the chloride reservoir in previous experiments. This procedure eliminates the need for handling hygroscopic chlorides. A gas supply was attached to introduce a small, controllable flow of HCl gas into the argon stream which flowed over the chromium metal, where the HCl reacted to generate CrCl₂ and hydrogen. This vapor flowed to the reaction zone, mixed with hydrogen, and impinged on the sample. The concentration of chromium was conveniently controlled by regulating the flow of the HCl gas. An added advantage was that the materials maintained in the furnace area were metallic and thus did not adsorb water. Purity of the gas stream should equal that of the reacting gases and in fact some impurities (such as nitrogen) may be gettered

out by the chromium metal. Deposits made using this source of $CrCl_2$ were shiny, and metallic in appearance, with no discoloration. A convenient, reliable control of the concentration of chromium in the gas phase was thus obtained.

As is often the case, obtaining uniform deposits was as difficult as controlling the reacting gases. When CrCl₂ is carried in an argon gas stream and mixed with hydrogen immediately above the sample, uniform mixing of the gases is essential for uniform deposits. For this reason, it was necessary to investigate the relation of jet design and behavior of reaction gas to the formation of uniform deposits. A device similar to an oil-burner nozzle, consisting of two concentric rings of nozzles, was settled on. Under the proper flow conditions and deposition rates this device produced a uniform deposit.

An initial series of the planned experiments was carried out as shown in Table V. When plotted, some of the values appeared anomalous in that the observed rates did not correlate with the expected values. Comparison of the rate data with the order in which the points were run revealed that deposition rates exceeded the expected values in runs immediately following the highest HCl gas flow. This was consistent throughout the plan. In addition, following the high HCl flow a small amount of smoke or vapor could be seen coming from the chromium reservoir, even when only argon was passing through the reservoir. It was obvious that in these cases more chromium chloride was formed in the reservoir than was carried out in the argon stream. Such a situation nullified most of the values taken at the high chloride points as well as values following these points or where chlorides were retained from previous experiments. In a re-run of the statistical plan this effect was eliminated by: (1) allowing additional time between runs, several minutes at least, during which argon was passed through the chromium reservoir; (2) increasing furnace temperature to increase vapor pressure and the amount of CrCl, vaporized; (3) decreasing maximum HCl flow passed over the chromium to less than the maximum amount that could be utilized.

The statistical plan for the chromium was re-run and the data analyzed by multiple regression and correlation analysis, and analysis of variance. The

TABLE V

Statistical Plan and Results for Chromium

Deposition Investigation

Run No.	×ı	× ₂		×3		У
101	0	0		•		21. 1.
191	0	0		0		24.4
192	-1	1 -1		-1 -1		7.0 18.0
193	-1					
194	0	1 0		1 0		10.2
195 196	-1	1		-1		25.5 13.9
197	1	1		-, 1		7.2
198	1	-1		' 1		7.2 53.7
199	0	0		0		16.1
200	-1	1		1		13.0
201	1	-1		-1		13.2
202	0	0		0		14.5
203	1	1		1		9.3
<u>Variable</u>		Level	<u>-1</u>		<u>o</u>	<u>1</u>
						
×į	Reacte	or Temperature (°C) 1200		1265	1325
× ₂	CrCl ₂	Concentration (%)	0.3		0.6	0.9
× ₃	Flow	Rate (½/min)	2.5		4	5.5
у	Weight Gain (mg)					

plan and results are given in Table VI. Results are displayed graphically in Figure 17. Figure 17(a) is a plot of the deposition rate as a function of the reactor temperature, Figure 17(b) shows the deposition rate versus the CrCl₂ concentration in the gase phase, and Figure 17(c) is deposition rate as a function of the total gas flow rate. In all cases, the higher levels for these variables yielded increased deposition rate. Figure 17(c), deposition rate as a function of chloride concentration, shows that the process was diffusion- or transport-limited, since deposition rate increased with increased flow. From an operational standpoint, this means that when processes work in these regions, flow must be carefully controlled to maintain control of deposition rate. The same comment obviously applies to temperature and concentration.

In titanium depositions flow rate was not as critical as temperature. This was particularly true of depositions at lower temperatures.

Chromium deposition data can also be represented in a three-dimensional plot, as in Figure 18. In this case, three of the variables are independent (temperature, flow, and concentration) so that the dependent variable (deposition rate) is not uniquely defined for any particular two values. However, deposition rate can be depicted as response surfaces, as shown in Figure 18, where the planes correspond to constant deposition rates. From this plot and the corresponding three-dimensional plot for titanium, experimental parameters were selected for the co-deposition of titanium and chromium which would yield alloys of various compositions. This, of course, is based on the assumption that the depositions will proceed independently, a factor shown later to be invalid in the range of experimental variables studied.

The chromium deposition plan was analyzed by multiple regression, as shown in Table VII, and also by an analysis of variance, as shown in Table VIII. Comparing the F values to that required for a 95% confidence level shows that all three independent variables are significant at this level. The correlation coefficient indicates a relatively good fit of the data by the equation.

TABLE VI

Second Statistical Plan and Results for
Chromium Deposition Investigation

Run No.	×1	× ₂	× ₃	У
358	0	0	0	0.98
359	-1	1	-1	0.81
360	-1	1	1	2.50
361	0	0	0	1.82
362	1	-1	-1	0.68
363	-1	-1	1	1.43
364	1	1	1	3.43
365	0	0	0	1.87
366	1	1	-1	1.23
367	1	-1	1	1.91
370	-1	-1	-1	0.42
372	0	0	0	1.54
Variable	Level	-1	0	1
× ₁ React	or Temperature (°C)	1200	1275	1350
× ₂ CrCl ₂	CrCl ₂ Concentration (Percent)		0.225	0.3
× ₃ Flow	Flow Rate (ℓ/\min)		4.0	5.5
	t Gain (mg/min)	_		

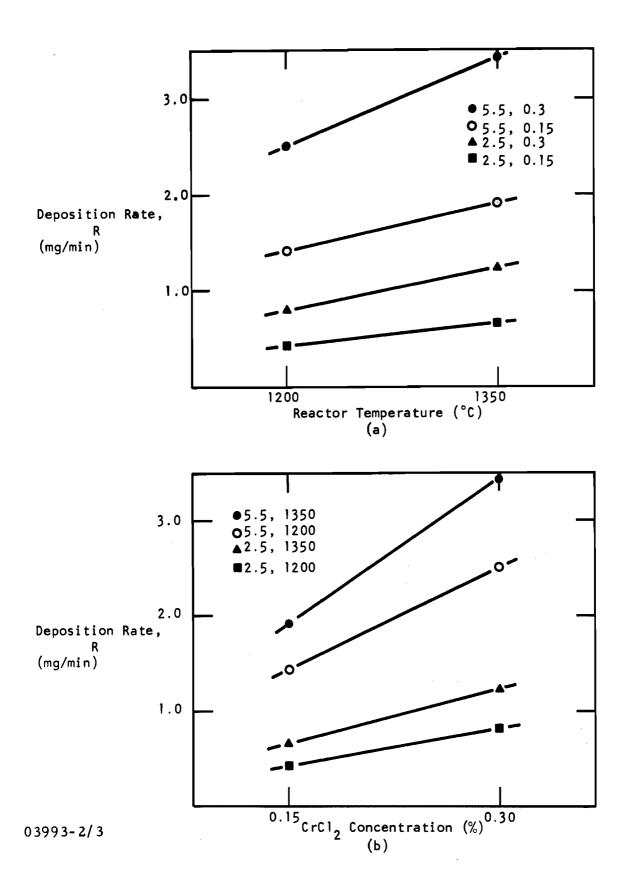


Figure 17 Plots of Deposition Rate of Chromium from Statistical Plan

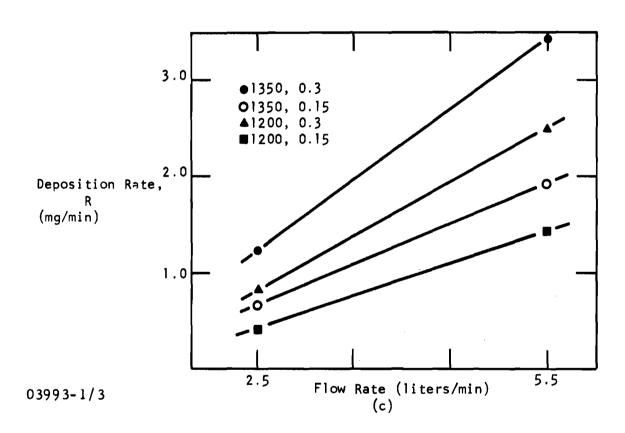
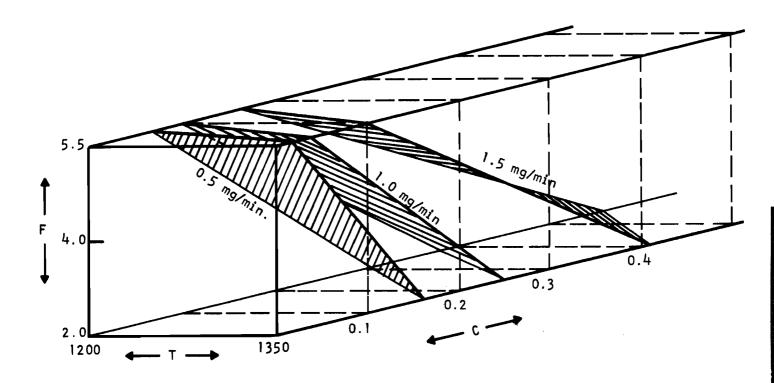


Figure 17 Plots of Deposition Rate of Chromium from Statistical Plan (continued)



03993-3/3

Figure 18 Planes of Constant Deposition Rate of Chromium

[F = Flow, liters/min; T = Reactor Temperature, °C; C = Concentration of CrCl₂, %.]

TABLE VII Multiple Regression and Correlation Analysis of Chromium Depositions

Variable	Regression Coefficient	Observed T Value	T for 95% Level
×ı	3.483×10^{-3}	1.99	2.36
× ₂	5.883	3.36	2.36
× ₃	5.108 × 10 ⁻¹	5.84	2.36

Intercept: -6.255

Standard Error of Estimate: 0.371

Multiple Correlation Coefficient: 0.9359

TABLE VIII

Factorial Analysis of Variance for Chromium Depositions

NUMBER OF VARIABLES 3
NUMBER OF REPLICATES 1

VARIABLE NO. OF LEVELS

1 2
2 2
3 2

GRAND MEAN

1.55125

SOURCE OF VARIATION	DEGREES OF FREEDOM	SUMS OF SQUARES	F VALUE *	F REQUIRED ☆☆
1	1	0.54601	3,187	10.1
2	1	1.55761	9.093	
3	1	4.69711	27.278	
12	1	0.04651	0.271	
13	1	0.06661	0.388	
23	1	0.34031	1.987	
RESIDUAL	1	0.01051	0.061	10.1
TOTAL	7	7.26469		

^{***} For 95% confidence level

This series of the statistical plan produced chromium layers uniform over the sample, except for the one deposited at the highest rate. The high rate produced some nodular or needle growth, which is characteristic of a high supersaturation ratio.

Analysis of the deposition rate on an Arrhenius type plot is shown in Figure 19, where the logarithmic deposition rate is plotted versus the inverse of absolute temperature. The data fall on four lines, each correpsonding to a particular combination of concentration and flow rate. Such a division emphasizes the independence of the deposition parameters. As was the case for the titanium depositions, the slope of the lines for the two higher temperatures yields a highest heat of reaction value, 12 kcal/mole, while the slope for the lower flow rate curves corresponds to about 9 kcal/mole. The small values of ΔH from this data indicate that the deposition is primarily diffusion-controlled, since reaction-controlled processes usually have heats of reaction larger than these values.

C. Comparison of Titanium and Chromium Deposits

The major differences between the deposition data for chromium and titanium were accompanied by the significant difference in appearance of the two deposits. The surface of the as-deposited titanium material was shown in Figure 14. The grain structure of the substrate material could be seen clearly. The chromium deposits, as shown in Figure 20, however, did not show this grain structure, but appeared microcrystalline, with the many growth steps more typical of vapor-deposited material. These differences were significant, since further examination showed them to result from different reaction mechanisms. This difference has important implications for the deposition of chromium.

Examination of the chromium-columbium interface reveals narrow-intermetallic compound interlayer, which might have tended to prevent continued propagation of the substrate grains into the coating. An additional, or perhaps alternative, explanation was suggested by a series of photographs of the surface after only a very short deposition, as shown in Figure 21. This picture shows isolated nodules

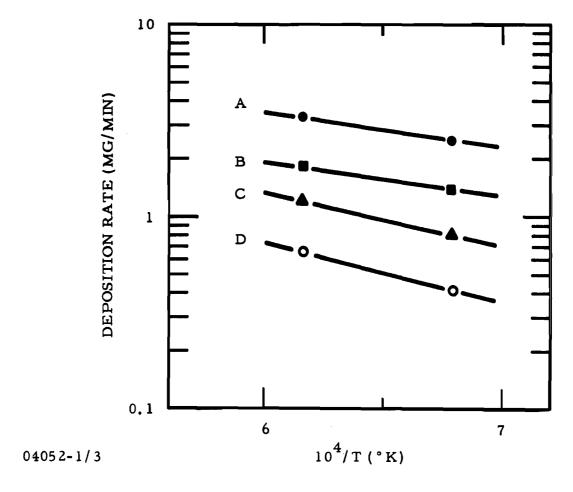


Figure 19 Arrhenius Plot of Chromium Deposition Data

<u>Line</u>	Flow	CrCl Conc.
Α	5.5	0.3
В	5.5	0.15
·C	2.5	0.3
D	2.5	0.15

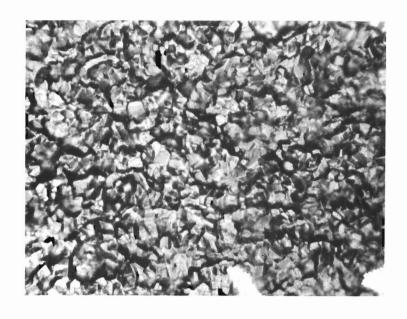


Figure 20 Surface of Chromium Deposited on Columbium (500X)

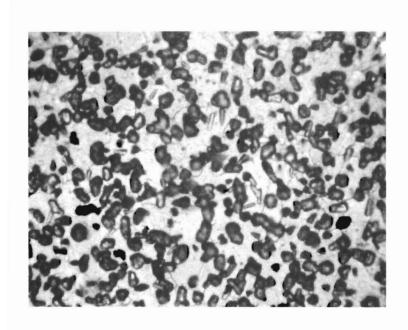


Figure 21 Columbium Surface After a Brief Deposition of Chromium (500X)

or particles which are crystallites growing on the surface of the columbium and yet leaving portions of the substrate bare. (It appears in this case that the crystallites are more concentrated in the grain boundary areas.) This may have occurred because the chromium chloride was very strongly adsorbed on chromium but only weakly adsorbed on columbium. Only when the crystals reach some critical size or population density do they contact and form a continuous coating.

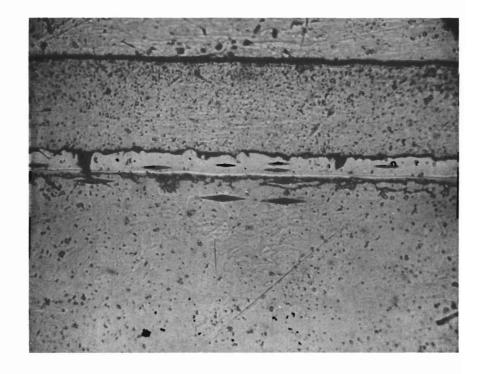
However, many times when chromium coatings were grown relatively rapidly, voids occurred in the coating, as shown in Figure 22(a). Those grown at a somewhat slower rate, such as those shown in Figure 22(b), have fewer voids, though the possibility for voids still exists. The type of mechanism observed for titanium deposition is obviously to be preferred.

It should be noted, however, that the chromium depositions reported here were made at temperatures higher than necessary for deposition of chromium alone. These temperatures were used so the data obtained could be compared directly with titanium deposition data and could be used to estimate co-deposition rates. For deposition of chromium alone, temperatures 500 to 700° lower would be used, and the interaction between columbium and chromium would be lessened. It is probable that a different mechanism would control low-temperature depositions.

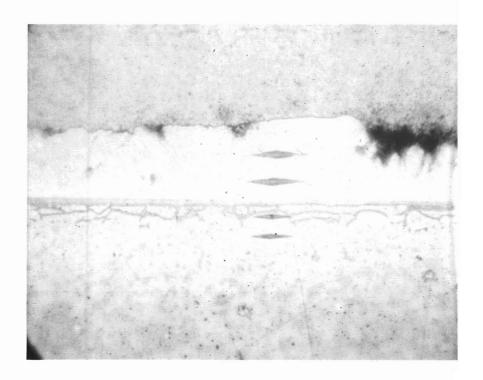
D. Alloy Codeposition Process Study

Several different processes were developed for codeposition of titanium and chromium in various alloy ratios. Since the various processes differed mainly in the manner by which the reducible metal chlorides were produced and transported to the reduction site, the major equipment modification on the coating reactor involved the chloride chamber.

Each process and the resulting deposits are discussed separately.



(a)



(b)

Figure 22 Cross Section of Chromium Deposits

1. Separate Production of Titanium and Chromium Chlorides

The processes which had been developed for depositing individual components were operated simultaneously. Reducible titanium chloride (TiCl₃) was formed from titanium metal immediately before the TiCl₃ was impinged on the substrate. Chromium chloride was produced by reacting HCl gas with chromium metal. Although these processes were adequate for depositing individual components, the coreduction process yielded deposits having several faults. Even though deposition concentrations, flows, and temperatures were varied over wide ranges, deposits generally lacked chemical and physical uniformity. This was probably because the two metal chlorides were fed through different nozzles and apparently were not uniformly mixed.

The deposits were characterized by a center portion of primarily titanium and an exterior portion consisting mainly of chromium. Deposition rates were approximately one-fourth those expected. This process failed to yield satisfactory coatings for the experimental conditions investigated. The chloride nozzles were clogged by metal deposits during deposition runs. The attempted use of this process is listed in Appendix A, runs 485 to 632.

Figure 23 shows a photomicrograph of the cross section of a coating prepared by this method. The coating is typical except that it is somewhat thicker than ordinary because of the higher hydrogen and HCl flows during this run. When etched, as shown in Figure 23, some grain structure could be observed in the coating, and interlayers were evident. An electron microprobe scan as shown in Figure 24 gave a qualitative concentration profile of the coating and indicated that the coating was mainly titanium and that some interdiffusion had occurred. The concentration of chromium was extremely low. The microhardness indicated two zones within the coating, which was in general relatively soft.

2. Production of Mixed Chlorides

a. From Chromium Metal and TiCl $_{L}$

Nonuniform mixing of the metal chlorides could be eliminated by producing them at the same place. One method was to prereduce the ${\rm TiCl}_4$ with chromium metal (rather than titanium) in the chamber above the sample.

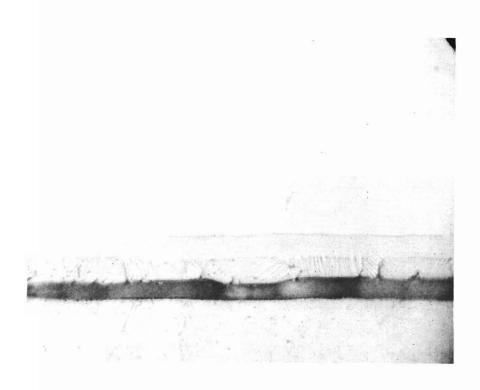


Figure 23 Codeposition of Titanium and Chromium Run 630 (X500)

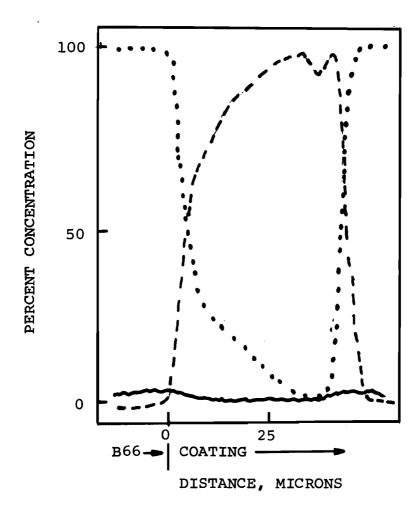


Figure 24 Electron Microprobe Profile Scan of Coating 630 Showing Qualitative Concentration

Temperature and concentrations were maintained at levels to keep the chlorides in the vapor state.

This process was more convenient operationally because only the ${\rm TiCl}_4$ flow had to be controlled, instead of both HCl and ${\rm TiCl}_4$. The possibility that unreacted HCl gas was available to inhibit subsequent reduction was also eliminated and resulted in higher deposition rates.

The reaction for the production of the chloride is

which is thermodynamically favorable in the temperature range 750°C to 850°C . In the presence of excess chromium, as was the case where the TiCl_4 was passed through a bed of chromium, the Ti:Cr ratio in the gas phase was 2:1. Deposit composition could be varied by controlled alteration of this ratio. The temperature of deposition altered slightly the ratio of metals in the deposit, high temperatures favoring titanium reduction, low temperatures favoring more chromium.

Experiments utilizing this source of reducible chlorides were carried out in runs 633 to 667 in Appendix A. The deposition rate was reasonable (about 0.1 mil/min), and the coatings were uniform across the samples. The photomicrograph of a cross section, Figure 25, shows two zones in the coating and an even top surface. This particular coating was harder than normal, which indicated the presence of impurities. A microprobe scan of the cross section, Figure 26, indicated a variation in the titanium and chromium distribution in the coating and little interdiffusion with the substrate. The concentration of chromium was higher than the desired equimolar mixture, but the process worked well for high chromium alloys.

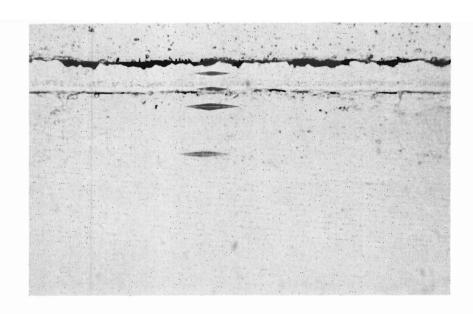
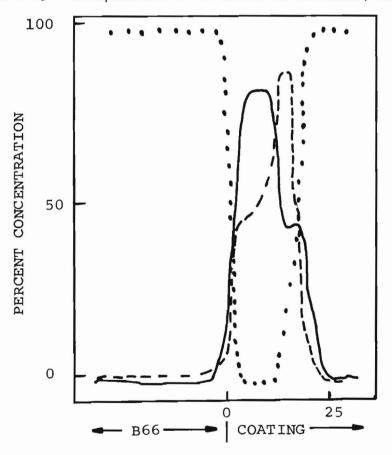


Figure 25 Codeposition of Titanium on Chromium, Run 643 (X500)



DISTANCE, MICRONS

Figure 26 Electron Microprobe Profile Scan of Coating 643 Showing Qualitative Concentration

[...Cb, ---Cr(X1.5), —
$$Ti(X1.5)$$
]

b. From a Mixture of Titanium and Chromium Metal

The possibility of increasing the Ti:Cr ratio in the gaseous chloride stream above the 2:1 value of the previous process was investigated by use of a prereduction mixture of titanium and chromium chunks. More titanium should appear in the vapor because formation of titanium chlorides is thermodynamically favored over formation of chromium chloride. However, in long-term use of this method it might be difficult to maintain a constant-composition gas phase because of the preferential consumption of titanium.

Experiments utilizing this source of reducible chlorides were carried out as listed in runs 668 to 686, in Appendix A. Many of the deposition rates were rapid (about 0.2 mil/min). Coatings typical of this process are shown in Figures 27 and 28. The coating in Figure 27 was prepared from a chloride source in which the TiCl₄ was contacted first with titanium, then with chromium, in the chloride generation chamber. The alloys thus deposited contained a high proportion of titanium, as indicated by their appearance and structure and as substantiated by the qualitative electron microprobe scan of Figure 29. From this scan, chromium concentration was estimated at about 5%. Interaction with the substrate is minimal, and the coating is uniform and soft.

The coating in Figure 28 was prepared with a chloride source which contacted the TiCl₄ with chromium metal first and then with titanium metal. The coating formed rapidly and exhibited an unusually uniform surface for the high chromium concentration (about 95%) indicated by the microprobe scan of Figure 30. The absence of voids in the chromium and the absence of interaction with the substrate are notable features.

This coating process and its two variations yielded interesting results but were apparently not satisfactory for depositing alloys of nearly equal concentrations of titanium and chromium.

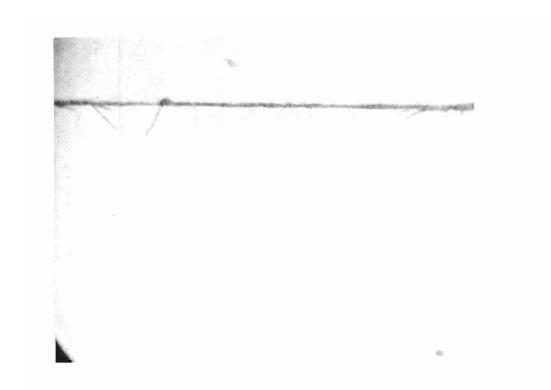


Figure 27 Codeposition of Titanium and Chromium, Run 672 (X500)

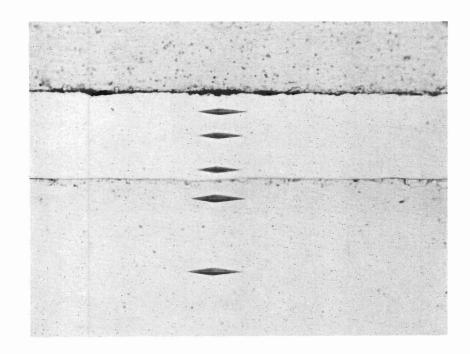


Figure 28 Codeposition of Titanium and Chromium, Run 686 (X500)

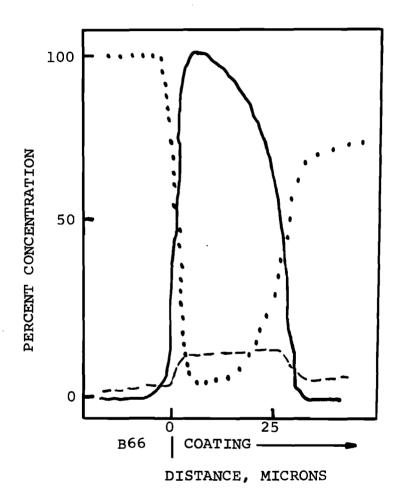


Figure 29 Electron Microprobe Scan of Coating 672 Showing Qualitative Concentration

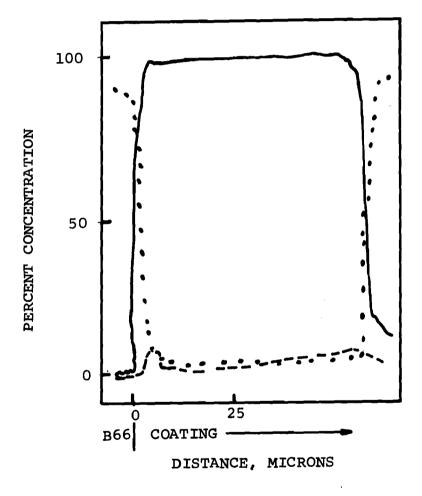


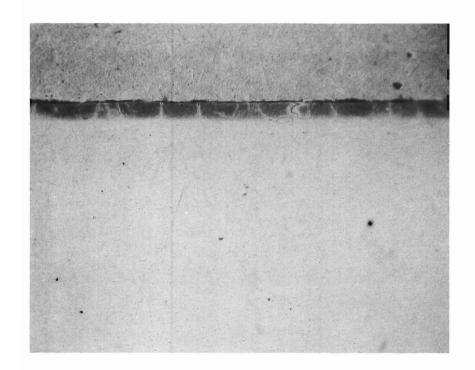
Figure 30 Microprobe Profile Scan of Coating 686 Showing Qualitative Concentration

c. From Titanium-Chromium Alloys

The previous methods for producing mixed titanium and chromium chlorides yielded deposits containing a high concentration of one of the elements. Obtaining deposits having ratios closer to the midrange of the binary system requires a different ratio of titanium and chromium chlorides in the gas phase. Reaction of TiCl₄ with alloyed titanium-chromium might serve to control this ratio. This method, if workable, could be conveniently used to include additional elements, such as vanadium, to form alloys of three or more components. Alloys of 60% Cr-40% Ti and 25% Cr-75% Ti obtained from Oregon Metallurgical Corporation were installed in the chloride generation chamber and depositions made.

Runs 778 to 781 of Appendix A utilized the 25% Cr-75% Ti alloy. Coating appearance and uniformity were good, and the surface appeared similar to titanium surfaces. Figure 31 is a photomicrograph of a cross section of a coating made in this manner. The deposition rate was slower than from pure metal chloride sources. Electron microprobe analysis (discussed later in this section) showed that these deposits were nominally in the range desired. This method thus demonstrated the capability of yielding mid-range composition alloys of Ti-Cr. The lowered deposition rate and slight variation in the composition occurring in the coating cross section may not be typical of this method in general and could be eliminated or overcome with additional experimentation.

Runs 782 to 788 in Appendix A utilized the 60% Cr - 40% Ti alloy as a chloride source, and again the deposition rate was slower than it was with the elemental (dual chamber) source. The coating surface had the "alloy" appearance, and uniformity was fair. Figure 32 is a photomicrograph of the coating cross section. Electron microprobe analysis showed the deposits were primarily chromium. This method of generating chloride would therefore be satisfactory for forming alloys of high chromium content rather than alloys in the mid-range of composition.



780 500 X

Figure 31 Ti-Cr Coating Prepared with 75% Ti - 25% Cr Alloy Chloride Source



788 500 X

Figure 32 Ti-Cr Coating Prepared with 40% Ti - 60% Cr Alloy Chloride Source

Samples from the alloy chloride source depositions were also analyzed. The 60% Cr-40% Ti alloy gave a coating of the composition shown in Figure 33. Since the chromium content is above the desired value for this study, this particular method is not preferred. Utilizing the 25% Cr-75% Ti alloy in the chloride generation chamber yielded deposits of the composition sketched. in Figure 34. These deposits are in the desired range for this study. A gradation in composition is noted for this sample, though this trend is not necessarily typical of the method. Since this process yielded coatings of the desired composition, it could be used for formation of the Ti-Cr-Si coating.

d. From Separated Titanium and Chromium

Mixed titanium and chromium chlorides were also generated over a wide range of Ti:Cr ratios by isolating the two metals in separate compartments in the chloride generation chamber (See Figure 35). The incoming ${\rm TiCl}_4$, carried in an argon stream, was divided in the desired ratio by varying the size of the openings into the compartments. Gases exiting from the compartments were fed into a common line and impinged on the sample through a single jet. As with other methods of chloride generation, the hydrogen for reduction is mixed with the chlorides in the space between the jet and the sample surface.

Experiments using this dual-compartment chloride-generation process were carried out as listed in runs 698 to 759, Appendix A. After some preliminary depositions, results were consistent and in the general range desired, so it was feasible to carry out a statistical plan series. The plan used was similar to the individual component deposition plans, and the independent variables were also similar to the previous plans, i.e., reactor temperature, flow of gases, and flow of titanium chloride (which was also proportional to the concentration of total chlorides in the gas phase). An additional process variable which could be studied is the proportion of TiCl₄ passed over the metals in the chloride generation chamber. This would require only a change in the design of the chloride chamber. The ranges of the variables were approximately the same as in previous plans.

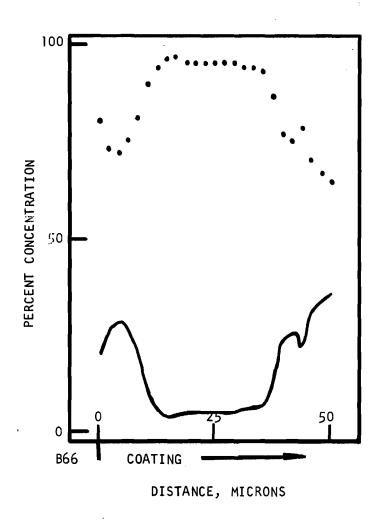


Figure 33 Electron Microprobe Profile Scan of Coating 788 Showing Qualitative Concentration (—— Ti, ... Cr)

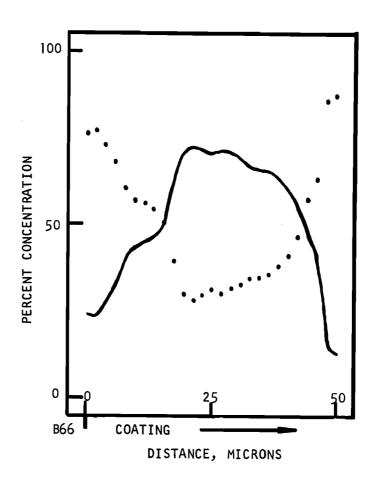


Figure 34 Electron Microprobe Profile Scan of Coating 780 Showing Qualitative Concentration (—— Ti, ... Cr)

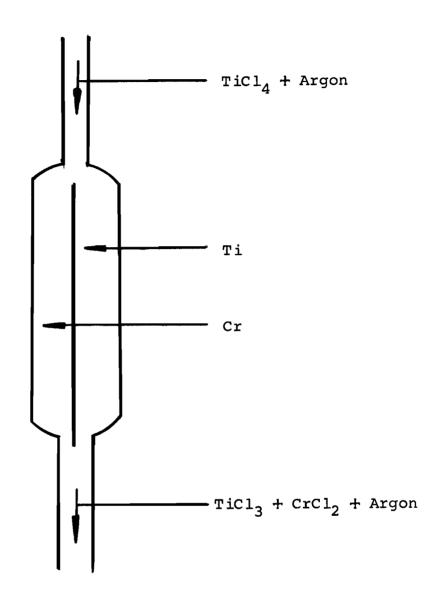


Figure 35 Sketch of Two-Compartment Chloride Generating Chamber

The runs in this plan and the resulting deposition rates are listed in Table IX.

Deposition rates from this plan are plotted as a function of the independent variables in Figure 36(a), (b), (c), and (d). All independent variables contributed positively and about equally to the deposition rate for the higher levels of the variables.

A multiple regression and correlation analysis of the data was run, and the results are given in Table X. The correlation coefficient showed a relatively good fit by this equation, though it was lower than might be expected from the apparent consistency of the plotted data. This could indicate that the linear model is not the best choice and that others could be found to give fit. The indicated F values in the table for a 95% confidence level show these variables are significant at this level.

The analysis of variance results are given in Table XI. They also indicate that the variables selected are significant at the 95% level and that the amount of interaction between variables is not significant at this level.

Figure 37 shows photomicrographs of cross sections of a portion of the plan. In general, coatings are of good quality and differ primarily in hardness and evenness of the top surface. Figure 38 is a photograph of a surface typical of the deposits of this series. It was quite different from the analogous photographs of titanium and chromium shown in previous reports. Codeposition was apparently controlled by a reaction mechanism different from that controlling single-element deposition, but the data are insufficient to precisely define this mechanism.

The electron beam microprobe was used to semiquantitatively analyze cross-sectioned samples from the statistical plan utilizing the dual chamber chloride source. Concentration counts for the individual elements were taken at 1-micron intervals. The profiles are plotted in Figure 39.

TABLE IX

Plan and Results for Cr-Ti Alloy Depositions

Run No.	×ı	× ₂	× ₃		У
756	0	0	0		1.73
734	-1	1	1		1.63
735	-1	-1	-1		1.03
736	1	-1	-1		1.27
740	0	0	0		1.40
741	-1	1	-1		1.40
742	1	1	1		2.04
743	0	0	0		1.58
744	1	-1	1		1.57
759	1	1	-1		1.84
750	-1	-1	1	1	
748	0	0	0		1.70
<u>Variable</u>	<u>Le</u> H ₂ Flow Rat	evel e (l/min)	<u>-1</u> 3	<u>0</u> 4	<u>1</u> 5
•	-		1200	1050	
×2	Keactor len	Reactor Temperature (°C)		1250	1300
× ₃	TiCl ₄ Flow	TiCl ₄ Flow Rate (mg/min)		330	410
У	Weight Gair	(mg/min)			

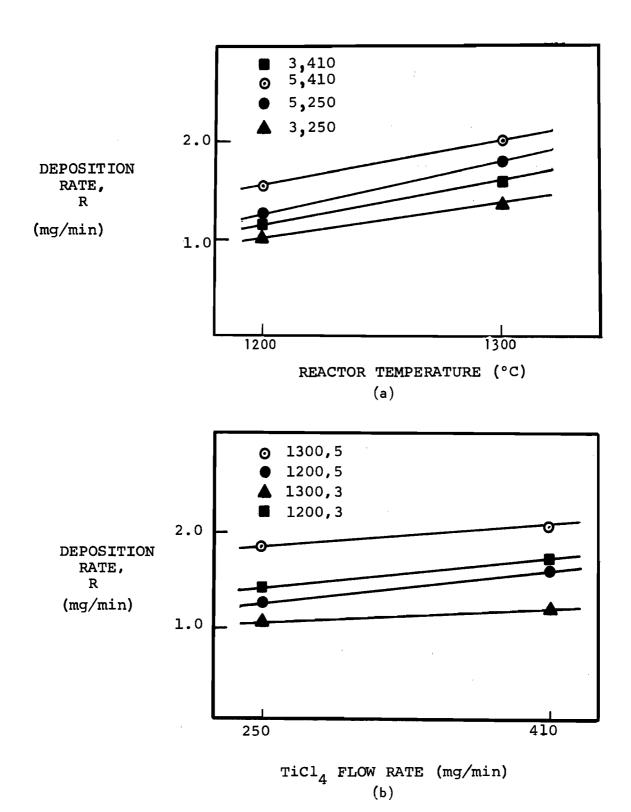
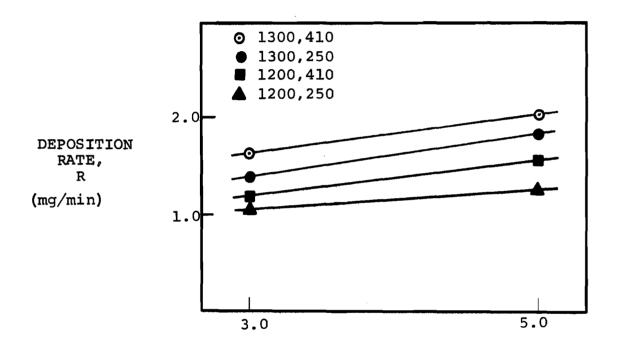


Figure 36 Deposition Rate versus Process Parameter



H₂ FLOW RATE (1/min)

Figure 36 (continued)

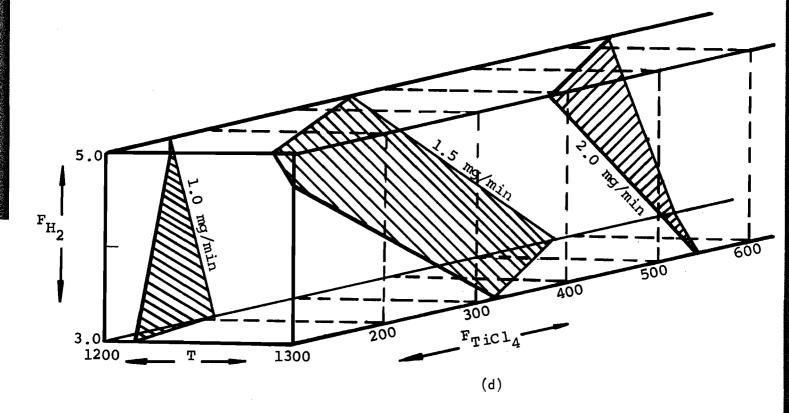


Figure 36 (continued) Planes of Constant Deposition Rate of Chromium-Titanium Alloy

(T = Reactor Temperature, °C; F_{H2} = Hydrogen Flow,

Liters/Min; F_{TiCl4} = Titanium Tetrachloride Flow,

Liters/Min)

<u>Multiple Regression and Correlation Analysis</u>
of Ti-Cr Alloy Depositions

Variable	Regression Coefficient	Observed T Value	T Value 95% Level
×ı	1.84 × 10 ⁻¹	4.45	2.36
× ₂	4.62×10^{-3}	5.60	2.36
× ₃	1.39×10^{-3}	2.69	2.36

0.17

Intercept: -5.43

Standard Error of Estimate:

Multiple Correlation Coefficient:

0.938

x, Total gas flow, 1/min

x₂ Reactor temperature, °C

x₃ TiCl₄ flow, mg/min

TABLE XI
Factorial Analysis of Variance for Alloy Depositions

NUMBER OF VARIABLES 3
NUMBER OF REPLICATES 1

VARIABLE NO. OF LEVELS

1 2
2 2
3 2

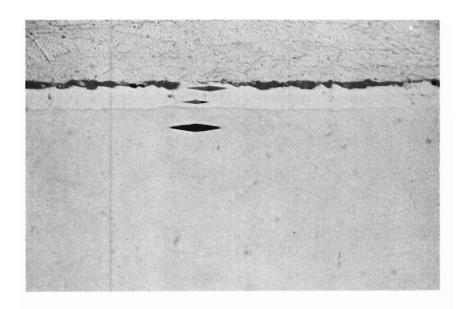
GRAND MEAN

1.49625

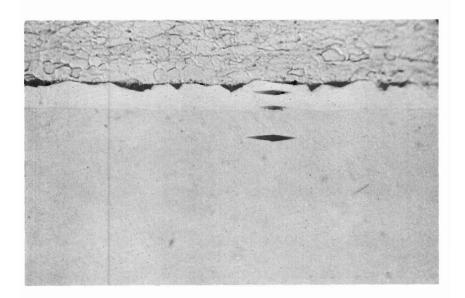
SOURCE OF VARIATION	DEGREES OF FREEDOM	SUMS OF SQUARES	F VALUE *	F REQUIRED **
1	1	0.21451	16.02	10.1
2	ì	0.04061	3.03	1
3	1	0.31601	23.57	
12	1	0.04351	3.25	
13	. 1	0.05281	3.91	
23	1	0.03781	2.81	
RESIDUAL	1	0.10351	7.72	10.1
TOTAL	7	0.80879		

$$* \frac{\Lambda^2}{\sigma} = 0.0134$$

^{**} For 95% confidence level

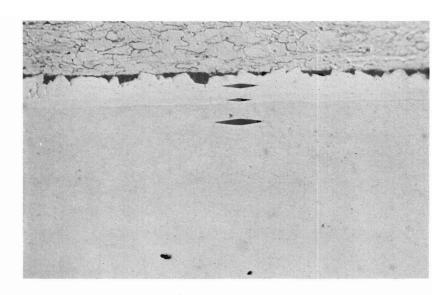


(a) Run 741

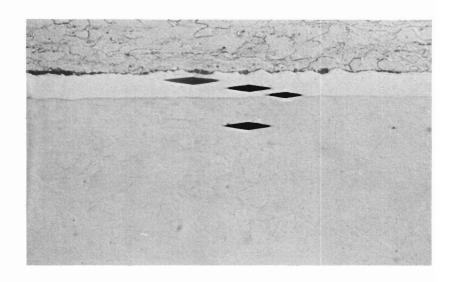


(b) Run 743

Figure 37 Codepositions of Titanium and Chromium (X500)

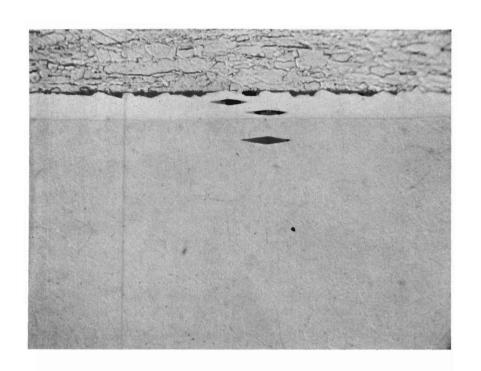


(c) Run 746



(d) Run 747

Figure 37 (continued)



(e) Run 745

Figure 37 (continued)

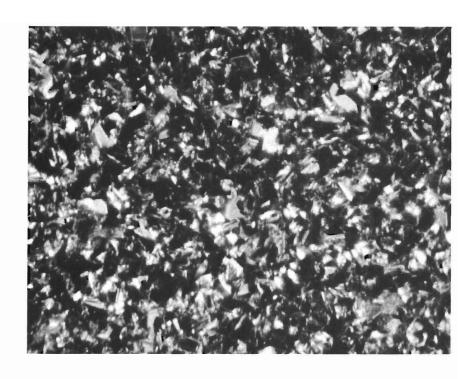


Figure 38 Surface Photograph of "as-deposited" Ti-Cr Alloy (X500)

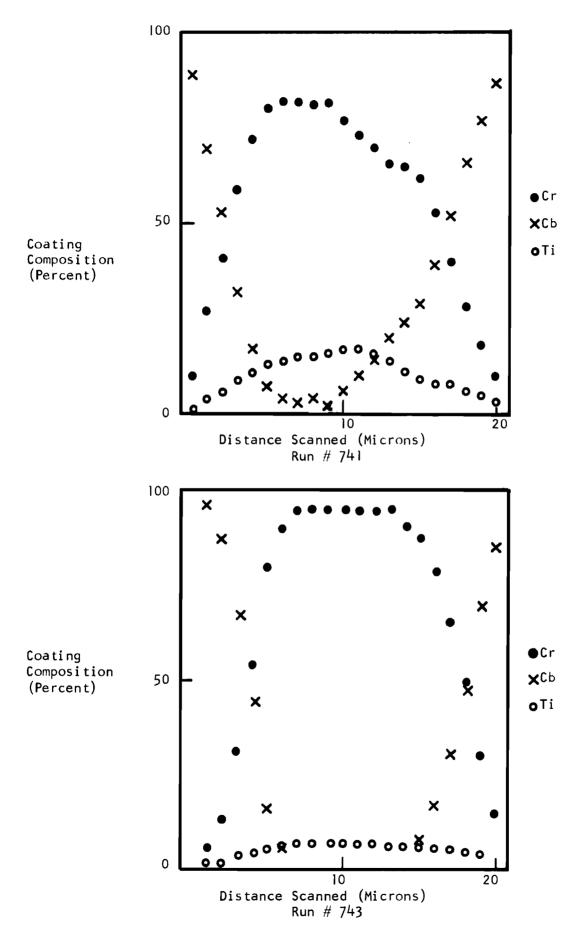
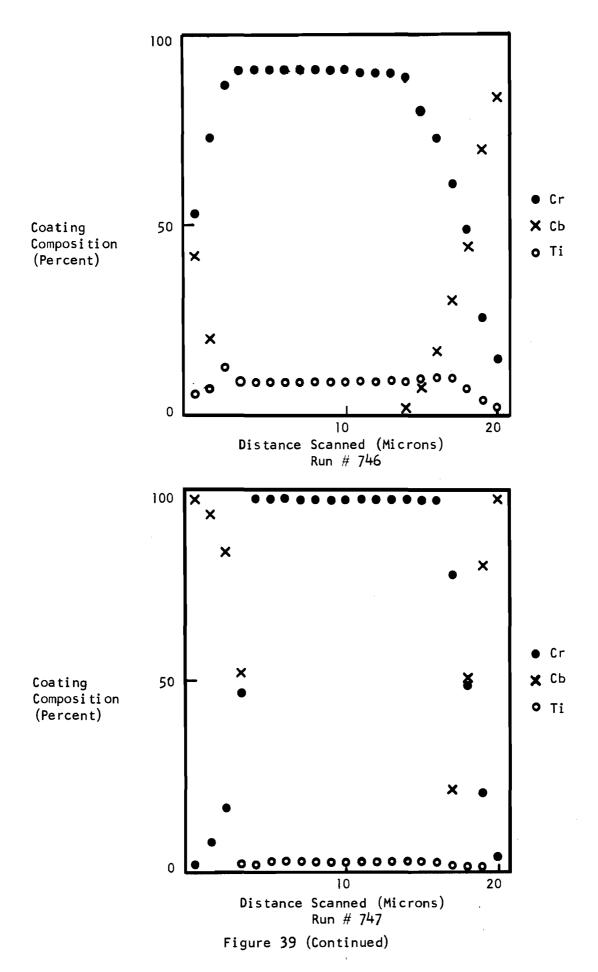


Figure 39 Profiles of Coating Composition Measured with an Electron Microprobe



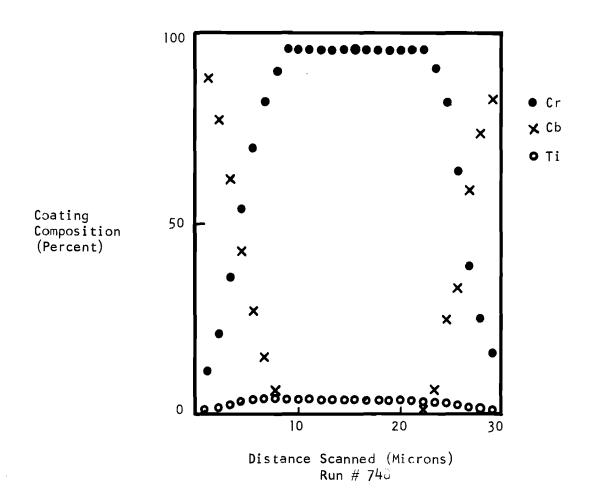


Figure 39 (Continued)

In general, concentrations were uniform through the coating, though for this particular series titanium concentrations were lower than desired. Alteration of the ratio of TiCl₄ flowing to the chromium and titanium compartments in the chloride generation chamber increased the titanium concentration in the deposits.

To test the reproducibility of the coating process, eight samples were coated on both sides. Each sample was coated on one side, removed from the reactor, weighed, then coated on the other side. The results, shown in Table XII, indicate fair reproducibility, with an average coating weight per side of 24.1 mg and an average deviation from this weight of $\pm 10\%$. This appears satisfactory for the present level of process development.

This test series was also used to determine the effect of continuous operation on the performance of the coating equipment. Performance was generally satisfactory, except that the sample temperature increased slightly as the series progressed. An improved temperature controller was installed, and the method of attaching the thermocouple that controls sample temperature was improved to permit more precise control and to facilitate changing thermocouples. This increased overall precision and control of alloy deposition. A Pt-Pt10% Rh thermocouple with a quartz protection sleeve was embedded in the heater element.

Figure 40 shows cross sections of the coatings on two of the samples. These two samples were sheared to obtain bend-test specimens, which were subjected to the 4-T bend test. Both bent to 90 degrees without fracture of coating or substrate. X-ray analysis showed a body-centered cubic structure, quite close to the lattice parameter for chromium. This would be expected, since the body-centered cubic chromium phase would occur above 67% chromium. Microprobe analysis run on these samples is reported in the analysis section of this report.

TABLE_XII

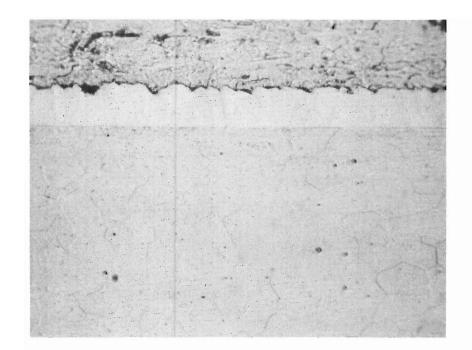
Alloy Deposition Reproducibility Series

	Coating Weight		
Run No.	Side 1	Side 2	
769	22.3	21.1	
770	20.7	22.2	
772	22.9	23.6	
773	23.6	24.9	
774	25.2	26.5	
775	15.9	28.2	
776	26.9	27.1	
777	27.6	26.7	

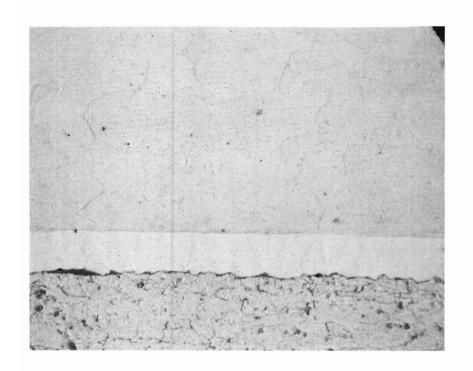
Average Weight Gain: 24.1 mg
Average Deviation: 2.5 mg

Experimental Parameters:

Temperature	1300° C
Hydrogen Flow	4.0 <i>l</i> /min
Argon Flow	4.0 <i>l</i> /min
Chromium-Titanium Temperature	750°C
Deposition Time	10 minutes

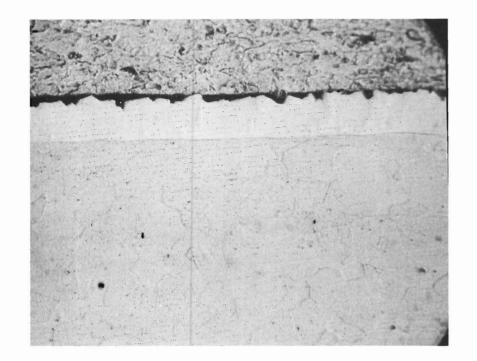


776 500 X

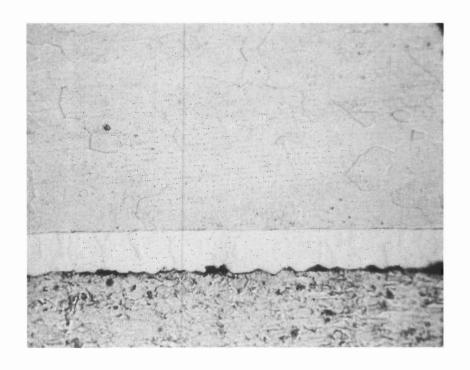


776 500 X

Figure 40 Ti-Cr Coatings on Two Sides of B-66 Substrates (Mag. 500 X)



777 500 X l in = 2 mils



777 500 X l in = 2 mils

Figure 40 (Continued)

E. Relationship of the Gas-Phase Concentration to Deposit Compositions

The investigations of codeposition of titanium-chromium alloys have further elucidated the gas-phase composition required to obtain coatings having specified metal concentrations.

Data obtained comparing alloy deposit composition and gas-phase compositions were analyzed. The plot in Figure 41 summarizes these results. The concentration of titanium is about 14 times greater in the gas than in the coating for the range of experimental variables investigated. This is to be expected in view of the relative thermodynamic stability of the chlorides. Deposit composition was slightly temperature-dependent at any gas composition, within the limits noted on the plot. In general, higher temperatures favored a higher titanium composition, which implies a higher "effective" entropy change for the titanium reduction than for the chromium reduction.

F. Silicon Deposition Process

Deposition of silicon was an established process and thus required little preliminary investigation. To check the process, deposits of silicon were made on several substrate materials by the hydrogen reduction of silicon tetrachloride. Runs 800 through 814 listed in the appendix were made in this way. Figure 42 is a cross section of silicon on tantalum; Figure 43, silicon on molybdenum; and Figure 44, silicon on titanium-chromium alloy deposit from Run 773, which before siliciding would have had a cross section similar to the Run 776 figure. Only on the Ti-Cr alloy did the silicon interact extensively with the substrates, as evidenced by the existence of several zones in the coating. Silicon content of this coating was approximately 10 mg/cm². The standard silicon deposition process was satisfactory for preparation of coatings, though the best diffusion time to interact the silicon with the alloy coating was not investigated.

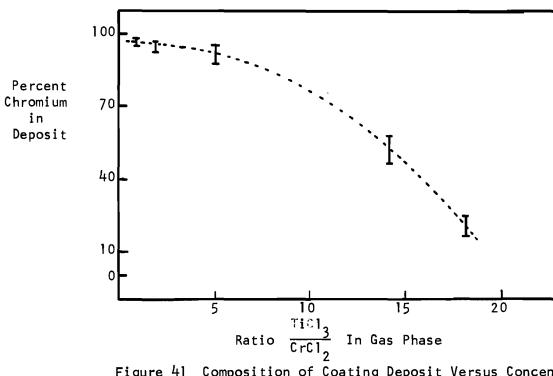


Figure 41 Composition of Coating Deposit Versus Concentration of Gas Phase for Codeposition of Titanium-Chromium Alloys

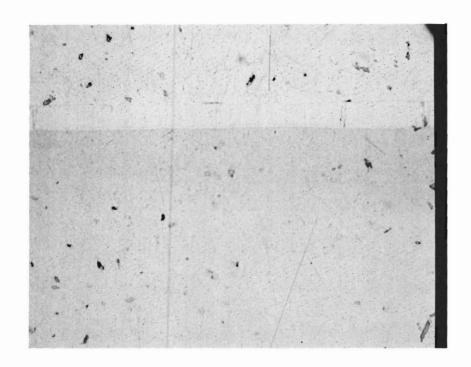


Figure 42 Silicon Coating on Tantalum (Run 808, 500X)

Figure 43 Silicon Coating on Molybdenum (Run 806, 500X)

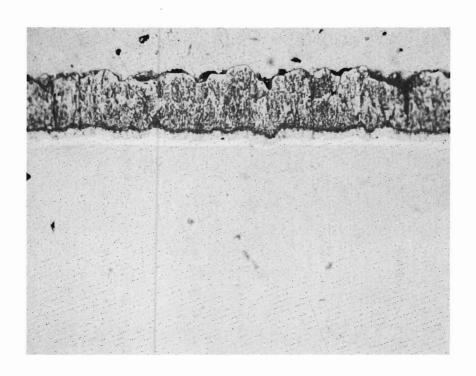


Figure 44 Titanium-Chromium-Silicon Coating on B-66 (X500)

SECTION IV

COATING PROTECTIVENESS STUDIES

The purpose of this phase of the work was to prepare and evaluate coatings to determine the applicability of the CVD process for preparing the titanium-chromium-silicon coating. Previous efforts (Phase I) had resulted in several processes by which the metal alloy coatings could be prepared. To obtain the maximum range of possible compositions, the two-chamber process was selected for the alloy deposition. In this process the titanium-to-chromium ratio is adjusted mechanically and thus remains constant throughout a series.

The overall plan was to evaluate a number of series, each containing several samples. This approach yielded a more realistic view of the relation of performance to coating procedure and also served to test the capability to prepare coatings reproducibly. Six samples were coated in each series and subjected to test and evaluation as follows:

Sample No.	
1	Retained for Reference
2	Oxidation Test 2300°F
3	Oxidation Test 2500°F
4	Cross Section Analysis
5	Oxidation Test 2300°F
6	Oxidation Test 2500°F

The coating was formed by depositing the desired alloy coating on one side, removing and weighing the sample, then alloy coating the other side. The samples were then siliconized by depositing and reacting silicon using hydrogen reduction of $SiCl_4$.

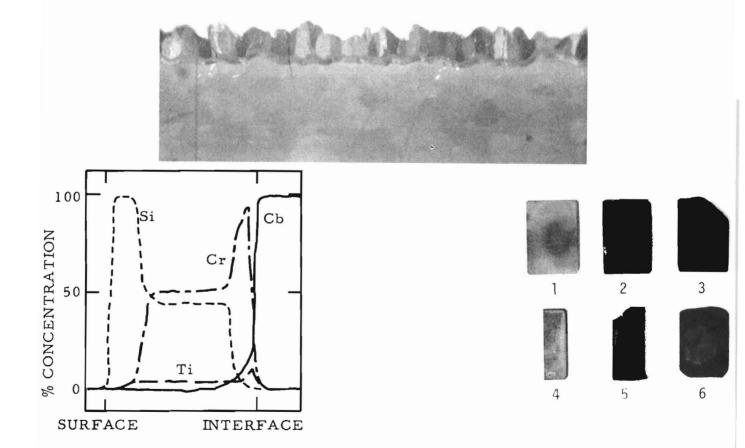
Samples were oxidized in a silicon carbide element electric form or in slowly moving air. They were supported on Leco zirconia boats in a mullite tube. The boat was inserted into the hot furnace, rapidly removed at approximately 1 hour intervals during the day for visual examination and weighing,

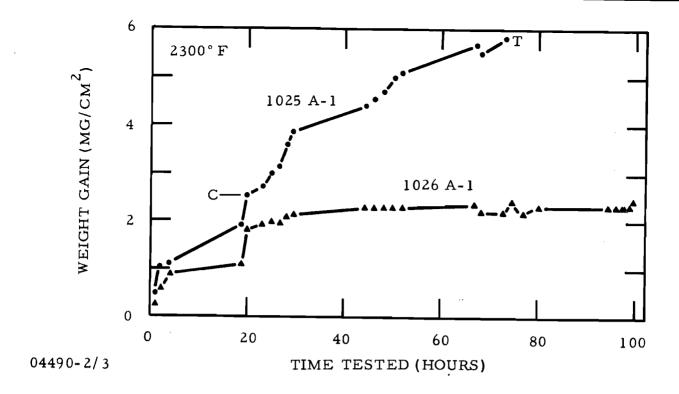
then returned to the hot furnace. The samples were allowed to soak overnight at the test temperature.

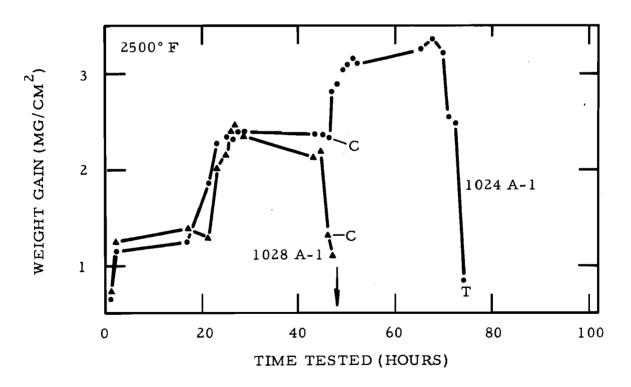
Results of the evaluation for each series are shown in the following pages and are summarized in Table XIII.

SERIES A-1

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1023	18.4	15.0		
2	1026	17.6	8.2	2300	100
3	1028	18.6	8.0	2500	70
4	1027	18.6	9.0		
5	1025	19.0	8.6	2300	74
6	1024	19.2	7.1	2500	89



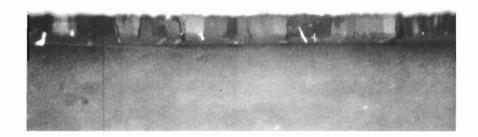


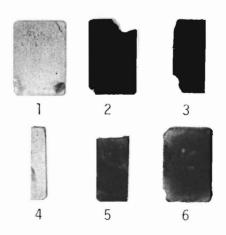


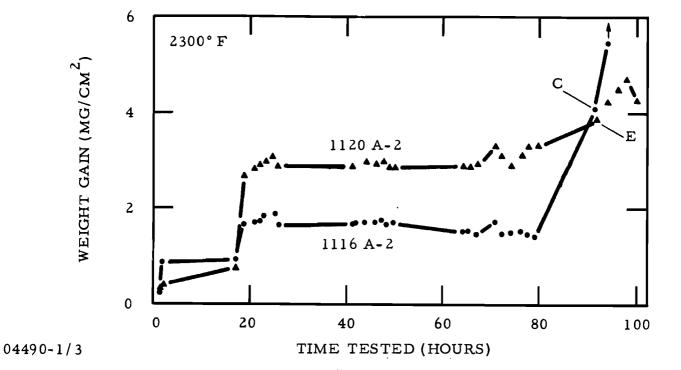
Weight Gain of Coatings in Air for Series A-1

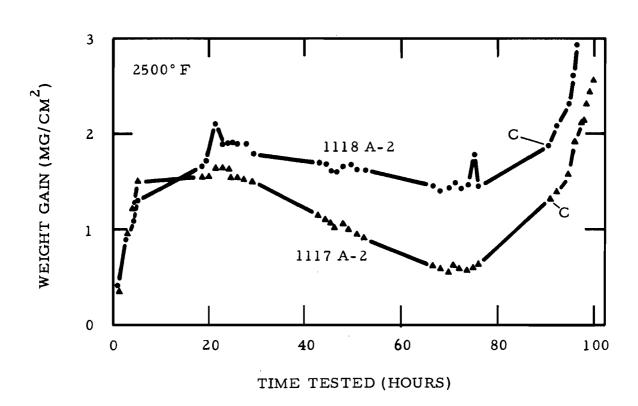
SERIES A-2

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1 2	1119 1120	6.8 6.3	8.3 7.4	2300	100
3	1118	6.0	9.8	2500	100
4	1115	5.9	9.1		
5	1116	6.0	9.1	2300	100
6	1117	6.2	10.4	2500	100





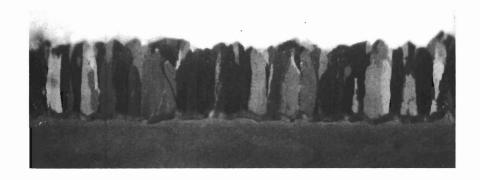


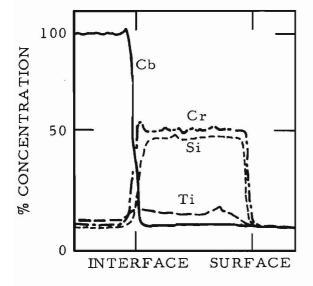


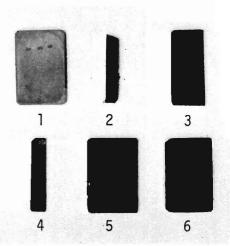
Weight Gain of Coatings in Air for Series A-2

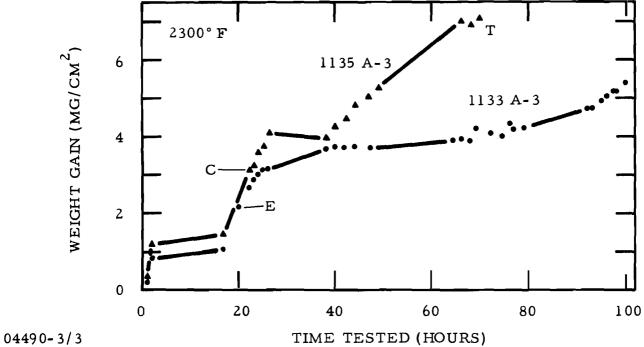
SERIES A-3

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1131	11.0	12.8		
2	1135	11.8	13.8	2300	70
3	1136	10.7	13.4	2500	100
4	1137	11.3	13.6		
5	1133	10.7	12.8	2300	100
6	1129	10.3	12.0	2500	100

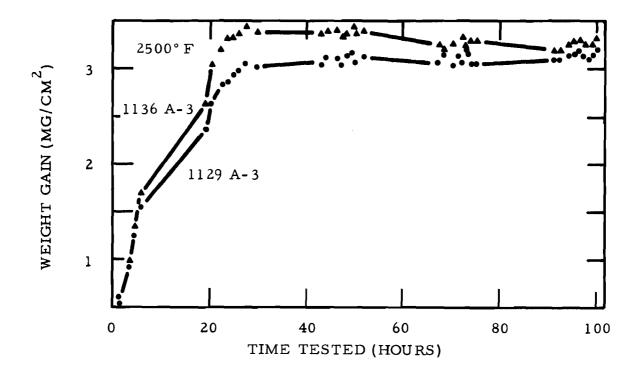








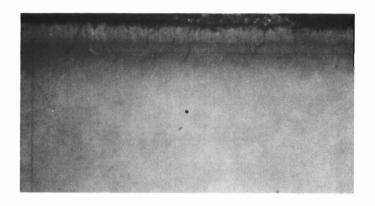


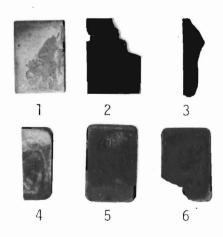


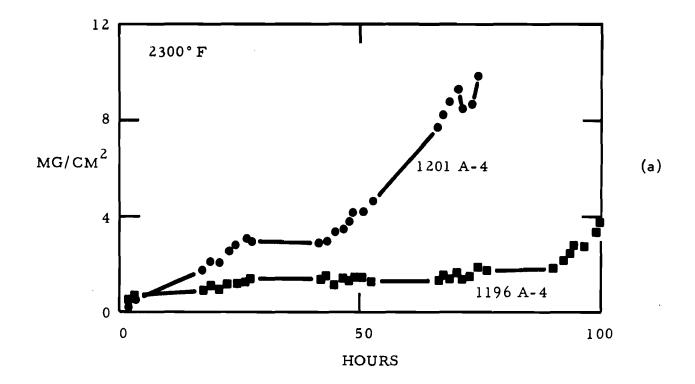
Weight Gain of Coatings in Air Series A-3

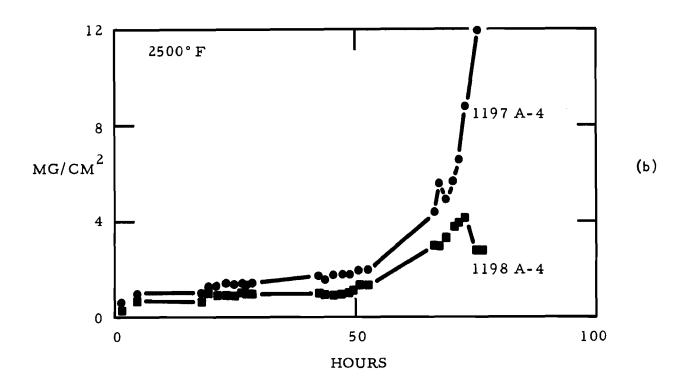
SERIES A-4

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1200	8.0	8.2		
2	1201	6.6	9.1	2300	75
3	1198	8.4	10.9	2500	90
4	1199	9.1	7.5		
5	1196	6.1	9.1	2300	100
6	1197	7.8	9.5	2500	76





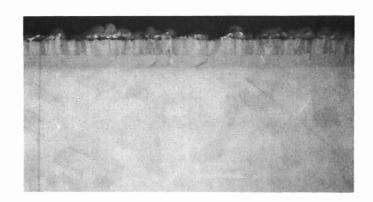


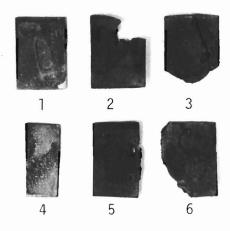


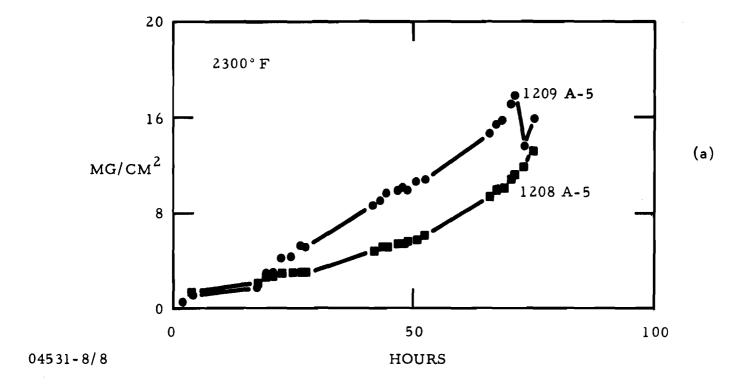
Oxidation Test Results for Ti-Cr-Si Coating (Weight Gain vs Time) Series A-4

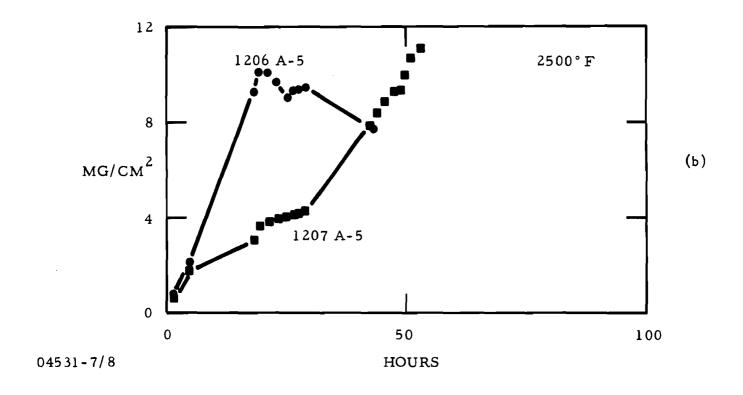
SERIES A-5

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1204	12.6	7.4		
2	1209	10.7	10.4	2300	76
3	1207	11.6	12.5	2500	67
4	1205	14.2	13.2		
5	1208	10.3	8.8	2300	76
6	1206	14.0	15.0	2500	44





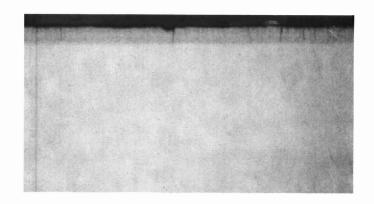


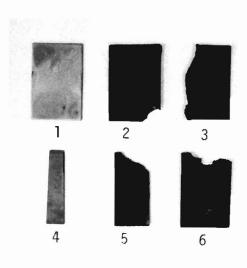


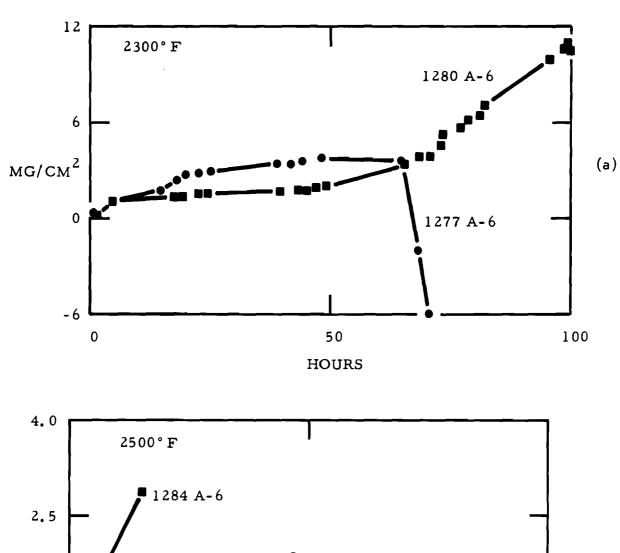
Oxidation Test Results for Ti-Cr-Si Coating (Weight Gain vs Time) Series A-5 $\,$

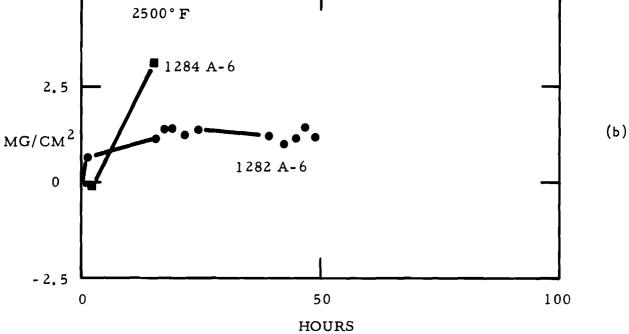
SERIES A-6

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1281	9.0	5.6		
2	1280	10.1	5.8	2300	100
3	1284	10.1	6.4	2500	15
4	1283	12.0	6.3		
5	1277	10.0	7.4	2300	70
6	1282	9.4	8.0	2500	49





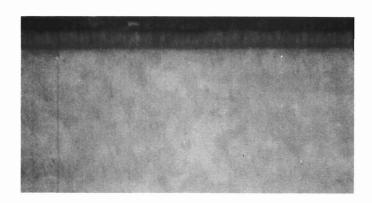


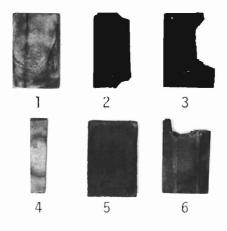


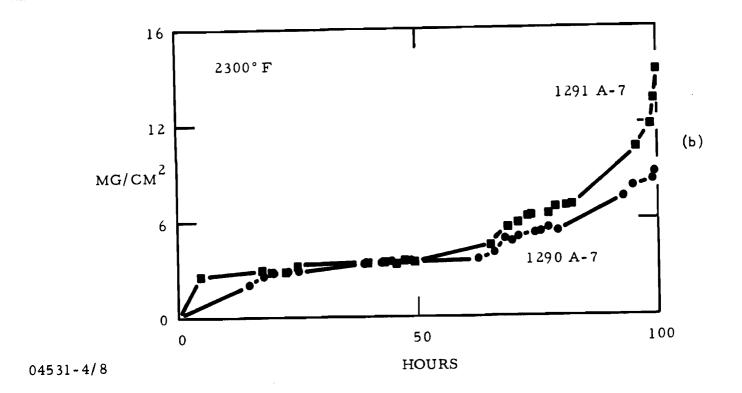
Oxidation Test Results for Ti-Cr-Si Coating (Weight Gain vs Time) Series A-6 $\,$

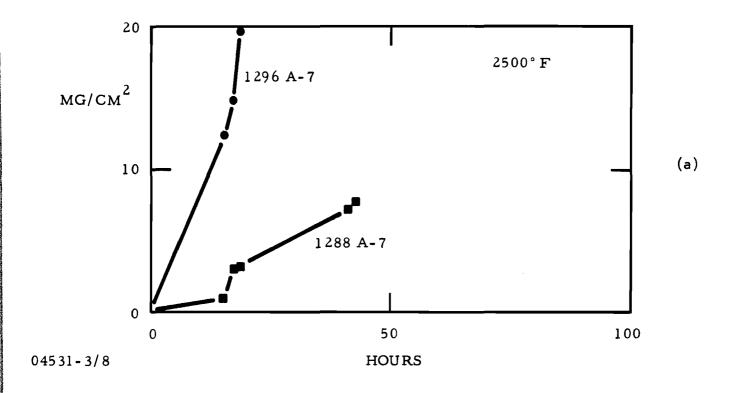
SERIES A-7

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1289	10.4	5.4		
2	1291	8.8	10.2	2300	100
3	1296	9.8	11.5	2500	18
4	1286	10.8	5.5		
5	1290	9.0	9.6	2300	100
6	1288	11.7	6.8	2500	43





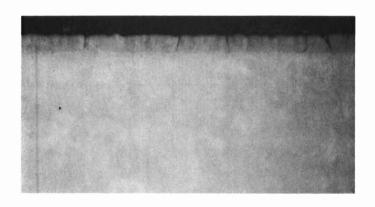


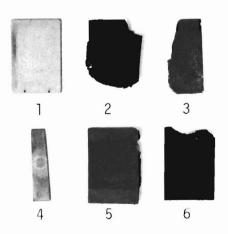


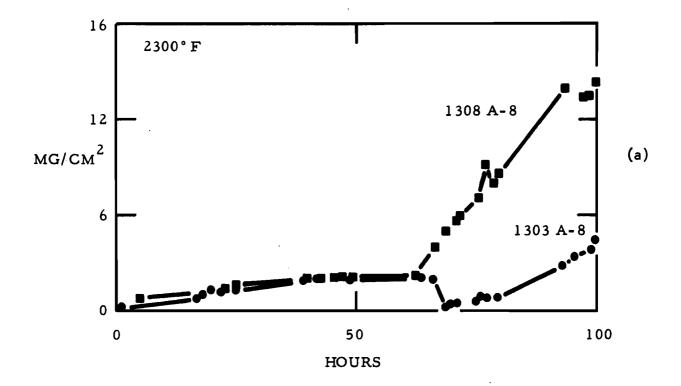
Oxidation Test Results for Ti-Cr-Si Coating (Weight Gain vs Time) Series A-7

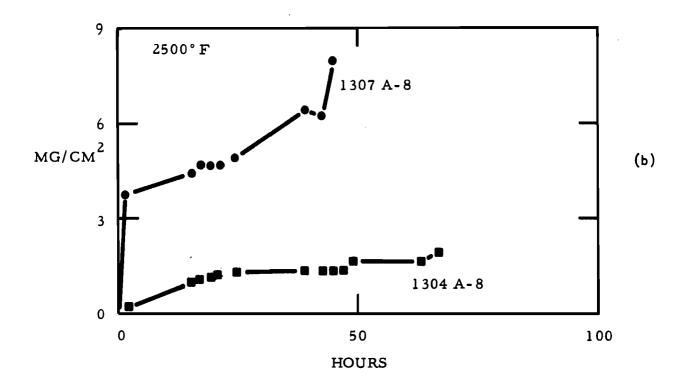
SERIES A-8

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1301	8.0	7.4		
2	1308	8.6	7.6	2300	100
3	1307	9.2	8.4	2500	49
4	1309	8.8	7.4		
5	1303	8.6	9.9	2300	100
6	1304	8.6	9.2	2500	67





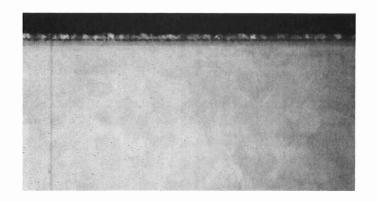


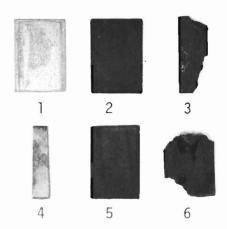


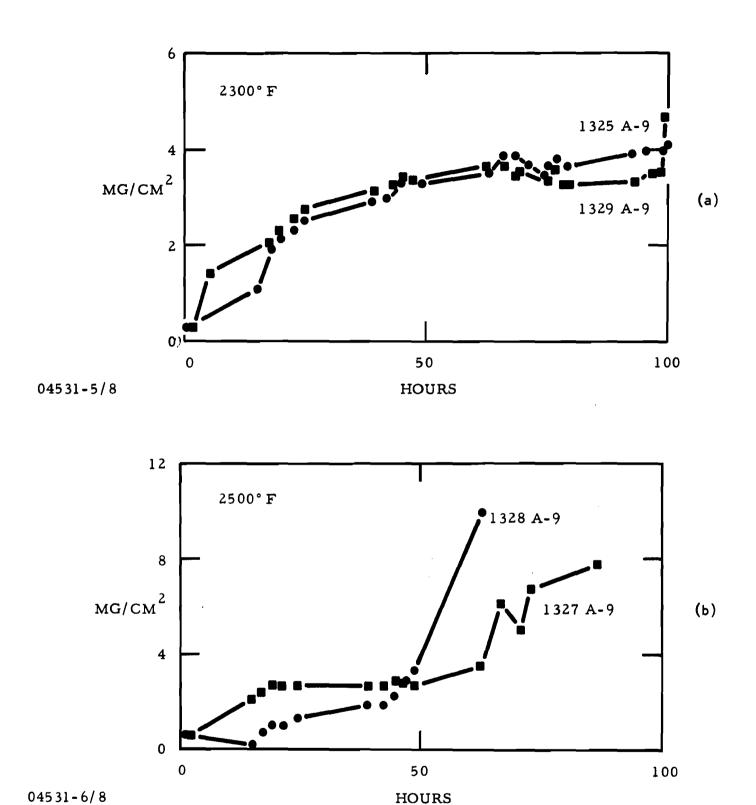
Oxidation Test Results for Ti-Cr-Si Coating (Weight Gain vs Time) Series A-8

SERIES A-9

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1324	9.5	8.6		
2	1329	10.9	8.5	2300	100
3	1328	10.7	8.8	2500	63
4	1326	10.1	8.3		
5	1325	10.4	8.3	2300	100
6	1327	10.5	9.1	2500	88



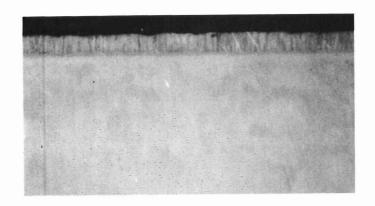


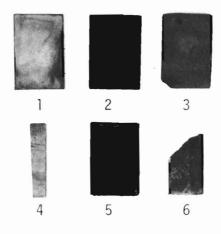


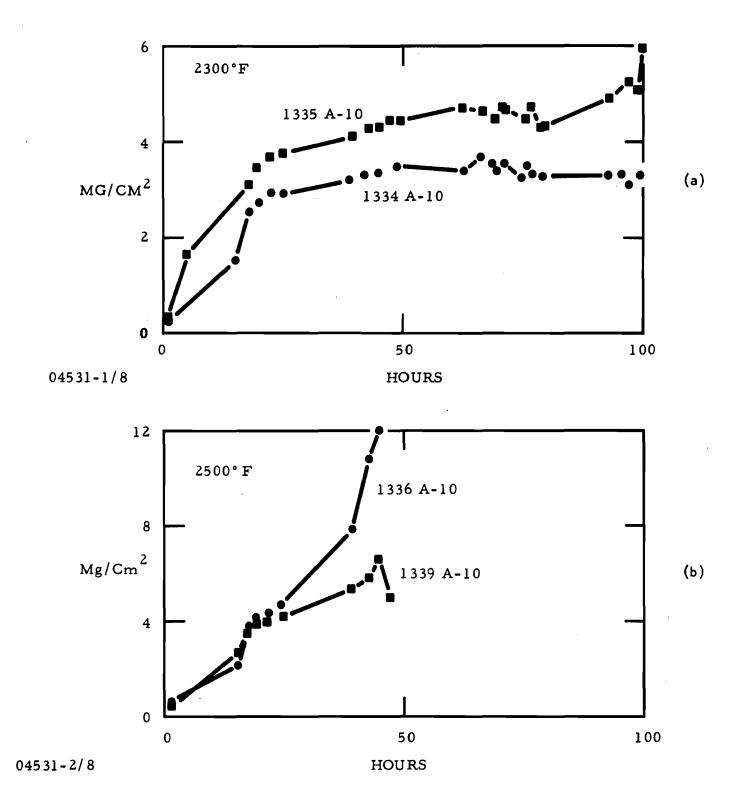
Oxidation Test Results for Ti-Cr-Si Coating (Weight Gain vs Time) Series A-9 $\,$

SERIES A-10

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1337	13.4	7.9		
2	1335	14.3	11.9	2300	100
3	1339	12.2	16.4	2500	63
4	1340	11.7	11.3		
5	1334	15.9	10.5	2300	100
6	1336	12.7	16.4	2500	63



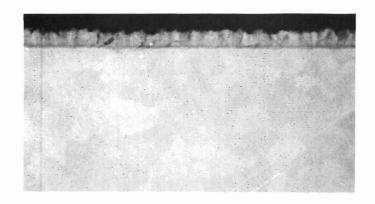


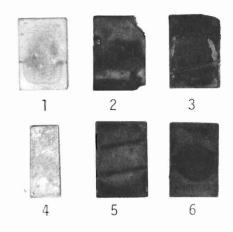


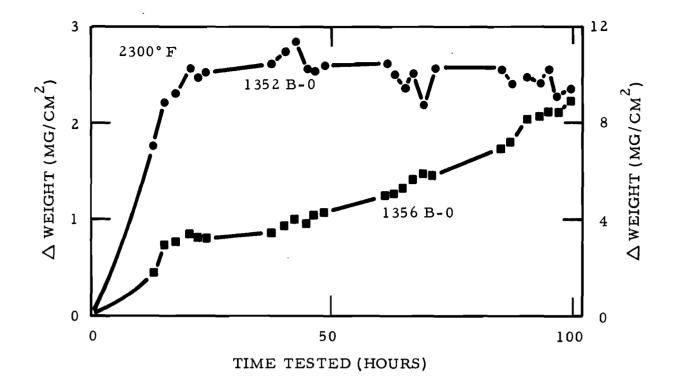
Oxidation Test Results for Ti-Cr-Si Coating (Weight Gain vs Time) Series A-10

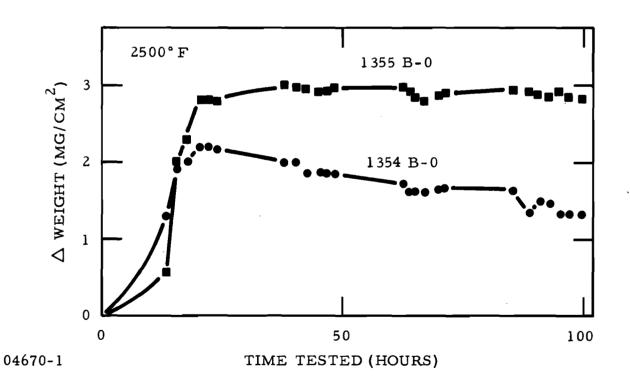
SERIES B-0

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1357	11.7	10.3		
2	1356	11.1	8.9	2300	100
3	1355	11.6	6.7	2500	100
4	1353	10.8	5.9		
5	1352	13.0	6.7	2300	100
6	1354	10.3	7.5	2500	100



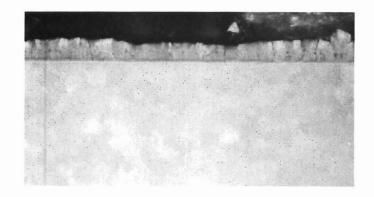


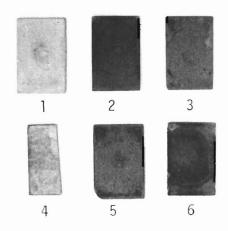


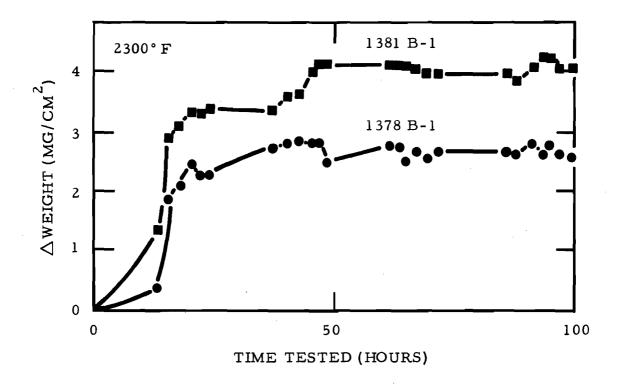


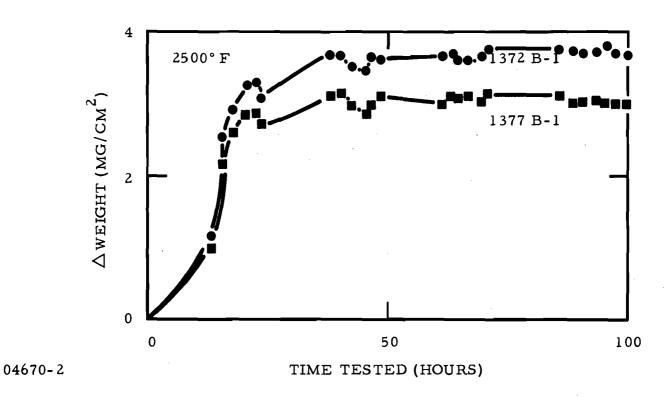
SERIES B-1

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
140.	NO.	(iig/ciii)		()	(Hours)
1	1380	13.6	8.4		
2	1381	13.0	8.0	2300	100
3	1377	12.4	7.0	2500	100
4	1379	12.8	8.0		
5	1378	12.7	8.3	2300	100
6	1372	13.6	7.8	2500	100



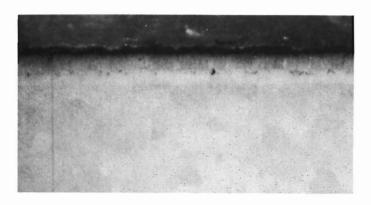


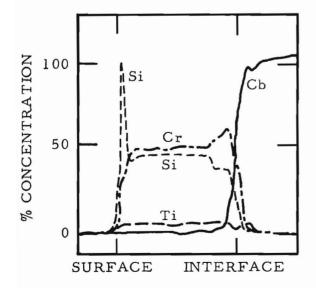


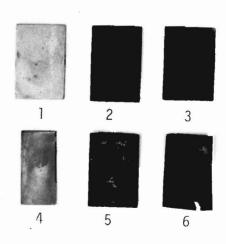


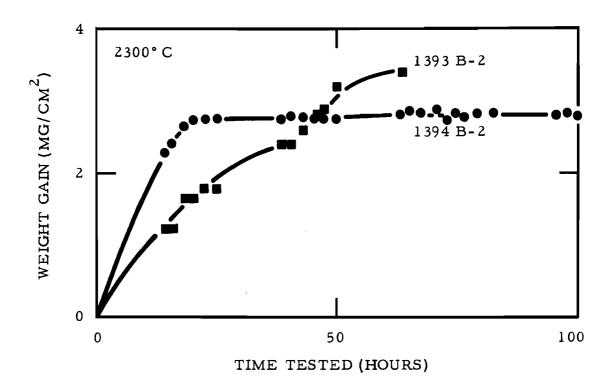
SERIES B-2

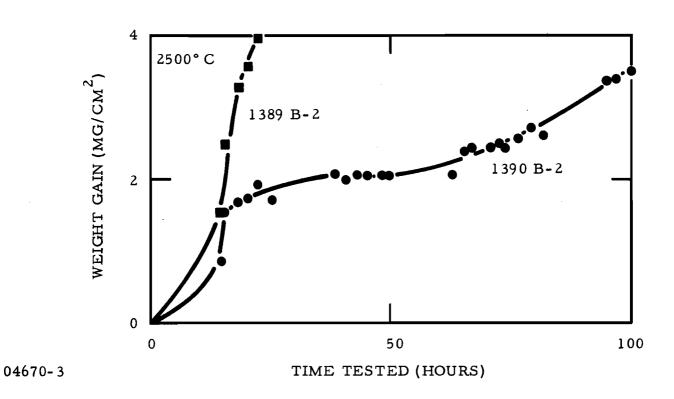
No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1395	15.0	10.4		
2	1394	12.6	8.7	2300	100
3	1390	13.4	7.3	2500	100
4	1391	12.0	7.8		
5	1393	10.2	6.7	2300	68
6	1389	8.8	7.4	2500	38





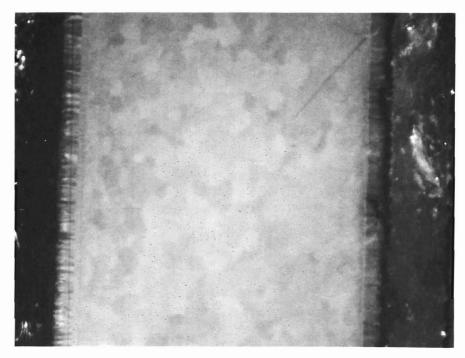


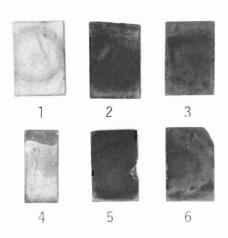


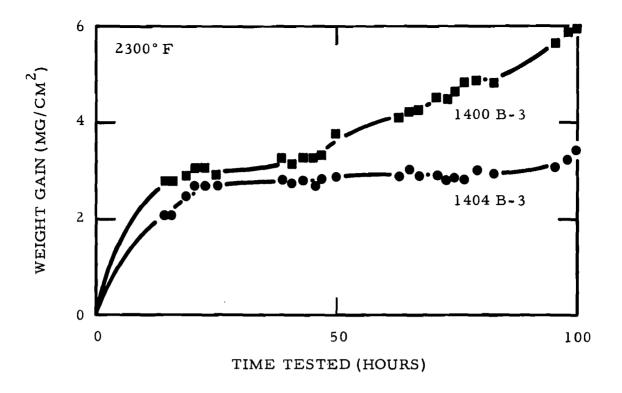


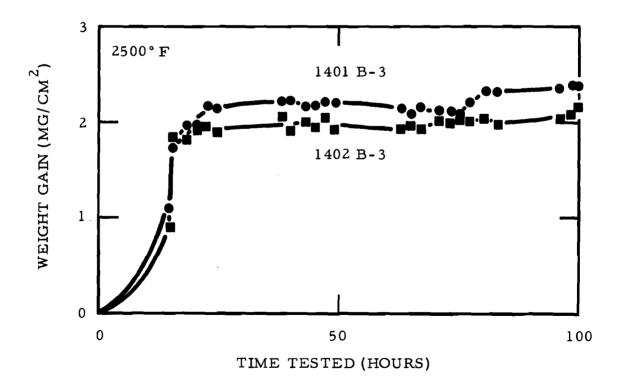
SERIES B-3

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1403	9.6	10.5		
2	1404	10.2	11.5	2300	100
3	1402	8.4	8.1	2500	100
4	1398	9.0	10.7		
5	1400	8.4	9.1	2300	100
6	1401	11.4	7.0	2500	100





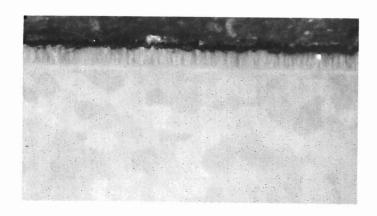


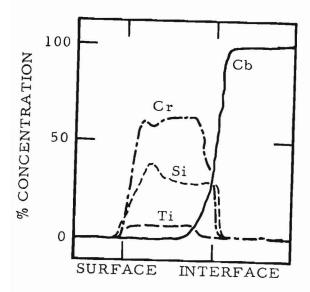


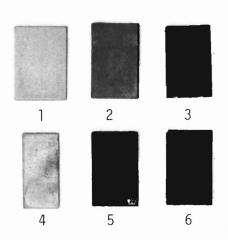
04670-4

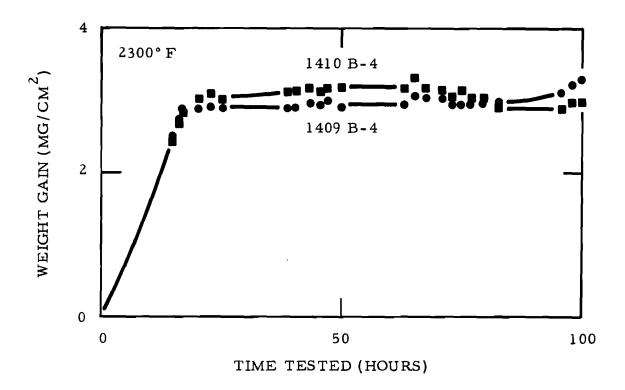
SERIES B-4

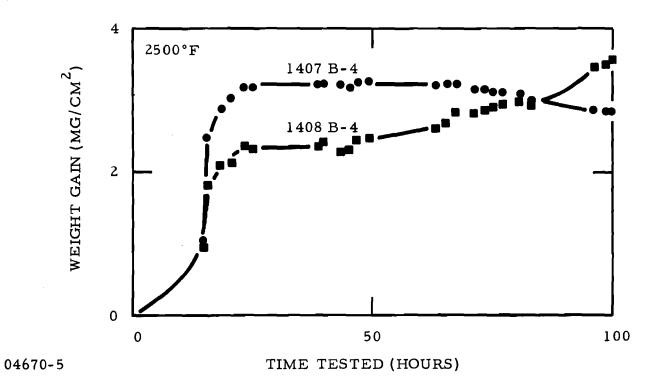
No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1412	10.6	10.1		
2	1410	9.8	10.1	2300	100
3	1408	8.8	10.1	2500	100
4	1411	10.5	7.6		
5	1409	9.5	11.6	2300	100
6	1407	10.3	8.6	2500	100





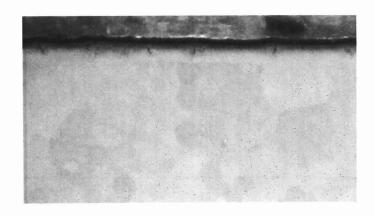


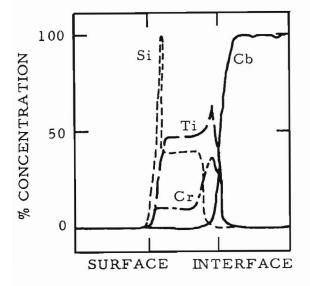


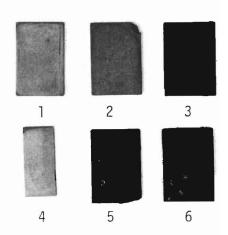


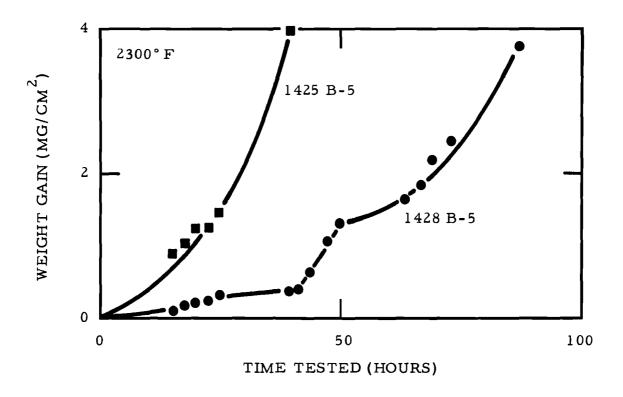
SERIES B-5

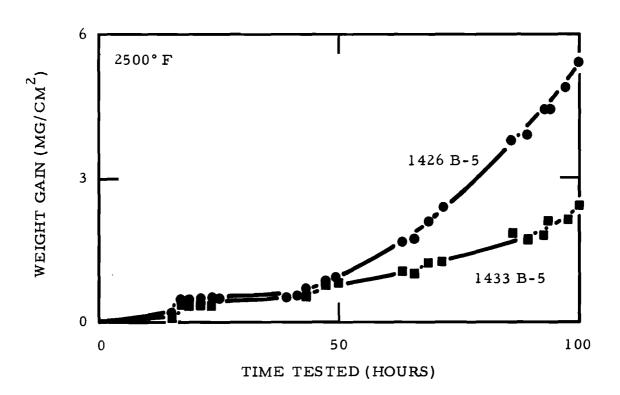
No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1427	9.2	4.4		
2	1428	9.7	5.2	2300	87
3	1433	10.8	6.8	2500	100
4.	1434	8.4	3.2		
5	1425	10.3	5.8	2300	40
6	1426	9.4	7.4	2500	100







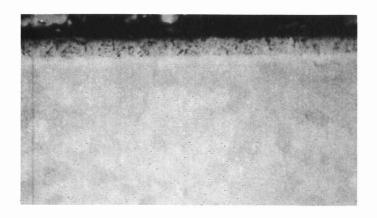


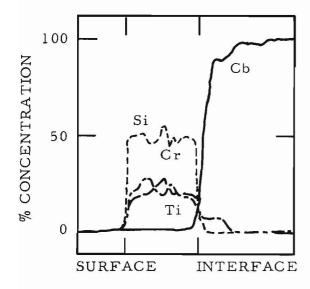


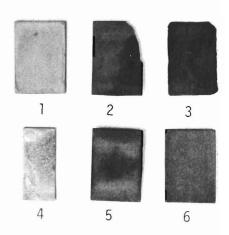
04670-6

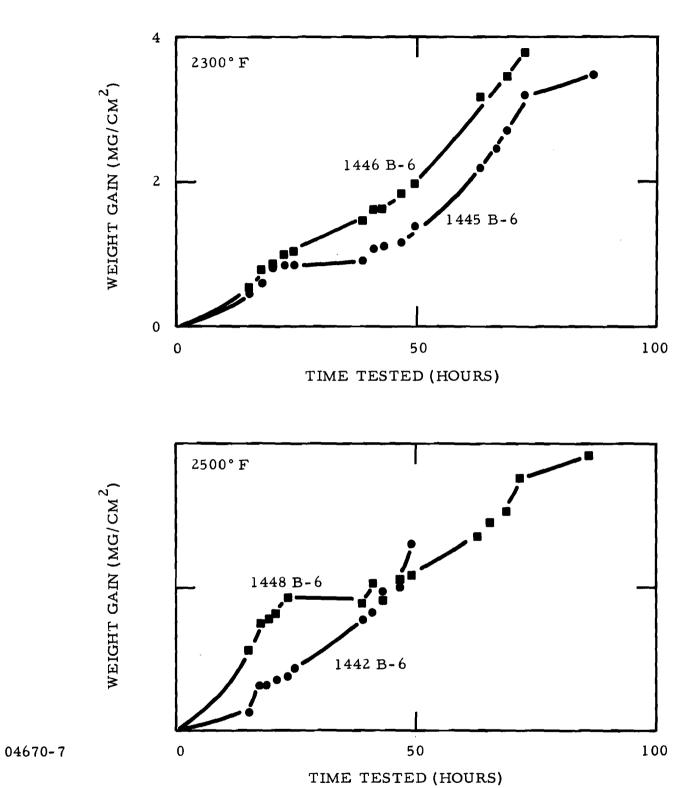
SERIES B-6

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1443	10.4	3.8		
2	1446	8.2	6.3	2300	86
3	1448	8.3	8.6	2500	86
4	1444	10.1	4.6		
5	1445	8.3	8.4	2300	92
6	1442	9.9	6.9	2500	63

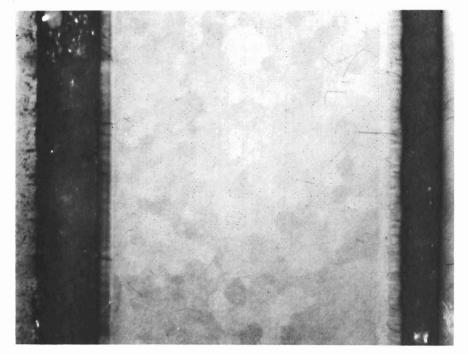


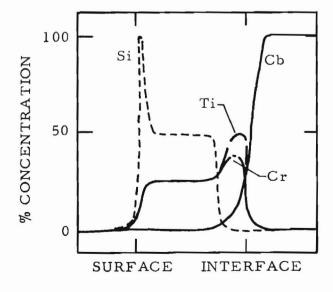


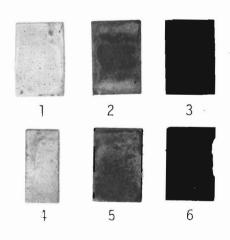


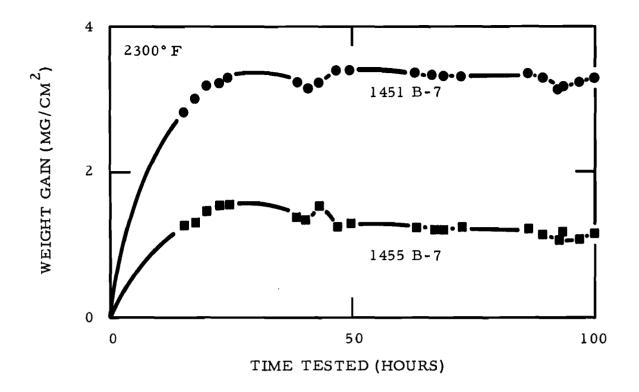


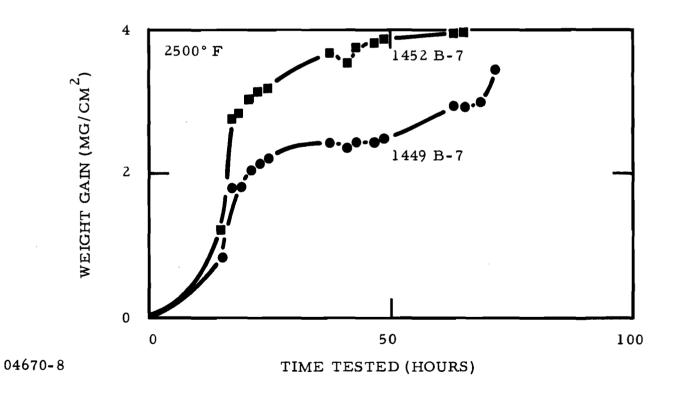
No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1454	9.0	5.2		
2	1455	8.9	6.7	2300	100
3	1452	10.3	9.1	2500	86
4	1453	9.5	6.8		
5	1451	8.9	11.2	2300	100
6	1449	8.1	7.4	2500	86





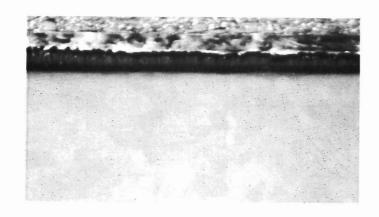


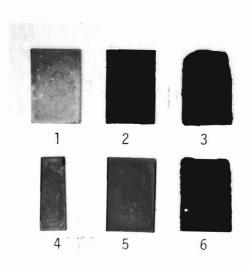


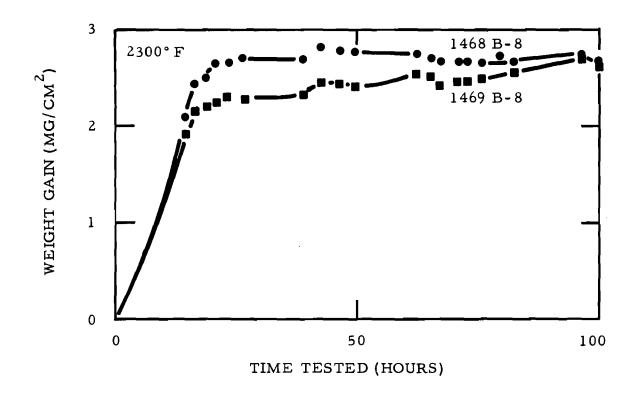


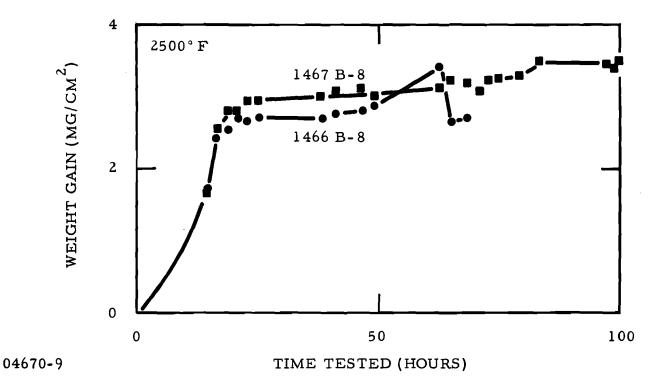
SERIES B-8

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1 2 3 4 5	1465 1469 1466 1464 1468 1467	10.0 8.0 8.8 10.0 9.3 8.1	8.7 7.9 9.1 5.1 8.1 8.3	2300 2500 2300 2500	100 73 100 100



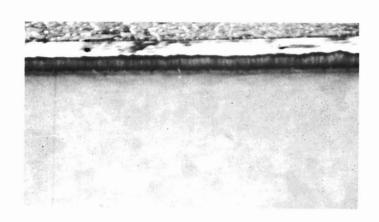


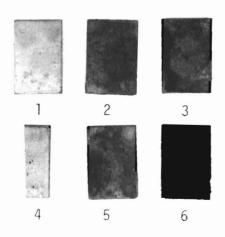


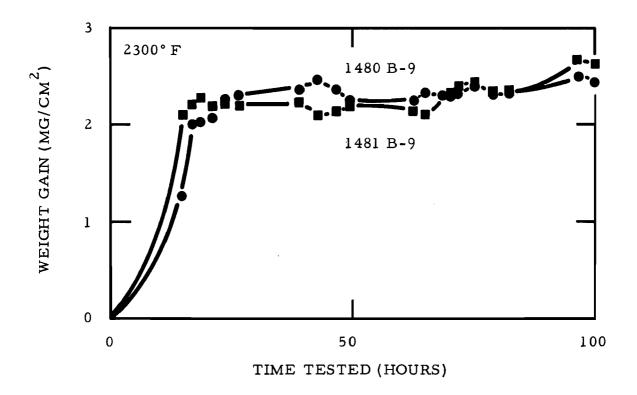


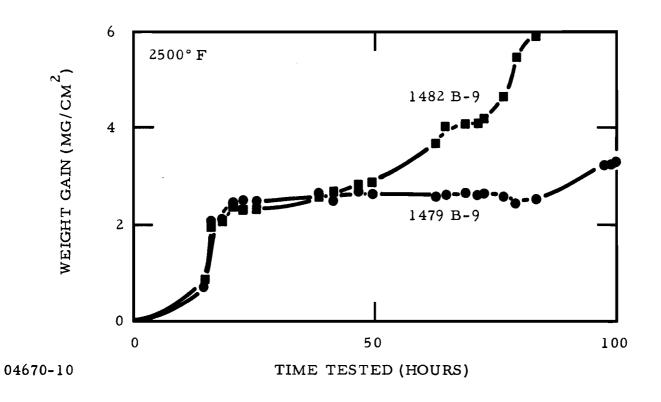
SERIES B-9

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1 2 3 4	1476 1481 1482 1478	8.0 8.1 8.2 9.6	3.1 7.7 7.8 4.6	2300 2500	100 83
5 6	1480 1479	8.8 8.4	6.9 6.6	2300 2500	100 100



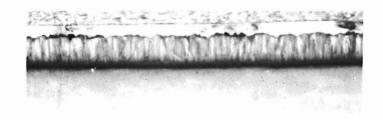


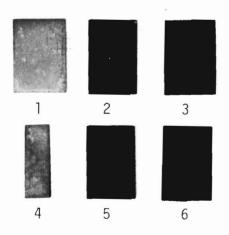


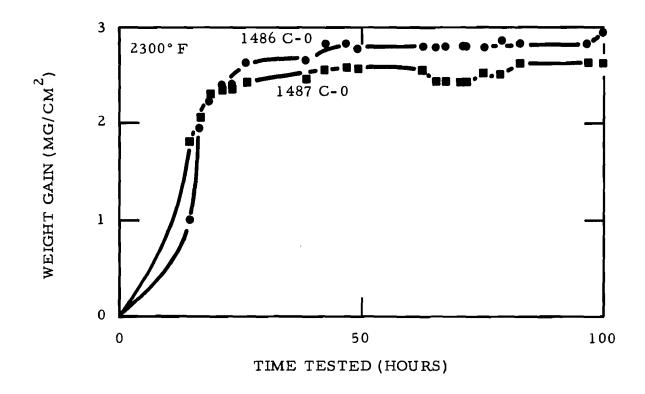


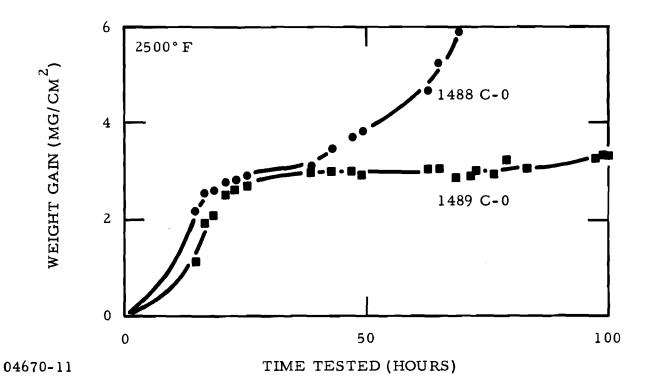
SERIES C-0

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1484	10.8	11.3		
2	1487	8.9	9 9	2300	100
3	1489	8.5	11.7	2500	100
4	1485	9.7	9.3		
5	1486	9.3	9.8	2300	100
6	1488	9.0	9.8	2500	73



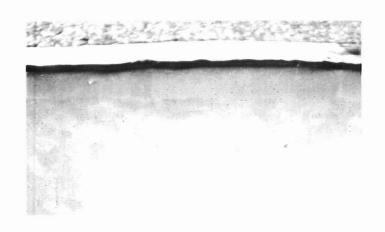


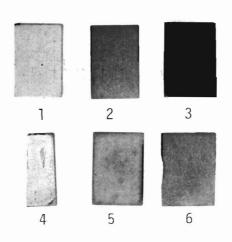


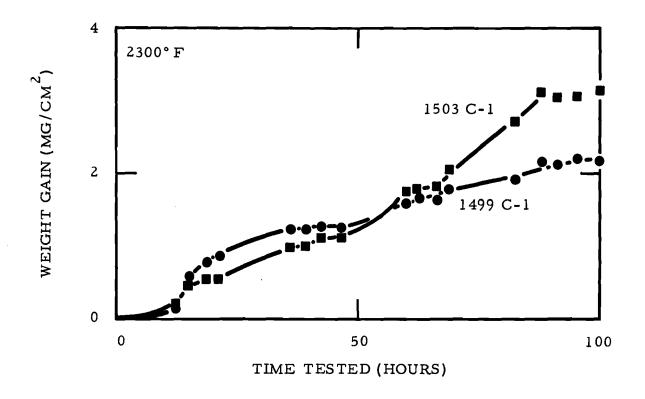


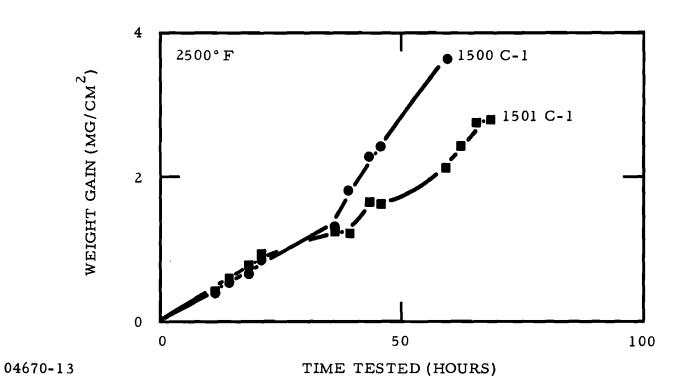
SERIES C-1

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1502	7.5	3.5		
2	1503	6.6	2.5	2300	100
3	1501	8.7	5.3	2500	83
4	1498	9.8	4.3		
5	1499	9.6	4.2	2300	100
6	1500	9.5	5.2	2500	60





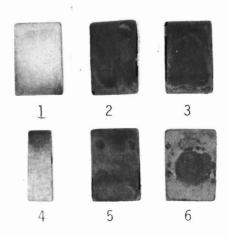


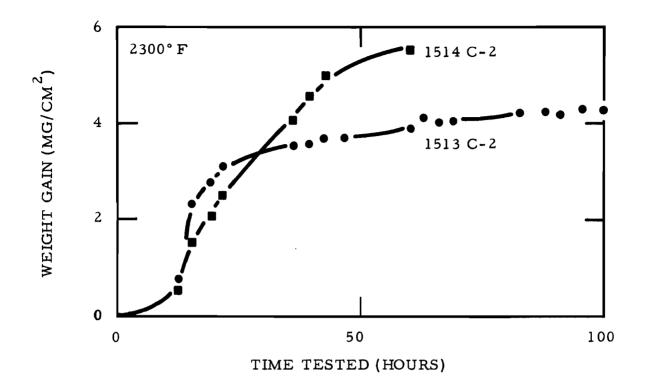


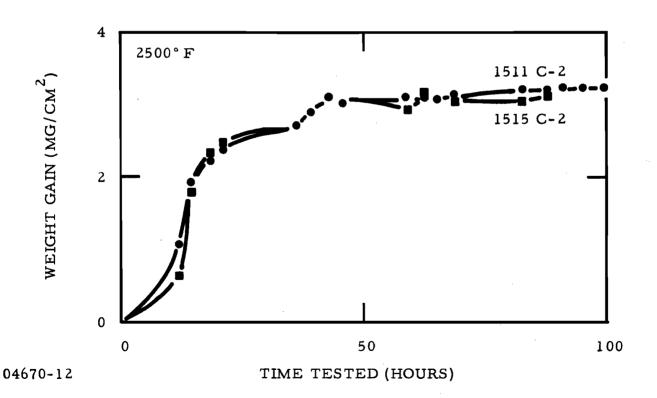
SERIES C-2

_No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1510	10.3	5.6		
2	1514	13.0	6.3	2300	51
3	1515	13.2	8.1	2500	100
4	1512	12.6	6.9		
5	1513	13.2	7.8	2300	100
6	1511	13.2	6.6	2500	100





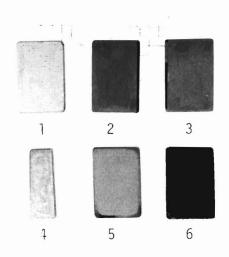


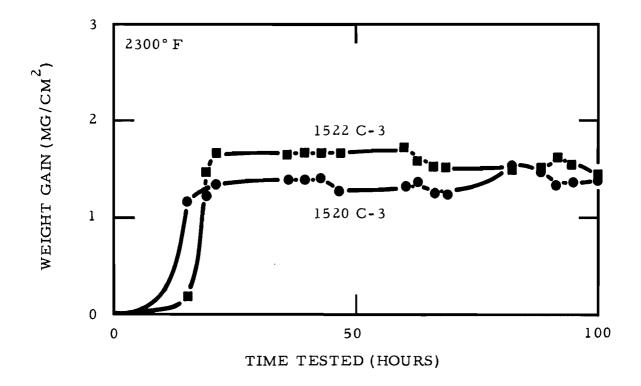


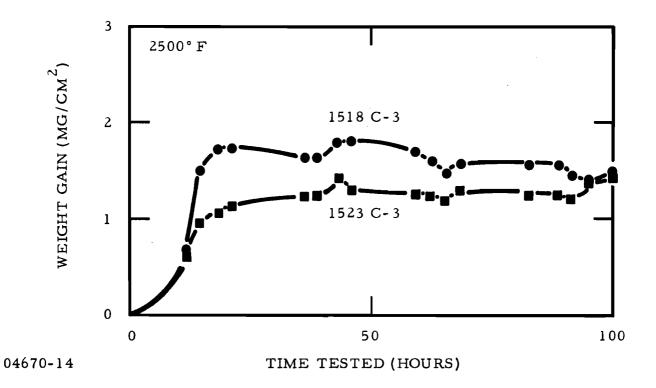
SERIES C-3

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1519	10.2	1.8		
2	1522	8.9	5.7	2300	100
3	1523	9.3	6.0	2500	100
4	1521	9.3	2.5		
5	1520	10.5	2.2	2300	100
6	1518	9.5	3.8	2500	100





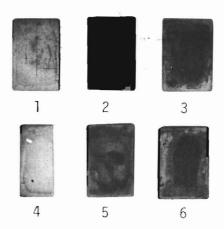


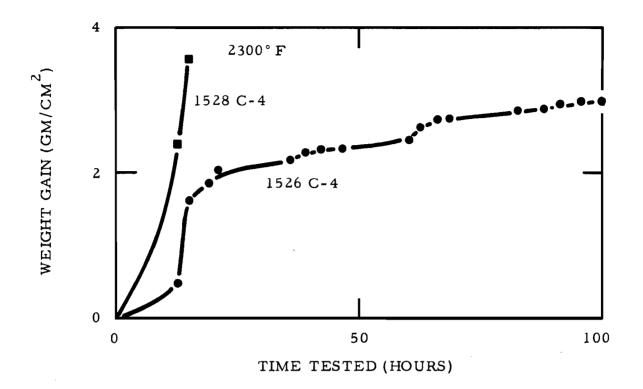


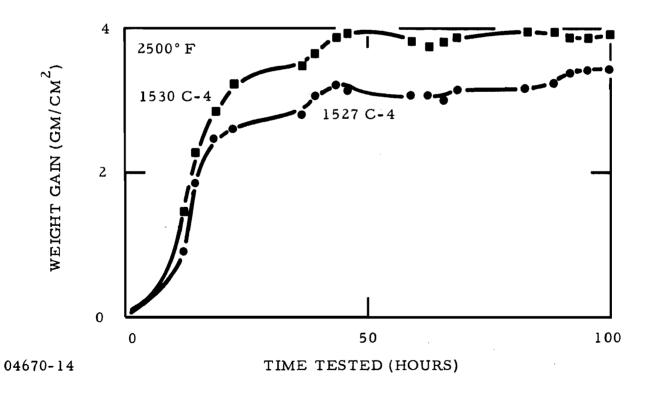
SERIES C-4

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1524	13.4	5.7		
2	1526	14.8	9.2	2300	100
3	1527	14.9	11.0	2500	100
4	1525	15.6	7.1		
5	1528	15.2	7.0	2300	19
6	1530	15.8	10.9	2500	100



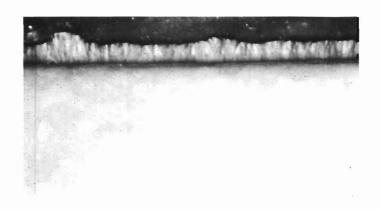


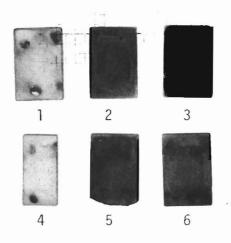


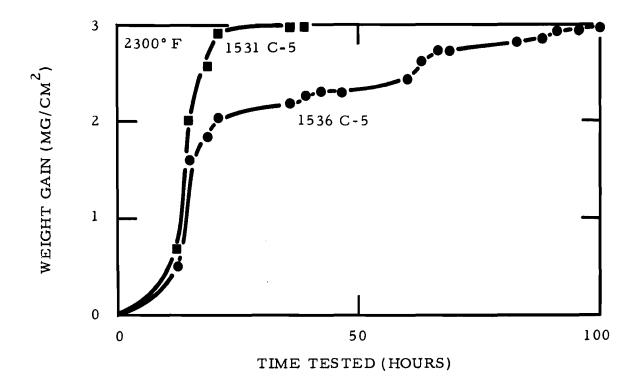


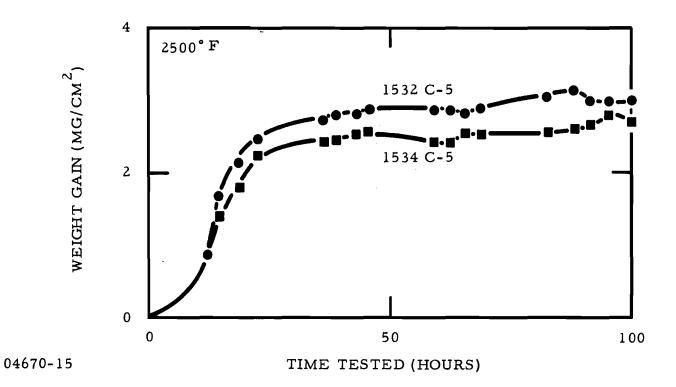
SERIES C-5

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1 2 3	1533 1536 1534	12.2 13.0 12.9	6.3 7.7 10.9	2300 2500	100 100
4 5 6	1535 1531 1532	13.1 12.9 12.1	8.4 10.2 7.4	2300 2500	47 100



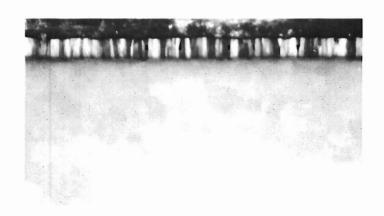


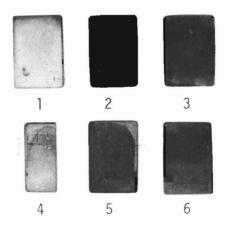


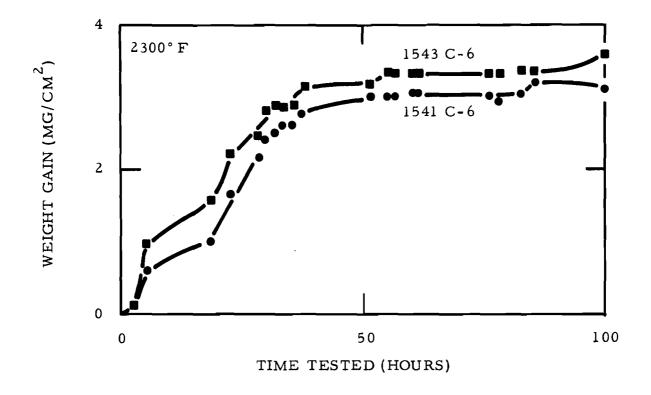


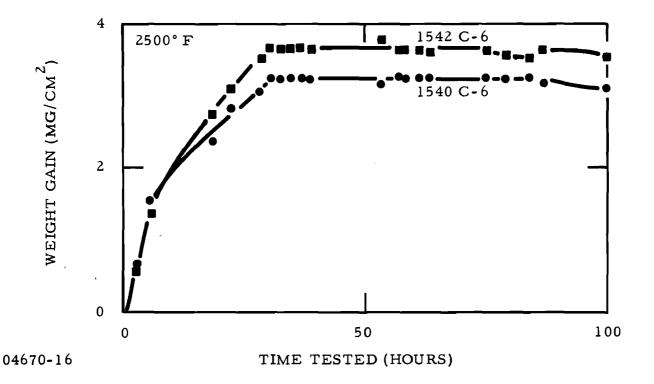
SERIES C-6

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1544	13.4	7.5		
2	1543	12.9	10.8	2300	100
3	1542	13.6	10.4	2500	100
4	1545	13.3	6.7		
5	1541	14.4	9.8	2300	100
6	1540	15.0	8.0	2500	100



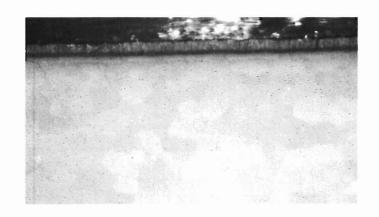


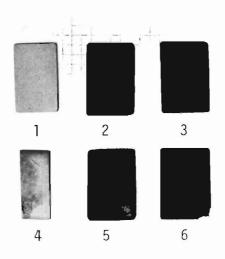


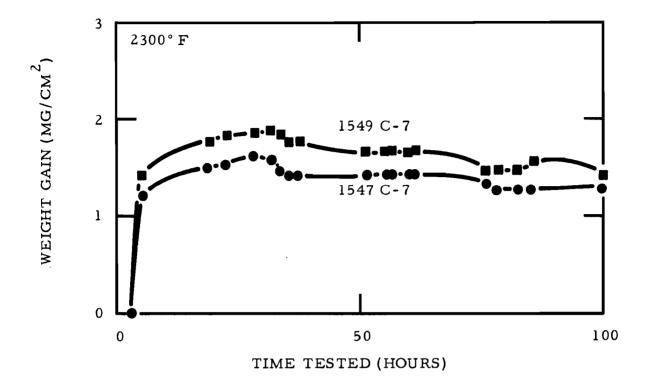


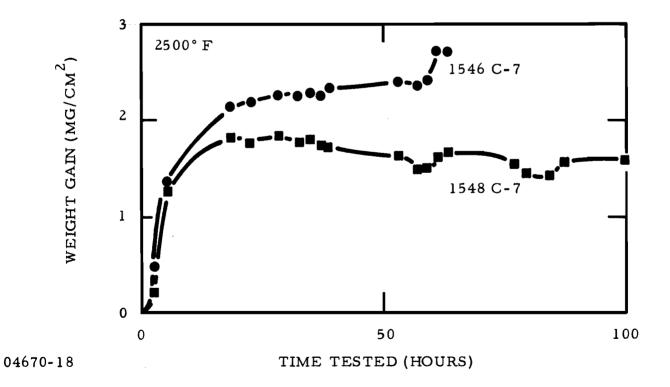
SERIES C-7

No.	Run No.	Average Weight Cr - Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1551	7.9	3.4		1.00
2	1549	8.8	6.9	2300	100
3	1548	9.2	5.9	2500	100
4	1550	8.6	3.5		
5	1547	10.2	5.8	2300	100
6	1546	10.4	5.7	2500	78



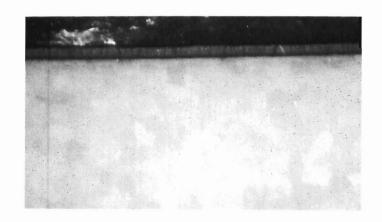


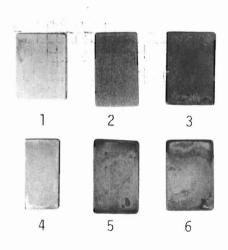


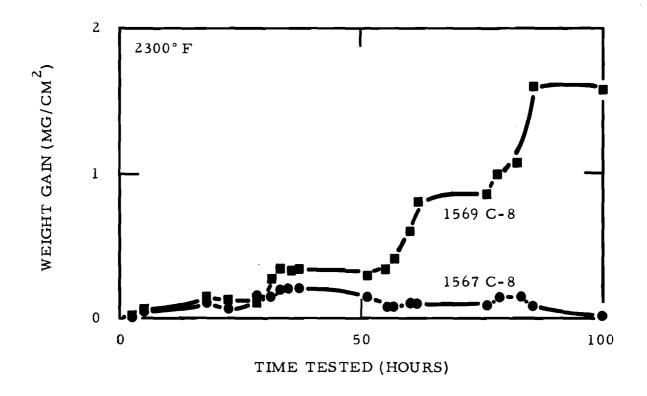


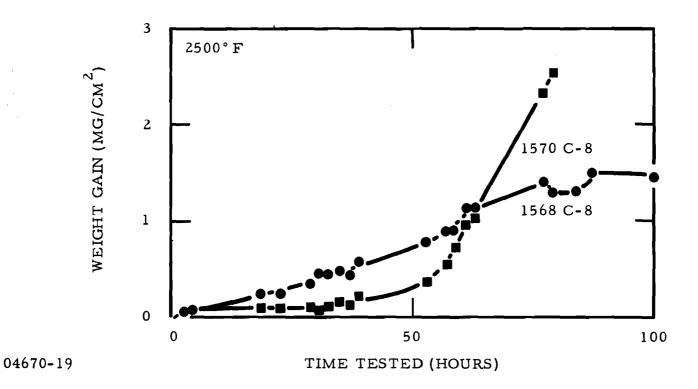
SERIES C-8

No.	Run , No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1 2 3 4	1571 1569 1570 1566	8.1 8.0 8.8 8.7	2.4 4.4 4.8 3.4	2300 2500	100 87
5 6	1567 1568	8.5 8.1	4.4	2300 2500	1 00 1 00



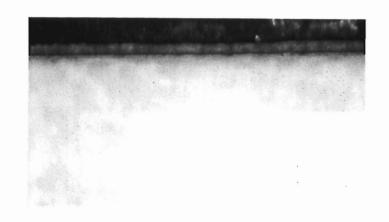


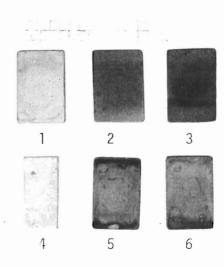


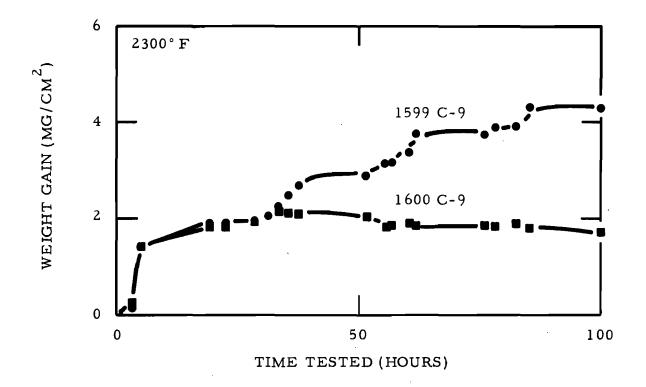


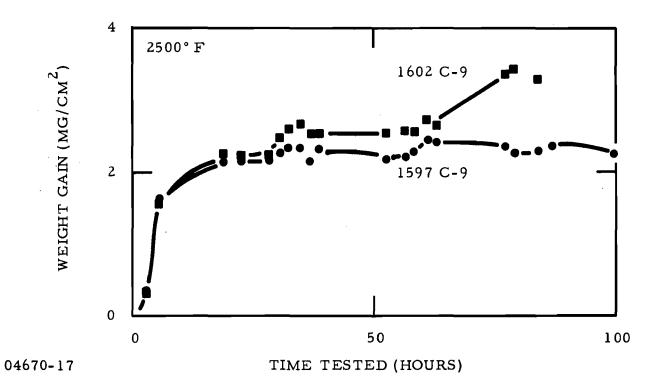
SERIES C-9

No.	Run No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1601	11.4	3.4		
2	1600	10.5	4.7	2300	100
3	1602	11.0	5.7	2500	87
4	1598	10.8	5.6		
5	1599	11.2	5.5	2300	ì 00
6	1597	10.4	5.0	2500	100









SECTION V

IRREGULAR SHAPES

Several investigations were carried out in an attempt to establish criteria for scaling-up the process and to assess the applicability of the coating process for samples other than the flat coupon shape. Two different sample shapes were run. One was a flat sample containing a lap-welded joint and the other a curved (bent) sample.

A. Welded Joint (Series D-0)

Samples having a lap joint were made from regular coupons sheared in half and then spot welded together (see the cross section in series D-O, page 154), yielding a faying surface. The mating surfaces received a considerably reduced quantity of the coating gases because the gases were impinged normal to the surface rather than in the more desirable edgewise direction. This is a severe test of ability of the coating process to impart a protective coating. These samples were coated first with titanium-chromium alloy, then with silicon. The photomicrograph of one such sample in series D-O shows a diminished amount of coating in the faying area. Oxidation tests were carried out at 2300°F and 2500°F to test the amount of protectiveness obtained. As the cyclic testing proceeded, there was no observable failure until a catastrophic, gross oxidation at the point when testing was terminated. The shape of the weight gain curves shows that a truly parabolic rate was not attained, though the total weight gain in these tests was not excessive.

These tests demonstrate that at least a short protective lifetime was imparted even under very nonideal coating conditions and indicated that a potential to provide more useful lifetimes definitely exists.

B. <u>Curved Samples (Series D-1)</u>

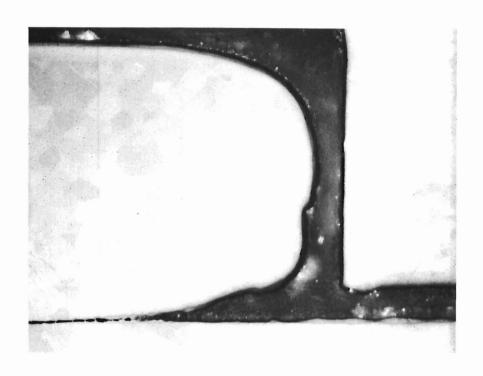
Some of the coupons were bent approximately 90-degrees around on a 3/8-inch diameter radius to form a shape resembling a leading edge. To conserve time, these samples were coated by the trideposition process described in the next section. This process provided an equivalent assessment of the ability to

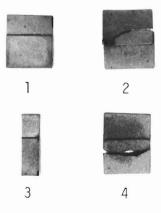
protect curved shapes, since the equipment and conditions utilized were designed and developed specifically for flat coupons. Thus, the flow over the surface and the temperature of the sample could have been nonuniform. Sample 1637-D-1 contained a longitudinal lap-weld joint in addition to the curvature. In tests of these coatings the curves at 2300°F show good protection for the 100-hour tests and a relatively low total weight increase. Appearance of the sample after testing was good. The lap joint sample tested at 2500°F began to fail after about 25 hours at the coating irregularity (caused by poor nozzle design) under the lap joint. Elimination of this irregularity would be expected to improve the useful lifetime.

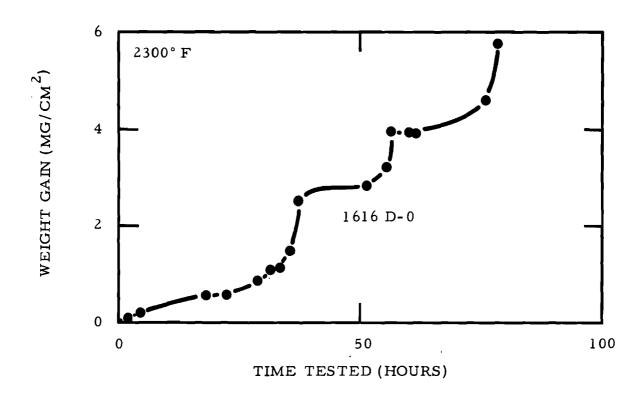
Though the time to failure at 2500°F was less than desired, the demonstrated ability and the indicated potential of this process to provide protective coatings encourages further investigation.

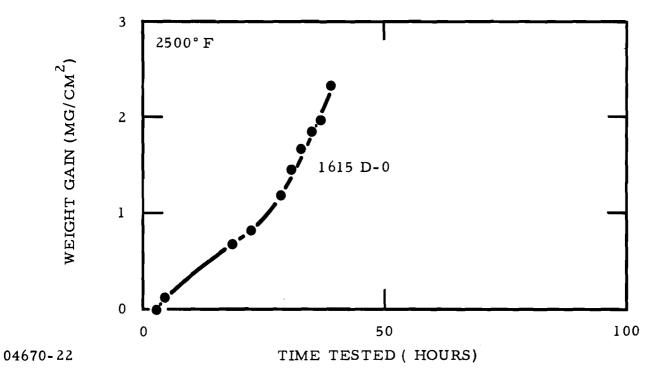
SERIES D-0

No.	Run , No.	Average Weight Cr-Ti (mg/cm ²)	Average Weight Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1617	8.5	1.4		
2	1616	9.7	2.7	2300	80
3	1614	10.2	1.6		
4	1615	9.2	2.4	2500	40



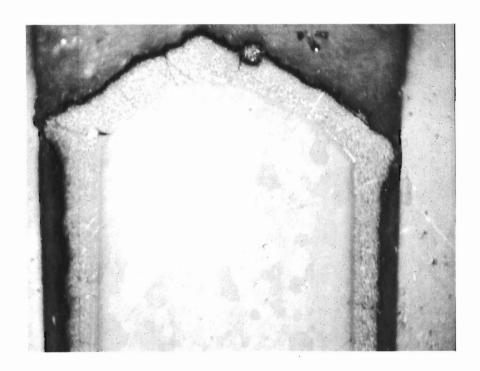


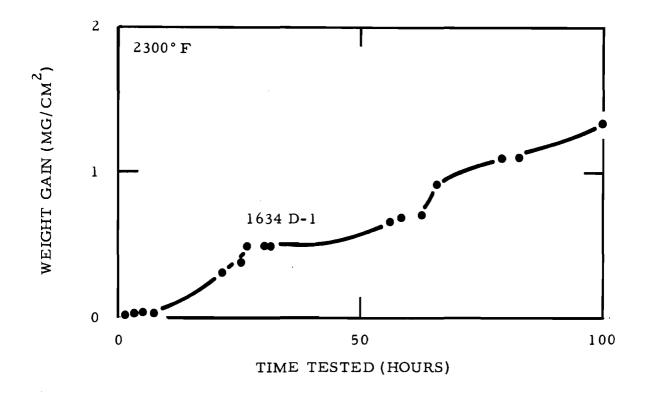


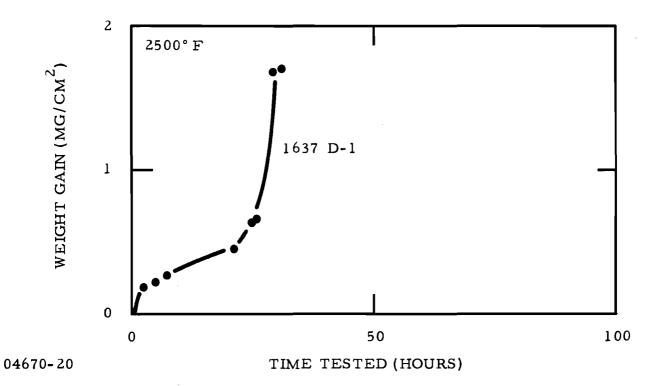


SERIES D-1

No.	Run No.	Average Weight Cr-Ti-Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1636	17.6		
2	1634	21.0	2300	100
3	1635	18.7		
4	1637	17.1	2500	31







SECTION VI

TRIDEPOSITION PROCESS

As discussed in the Introduction, one of the major assets of the CVD process is its versatility in handling different compositions of materials, as well as the sequential combinations which can be deposited. The major work on this process study contract was to produce coatings comparable with vacuum pack coatings. To permit a reasonable comparison of performance results obtained with the two processes, the coatings had to be fabricated in a similar mode. Therefore, the pre-established order for most of the coating preparation was titanium-chromium alloy deposition followed by a siliciding process. In this procedure, the capability to manipulate and control the deposition of metal alloys by simultaneous reactions was demonstrated.

The possibility of codepositing all three elements simultaneously to form a mixed silicide refractory compound could lead to a one-step, rapid coating process. Therefore, in the latter part of the program simultaneous codeposition of the three constituents titanium, chromium, and silicon was given cursory investigation. It was found that the process worked at least as well as expected and demonstrated potential for a number of significant improvements over more common coating processes being utilized or investigated for refractory metal coatings.

Trideposition was accomplished by adapting the alloy deposition equipment by adding an inlet for injection of SiCl_4 into the halide stream. These gases then premix with a hydrogen stream in the region just outside the nozzle and are swept on to contact the heated sample. A sketch of the nozzle sample geometry is shown in Figure 45. As expected from this rather elementary arrangement, the sample area immediately under the nozzle was not coated uniformly, though the remainder of the surface was well coated. Inclusion of SiCl_4 at a more suitable point in the halide stream would remove this discontinuity.

The experimental parameters of the trideposition coating are given in the Appendix, runs 1618 through 1622' and 1631 through 1638'. Deposition rates observed with the use of this apparatus were more rapid than desired for the

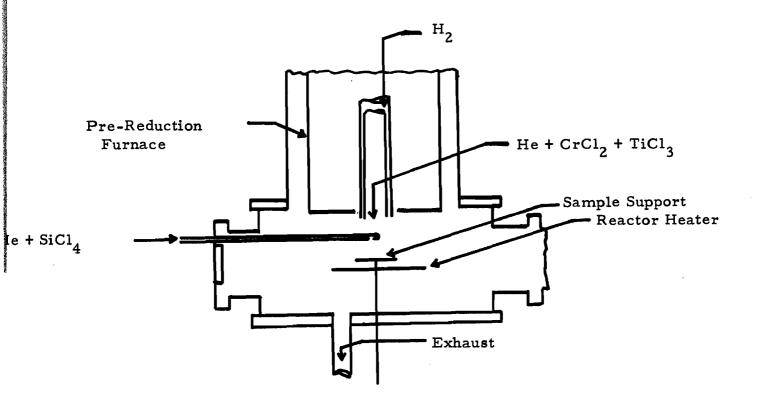
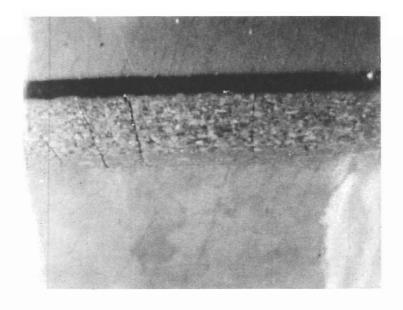


Figure 45 Schematic Diagram of Coating Reactor used to Trideposit Ti-Cr-Si Coating

reaction conditions (which were similar to those used for alloy depositions). Therefore, the coating temperature was lowered by approximately 150 degrees. Except in the nozzle areas, the coatings were uniform in appearance, and edge coverage was good.

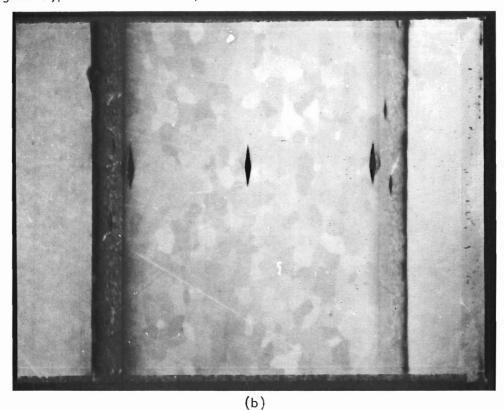
The photomicrographs in Figure 46 illustrate the general appearance of some of the coatings. A very narrow interaction area is observed, which indicates minimum diffusion into the substrate. The hardness indentations also show this. The fine grain (~ 2 micron diameter) structure is unusual for CVD deposits but is more desirable than the expected columnar grain structure. The thicker coatings near and around the edges of the samples apparently result to some extent from the double coating the edges received when the samples were coated, then turned and coated on the other side. This illustrates that the process has a high degree of "throwing power," since the coating is "lapped under" on areas not directly exposed to the gas stream.

Oxidation tests were performed on a number of the coatings, even though the nonuniformity due to the nozzle could potentially reduce the measured lifetimes. Tests series D-2 and D-3 were trideposition coatings. Two of the samples failed in the nozzle area; the other two samples completed the 100-hour test at 2300°F and 2500°F. No failures occurred at other than the nozzle nonuniformity.



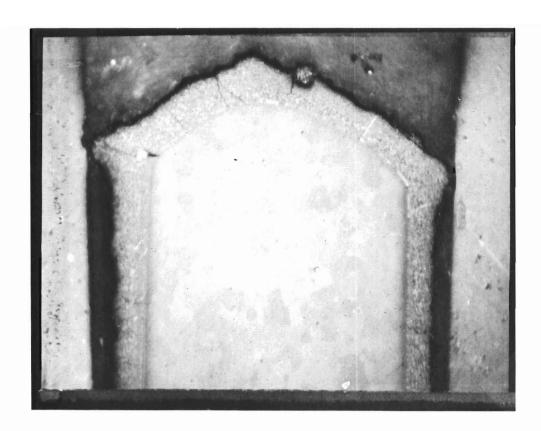
(a)

Fine Grain Size (\sim 5 x $10^5/\mathrm{mm}^2)$ and Slight Substrate Interaction Region Typical of this Preparation Method.



Coating Hardness $\sim 900~\text{kHN}_{100}\text{, Substrate 210-235 kHN}_{100}$ Unchanged from Uncoated Values (X135)

Figure 46 Cross Sections of Ti-Cr-Si Coatings on Curved Sample No. 1635 Prepared in the Trideposition Process



(c)

Excess Coating Buildup Noted on Edges of Curved Samples Due to Lapping Under on This Configuration. Edges well coated (X135)

Figure 46 (Continued) Cross Sections of Ti-Cr-Si Coatings on Curved Sample No. 1635 Prepared in the Trideposition Process

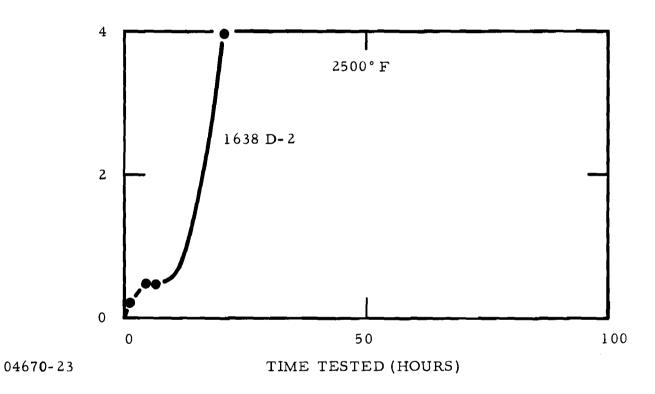
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SERIES D-2

No.	Run No.	Average Weight Cr-Ti-Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1 2	1639 1638	10.1 13.5	2500	21

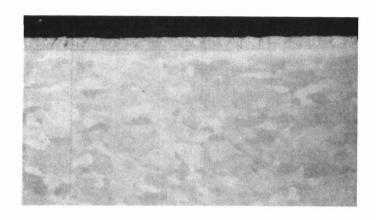


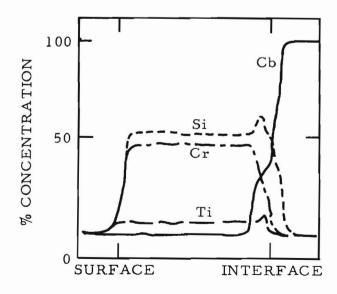


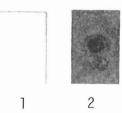


SERIES D-3

No.	Run No.	Average Weight Cr-Ti-Si (mg/cm ²)	Test Temperature (°C)	Time Tested (hours)
1	1619	21.6		
2	1622	20.3	2500	100
3	1631	12.4		







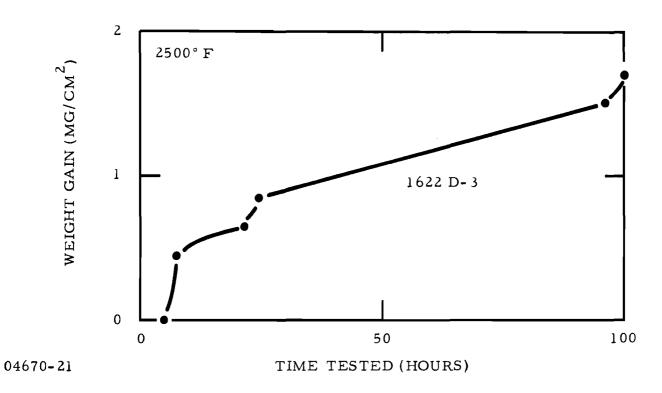


TABLE XIII

Results of Oxidation Testing at 2300°F

	T!	14 * 1 4	
C 1 -	Time	Weight	1
Sample Number	Tested	Gain	Location of
Number	<u>(hrs)</u>	<u>(mg/cm²)</u>	<u>Failure</u>
1025-A1	74		Corner
1026 - A1	100	2.7	
1116-A2	100		Edge
1120 - A2	100		Corner
1133-A3	100		Edge Split
1135 - A3	70		Edge
1196-A4	100	3.7	Protective but Surface Blister
120 <u>1</u> -A4	75		Corner
1208 - 45	76		Edge
1209-A5	76		Surface
1277 - A6	70		Edge
1280-A6	100		Corner
1290 - A7	100		Protective but Surface Blister
129 1- A7	100		Corner
1303-A8	100		Edge
1308-A8	100		Edge
1325 - A9	100	4.1	
1329 - A9	100	4.6	
1334 - A10	100	3.3	
1335 - A10	100	6.2	Protective but Surface Blister
1352 - B0	100	3.3	
1356-B0	100		Corner
1378-B1	100	•••	Corner
1381 - B1	100	4.0	
1393 - B2	68		Surface
1394-B2	100	2.7	
1400-B3	100	6.0	
1404-B3	100	==	Edge
1409-B4	100	3.2	
1410-B4	100	3.0	
1425-B5	40		Corner
1428 - B5	87		Corner
1445-B6	92	-	Surface
1446-B6	86		Edge
1451-B7	100	3.2	*** **
1455-B7	100	1.1	
1468-B8	100	2.7	••
1469-B8	100	2.7	en en
1480-B9	100	2.4	
1481-B9	100	2.6	
1486-c0	100	2.9	
1487-c0	100	2,6	
1499-01	100	2.1	
1503-C1	100	3.1	
1513-C2	100	4.2	
1514 - C2	51		Surface

TABLE XIII (continued)

	Time	Weight	
Sample	Tested	Gain	Location of
Number	<u>(hrs)</u>	<u>(mg/cm²)</u>	<u>Failure</u>
1520-C3	100	1.4	
1522-C3	100	1.4	
1526 - C4	100	3.0	
1528 - C4	19		Surface
1531 - C5	47		Edge
1536 - C5	100	3.0	
1541 - C6	100	3.2	
1543 - C6	100	3.8	==
1547-C7	100	1.4	
1549 - C7	100	2.0	*
1567-C8	100	0.1	
1569 - C8	100	1.0	
1599-09	100	4.1	n -
1600 - 09	100	2.0	-
1616-D0	80		Under Lap
1634 - D1	100	1.2	

TABLE XIV

Results of Oxidation Testing at 2500°F

	Time	Uni chi	
Sample	Tested	Weight Gain	Location of
Number	(hrs)	(mg/cm ²)	Failure
		(mg/ cm /	
1024-A1	89	10 ap	Edge
1028-A1	70		Corner
1117-A2	100		Corner
1118-A2	100		Corner
1129-A3	100	3.4	
1136-A3	100	3.2	
1197-A4	76		Corner
1198-A4	90		Edge
1206-A5	44		Edge
1207 - A5	67		Surface
1282-A6	49 15		Edge
1284-A6	15	-	Corner
1288-A7	43 19		Edge
1296-A7	18 67		Edge
1304-A8	67 49		Edge
1307 - A8 1327 - A9	49 88		Edge Surface
1327 - A9 1328 - A9	63		
1336 - A10	63		Edge Corner
1339-A10	63	1.2	Corner
1354-B0	100	1.3	 0
1355-B0	100	 2	Corner
1372-B1	100	3.6	
1377-B1	100 38	3.0 	 Cuufaca
1389-B2 1390-B2	100		Surface
1401-B3	100	3.6	 Compan
1401-83 1402-83	100	2.2	Corner
1402-83 1407-84	100	2.9	
1407-B4 1408-B4	100	3.5	
1426-B5	100	5.4	
1433-B5	100	2.4	
1442-B6	63	2.7 ==	Surface
1448-B6	86		Edge
1449-B7	86	==	Edge
1452-B7	86		Surface
1466-B8	73		Edge
1467-B8	100	3.5	Protective but Surface Blister
1479 - B9	100		Edge
1482-B9	83		Surface
1488-C0	73		Surface
1489-C0	100	3.4	
1500-C1	60	==	Edge
1501-C1	83	₩=	Edge
1511 - C2	100	3.1	J-
1515-C2	100	3.1	
1518-c3	100	1.4	

TABLE XIV (continued)

Sample Number	Time Tested (hrs)	Weight Gain <u>(mg/cm²)</u>	Location ofFailure
1523-C3	100	1.4	
1527 - C4	100	3.5	
1530-C4	100	3.9	
1532-C5	100	3.0	
1534-C5	100	2.7	
1540-C6	100	3.8	
1542 - C6	100	3.6	-
1546 - C7	78		Corner
1548 - C7	100	3.6	
1568 - c8	100	1.4	35 00
1570 - C8	87		Protective but Surface Blister
1597 - 09	100	2.1	
1602-C9	87		Protective but Surface Blister
1615-D0	40		Under Lap
1637-D1	31		Under Nozzle Spout
1638-D2	21		Under Nozzle Spout
1622-D3	100	1.8	

SECTION VII

PROTECTIVENESS EVALUATION

A. <u>Process Evaluation</u>

In an attempt to evaluate the reliability of the coating process in its present state of development, the results from the cyclic oxidation testing were plotted in a Weibull analysis method. The latter one-third of the normal series and the complex shape series (i.e., series C-O through C-9 and D-O through D-3) were treated as a single, equivalent sample, grouping. Earlier series (A and B) were excluded, since they were prepared mainly for parameter influence investigation and therefore differed more than later samples. Grouping the different series in this manner implies they are similar, which obviously is not entirely true. However, since the criterion of similarity is oxidation performance, the analysis is perhaps somewhat more justified.

Plots of failure occurrence are shown in Figures 47 and 48 for 2300°F and 2500°F oxidation temperature, respectively. The fact that the total percentage of failures was small in the 100-hour test time limit influences the appearance of the plots and makes the determination of the slope less accurate.

For the case of the 2300°F data, only four failures (16%) occurred within the 100-hour test limit. A straight line through the first and last points appears to adequately represent the data in this particular region. The indicated line has a slope, β , corresponding to \sim 0.93, which could be interpreted as indicating that a true "wear-out" type failure was not being observed. Thus, the true potential protection of the coating was probably not being evaluated, but rather experimental variation in the preparation procedure. This is further substantiated by noting that the failures were at visibly distinct areas for series C-2 and C-4 and under a faying surface in series D-0. The expected lifetimes at three confidence levels for this "batch" of coatings (series C-0 through D-1) obtained from the line in Figure 47 is given in Table XV.

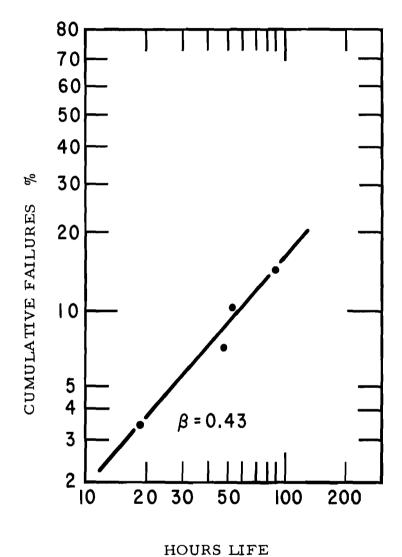


Figure 47 Percent Cumulative Failures vs Cyclic Oxidation Life at 2300°F for all Samples of C- and D- Series

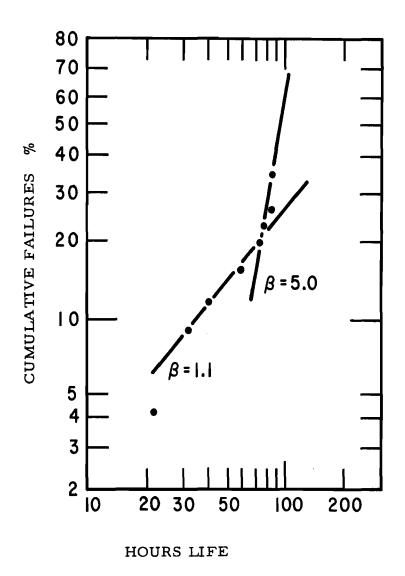


Figure 48 Percent Cumulative Failures vs Cyclic Oxidation Life at 2500°F for all Samples of C- and D- Series

TABLE XV

Coating Life (Hours) at Specified Levels of Reliability*

		<u>7%</u>	<u>90%</u>	<u>95%</u>
2300°F		205	57	28
2500°F	β = 1.1	120	34	19
2500°F	β = 5	82	62	55

Taken from data of Figures 47 and 48.

For the data on failures at 2500°F, as shown in Figure 48, a single straight line is less representative than the two shown. For the line of lowest slope of β = 1.1 the failure mode is not clearly of a "wear-out" nature. This again would be expected, since the two lowest time failures are due to coating irregularities caused by the nozzle design in the trideposition experiment. The higher slope line (β = 5) indicates a wear-out mode. The expected lifetime at 2500°F for three levels of reliability are given in Table XV for both lines. Thus, removal of the process causing coating irregularities would raise the expected lifetimes at higher confidence levels to those of the higher slope curve.

B. <u>Program Assessment</u>

A comparative analysis of the original purpose and objectives for the program and the results obtained shows that the program goals were achieved.

l. A process was designed and developed to apply high temperature oxidation resistant coatings to refractory metals. This process was demonstrated for application of the TiCrSi coating and would have general applicability for other coating compositions, or the individual steps can be utilized to provide coatings of metals such as chromium, titanium, or alloys, or silicon coatings.

The development of the process to rapid trideposition of the TiCrSi coating is considered a significant contribution, since it allows formation of the entire coating in a few minutes process time. The constant control of the coating process could be further exploited for formation of graded composition coatings or duplex compositions by altering the reactive gases as the coating operation is taking place.

Coatings prepared by either process were capable of controlled variation over wide ranges of composition and were low in impurity content. Both coating processes formed coatings without embrittling the substrate, as shown by the 4T bend tests.

- 2. The effects of process variables on the coating system were investigated according to statistically designed and analyzed experiments, as reported in the experimental sections of this report. The effects are primarily related to growth rate and composition of the coatings. Composition has one of the major influences on performance. Other variables which are ancillary to the primary process variables, such as uniformity and edge preparation, also influenced performance, as indicated by the shape of the Weibull analysis plots.
- 3. The potential ability to coat complex shapes and to protect faying surfaces was shown for an airfoil shape and a lapped, spot-weld joint. The trideposition process was utilized for the airfoil shapes, and within the constraints of the coating equipment design, afforded protection to these specimens. This is further significant in view of the fact that the equipment was specifically designed for flat specimens, yet was used without modification for curved specimens. The greater thickness observed at the edges of the samples indicated a "lapping under" effect and shows that surfaces not directly impinged by the gas stream are coated and protected. This potential for coating more complex shapes is shown, particularly where the coating equipment is designed with these shapes in mind. Performance of the lapped, welded samples was not as good as that of flat specimens, though the two-step process gave considerable protection and the only failures observed in the trideposition stemmed from coating nozzle design.
- 4. Equipment design and process parameters were established for the TiCrSi coating, and parameters for the control of the range of compositions in this system were defined. The methods and process have general applicability to future coating systems for materials of this general category. The specific application to additional materials has been shown in work outside this contract-supported effort.

The trideposition method and techniques for forming coatings have potential capability to form other than diffused or fused coatings as well as to form multilayer coatings in a single process. Alloy coatings are under consideration because of their protective ability, and the present process was shown capable of codepositing alloys over wide ranges of controlled composition.

5. Many of the necessary design and processing criteria were established as a basis for future scale-up operation. Over-all, the size, expense, and complexity of the equipment is comparable to other coating processes. The combination of short process time and moderate temperature requirements would permit higher equipment utilization in a production operation. The temperature for deposition was shown to be $\sim 1250^{\circ}\text{C}$ (2200°F) and could be lower for the trideposition process. The gas introduction nozzles were designed for the metal deposition to yield uniform coatings on flat and curved samples. Nozzle design for the trideposition process was somewhat simpler and resulted in mixing of the reactive gases in the region just exterior to the nozzle rather than closer to the sample. Thus, the nozzle-to-sample distance is not as critical a factor in design and operation of the trideposition system.

We did not investigate the potential size of parts that could be coated by this process. However, it is expected that coatings of this type could be deposited in reactors now used on a commercial, routine basis to deposit refractory coatings on complex shapes 2 feet x 1.5 feet x 2 inches.

SECTION_VIII

CONCLUSIONS

- 1. The basic processes of a manufacturing method for refractory metal coatings for oxidation resistance were developed and demonstrated through fabrication of the typical coating, TiCrSi. The extension of chemical vapor deposition technology to this field represents an advance in the state of the art in refractory metal coatings by permitting continuous control of the coating processes occurring during actual coating by means of reactant and parameter control.
- 2. The methods and procedures are applicable to processing a variety of materials (as well as for other purposes) and are potentially capable of developing into controlled and efficient manufacturing methods.
- 3. Reproducibility, reliability, and control of variations through process parameter control were high, particularly for this "first effort" program.
- 4. One of the most significant developments was the demonstration in cursory experiments of a trideposition process by which the three-component coatings could be rapidly prepared by simultaneous codeposition to result in a fine grained coating. The preliminary oxidation results indicate protection at least comparable to coatings formed in two-stage processes.
- 5. The overlayer concept of coating formation was demonstrated by the CVD formation of coatings of titanium, chromium, alloys of titanium-chromium, and titanium-chromium-silicon with a minimum of interdiffusion and minimal substrate in the coating.
- 6. CVD was demonstrated to be usable as a process tool by preparing a relatively large number of samples reproducibly by this method. The rapid coating rate obtained would aid in production.
- 7. As a laboratory process, CVD showed the capability to deposit varied, controlled compositions of coatings.

- 8. Hydrogen reduction processes are acceptable for forming coatings for refractory materials without substrate embrittlement. Other impurities (primarily nitrogen and oxygen) can be excluded to produce high-purity materials.
- 9. The simultaneous codeposition of titanium and chromium alloys was demonstrated to be possible and was controllable over essentially the whole range of binary composition. Since titanium and chromium alloys vary widely in thermodynamic stability, it is expected that many metal combinations more similar in stability could be deposited by a similar process.
- 10. Silicon coatings can be deposited by the CVD process to produce unreacted layers, or they can be reacted and diffused to form silicides.
- 11. Oxidation testing of the coatings has shown performance nearly comparable to that provided by current coating methods. This is considered quite good for the relative amount of effort in the respective methods.

SECTION IX

RECOMMENDATIONS FOR FUTURE DEVELOPMENT

CVD processes have shown a particularly valuable and unique potential for rapid formation of coatings of controlled composition. To continue the process development to practical utilization, the following additional work is needed:

- 1. Further development of the trideposition process and evaluation of the oxidation performance, substrate interaction, and reliability.
- 2. Acquiring capability to protect faying surfaces and hidden areas.
- 3. Development of equipment to permit complete coating of small samples in batches of several samples in a single step.
- 4. Optimization of the coating procedures and process parameters.
- 5. Determination of feasibility of continuous coating of small parts.
- 6. Scale-up to coating of larger prototype parts.
- 7. Further extension of the coating process to other coatings of similar nature and to different types and new concepts in coatings.

<u>APPENDIX</u>

PARAMETERS AND RESULTS OF ALL COATING DEPOSITION EXPERIMENTS

Parameters and Results of Titanium Depositions

Experi- ment No.	He Flow Through CrCl ₃ Tube (ml/min)	H ₂ Flow Through CrCl ₃ Tube (ml/min)	H ₂ Flow Through Bubbler (ml/min)	Argon Flow Through Bubbler (ml/min)	TiC14 Flow <u>(mg/min)</u>	% Conc. TiCl ₄ at Sample	Preheat Furnace Temp. ('C)	L Weight (mg)
		(Samples 1 Th	rough 21 Were	e Used to Che	ck Gas Purit	y)		
22	800	500	1500	0	120	0.70	200	1.0
24	800	1000	1500	0	120	0.60	200	1.2
26	0	1000	1500	0	120	0.80	200	5.0
28	0	1000	1000	0	80	0.65	200	2.0
30	0	1000	1500	0	120	0.80	200	5.0
32	0	1000	1000	0	80	0.65	200	1.5
33	0	1000	1000	0	80	0.65	200	3.3
37	0	500	500	0	45	0.70	200	1.2
38	0	0	2000	0	160	1.32	200	2.5
40	0	1000	1000	0	80	0.65	200	4.4
42	0	0	1500	0	120	1.33	200	0.6
43	1000	0	1500	0	120	0.80	200	0.5
44	0	0	2500	0	200	1.32	200	1.0
45	1000	0	1000	0	80	0.65	200	0.6
46	0	0	1500	0	120	1.32	200	0.7
47	0	0	1000	0	80	1.30	200	0.3
48	0	0	500	0	45	1.32	200	0.4
49	0	0	2500	0	200	1.32	200	2.5
50	0	0	2500	0	200	1.32	600	3.1
51	0	0	2500	1000	200	0.94	700	1.0
52	0	0	1000	1000	80	0.65	700	4.2

Experi- ment_No.	H ₂ Flow (ml/min)	TiCl ₄ Flow <u>(ml/min)</u>	% Conc. 	Preheat Furnace (°C)	∆ Weight (mg)
53	500	0.36	10	700	+3.8
54	500	0.71	20	700	+4.4
55	500	1.25	30	600	+7.5
56	500	1.25	30	600	+2.1
57	500	1.25	30	450	+4.4
58	500	0.36	10	400	+1.7
59	2000	0.36	2.5	200	+0.5
60	2000	0.71	5	200	+0.8
61	100	0.26	42	200	-5.7
62	100	0.14	29	200	+1.0
63	100	0.36	50	200	-66.2
64	100	0.50	59	200	-85.6
65	100	0.36	50	200	-55.2
66	500	0.36	10	200	+0.2
67	1000	0.15	3	700	+11.1
68	1000	0.15	3	700	-6.1
69	2500	0.15	1	700	+9.1

All experiments were made at $1350\,^{\circ}\text{C}$ and for 10-minute deposition time.

<u>Titanium Depositions (continued)</u>

Experi- ment No.	He Flow Through CrCl ₃ Tube (ml/min)	H ₂ Flow Through CrCl ₃ Tube (ml/min)	H ₂ Flow Through Bubbler (ml/min)	Argon Flow Through Bubbler (ml/min)	TiC14 Flow (mg/min)	% Conc. TiCl4 at Sample	Preheat Furnace Temp. (°C)	۵ Wei ght (mg)
70	0	0	2500	0	200	1.32	700	+18.0
71	0	0	2500	0	200	1.32	700	+17.1
72	0	0	3600	0	285	1.32	700	-
73	0	0	3600	0	285	1.32	700	+24.0
74	0	0	2500	0	200	1.32	800	+16.4
75	0	0	3600	0	285	1.32	800	+21.1
76	0	0	2500	0	200	1.32	900	+14.4
77	0	0	3600	0	285	1.32	860	24.5
78	0	0	3600	1400	525	1.32	800	10.9
79	0	0	3600	1400	525	1.32	800	11.5
80	0	0	3600	1400	525	1.32	750	9.8
81	0	0	2500	0	200	1.32	800	5.0
82	0	0	3600	0	285	1.32	800	25.8
83	0	0	2500	0	200	1.32	800	7.1
84	0	0	2500	0	200	1.32	800	6.0
85	0	0	2500	0	200	1.32	700	6.8
86	0	0	4000	0	320	1.32	700	19.1
87*	0	0	2500	0	200	1.32	700	7.0
88**	0	0	3300	0	260	1.32	750	10.5
89*	0	0	2500	0	200	1.32	800	3.0
90**	0	0	3300	0	260	1.32	750	7.7
91	0	0	2500	0	200	1.32	800	3.7
92	0	0	2500	0	200	1.32	700	10.4
93***	0	0	3300	0	260	1.32	750	8.8
94	0	0	4000	0	320	1.32	800	19.2
95*	0	0	4000	0	320	1.32	800	7.8
96	0	0	2500	0	200	1.32	800	11.6
97*	0	0	4000	0	320	1.32	700	8.3
98**	0	0	3300	0	260	1.32	750	12.8

All experiments 10 minute deposition time and 1350 $^{\circ}\text{C}$ except \pm 1200 $^{\circ}\text{C}$ reactor temperature and $\star\!\star\!\star$ 1275 $^{\circ}\text{C}$ reactor temperature.

Parameters and Results of Chromium Depositions

Run	Ar Flow Through CrCl3 Tube	H ₂ Flow Through CrCl ₃ Tube	H ₂ Flow Through Bubbler	Ar Flow Through Bubbler	Preheat Furnace Temperature	∠ Wt.
No.	(ml/min)	(ml/min)	(ml/min)	(ml/min)	(°C)	(mg)
99	1800	0	2000	0	500	11.6
100		ı	1 2000	l ĭ l	500	2.1
101					500	
102			ļ		775	2.0
103					775 775	3.6
104					840	1.8
105					900	2.7
106					850	58.0
107	1800				850	5.2
1 08	3600]		800	5.2 9.2
1 09	3600		2000		800	6.8
110	900		1800		775	102.2
111	900		1800		775	-31.5
112	900		1800	[]	775	147.0
113	500		2000		725	108.8
114	1800				690	14.3
115	2500					5.5
116	2500		2000		()	8.0
117	2500		500		690	
118	1800		2000		675	4.4
119	2500		2500		675	5.0 4.7
120 121	2500		2500 2700		675 690	7.5
122	2700		4700 4700		090	7.7
123			4300			9.2
124	2700		2000		1	5.7
125	1800		2000			106.6
126	500		1000			137.0
127	l 1 1		1000			169.8
128			1000			54.5
129			4600			91.9
130	500		4600		690	82.8
131	100		2000		850	16.2
132	100]	890	7.2
133	500			}	875	37.5
134	500				900	68.2
135	200		2000			110.7
136	200		4700			15.2
137	200		4700		900	108.1
138	100		4700		830	53.5
139	500		6500		850	39.8
140	125	ļ	500		820	68.0
141	75	0	500	0	840	18.8

Run No.	H ₂ Flow (<u>l/min</u>)	Ar Flow (2/min)	CrCl ₂ Temperature (°C)	Reactor Temperature (°C)	∆ Wt. (mq)
		Arg	on Bubbled Throu	igh Hot CrCl ₂	
142 143 144 145 146	0.5	1.15 1.15 1.4 to 0.0 1.5 to 0.8 0.7 to 0.3	800 800 800 775 800	1200 1200 1100 800 925	31.3 36.9 36.0 34.3

Run No.	HC1 Flow (ml/min)	Ar Flow Through Cr (¿/min)	Ar Flow Through Ti (//min)	H ₂ Flow (½/min)	Cr Temperature (°C)	Reactor Temperature (°C)	∧ Wt (mg)
		Argon -	+ HCl Passed	Over Hot Chro	mium Using Laval	Orifice for CrCl ₂	
147	5	3.5	0.0	1.5	800	1050	62.5
148	2	1	1	0.5			Dropped
149	2			0.5			3.1
150	5	1		1.0			2.7
151	10			1.0	800		2.6
152	10	3'. 5	0'. 0	1.0	900	1050	3.7

		Ar Flow	Ar Flow	<u> </u>		<u></u>	Т
_	!	Through	Through	,, ,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	Cr	Reactor	
Run No.	HC1 Flow (ml/min)	Cr (<i>t</i> /min)	Ti (//min)	H ₂ Flow (<i>L</i> /min)	Temperature (°C)	Temperature (°C)	۷ Wt (mg)
153	10	3.5	0.0	2.0	900	1050	2.8
154	20	i	i	i	1		2.7
155	30	. _					2.8
156 157	75	3.5 2.0		2.0			3.7 12.0
158		1		3.0			48.8
159	1			3.0			14.9
160 161	75 25	مام	0.0	0.5	900	1050	4.7
162	25 25	2.0 2.0	2.0 2.0	1.0 1.0	800 800	1 050 1 050	7.7 4.1
163	25	i	i	2.0	800	1050	3.8
164	50	_ _		2.0	900	800	20.9
165 166	50 30	2.0 3.0	2.0 1.5	0.5 3.0			10.9 5.8
167	JU I	1.0	2.0	4.7		800	18.3
168	1	1.0	2.0	4.7		920	23.8
169		0.5	2.0	4.7			8.8
170 171		1.0 i	1.0	4.0			Dropped 26.5
172	ļ	ľ			900		37.5
173		1.0			850		39.0
174 175		1.5 2.0			850 850	 920	31.2 6.9
176		1.0			900	1050	37.4
177	<u>l</u>	2.0			900	1050	20.1
178	30 20	0.5	1.0 0.8	4.0	850	1050	32.9
179 180	60	1.0	1.2	3.2 4.8		1265 1200	6.9 25.8
181	4		0.4	1.6		1200	18.1
182	4		0.4	1.6		1325	9.2
183 184	20 12		0.8 1.2	3.2 4.8		1265 1200	31.9 8.3
185	20		0.4	1.6		1200	13.2
186	20		0.4	1.6		1325	13.7
187 188	20 12		0.8 1.2	3.2 4.8		1265	8.3 5.6
189	60		1.2	4.8		1325 1325	10.3
190	20	1.0	, 0.8	3.2	<u>8</u> 50 .	1265	Dropped
191 192	20 60	1,0	0.8	3.2	1	1265	24.4
193	4		1.2 0.4	4.8 1.6		1200 1200	7.0
194	4	ľ	0.4	1.6		1325	18.0 10.2
195	20 12		0.8	3.2		1265	25.5
196 197	60	1.0	1.2 1.2	4.8 4.8		1200 1325	13.9
198	20	1.0	0.4	1.6		1325	7.2 53.7
199	20		0.8	3.2		1265	16.1
200 201	12 20		1.2 0.4	4.8 1.6		1325	13.0
202	20		0.8	3.2		1200 1265	13.2 14.5
203	60	1.0	1.2	4.8	850	1325	9.3
		Adapter Plate Us	ed to Reduce Si	ize of Ha + Arg	on Outlet		
204	30	1.0	1.0	4.5	<u></u> 850	1000	
205	5	2.0	1.0	4.5	Ĭ	1325	101.9
206	ĺ	3.0	1.0	2.0			38.3 16.8
207			1,6	2.0			13.2
208 209				1.0 1.0		Ì	9.5
210			1.6	0.5	1	1325	19.7 8.7
211			1.0	4.5		1200	9.1
212 213	·		1.0 1.6	2.0 2.0	1		8.9
214			1.6	1.0			10.5
215			1.6	0.5			8.2 6.1
216 217	1		0.0	1.0 4.5			5.7
21 <i>7</i> 218	5		1	0.5			5.8
219	20			0.5		1200	21.9 41.2
220	5	3.0	رار	4.5 4.5	850	1325	7.7
221	10	2.5	0.0	7.7	-,0	1325	22.3

Run	HC1 Flow	Ar Flow Through Cr (L/min)	Ar Flow Through Ti	H ₂ Flow	Cr Temperature (°C)	Reactor Temperature (°C)	ΔWt
No.	(ml/min)		(¿/min)			Through 1/2" Ope	(mg) ening
222 223 224 225 226 227 228 229 230 231 232 233 234 235 236	5 15 5	2.0 3.0 3.75 3.75 3.35 3.0 2.6 2.5 2.0 2.0 3.3	1.0 1.6 1.6 0.0	4.5 2.0 0.5 1.0 5.0 7.0	850	1325 1325 1200 1200 1325 1200 1325 1200	28.2 15.3 11.9 16.1 8.1 5.9 14.4 5.9 20.0 13.2 9.1 7.8 Dropped 3.6 9.7
237 238 239 240 241 242 243	5 20 60	3.3 0.3 0.3 0.5 0.4	0,0	7.0	850	1200 1325 1325 1325 1200 1200	12.2 12.6 4.5 6.6 7.2 10.5 31.8
244 245 246 247 248 251 251 252 253 255 255 261 266 267 272 273 274	5 50 5 50 50 50 55 55 0 51 50 50 50 50 50 50 50 50 50 50 50 50 50	Nozz le Used : 2.0	That Forces 0.0 3.0 3.0 3.0 3.0 4.0 4.0 4.0	3.0 3.0 2.0 4.0 4.7 4.0 6.5 4.0 6.5 3.2 1.6 3.2 1.6 4.0	800 800 800 850 850 850 800 800 800 800	1325 1325 1325 1325 1325 1250 1250 1250 1275 1200 1350 1275	55.137168844688259586767498389055540 114.899.6891.13161.6.7498389055540 114.94.680767498318106.60
274 275 276 277 278 279 280 281 282 283 284 285 286 287 288 289 290	15 20 20 7.5 40 20 7.5 20	1.0 2.0 2.0 1.0 2.0	0.0	4.0 1.0 6.5 4.0 4.0 4.0 4.0 3.5 1.0 6.0 3.5	850 800 800 800 850 850	1275 1200 1200 1275 1350 1200	40.0 10.0 19.2 6.6 15.2 23.2 16.5 8.0 11.3 10.2 10.1 4.0 5.8 22.2 10.9 29.4

		Ar Flow	Ar Flow				
h	HC1 Flow	Through	Through	H ₂ Flow	Cr	Reactor	Λ Wt.
Run	(cm ³ /min)	Cr	Ti	_	Temperature	Temperature	(mg)
No.		(£/min)	(L/min)	(£/min)	(°C)	(°C)	
291	40	2.0	0.0	6.0	850	1350	28.1
292 293	20	Check - A1 2.0	1 Gasses 0.0	3.5	850	1275	10.2
294	15	2.0	0.0	1.0	850	1350	11.7
295	20	i	ĭi	6.0	i	1350	5.4
296	15			1.0		1350	12.9
297	15			1.0		1200	14.4
298	20_	ì	1	3.5		1275	2.7
299 300	7.5 40			1.0 6.0		1350	3.5
301	40			6.0		1350 1200	3.0 3.9
302	20	ł	ŀ	6.0		1200	1.0
3 03	Ī			3.5		1275	0.5
3 04				1			15.6
3 05	1		_!_	_	1		15.0
306	20	2.0	0.0	3.5	850	1275	12.5
307 308		Check - All Check - Ar +					
309		Check - Ar	112				
310		Check - No	ases —				
311	Purity	Check - H2					
312		Check of HC1					
313	20	2,0	0.0	3.5	850	1275	8.0
314 315	15 15			1.0		1275 1200	7.7 8.2
316	40	2.0		6.0	ľ	1200	28.1
317	20	1.0	ĺ	3.5		1275	13.9
318	20	2.0		3.5		1275	16.9
319	15	Ì	ì	1.0	ľ	1275	15.6
320	15			1.0 1.0		1200	15.5 12.3
321 322	15 20			3.5		1350 1275	15.8
323	7.5	Ï	1	1.0		1200	4.8
324	7.5	2.0	0.0	1.0	850	1350	5.3
325	7.5	2.0	0.0	1.0	850	1350	5.3
326	20			3.5		1275	4.5
327	20			3.5		1275	6.0
328	15		1	1.0	l	1200 1200	2.2 2.5
329 330						1050	2.6
331		l		1		1350	3.4
332	15	Ţ	l	ł	Į.	1200	2.2
333	40			1.0	850	1200	2.8
334	20		l .	3.5	750	1275	19.0 19.5
335 336	Į	ļ	l i			1275 1275	15.4
337		ı		3.5	750	1350	17.9
338			l	0.5	800	آ	15.3
339	I	[ļ	1	800	1	13.3
340	20			را	900		16.3
341 342	5 20			0.5 3.5	900 850	1 1350	5.2 11.9
342 343	20	l l	ļ	3.5 3.5	850	1275	11.0
344	40		ı	3.5	900	1275	21.2
345	15			1.0	1	1200	7.4
346	15	Į.	ł	1.0		1125	3.8
347	15		ı	1.0	1	1275	6.0
348 349	40 15		ı	3.5 1.0	900 850		19.5 6.5
349 350	·15	2.0	{	1.0	1	ŀ	8.3
351	40	3.0	ı	3.5			16.7
352	20	2.0		1.0			9.4
353	15	2.0	,	1.0	<u> </u>	, l	14.0
354	40	3.0	0.0	3.5	850	1275	21.9

Run No.	HCl Flow (cm ³ /min)	Ar Flow Through Cr (//min)	Ar Flow Through T1 (//min)	H ₂ Flow (1/min)	Cr Temperature (°C)	Reactor Temperature (°C)	Δ Wt. (mg)
355 356 357 358 359 360 361 362 363 364 365 366 367 368 370 371 372	15 40 40 18 15 33 18 7.5 16.5 33 18 15 7.5 16.5 7.5	2.0	0.0	1.0 3.5 2.0 0.5 3.5 2.5 3.5 2.5 3.5 0.5 5.5 0.5 0.5	850 	1275 1275 1200 1200 1275 1350 1200 1350 1275 1350 1200 1200 1200	4.8 17.4 26.2 9.8 8.1 25.0 18.2 6.8 14.3 34.3 19.1 11.0 8.4 15.2 4.2
Run	H ₂ Flow Through Bubbler & Ti (2/min)	H ₂ Flow Through Ti Only ((/min)	Ar Flow Through Cr (½/min)	HCl Flow	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	∆ Wt.
No. 373 374 375 377 378 381 382 383 384 3886 3889 3991 3992 3993 3994 400 400 400 400 400 400 400 400 400	3.5 3.5 3.5 3.5 3.5 3.5 3.5 2.0 2.0 2.0 2.0 2.0 2.0 2.0 2.0 2.0 2.0	0.0 1.0 0.0 2.5 0.0	0.0 2.0 1.0 0.0 2.0 0.0 2.0 0.0 0.0 1.0 0.0	0.0 20.0 0.0	800 850 850 850 850 850 850 850 850 850	1275 1275 1200 1200 1200 1350 1075 1275 1275 1350 1200 1275 1275 1275 1275	4.3 3.7 2.0 2.5 2.8 0.2 1.1 2.5 2.8 0.2 1.1 2.7 4.8 3.1 8.2 1.1 0.6 6.3 0.6 2.7 2.7 2.8 3.1 1.9 2.7 2.7 2.7 2.7 2.7 2.7 2.7 2.7 3.0 3.1 3.0 4.2 3.0 3.0 4.2 3.0 4.3 3.0 4.3 3.0 4.3 3.0 4.3 3.0 3.0 4.3 3 3.0 4.3 3 3.0 4.3 3 3.0 4.3 3 3.0 4.3 3 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3.0 4 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3

				·			
([H ₂ Flow Through	H ₂ Flow Through	Ar Flow Through		Cr & Ti	Reactor	1
Run	Bubbler & Ti	Ti Only	Cr (e/min)	HCI Flow (e/min)	Temperature (°C)	Temperature (°C)	∧ Wt. (mg)
L No.	(e/min)	(e/min)	0.0	0.0			
421 422	6.0 8.0	0.0 	Ϋ́İ	i	700 750	1275 1275	2.7 1.6
423	6.0				775	1270	2.3
424 425	j	1			800 800	ļ	1.1 0.3
426	6.0	Ì			750		1.0
427 428	4.0 	1		1	750 875		1.1 5.2
429 430		1	ļ		825 800		6.6 10.2
431		}	ł		700		10.2
432 433	4.0			·	825 750	ł	8.1 13.5
434	2.5				675	ľ	7.0
435 436	2.5 4.0	0.0			800 750		5.0 5.6
437	2.0	2.0			750	J	22.5
438 439	3.0 3.0	3.0 1.0			775 750	ļ	21.0 21.1
440	1.0	3.0			775	ļ	Dropped
441 442	2.0 2.0	2.0 2.0	Ì		775 750	ĺ	3.2 2.2
443 444	2.0	2.0 0.0	Ì		725 700		1.3 1.2
445	4.0 4.0	0.0			800	Ì	4.2
446 447	2,0	2.0	0.0 2.0	0.0 20.0	700 725		3.6 13.9
448	ļ	-	2.0	10.0	750	Ì	17.9
449 450	2,0	 2.0	0.0 0.0	0.0 0.0	750 725	1 1270	12.9 12.6
451	1	1	1.0	0.0	725	1	19.4
452 453		ļ	1.0 2.0	20.0 10.0	750 725		15.0 13.6
454	Į		2.0	0.0	750	ļ	2.0
455 456	1	ĺ	2.0 1.0		(2.1 3.1
457			1.0		1	ł	5.6
458 459	ĺ	Ì	1.0 0.0			Į	3.3 2.2
460 461	ľ				j	 1270	3.4 4.2
462	1)		_]	}	1275	5.4
463 464	2.0	2.0	0!0 1.0	0.'0 5.0		1300 1300	7.2 6.2
465	2.0	2.0	2.0	0.0	750	1275	11.6
466 467	2.0 0.0	2.0 6.0	2.0 0.0			ĺ	8.1 3.6
468 469	}	6.0	0.0			}	2.4
470	0.0	6.0 4.0	1.0 0.0	}		1	2.0 4.2
471 472	2,0 2,0	2.0 2.0	2.0 2.0			}	5.6 3.9
473	2.0	2.0	0.0			1	9.7
474 475	0.0 4.0	4.0 0.0	0.0 1.0	0.0 5.0	750 775		5.5 0.5
476 477	4.0 2.0		2.0	20.0	775		12.7
478	2.0	0.0		10.0 10.0	775 800		9.7 2.1
479 480	0.0 0.0	2.0 2.0	2.0 1.0	0.0 0.0	800 800	l 1275	1.2 1.4
481	2.0	2.0	1.0	0.0	800	1275	4.0
482 483	`	2.0 2.0	1.0 2.0		1	J	8.8 9.0
484	2.0	3.0	Ī	0.0	1]	8.9
485 486	4.0 4.0	0.0 0.0	ľ	10.0 10.0	1]	1.6 6.2
487 488	4.0 2.0	0.0 2.0		20.0 0.0	800 700	1	9.8
489	1	2.0	ļ.	1	750 750	j	3.9 8.2
490 491	1	2.0 5.5	2.0 0.0		1	l 1275	10.4 .7.1
492	2.0	2.0	2.0	0.0	750	1300	i1.4

			II OIIII UIII _	Depositio	iis (continued)		
Run No.	H ₂ Flow Through Bubbler & Ti (£/min)	H ₂ Flow Through Ti Only (1/min)	Ar Flow Through Cr (£/min)	HCl Flow (L/min)	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	∆ Wt. (mg)
493 494 495 496 497 499 500 500 500 500 500 500 501 501 501 501	2.0 0.0 2.0 2.0 0.0 3.0 0.0 3.0 0.0 1.5 3.0	3.0 4.5 2.0 4.0 2.0 0.0 3.0 3.0 1.5 0.0	2.0 0.0 0.0 2.0 0.0 0.0 1.0 0.0 0.0 1.0	0.0 10.0 20.0 0.0 20.0	750 600 700 750 750 750 750 650 750 750 750 750 750 750	1300 1300 1250 	8.7091979 6.7672.749597001.586048
522 523 524 525 526 527 528 529 531 532 533 534 535 536 537 538 539 540	3.0 0.0 3.3 2.0 3.0 6.0 0.0	0.0 3.0 3.3 2.0 0.0 0.0 3.0 4.0	1.0 0.0 1.0 0.0 1.0 0.0 1.0 0.0	0.0 20.0 0.0 30.0 75.0 90.0 90.0 75.0	750 750 700 800 800 750 750 825 825 825 775 	1275 1200 1350 1350 1350 1250	1.7 1.2 0.9 30.8 5.9 6.2 1.2 2.2 2.7 2.6 3.9 8.8 3.0 8.6 5.6 7.0
542 543 544 545 546 547 548 550 551 552 553	Purity Ch Purity Ch 0.0 0.0 3.0 3.0 5.0	eck 	1.0 0.0 0.0 1.0 1.0 0.0	75.0 75.0 30.0 20.0 0.0	750 800 750 750 800	1275	2.3 1.2 1.4 3.0 0.5 0.2 10.9 4.8 3.2 4.4

Γ	H ₂ Flow	lla et	I Am Elmi			<u> </u>	
	Through	H ₂ Flow Through	Ar Flow Through	HC1 Flow	Cr & Ti	Reactor	
Run No.	Ti & Bubbler (L/min)	Ti Only (L/min)	Cr (4/min)	(L/min)	Temperature (°C)	Temperature (°C)	∆ Wt. (mg)
554	5.0	0.0	0.0	0.0	0.0	1275	4.0
555	3.0	1	1.0	I.	i	1150	0.5
556 557	3.0 5.0					1275 	3.0 5.3
558	i i			1			6.7
559 560			1 1.0			J	8.6 10.0
561 562	5.0 4.0		0,0				8.8
563	i					1275 1200	3.2 2.6
564 565			0.0 1.0			1350	7.1
566	1	0.0	j		800 700		12.9 10.9
567 568		1.5 2.5		ļ	800		19.6
569		2.3 					23.8 22.5
570 568	Ì						22.7 16.2
569	ال						14.3
57 I 572	4.0 0.0	2.5 5.0	1 1.0	0.0 20.0			22.2 7.7
573	Ï	5.0	1.0				7.4
574 57 5		5.0 3.0	1.0 1.0				8.2 10.7
576		3.0	2.0	20.0			10.0
577 578		3.0 4.0	3.0 2.0	30.0 20.0			21.8 13.7
579 580	0.0 2.0	4.0 2.0	2.0 2.0	20.0 0.0			21.0
581	2.0	2.0	2.0	0.0	800	1350	3.6 2.4
582	4,0	2.0 2.0	1.0	0.0	800	1350	7.2
583 584		2.5					5.7 7.1
585 586	 4.0	2.5 2.5		 0.0			1.1 0.9
587	0.0	5.0		20.0			9.6
588 589	0.0 0.0	4.0 4.0		20.0 70.0		1 6	9.3 5.4
590	3.0 3.0	1.0 1.0		0.0		3	4.0
591 592	4.0	2.5	İ	ľ			5.5 1.1
593 594	4.0 3.0				 800		0.8 3.6
595	4.0				750	1	5.3
596 597	4.0 6.0	2.5 4.0				1	0.4 7.0
5 98	4.0	2.5				1	6.3
599 600	ľ	0.0 2.5	1.0				9.0 3.2
601		اً	2.0				8.9
602 603			0.0 1.0				6.7 1.8
604	 4.0	[2.5	1.0 2.0	0.0		2	3.6
605 606	0.0	5.0	1.0	20.0		1	1.5 5.7
607 608	3.0 0.0	1.0 4.0	Ī	20.0 20.0		3	5.8 7.0
609	4.0	2.5		0.0		1	3.8
610 611		I			j		6.4 2.6
612	,	ا ۽	ا	ار	_[_	1	3.8
613	4.0	2.5	1.0	0.0	750	1350 1	1.7

Run No.	H ₂ Flow Through Ti & Bubbler (4/min)	H ₂ Flow Through Ti Only (L/min)	Ar Flow Through Cr (L/min)	HC1 Flow	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	Δ Wt (mg)
614 615 616 617 618 620 621 622 623 624 625 626 627 628	4.0 0.0 0.0 4.0 0.0 3.0 3.0 3.0 4.0 4.0	2.5 5.0 3.0 2.5 2.5 3.0 0.0	1.0	20.0 20.0 20.0 50.0 0.0 20.0 0.0 20.0 0.0 20.0 30.0 20.0	750 750 750 700 750 800 800 750 750	1350 	12.9 6.8 7.9 12.4 16.2 23.9 12.2 11.2 8.5 11.7 10.4 7.5 6.3 6.4 26.2 12.1

Parameters and Results of Titanium-Chromium Alloy Deposition

		A	A 51	т			
	 	Ar Flow Through	Ar Flow Through	Argon	Cr & Ti	Reactor	
Run	H ₂ Flow	Bubbler & Cr	Cr Only	With H ₂	Temperature	Temperature	ΔWt
No.	(L/min)	(l/min)	(/min)	(l/min)	(°C)	(°C)	(mg)
632	4.0	2.5	0.0	0.0	800	1275	42.7
633	6.5	4.0		0.0	800		87.8
634	6.5	1.0		0.0	750	1	28.0
635	3.0			2.0	775		25.7
636	6.1		1	0.0	800	j	39.9
637 638	4.0		l	1.0			46.8 46.1
639	4.0 4.0	1.0	0.0	1.0 1.0	1 800	1275	41.1
640	4.0	1.0	0.0	0.0	600	1250	4.3
641	i	i	1.0	1.0	700	1250	16.7
642		J	0.0	1.0	700	1325	21.7
643	ļ			1.5	700	1325	16.6
644		1.0		1.5	800	1325	42.8
645	4.0	1.5	ł	1.5	1	1300	53.8
646	3.0	1.0		2.0		1	52.1
647	2.0	0.6	l	2.5			36.8
648	1.0	0.6	ſ	2.5	_ [ľ	40.0
649	2.0	0.3	Ì	2.5	800	Į.	23.6
650	5.8	0.6	ł	0.0	750	J	54.9
651	5.8	1.0	ì	0.0	700 700		44.3 36.2
652 653	5.8 4.0		0.0	0.0 1. 0	800		40.5
654	4.0	1.0	1.0	1.0	1	ł	42.3
655	2.0	0.5	0.0	2.5			16.0
656	2;°	0.6	1	2.5]	}	20.0
657		0.6	ļ ,	2.0			28.7
658		0.5		2.0			24.9
659		0.0	1 ,	2.0	ł	ł	(Purity)
							Check
660	1	0.5	1	2.2	800	1	21.6
661		0.5	١	2.4	740	[12.8
662	_1_	0.5	0.0	2.4			10.3
663	2.0	0.7	2.0	2.4	ŀ	}	16.9
664	0.5	0.7	2.5	0.0			16.3
665 666	2.0	0.7 1.0	0.0 0.0	2.0 2.0	740	1	11.2 19.8
667	2.0 2.0	1.0	0.0	2.0	700	1300	19.3
007	2.0	1.0	0.0	2.0	/00	1300	17.7

Run No.	H ₂ Flow (e/min)	Ar Flow Through Bubbler & Cr-Ti (£/min)	Ar Flow Through Ti-Cr Only (4/min)	Ar Flow With H ₂ (£/min)	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	∆ Wt. (mg)
668 669 670 671 672 673 674 675 676	2.0 2.0 4.5 2.0 1.5	1.0 1.5 2.0 2.0 2.5 2.5 2.0	0.0 	2.0 2.0 0.0 2.0 2.2	800	1250 1250 1325 1325 1325 1325 1250	4.4 4.7 10.0 10.9 11.3 5.8 8.9 12.1 11.8 10.4
678 679 680 681 682 683 6 84 685 686	1.0 1.0 2.0 2.0 4.5 4.0 4.8	2.0 1.0 1.0 2.5 2.5 2.5	1.0	2.2 2.2 2.0 2.0 1.0 1.5 0.0	800 840 	1350 1350 1250 1250	11.0 10.0 8.2 4.6 7.8 4.7 23.4 14.3 29.9
Run No.	H ₂ Flow (L/min)	Ar Flow Through Cr Only (£/min)	Ar Flow Through Bubbler (£/min)	Ar Flow With H ₂ (l/min)	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	Δ Wt.
687 688 689 690 691 692 693 694 695 697 698 699 700 701 702 703 704 707 708 710 711 712 713	2.0 2.0 2.0 5.0 5.0 2.0 5.0 2.0	0.0	1.0 2.0 2.0 1.0 2.0 1.0 2.0 2.0 2.0 2.0 2.0 2.0 2.0 2.0 4.0 4.0 4.0	2.0 2.0 3.0 3.0 2.0 2.0 2.0 2.0 2.0 2.0 2.0 2	840 910 910 840 860 860 860	1250 	24.9 12.0 8.3 18.9 20.2 10.3 17.7 14.5 22.7 4.4 9.4 9.7.1 7.0 8.8 60.0 30.3 25.5 47.8 532.2 13.3 6.9
714 715 716 717 718 719 720 721 722 723 724 725	2.0 4.0 2.0 1.0 2.0 4.0	0.0	2.0 4.0 3.0 2.0 1.0 1.0 4.0 2.0 1.0 2.0	2.0 4.0 2.0 2.0 0.0 1.0 2.0	800 725 550 800	1325	7.3 5.7 3.7 6.1 4.2 18.0 3.6 1.3 3.3 6.9 3.0

		Ar Flow	Ar Flow				
		Through	Through	Ar Flow	Cr & Ti	Reactor	
Run	H ₂ Flow	Cr Only	Bubbler	With H ₂	Temperature	Temperature	∆ Wt.
No.	(ℓ/min)	(l/min)	(£/min)	(ℓ/min)	(°C)	(°C)	(mg)
726	4.0	0.0	4.0	2.0	800	1325	5.0
727	1	Ì	4.0	0.0	1	Ī	16.5
728			5.5	Ì			14.2
729	ľ	0.0	4.0			1325	18.9
730	4.0	2.0	1			1300	10.7
731	5.0	2.8				1300	11.3
732	5.0	0.0				1300	15.4
733	4.0		4.0			1250	11.1
734	3.0		5.0			1300	16.3
735	3.0		3.0			1200	10.3
736	5.0		3.0			1200	12.7
737	4.0		4.0			1250	8.5
738	3.0		3.0			1300	31.2
739	4.0		4.0			1250	8.6
740	4.0		4.0			1250	14.0
741	3.0		3.0			1300	14.0
742	5.0		5.0			1300	20.4
743	4.0		4.0			1250	15.8
744	5.0		5.0		ľ	1200	15.7
745	5.0		3.0			1300	
746	5.0		3.0			1300	22.5
747	3.0		5.0			1200	15.1
748	4.0		4.0			1250	17.0
749	3.0		5.0			1300	18.7
750	3. 0		5.0			1200	11.9
75 I	4.0		4.0			1250	9.8
752	5.0	1	3.0			1300	22.6
753	4.0		4.0			1250	18.8
754	5.0		5.0			1300	17.0
755	5.0	l	3.0			1300	16.3
756	4.0	0.0	4.0			1250	17.3
75 <u>7</u>	3.0	2.4	0.6			1300	8.2
758	3.0	2.4	0.6	I	_1	1200	5.7
759	5.0	0.0	3.0	0.0	800	1300	18.4

				,			
		Ar Flow	Ar Flow				1
		Through	Through	Ar Flow	Cr & Ti	Reactor	
Run	H ₂ Flow	Cr & Ti Only	Bubbler	With H ₂	Temperature	Temperature	ΔWt.
No.	(l/min)	(l/min)	(l/min)	(l/min)	(°C)	(°C)	(mg)
760	4.0	0.0	4.0	0.0	730	1250	11.4
761	7.0	0,0	4.0	0.0	650	1250	4.0
762				J	650		4.2
763					800		1.4
764					i		
760'							29.5
765							26.1
766							23.7
766 '					800		16.8
					750		16.6
767 768					750		20.0
769							22.3
770							20.7
769'							21.1
770'		ł					22.2
770			ĺ		•		15.0
771							22.9
772 772							23.6
773 772 '							23.6
							25.3
774 773 '							24.9
			1	1			15.9
775 776			1				26.9
776 774'			1				26.5
775'			1				28.2
			1				27.6
777 776'			1				27.0 27.1
			1				
777'	, ,	0 0	<i>i</i> . 0	0.0	750	1250	26.7
778	4.0	0.0	4.0	0.0	750	1250	3.9

Run No.	H ₂ Flow (½/min)	Ar Flow Through Cr & Ti Only (&/min)	Ar Flow Through Bubbler (L/min)	Ar Flow with H ₂ (<i>L</i> /min)	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	∆ Wt. (mg)
779	5.0	0.0	4.0	0.0	750	1250	6.9
780	5.0		5.0	1	,]	1300	10.5
781	5.0		6.0			1,700	8.9
782	4.0		4.0				6.7
783	5.0		4.0	1			7.6
784	5.0		5.0			1300	7.0 9.1
785	4.0		4.0			1250	8.5
786	5.0		4.0			-	
787	5.0		6.0	ĺ		1250 1300	8.6
788	4.0	ł	4.0				10.3
789	1		7.0			1200	13.3
790	ļ					1230	11.2
791				ľ		1230	10.9
792						1300	8.8
793						1250	8.0
794					1	1250	8.2
795			i	ľ		1240	9.1
796						1240	9.2
797				ļ		1265	4.9
			J			1265	7.3
798	, ,	ماء	. ا			1270	6.8
799	4.0	0:0	4.0	0.0	750	1260	

Parameters and Results of Silicon Depositions

Run No,	Substrate	H ₂ Flow	Ar Flow To Reactor	Ar Flow Through Bubbler	Reactor Temperature (°C)	∆ Wt. (mg)
800 F	Мо	7.0	7.0	1.0	1000	11.4
800 C					1 000	11.8
800 B					1000	8.8
801			7.0		1 05 0	6.9
802			15.0	1.0	1	6.4
803			15.0	0.5		5.2
804	1	7.0	7.0	0.5	1 05 0	4.5
805		4.5	1	1.0	1250	16.9
806	Mo	7.0		0.2	Ĺ	8.2
807	Ta	i		0.2		8.3
808	Ta			i		12.0
809	Ta					7.6
810 F	Ti on B66 #437					19.8
810 C	Cr on B66 #667					
810 B	Cr-Ti on 866 #773	6	, i			35.6
811	Ta					2.4
812	Ta					9.6
813 F	Cr-Ti on B66 #770	6				33.5
813 C	Cr-Ti on B66 #7731	5				21.5
813 B	Cr on B66					
	#362	7.0	7.0	0.2	1250	9.6

		1					
		Ar Flow	Ar Flow	_			
	H ₂ Flow	Through	Through	Ar Flow	_ Cr & Ti	Reactor	
Run	_	Cr & Ti Only		With H2	Temperature	Temperature	Λ Wt.
No.	(/ min)	(//min)	(£/min)	(/min)	(°¢)	(°C)	(pm)
814	4.0	0.0	4.0	0.0	750	1300	
815	4.0	1	1	1	750	1	4.9
816	4.0				750		4.4
817	5.0				650	ļ	1.2
818	4.0		4.0		750	1300	1.7
819	5.0	ا	5.0			1200	1.3
820	,	0.0	5.0			1350	3.0
821 822		3.0	2.0				6.0 4.5
823		3.0 3.5	2.0 4.0	0.0			3.2
824		3.0	2.0	4.0		1350	3.1
825		ں. ر ا	2.0	0.0		1400	7.0
826				1		1400	8.7
827							5.8
828		ŀ					5.9
829			2.0				5.1
830			4.0				8.7
831	[3.0	2.0	Ĭ			12.5
832		0.0	2.0			1400	11.8
833 834			2.0			1250	4.4
835	5.0		5.0 1.0			1400 1400	5.5 18.2
836	2.0		1.0			1400	12.1
837	2.0		1.0			1450	12.1
838	5.0		2.5		750	1250	3.9
839	1		ĺ		840	1250	
840					750	1300	4.7
841				1	750	1300	6.0
842			ا		800	1350	16.1
843 844			2.5		75 O	1350	7.7
845	ł		4.0 2.5	1	750 800	1350	4.8
846			2.5		700	1220 1350	9.3 3.5
847	ľ		2.5		760	1300	5.9
848			4.0	ļ	700	1350	3.0
849			1.0		700	1220	7.1
850			1.0		825	1220	9.1
851			2.5		760	1300	7.7
852			1.0		700	1350	12.9
853			4.0		825	1350	10.3 8.0
854			2.5 4.0		760 825	1300 1220	6.5
855 856			1.0		825	1350	16.5
857			2.5		720	1250	3.4
858	l		2.5	1	720	1300	5.5
859			2.5		820		16.5
860			1.0				15.7
861			0.5			1300	8.4
862 863			1.0 2.5			1350	13.1 12.0
864	ļ		4.0				21.2
865			i			1350	12.4
866			1			1300	
867			4.0			1	15.0
868			5.0				18.0
869			5.0				17.2
870			6.0				20.6
87 1 872	Į.		1.0	1	J		10.7
873			2.0 3.0				8.7 16.2
874			6.0				7.3
875			1.0		820		9.0
876	5.0	ļ	1.0	I	900	I	11.0
877	5.5	0.0	5.0	0.0			28.1

		Ar Flow	Ar Flow				
	H Elas	Through	Through	Ar Flow	Cr & Ti	Reactor	
Run	H ₂ Flow	Cr & Ti Only	Bubbler	With H ₂	Temperature	Temperature	Δ Wt.
No.	(/min)	(£/min)	(/min)	(/min)	(°C)	(°C)	(mg)
878	5.5	0,0	5.0	0.0	900	1300	92.4
879	5.0					1000	28.2
880 881					900	900	27.5
882				J	840	1100 1000	31.5 12.9
883					900	1000	29.0
884					900	1200	36.2
885					900	1300	37.5
886 887					875 850	825	22.1
888					850 950	1 000 1 000	31.4 42.6
889					850	1000	22.4
890	ļ			l	900	1 000	34.8
891	•				850	1 000	75.3
892 893			F 0		850 850	850 1350	84.7
894			5.0 4.0		800	1250 1300	33.1
895			4.0		Ĭ	1	33.1
896			5.0		1		38.0
897		ا	3.0		800		12.5
898 899		0.0 4.0	1.0		750 800		28.5 14.2
900		0.0		1	75 O		15.0
901		I	1.0		800	ļ	15.5
902			1.5		1	1300	12.6
903			1.5			1200	4.8
904 905			1.0		800 900	1270	7.6 10.1
906			l		950		11.7
907					1000	1270	8.6
908		,		ł	900	1325	16.0
909			١			1375	17.3
910 911			1.0 4.0		900	1300 	19.0 12.9
912	l		7.0		1000		16.9
913					I		18.1
14			اا				11.6
915 916			4.0 1.0				14.1 14.1
917			5.5				11.1
918			5.0				11.4
919			5.0				9.7
920			3.0				13.1
921 922			4.0 4.0	l		1300	16.2 16.2
922 923			4.0	1		1300	15.4
924			3.0		l	1	11.6
925			1.0	1	1000		8.6
926			1.0		1100 1100		9.7
927 928			4.0 1.0		1100	1	13.5 11.2
929		i i	4.0		1000	1200	7.7
930	5.0		5.5			1250	7.3
931	1.0		4.0		1000	1	9.8
932	1.0		3.0		1000 900	l l	9.2
933 934	5.0 5.0		1.0 3.0		900		9.7 8.6
935	4.0		4.0		800		5.0
936	5,0		1.0		1		8.4
937		_[_	2.0	_ ا _	। 800	1050	9.0
938	5:0	0.0	3.0	0:0	900	1250	6.5

		Ar Flow	Ar Flow	,	1		,
} }		Through	Through	Ar Flow	Cr & Ti	Reactor	
Run	H ₂ Flow	Cr & Ti Only	Bubbler	With H ₂	Temperature	Temperature	ΔWt.
No.	(ℓ/min	(L/min)	(/min)	(l/min)	(°C)	(°C)	(mg)
939	5.0	0,0	2.0	0.0	 850	1200	8.5
940	" ار	Ϋ́I	5.5	Ϋ́İ	850	1250	6.3
941			4.0		900	1275	10.7
942	ł	ł	1.0	ł	900	1300	12.5
943			4.0	ļ l	1 000	1300	17.0
944			4.0		1000	1200	12.2
945 946			2.0 1.0		1 000 900	1200	11.8 16.9
947			4.0		900	1350 1350	18.2
948			1		900	1000	3.8
949	l	1		l	1 000	1000 /	3.8
950			1			1300	19.9
951	_ l _		4.0			1	22.7
952	5.0		1.0				16.3
953	2.0	١	1.0 1.0				15.4
954 955	1.0 5.0	0.0 1.0	3.0				13.6
956	5.0	2.0	1	ł		1300	14.6
957	4.0	1				1275	
958	Ì	J				1275	17.1
959	. [1 000	1300	17.6
960	1	١	3 0	}	900 1 000	1250	10.5 13.5
961 962	4.0 I	2.0 2.0	3.0 4.0		1000	1250 1250	16.0
963	4.0	1.0	4.0	ľ		1275	18.0
964	3.0	1.0	4.0			1275	17.8
965	5.5	1.0	4.0		1 000	1275	18.1
966	5.0	0.0	1.0		800	1300	
967		I	1		900	1300	19.9
968					900 1 000	1250 1200	15.4
969 970			1.0	ł	1000	1000	17.7 7.3
970 971			4.0			1200	25.4
972			4.0			1200	25.5
973	5.0	0.0	4.0	o'.o	1 000	1300	30.7
974	5,0	0.0	1.0	4.0	900	1000	20.7
975	1	1	1.0				20.1
976	- I	-	4.0	Ì	1	1 1300	19.6
977 978	5.0 3.0		1.0 2.0		900 1 000	1250	19.0 15.5
979	3.0		1		1 000	1250	22.6
980	3.0					1250	14.2
981	2.0					1200	13.6
982	3.0					1300	20.9
983	3.0	1	2.0	ľ		1250	17.2
984 985	5.0 3.0		1.0 2.0			1250 1300	11.3 26.4
986 986	5.0		ا .			1300	11.5
987	j.						8.7
988							6.9
989	_		1				8.4
990	5.0	İ	1	1		ĺ	7.3
991	3.0		1		1 000		13.2 21.5
992 993	3.0 3.0		2.0		900]	15.6
994	5.0		1.0		ĺ		9.2
995	5.0		1.0				9.1
996	3.0]	2.0		900		10.3
997	3.0	İ	2.0	1	1 000	ľ	18.6
998	5.0		1.0	Ì		1200	12.6
999	4.0		·			1300 1350	11.0 12.3
1 000 1 001	3.0 3.0) J U	13.6
1 002	5.0						13.8
1 003	1.0		1.0				12.8
1004	2.0		2.0	İ	1		18.0
1 005	3.0		3.0				20.9
1006	3.0	1	3.0	1	1 1000	1250	19.2
1007	4.0	0.0	3.0	0.0	1000	1350	22.1

		Ar Flow	Ar Flow		1		Т
_		Through	Through	Ar Flow	Cr & Ti	Reactor	
Run No.	H ₂ Flow (L/min)	Cr & Ti Only (e/min)	Bubbler (£/min)	With H ₂ (£/min)	Temperature (°C)	Temperature (°C)	/ Wt.
1008	4.0	0.0	3.0	0.0	1000	1350	(mq)
1009	5.0	i	5.0	i	1000	1350	20.0 25.2
1010	7.0						26.0
1011 1012	3.0 1.0				 1000		25.8
1013	1.0				950		24.3 19.3
1014	1.0		5.0		Ť		17.4
1015 1016	5.0 4.0		1.0 5.0				12.2
1017	7.0). 		950		21.6 23.3
1018	4.0				1 000		28.2
1019 1020							28.1 27.5
10181							30.0
10191					1000		29.5
1021 1022			5.0 2.0		900 900	1	25.2 22.1
1 023			5.0		1000		44.0
1024							47.6
1025 1023 '				*			50.4 47.8
1024	ļ						48.2
1025		ľ					44.7
1 026 1 02 7]	42.4 47.3
1028							47.4
1026						[45.3
1027' 1029							45.8 47.6
10281	اب						45.4
1029' 1030	4.0 5.0		5.0 4.0		1 000 900		47.0
1031	2.0		5.0		900	1350	7.5 3.3
1 032	7.0				900	1350	12.0
1 03 3 1 03 4	7.0 3.0		1	0:0 3.0	1000	Ī	12.3 7.8
1034	2.0	-		0.0		1350	2.7
1036	2.0		5.0	2.5		1450	7.3
1 03 7 1 03 8	5.0 2.0		4.0 4.0	0.0		1350	12.1 14.0
1039	5.0		2.0				10.4
1040	2.0	0.0	5.0				13.7
1 041 1 042	2.0 2.0	2.0 2.0	3.0 1.0				11.8 7.8
1 043	4.0	2.0	3.0				14.7
1044	5.0	3.0					13.8
1 045 1 046	7.0 4.0	2.0 2.0					12.2 12.9
1047	ĺ	1.0					15.0
1 048 1 049	 4.0	1.0 2.0	Į.				16.9 13.1
1050	2.0	2.0					11.2
1 05 1	7.0	0.0					17.2
1 052 1 053	1.0 4.0	2.0					8.8 10.8
1 054	4.0						8.0
1 055	4.0	2:0					7.4
1 056 1 05 7	7.0 5.0	0.0 2.0					12 .6 5.4
1 058	10.0	2.0			Ì		5.2
1059		0.0	3.0				9.5
1060 1061	10.0		5.0 5.0				9.6 6.9
1062	4.0		4.0	•			3.0
1063	2.0		5.0	0.0	1000		1.4
1 064 1 065	7.0 4.0	0.0	4.0 4.0	2.5 0.0	1 000 1 1 00	 1350	6.9 2.3
	- • -		• -				,

Run No.	H ₂ Flow (_ℓ /min)	Cr & Ti Only (<u>/</u> /min)	Ar Flow Through Bubbler (¿/min)	Ar Flow With H2 (L/min)	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	Δ Wt. (mg)
1 066	4.0	0.0	4.0	0.0	1000	1350	33.2
1067	5.0	. 1	2.0	i	1	1300	26.2
1068	5.0		ï			1300	22.3
1069	5.0					1250	21.2
1070	7.0		}	ì		1300	27.3
1071	7.0					1300	23.9
1072	7.0	0.0	2.0	1		1325	30.8
1073	5.0	3.0	0.0			1325	0.4
1074	7.0	0.0	2.0			1300	37.6
1 075	, i	ĺ	1.0			1250	35.5
1076			i			1275	34.9
1077							27.0
1078							28.8
1079		Ì					26.8
1078	l						18.0
1079							1.5
1080			1.0	1		1275	16.9
1081		j	3.0			1350	39.5
1082			1			1300	35.6
1 083		l	1			1275	32.7
1 084	7.0		3.0			1275	28.7
1 085 - 1	5.0		5.0			13,00	26.8
1085-2	Ĩ		i I				24.9
1085-3						1	8.1
1 085 -4	5.0	0.0	5.0	0.0	1000	1300	18.4

Run No.	H ₂ Flow	Ar Flow Through Cr & Ti Only (L/min)	Ar Flow Through Bubbler (1/min)	Ar Flow With H ₂ (£/min)	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	∆ Wt. (mg)
1 086	5.0	0.0	3.0	0.0	1 000	1300	25.6
1 087	1					1325	25.8
1 088						1300	21.5
1 089		1			1000	1350	21.7
1090			3.0		940	1	16.8
1091	Ì		5.0		950		19.2
1092			3.0	ł	-1		17.3
1 093			1				18.0
10901	1						18.2
1 092 1							18.0
1 093 1	5.0				ĺ		20.3
1 094	4.0		·				8.5
1 095	4.0		-				8.3
1 096	10.0		3.'o		1		88.1
1097	3.0		2.0	1			7.6
1098	5.0		1.0				6.4
1099	-		3.0			1350	7.4
1100			3.0		950	1400	8.8
1101	5.0		5.0		1000	1450	9.5
1102	3.0		5.0		Ĭ	1450	8.5
1103	4.0		4.0			1450	11.7
1104	5.0		5.0			1480	13.3
1105	5.0		4.0	1		1510	11.2
1106	5.0		5.0		ĺ	1540	11.8
1107	2.0		1			,5,70	13.2
1108	i			1			25.4
1109			i	1			37.9
1110						1540	7.2
1111						1480	16.2
1112	2.0	0.0	5.'0	o¦o	1000	1400	15.9

Run No.	H ₂ Flow (½/min)	Ar Flow Through Cr & Ti Only (L/min)	Ar Flow Through Bubbler (£/min)	Ar Flow With H ₂ (£/min)	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	Δ Wt.
1113	7.0	0.0	5.0	0.0	1000	1400	15.3
1114	2.0	İ	i	l		1400	15.2
1115	1					1250	14.0
1116			ļ	l l		ĺ	13.7
1117		ľ					14.3
1115'				J			15.4
11161						Į .	16.4
1118		1	ì				14.6
1117'						ľ	16.9
1119							16.9
11181	{		1	1)	15.3
1120				ł			15.2
11191	1						16.9
1121			l			ļ	15.8
11201							16.3
11211		į.		ł	1000	1250	17.4
1122	2.0		5.0		800	1400	4.2
1123	. 5.0	I	1.0		800	Ì	5.1
1124	5.0	1	1.0		900		7.0
1125	5.0		5.0		900		9.9
1126	2.0		5.0	Į	900	l l	8.7
1127	5.0		5.0		1000		9.7
1128	5.0	0.0	1.0	0.0	1000	1400	7.2

Run No.	H ₂ Flow (L/min)	He Flow Through Cr & Ti Only (L/min)	He Flow Through Bubbler (£/min)	He Flow With H ₂ (£/min)	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	Δ Wt.
1129	5.0	5.0	5.0	0.0	950	1400	24.5
1130	5.0	10.0	i	ĺ	Ì	ļ	14.9
1131	7.0	5.0		ł		ļ	26.7
1132	5.0	0.0					27.8
1133	7.0	5.0	ł				25.8
1134	i	i	ł	Ì	,	Ì	27.9
1135							28.7
11291						1	27.0
11311	(ļ	l l	ļ			28.1
1133'			1				27.9
11341		1					28.5
1136							26.1
1137	1	ì)	Ì		Ĭ	29.5
11351							30.2
11361		1					27.4
1137'	7.0		<u> </u>	Į.		{	27.0
1138	3.0		5.0	1			36.2
1139	5.0	5.0	1.0				14.8
1140	5.0	9.0	1.0	o¦o	950	1400	11.7

Parameters and Results of Silicon Depositions

Run No.	Substrate No,	H ₂ Flow (L/min)	He Flow (1/min)	H ₂ Through Bubbler (L/min)	Reactor Temperature (°C)	Δ Wt.
1141-1	B-66	7.0	7.0	0.2	1250	17.9
1141-2	ĺ	i	1	1	Ĭ	21.9
1142-1			l			16.0
1142-2						15.5
1142-3	B-'66	ļ				16.9
1143-1	889					19.5
1143-2	887	i	1			20.8
1143-3	882	7.0	7.0	0.2	1250	24.1

Silicon Depositions (continued)

				H ₂ Through	Reactor	
Run	Substrate	H ₂ Flow	He Flow	Bubbler	Temperature	Δ Wt.
No.	No.	(Æ/min)	(l/min)	(<i>l</i> /min)	(°C)	(mg)
1144-1	1115	7.0	7.0	0.2	1250	31.9
1144-2	1116	í,	'i'	i	ĺ	31.2
1144-3	1117					34.3
1144-4	1118					33.1
1144-5	1119					34.7
1144-6	1120					32.9
1145-1	1121					36.1
1145-2	1129					40.1
1145-3	1131					41.5
1145-4	1133					40.4
1145-5	1134					45.0
1145-6	1135				ŀ	41.6
1146-1	1115					14.6
1146-2	1116					14.5
1146-3	1117			ì		17.9
1146-4	1118					15.7
1146-5	1119					16.9
1146 <i>-</i> 6	1120				ſ	14.3
1147-1	1121					20.8
1147-2	1129					19.7
1147-3	1131					22.5 23.8
1147-4	1133					33.l
1147-5	1134			١		27.3
1147-6	1135			0.2 0.1		17.6
1148-1	1024			0,1		21.2
1148-2	1025					23.7
1148-3 1148-4	1029 1028	l	İ		ì	21.2
1148-5	1027		ļ			22.8
1148-6	1026]	0.1		19.6
1149-1	1137			0.2		37.8
1149-2	1023		Ì	0.2		37.2
1149-3	1136			0.2		37.1
1150-1	1024			0.1]	17.7
1150-2	1025					22.0
1150-3	1029]	23.6
1150-4	1028			1		18.6
1150-5	1027					21.8
1150-6	1026	1		0.1		20.9
1151-1	1137			0.2		30.0
1151-2	1023		1	0.2		27.4
1151-3	1136			0.2		30.1
1152-1	766		1	0.1		29.0
1152-2	769					31.1
1152-3	774					35.3
1152 '-1	766					20.8
1152 ' -2	769	_ا_	_1_	. ا		16.5
1152 ' - 3	774	7.0	7.0	0.1	1250	10.0

Parameters and Results of Titanium-Chromium Alloy Depositions

		He i	Flow (//min)		Cr and Ti	Reactor	∆ Wt. (mg)
Run No.	H ₂ Flow (½/min)	Cr and Ti Only	Through Bubbler	With H2	Temperature (°C)	Temperature (°C)	
1153	5.0	5.0	5.0	0.0	1000	1400	12.1
1154	5.0	5.0	10.0	ſ		1	12.1
1155	5.0	7.0	7.0				19.9
1156	7.0	4.0	7.0				-
1157	7.0	3.0	3.0			1400	27.4
1158	5.0	5.0	5.0			1300	32.8
1159	5.0	5.0	5.0			1450	29.7
1160	7.0	3.0	3.0			1400	30.0
1161	ì	5.0	1.0			i	17.3
1162	ľ	3.0	3.0	ı		•	29.4
1163	7.0	3.0	3.0	0.0	1 000	1400	33.7

	<u> </u>	He	Flow (L/min)		Cr and Ti	Reactor	
Run No.	H ₂ Flow (!/min)	Cr and Ti Only	Through Bubbler	With H2	Temperature (°C)	Temperature (°C)	Δ Wt.
1157' 1160'	7,0	3.0	3.0	0.0	1000	1400	(mg) 30.9
1164		ı	1	1	1	1	31.6 35.0
1 165 1 162 '				· [[38.7 37.2
1163 ' 1164 '							37.6 37.7
ا 1165			- [1000		38.9
1166 1167					930 		23.2 21.2
1168 1166 '					ł	l	20.7 16.4
1169 1167'		Ì					16.3
1170					930 950		12.0 14.3
1168' 1169']	1 3.0		960 980	1400 1370	7.7 9.6
1171 1172			6.0 6.0		Ĩ		28.6
1173			6.0]		39.4 47.4
1174 1175			5.0 4.0		 980		42.0 35.4
1176 1177		í 3.0	Ì		960		29.4 24.5
11711		3.5					19.7
1178 1177		5.0	Ì]		21.3 19.0
1176 ' 1178 '							22.0 21.1
1179		Ì				}	23.2
1180 1179'		,					26.4 25.2
1180' 1181		5.0 3.0	4.0 6.0	1		Ì	26.1 38.9
1 182 1 183		3.0	4.0 6.0				35.4 42.1
1184		0.0	4.0	}	,		35.3
1185 1186		0.0 3.0	2.0 4.0				17.8 30.6
1187 1188		1	4.0 4.0		Ì	Ì	- 29,6
1191			6.0				7.9
1192 1193		3.0	6.0 8.0		1	ì	15.7 9.7
1194 1195		0.0 3.0	8.0 6.0		960 1000	1370 1400	11.0 9.6
1196 1197			i				16.0 16.0
1198							20.1
1199 1196 '			1			Ì	12.9 14.5
1200 1201			1				16.8 14.8
י 1192			}	})	8.5
1197' 1198'							23.1 22.2
1199' 1200'			1	ł		}	32.9 23.5
1201	7.0	3.0	6.0	0.0	1000	1400	18.2

			Flow (L/min)		Cr and Ti	Reactor	
Run No.	H2 Flow (#/min)	Cr and Ti Only	Through Bubbler	With H2	Temperature (°C)	Temperature (°C)	Δ Wt. (mg)
1202 1202-1 1202-2 1203 1204 1205 1206 1207 1205	7.0 7.0 7.0 4.0	3.0 3.0 3.0 0.0	6.0 6.0 9.5 6.0 4.0	0.0	1000	1400 1380 1400	23.0 16.7 16.3 18.7 30.4 32.1 31.6 38.9 32.6
1208 1206 ' 1209 1207 ' 1208 ' 1209 ' 1210 1211 1212 1213	4.0 7.0 4.0	0.0 3.0 0.0	4.0 6.0 4.0			1400 1380 1420 1380	34.8 37.7 35.4 26.6 16.7 18.1 12.4 16.6 24.8
1214 1215 1216 1212' 1217 1218 1219 1220 1221 1222 1228	4.0 5.0	0.0 3.0 0.0 0.0 3.0	4.0 6.0 4.0		900 800	1380 1400	25.3 19.6 15.8 20.0 15.3 29.0 28.0 39.0 31.0 20.4 7.2
1229 1230 1231 1232 1233 1234 1235 1236 1237 1238	5.0 7.0 5.0 9.0 9.0 3.0 4.0 4.0 7.0	0.0 0.0 3.0 0.0	4.0 7.0 4.0 10.0 10.0 4.0 6.0 10.0		820 850	1400 1380 1400	4.9 6.5 8.1 17.3 17.3 11.2 9.9 17.0 15.6 9.2
1239 1240 1241 1245 1246 1247 1248 1249 1245' 1250	4.0		6.0 10.0 10.0 6.0		850 875 		19.1 23.7 25.0 29.2 42.6 29.1 23.3 30.1 26.0 37.0 32.2
1249' 1250' 1260 1261 1262 1263 1264 1265 1266 1267	4.0 5.0 4.0 4.0	0.0 3.0 0.0			875 825 800 800 850	1400 1300 	37.7 23.1 10.4 6.8 - 5.0 5.9 5.2 4.7 7.1
1268	7.0	0.0	6.0	0.0	850 850	1400	6.9

Parameters and Results of Silicon Depositions

			D. 1				
Run	Substrate	H ₂ Flow	Dilution Flow	Bubbler Flow	Reactor Temperature	ΔWt.	Time
No.	No.	(£/min)	(<i>l</i> /min)	(£/min)	(°C)	(<u>mg</u>)	(Minutes)
1189-1	1171	7.0	7.0	0.1	1225	_	10
1189-2	1177	,,,	1	1	1	-	,,
1189-3	1178			ĺ		-	ſ
1189-4	1179	l				-	
1189-5 1189-6	1180 1176	1		ì	j	-	
1199-6	1176 1165			ĺ		-	10 12
1190-2	1157					-	14
1190-3	1163					-	J
1190-4	1164	1		Ì	1	-	
1190-5 1190-6	1162 1160			ļ	1225	-	ļ
1189'-1	1171				1225 1200	-	12 15
11891-2	1177	l			1	-	19
11891 3	1178					-	ľ
1189'-4	1179	[ļ	J	•	
1189'-5 1189'-6	1180 1176				ر 1200	•	1
1190'-1	1165	ļ			1180	-	1
1190 -2	1157	J			i i	-	
11901-3	1163				1	-	1
1190'-4	1164				[-	
1190'-5 1190'-6	1162 1160	}		}	1 1180	-	}
1223-1	B-66				1200	5.8	
1223-2	Ĩ	}			Ī	7.3	1
1223-3	[3.5	15
1223'-1 1223'-2	1	Ì	}	ĺ	1	8.2	4 0
12231-3	1	l			1	7.2 5.5	
1224-1	1				1	6.0	ł
1224-2	1			l	{	5.3	l
1224-3 1224'-1]]				5.0 8.4	
1224'-1	ľ					8.1	l
1224'-3		ļ		1	1200	7.6	20
1225-1				ļ	1215	15.7	25
1225-2) n ((12 15	11.3	25
1225-3 1226-1	B-66 1138			1		9.7 15.0	25 20
1226-2	1132	ł	1			10.8	20 I
1227-1	1139	\	1	1		17.1	Ì
1227-2	1140				1215	15.8	
1242-1 1242-2	в-66 I	J		ł	1120 1120	12.4	
1242-3	J				1120	7.9 5.5	
1243-1		Ì	1	Ì	1205	7.9	ĺ
1243-2				ľ	1205	9.5	
1243-3 1244-1	l	ľ			1205 1250	8.7	
1244-1		l		t t	1250	7.2 15.8	Į.
1244-3		}			1250	12.1	
1251-1				ļ	1210	17.3	1
1251-2				۱,	1210	16.4	
1251-3 1252-1				0.1 0.15	12 10 1225	13.2 12.5	
1252-2	ľ	Ì		J. 15	1225	14.1	
1252-3				1	1225	10.7	
1253-1			1		1215	17.9	ĺ
1253-2 1253-3		1		ł	12 15 12 15	17.0	ľ
1253-3					1215	10.1 12.7	
1254-2					1200	13.6	
1254-3					1200	12.1	
1255-1		Ì		1	1220	11.2	Ì
1255-2 1255-3	B-66	7.0	7.0	0.15	1220 1220	15.5 15.9	1 20
1-22-2	D-00	7.0	7.0	V. 15	1220	13.3	20

							
]			Dilution	Bubbler	Reactor		
Run	Substrate	H ₂ Flow	Flow	Flow	Temperature	△ Wt.	Time
No.	No.	(l/min)	(l/min)	(l/min)	(°C)	(mg)	(Minutes)
1256-1	B-66	7.0	7.0	0.15	1240	25.1	20
1256-2	I I	7.0	7.0	0.15	1240	30.3	I I
1256-3						20.4	
1257-1						15.8	
1257-2			i	ĺ		14.2	
1257-3	B-66			Ì	1240	20.0	
1258-1	1199			ľ	1250	21.4	
1258-2 1258-3	1200 11 9 8					16.4 26.2	
1258-4	1197	[i			22.9	i
1258-5	1196		Į.			27.1	
1258-6	1201					21.9	
1259-1	1207					52.2	
1259-2	1209	[Í	ĺ		42.1	1
1259-3	1205					50.5	
1259-4 1259-5	1206 1208					55.5 33.5	
1259-6	1204					37.2	
12581-1	1199		[ĺ		16.2	İ
12581-2	1200					24.7	
12581-3	1198					28.4	
12581-4	1197					24.4	
1258'-5 1258'-6	1196 1201	[[[18.2 23.5	i
1259'-1	1207					10.5	
12591-2	1209		1			10.0	
12591-3	1205					15.9	
1259'-4	1206			[1	19.6	ĺ
1259'-5	1208					10.5	
1259'-6 1269-1	1204 1240					20.3	
1269-2	1241	J	J			21.0	
1269-3	1236					19.0	[
1269-4	1222			- 1		18,6	
1269-5	1239		Ì			15.2	
1269-6	1221	Į.		J	,	18.3	
1269-7 1269'-1	1200 1240					19.6 19.7	
12691-2	1241					37.5	
12691-3	1236					25.7	
12691-4	1222	1]		32.7	
1269'-5	1239					12.4	
1269' -6 1269' - 7	122 î 1220					18.7 13.7	ł
1270-1	1215]		34.9	}
1270-3	1214	1	ļ	1	1250	31.4	Į
1285-1	1271				1235	39.2	
3,	1274	[30.8	ļ
4	1273 1272	[1 1235	22.0 22.6	
1285'-1	1272	1		}	1235	22.6 28.0	
3	1274				1200	22,2	j
4	1273					17.1	
6	1272				1200	10.0	}
1292-1	1212	1			1210	16.8	1
3 4	1211 1210				l	14.2 18.4	
6	1210				1	21.5	Ì
1292'-1	1212			Ì	İ	15.4	[
3	1211	1				14.4	1
4	1210					8.3	1
1202-1	1203]			6.7	
1293-1 3	1001 1002			Ì	l	20.0	ĺ
4	1002	ł	1	ł	}	18.3 35.6	1
6	1000	7.0	7.0	0.15	1210	15.6	20

		•	Dilution	Bubbler	Reactor	_	,
Run	C.,ha hanna	H ₂ Flow	Flow	Flow	Temperature	Δ Wt.	Time
No.	Substrate 1001	(<u>l/min)</u> 7.0	(<i>l</i> /min) 7.0	(<i>L</i> /min) 0.15	("C) 1210	(mg) 7.8	(min) 20
3	1002	i,i	,;, 	1	1	6.7	.20
4	1006 1000				 1210	6.0 5.0	
1297-1	1232				1200	27.0	
3 5	1237 1233	ľ	Ì			2 0.0 11.0	}
1297'-1	1232					9.3	
3 5	1237 1233	7.0				11.1 8.3	
1298-1	1238	6,0			·	16.2	
3 5	1235 1234		ľ		{	14.0 11.2	
12981-1	1238					10.6	
3 5	1235 1234			1		8.6 8.3	
1299-1	1294					15.2	
3 5	1295 1287	l 6.0	ĺ			11.7 12.0	
1305-1	1106	5.0	Ĭ])	10.1	
3 5	1107 1110					9.9 10.7	
1306-1	1114	İ				11.8	
3 5	111 3 1111	J 5.0				12.8 7.2	
1314-1	86 0	4.0		-	1	9.1	
3 5	859 871					6.9 7.0	
1315-1	909					16.3	
3 5	910 908	4.0	1	}	1200	- 6.7	
1322-1	958	7.0	1		1210	26.9	
3 5	962 959	Ì		Ì	1210 1210	25.8 15.3	Ì
1323-1	965 964	l	•		1200	21.0	ľ
3 5	963	ł				30.5 17.5	
1323'-1 3	965 964	ĺ				15.9 11.1	
5	963	ļ		ļ	1200	7.1	
1330-1 3	961 954	İ			1215 	20.6 17.3	
5	956					13.6	
1330'-1 3	961 954					15.0 12.6	
5	956	[10.1	
1331-1 3	953 946	Ì		Ì		15.1 6.9	
5	952				1215	15.0	
1332-1 2	944 942				1205	-	
3	941					23.1	
4 5	920 915					19.5 20.9	
6	916					18.6	
7 1332'-1	92 1 920					21.0 7.1	
3	941					6.0	
5 6	915 916			l		4.7 3.6	ļ
7	921	7.0	7.0	0.15	1205	3.5	20

Run No.	Substrate	H ₂ Flow (<i>l</i> /min)	Dilution Flow (L/min)	Bubbler Flow (L/min)	Reactor Temperature (°C)	ΔWt.	Time (min)
				0.15	1215	22.4	20
1333-1	862 867	7.0	7.0	U.15	1215	27.3	
2	%8	i				21.7	
3 4	901			1		21.1	
5	978		Ì			18.6	
	983			ì	Į	20.2 16.2	
7	969		1			7.0	
1333'-1	862 867					12.0	
2 3	968					7.6	
4	901	i				9.1	
5	978	ļ				8.3	
6	983	ŀ			1015	11.3	
7	969				1215 1200	8.1	
1342	- 1278				1200	7.9	
3	12/6				1200	-	
1343-1	1192				1215	14.0	
2	1166					18.4	
3	1090			ŀ		16.4 17.5	
3 4 5 6	1092					12.3	l l
5	1250 1247					11.8	ļ
5 7	1247					11.4	
1343 '-1	1192					10.0	Ì
2	1166					6.5	
3	1090					11.0	
4	1092					11.2	*
3 4 5 6	1250					7.2	
6 7	1247 1249					5.8 6.4	
13 44- 1	1167					10.5	
2	1217					15.4	
3 4	1093					19.5	
4	1216					9.6	
5 6	1078	,		ĺ		12.3 7.9	
7	1219 1245					12.9	
1344'-1	1167					6.9	
2	1217					11.3	
3	1093					17.7	
3 4 5 6	1216					10.3	
5	1078 1219					12.4 12.5	
7	1245				1215	9.6	
1345-Í	1089				1200	11.6	
3	1087				1200	15.4	
5	1088				1200	16.5	
1346-1	1296				1210	36.9	
2 3	1284 1288					22.7 20.6	
4	1283				1210	24.9	1
1347 - i	1277				1200	26.5	1
2	1282					25.1	1
3	1281				1	15.0	
12461 1	1280				1200	17.4	1
1346'-1 2	1296 1284				1210 	20.9 9.2	1
3	1288		1	I		13.4	I
4	1283	7.0	7.0	0.15	1210	6.8	20
4	1203	/.··	/.∪	0.15	1210	0.0	20

Run		H ₂ Flow	Dilution Flow	Bubbler Flow	Reactor Temperature	ΔWt.	Time
No.	Substrate	(//min)	(<u>l/min)</u>	(L/min)	(°c)	(mg)	(min)
1347'-1 2	1281 1282	7.0	7.0	0.15	1200	12.8 12.7	20
4	1277)	ì		10.6	
6	1280	l l				11.5	ı
1348-1	1328)	1		16.8]
2 4	1327 1329					17.3 17.1	
5	1325	1)	ì		16.4	Ì
	1326					16.4	1
7 1349-1	1324 1304	1		}	1	17.7 24.2	20 40
2	1303					26.2	i i
4	1307		ļ	}	}	20.9	}
5 6	1308 1301	1				18.1	
7	1301	ļ		ļ	ļ	19.9 22.1	40
1348'-1	1328	ļ				27.1	20
2	1327		ļ	ļ	-	28.0	}
4 5	1329 1325		ĺ			25.5 25.4	
5 6	1326			Į.	1	25.3	l
7	1324		1	ſ		25.5	20
1349'-1 2	1304 1303	1	Į		l	21.7 23.4	40
4	1307		1			21.2	
5 6	1308	1			}	20.1	
6 7	1301 1309	ľ		1	J	17.2 15.1	ľ
1350-1	1336	ł				53.6	
2	1339	ľ	1	Ì		51.8	
4	1335 1334				İ	28.2 25.4	
5 6	1340			ì		28.1	
7	1337					19.5	Ì
1350'-1	1336 1339			1		28.4 30.0	
1350'-4	1335					31.3	
5	1334	ſ	Ì	1		26.9	j
6	1340 1337		ĺ			28.2 20.1	40
1351-1	1291	1)	1		26.8	30
2	1290					25.1	
4 6	1289 1286	ł	ì	ł		18.7 13.5	ì
1351'-1	1291					24.2	ľ
2	1290		ł	ſ		23.1	}
4	1289 1286				1200	8.3 14.1	
1373-1	1191	l l	{		1215	5.4	
3	1193				ľ	5.1	
6 1374-1	11 <u>9</u> 4 1080	ļ	1	-		5.2 13.1	
2	1079			ļ		10.1	
3	1077	Ţ	1	-		12.2	}
4 5	1084 1081	ľ				13.7 9.6	
6	1109	Į	[- (l	18.2	}
7	1112			1	1215	4.1	
13741-2	1079 1077	l	Ţ	,	1200	33.3 33.0	
4	1084			1		37.3	
5	1081	ı	[29.5	ļ
6 7	1109 1112				1 1200	51.1 42.2	1 30
1375 - 1	1099		[Į	1175	76,6	10
2	1102	Ì	l		ĺ	-	- 1
3 4	1100 1101	1				-	
	1103	1	1	1		•	ĺ
5 6	1104	7.0	7.0	0.15	1175	-	10

		31110011	Deposition	13 (00116)			
Run No.	Substrate	H ₂ Flow (L/min)	Dilution Flow (L/min)	Bubbler Flow (£/min)	Reactor Temperature (°C)	Δ Wt. (mg)	Time (min)
1376-1	1098	7.0	7.0	0.15	1125	16.7	10
1370-1	1097	,,,,	1	T	1	7.6	- 1
3	1096					17.0 14.7	
4	1095	l l	j	J		14.7	1
5	10 <i>9</i> 4 1074				1125	10.6	
1382-1	1069				1100	13.5	
2	1068			ì	1	28.3	
3	1067					12.2 12.9	
4	1070		l			12.9	
6	1075 1076		ł	1	}	11.7	1
1383-1	1365		l			12.1	
2	1363		ĺ		ľ	22.5	
3	1364	1	1			11.0 11.4	
13831-1	1365					16.3	
2	1363 1364					10.0	
1384-1	1354	ĺ	1	Ì		18.2	
2	1356					22.0	
3	1357					35.0 21.0	
4	1352			l		16.9	
5 6	1355 1353				1100	11.8	10
13841-1	1354	}	ļ	J	1080	19.1	15
2	1356					22.5	
3 4	1357					16.5 12.6	
	1352					16.7	l
5 6	1355 1353				1080	17.7	15
1385-Ĭ	1380				1090	17.1	10
2	1379	· }			1	15.6	
3 4	1381					17.4 16.9	
4 5	1378 1372					18.2	l
6	1377	J		ļ	1090	17.9	10
1385'-1	1380			i	1080	25.0	15
2	1379			1	1	24.3	
3 4	1381	1	1	l		22.2 24.8	
4	1378 1372					24.8	
5 6	1377	7.0	7.0	0.15	1080	16.9	15
	.311	, • •	,				

Parameters and Results of Titanium-Chromium Alloy Depositions

		He	Flow (l/min)		Cr and Ti	Reactor	
Run No.	H ₂ Flow (L/min)	Cr and Ti Only	Through Bubbler	With H2	Temperature (°C)	Temperature (°C)	Δ Wt. (mg)
1271	4.0	0.0	6.0	0.0	850	1350	17.8
1272	ĺ	1	Ī	1	850	1350	16.3
1273					850	1350	38.1
1274					800	1375	36.5
1275	ļ				800	1	32.2
1276	4.0				750		25.5
1277	7.0		6.0		750		28.8
1278	i l	ł	4.0	1	800		29.7
1279			6.0		750		21.8
1280	ļ	Į.	•	l	1	1	27.0
1281	7.0	0.0	6.0	010	750	1375	25.7

		He F	low (L/min)		Cr and Ti	Reactor	
•	H ₂ Flow (L/min)	Cr and Ti Only	Through Bubbler	With H ₂	Temperature (°C)	Temperature (°C)	Δ Wt (mg)
82	710	010	610	0,0	750	1375	24.6
, ı , ı]	21.1
1 8	l li			ļ			23.4 19.3
				ľ		ļ	25.9
		1	1				23.9 22.2
			6.0				33.8 26.7
			3.5		750 700	ł	27.5 19.7
	Ī				750		23.2
							22.2 25.7
							26.7 26.5
			1 3.5				26.4 29.6
			3.5				14.1
			1				13.8 22.0
						ľ	19.1
			3.5			1375	17.4 27.1
			3.0 6.0			1400 1400	14.9
			8.0 6.0			1415 1400	24.0 23.3
			Ī			Ī	22.7
			}			}	24.8
							17.3 19.6
							20.1 22.7
							26.1 18.2
	- }			}		1400	17.8
						1425 	17.4 15.9
						 1425	18.6 17.3
						1400	25.4 26.4
	1		\			1	22.5
							24.8 22.5
						1	23.3 28.3
							21.2 27.4
					Ì	1	20,8
	7.0				 .	1400	24.6 15.4
	4.0					1375	29.7 29.5
							24.6
					Ì	1	25.1 26.2
						1	27. ¹ 28.
							27.1 27.8
	4.0	o.'o	6.0	0.0	750	1 3 75	26.0

			low (&/min)		Cr and Ti	Reactor	
Run No.	H ₂ Flow (l/min)	Cr and Ti Only	Through Bubbler	With H ₂	Temperature (°C)	Temperature ("C)	∆ Wt. (mg)
1328' 1329' 1334 1335 1336 1337 1338 1339 1334' 1335' 1336' 1339'	4.0 4.0 5.0	0.0	6.0	0.0	750	1375 1375 1400	27.2 26.3 42.9 41.6 30.3 35.8 33.3 35.0 26.7 30.0 31.0 33.2 26.1
1340 1341 1352 1353 1352 1353 1355 1355 1356 1356 1356						1400	30.9 27.7 19.9 34.8 28.6 30.0 33.1 21.3 23.1 27.5 25.9 31.4 28.2
1357' 1358 1359 1360 1361 1362 1363 1364 1365 1366 1367 1368 1369	5.0 7.0 6.0 5.0 7.0 5.0		6.0 11.5 6.0 6.0 6.0 10.0		750 800 800 750 750 800	1350 1250 1250 1350 1350 1400 1450 1350	27.0 23.7 28.2 32.5 21.3 - 27.0 21.0 33.9 21.9 19.8 15.2 21.3 16.8
1370 1371 1372 1377 1378 1379 1380 1372' 1377' 1381 1378' 1380' 1381' 1381'	5.0 7.0		10.0		800 850		18.6 28.6 36.1 31.9 34.5 33.8 38.2 30.0 35.0 28.8 30.0 30.0 30.2
1387 1388 1389 1390 1391 1392 1393 1394 1395 1399 1399 1399	7.0	0.0	6.0	0.0	850 860 	1350 1375 	32.1 44.0 39.7 21.7 39.4 24.4 23.2 36.3 39.3 22.1 26.9 16.3 26.9 16.3 26.9

		He F	ow (L/min)		Cr and Ti	Reactor	
Run No.	H ₂ Flow (£/min)	Through Bubbler	Cr and Ti Only	With H ₂	Temperature (°C)	Temperature (°C)	ΔWt. (mg)
1396	7.0	6.0	0.0	0.0	880	1325	45.9
1397 1398			İ	- 1			45.8
1399			ه!ه		880		22.2 18.9
1400			3.0		860	i i	20.7
1401 1402							23.4
1403							24.1 26.3
13981							22.8
1400 ' 1401 '							21.3
1404							33.6 23.5
1402 '			l				18.0
1403 ' 1404 '	7.0		3. 0		 860	 1325	21.6 27.3
1407	5.0		0.0		850	1400	27.3
1408							23.7
1409 1410		ľ					25.1 26.5
1411							27.1
1407							23.6
1408 ' 1412						ì	20.2 26.8
1409							22.2
1410'							23.5
1411' 1412'		6.0				1400	25.3 26.3
1415		2.5				1375	30.6
1416		6.0				Ī	27.7
1417 1418		4.0 6.0				1375	
1423	5.0	4.0				1350	16.1
1424	7;0	4.0				1400	
1425 1426		4.5					26.5 27.1
1427							29.8
1425							24.8
1426 ' 1428							20.0 23.7
1429							13.5
1430							13.5
1431 1432						1400 1300	16.0 15.0
1433						1400	25.4
1434 1427'							23.6
1427							16.4 24.7
1433 '							28.4
1434 ' 1435	7:0 5.0	4. ¹ 5 6.0			850 875	1400 1375	18.2
1436	7.0	6.0			6/5 	1375 	6.1 13.4
1437	7.0	2.5]_	13.9
1438 1439	3.0 5.0	6.0 5.0			875 850	13 <i>7</i> 5 1350	3.7 6.8
1440	5.0	5.0				1400	14.2
1441	710	6.0					18.2
1442 1443							27.7 26.0
1444							24.1
1445							21.0
1446 1442 '							21.1 21.7
1 443 '							26.2
1 <i>444</i> ' 1 <i>44</i> 1 '		6.0				1400 1300	26.5
14471	7.0	5.0 5.0	٥.٥	0.0	850	1400	6.0 12.2
,			- • -				

		He F	low (l/min)		Cr and Ti	Reactor	
Run No.	H ₂ Flow (l/min)	Through Bubbler	Cr and Ti Only	With H2	Temperature (°C)	Temperature (°C)	Δ Wt. (mg)
1448 1449 1445 1446 1448 1450 1451 1452 1453 1454 1455 1451 1464 1452	7.0 5.0 7.0 7.0 7.0 5.0	5.0	0.0	0,0	850 900 850 850 850 900	1300 1300 1400 1400 1400 1300	20.2 19.9 20.6 20.0 21.1 11.3 25.6 22.2 23.9 23.4 20.4 21.5 19.0 29.1 29.1 29.2 23.4
1455 1455 1466 1466 1466 1466 1466 1466		5.0 7.0				1300	21.7 23.4 19.6 22.8 24.6 22.4 20.2 26.7 24.6 17.6 21.7 17.0 23.7 21.2 23.0 21.9 21.0
1476: 1478: 1478: 1482: 1480: 1481: 1483: 1484: 1485: 1486: 1487: 1488:	5.0	7.0				1350 1300	15.2 24.8 20.0 22.9 20.9 19.4 18.0 40.7 27.2 23.5 24.3 24.9
1489 1484 ' 1485 ' 1486 ' 1487 ' 1488 ' 1498 1499 1500	4.0 4.0 4.0 5.0	5.0 3.0 5.0 5.0	0.0			1300 1350	22.3 26.6 24.9 19.6 20.2 20.3 20.0 26.8 23.1 22.5
1501 1502 1503 14981 14991 15001 15011	5.0	4.0	2.5	0.0	900	1350	22.7 17.1 20.8 22.3 24.8 24.9 20.7 20.6

		He F	low (1/min)		Cr and Ti	Reactor	
Run No	H ₂ Flow (½/min)	Through Bubbler	Cr and Ti Only	With	Temperature (°C)	Temperature (°C)	4 Wt.
1503'	5.0	4.0	2.5	0.0	900	1350	(mg) 12.0
1509	4.0	5.0	0.0	i	880	,,,,,	19.2
1510	5.0]	1	ĺ			18.5
1511 1512			l	ł			26.7
1513	Ì	1	ļ	J			30.6 31.5
1514					1		29.4
1515		1					30.3
1510'					}		32.9
1511' 1512'							35.5
15131	Į.	[[32.5 34.5
15141		1					35.8
1515'		5.0 4.0	0.0		. 880		35.9
1518		4.0	2.5		900		25.5
1519		· {	}		1	}	23.7
1520 1521							23.7 23.6
1522	1	{	ļ	}			23.8
15181		1		ļ			21.9
15191	1	}	İ		ľ		27.5
1520'	ļ						28.9
1521' 1522'	}	}	ĺ		j		23.0 20.7
1523				- 1			23.4
1523'		ĺ		Ì		1350	23.0
1524		J				1300	38.9
1525		Ì	Ì	1			42.3
1526	1	ł	ļ			Į.	40.0
1527 1528	į	Ì		İ		ļ	37.2 35.4
1529							27.7
15241		4.0	ŀ			1300	27.9
1530		5.0	ļ	1		1350	36.6
1531		3.0				1375	34.0
1532 1533	ł	1					29.8 30.4
1534							30.3
1535	1	{	,		}	{	31.3
1536		1					32.2
1531'	}	}					30.8
1532 ¹ 1533 ¹							30.5 30.7
1535'		})	34.2
1534'			1				34.0
1536'		3.0				1375	32.5
1525 ' 1526 '	1	4.0	1			1350	35.8 35.8
1527	1	1)	})	37.4
1528		[38.4
1530'	Í	1			900	ľ	42.5
1540		[1	88 0		36.7
1541 1542		Ì)				38.5 34.4
1542		1	Į	l			34.4 34.5
1544	1	Ì			1		36.5
1545				Į	ļ		35.4
15401	1]			[38.1
1541	'	ł		1	ļ		33.7
1 542 ' 1 543 '	}						33.5 30.1
1545					l		30.7
15451		4.0			88'0	1350	31.0
1546	{	3.0	l		890	1375	29.8
1547]	1					29.0
1548	- I	ء ا	_			[26.0
1549	5.0	3.10	2.5	0.0	890	1375	23.4

Titanium-Chromium Alloy Depositions (continued)

1,550		[low (l/min)		Cr and Ti	Reactor	
1550		H ₂ Flow				Temperature (°C)	Temperature (°C)	
1546							<u> </u>	
1543 1548 1549 1549 1549 1551 1551 1551 1565 1566 1566 1566 1566 1566 1566 1566 1566 1571 1566 1571 1588 1586 1588 1586 1588 1586 1588 1586 1588 1586 1588		J. U).U	1	"	1		
1948		Ĭ					į .	21.9
1549 1550 1600 1600 1600 1600 1600 1600 1600 1600 1600 1600 1600 1659		[ţ	Ì	1	1	}	20,2
1550		}	-	}	ļ	j		20.7
1551	15501	}		i i		1	1	20.4
1551				[Ì	1	1	21.0
1565	1551	į	3 0	1	ì	890	1375	18.3
1566	1565	- {	4.0	ļ	j	870	1350	14.6
1567 1568 1569 1570 1571 1566 1571 1566 1570 1571 1566 1570 1570 1570 1588 1325 20.4 1586 1586 1.0		ļ	3.5		l l	880		22.8
1568 1569 1570 1571 1571 1566 1567 1568 1569 1570 1570 1570 1580 1571 1571 1571 1571 1571 1571 1585 5.0 5.0 880 1350 11.8 1587 4.0 6.0 910 10.8 1588 4.0 6.0 97.1 1589 5.0 4.5 990 1350 40.2 1599 1596 13.5 2.5 880 1325 31.7 1591 3.5 2.5 880 1325 31.7 1592 5.0 3.0 6.0 97.7 1592 5.0 3.0 6.0 12.5 15.9 1596 1597 1598 122.9 1599 1600 1601 1601 1601 1601 1601 1601 16		Į į	,,,	ĺ	1		1	21.8
1569		ł	1	1	ļ			20.3
1570 22,4 20,1 1566 1567 20,8 20,7 20,2 20,0 20,1		Į)	ļ	l l			
1571 1566 1567 20.8 20.7 20.1568 20.7 20.2 20.0 20.8 20.7 20.2 20.0	1505	Į			{			
1566 20.8 20.7 20.2 20.9 20.7 20.2 20.0 20.1		ĺ		1	ł			
15667 20.7 20.7 20.8 20.7 20.1 1570 20.1 1570 20.1 1571 3.5 1325 20.4 1588 4.0 4.0 910 10.8 10.7 1588 4.0 6.0 910 1350 13.3 10.7 1588 4.0 6.0 910 1350 13.3 1599 3.5 2.5 880 1325 31.7 1599 1596 1599		}	}	į	j)		
1568 20.2 20.0 1571 1325 20.4 1585 5.0 5.0 880 1350 11.8 1585 5.0 5.0 880 1350 11.8 1587 4.0 6.0 910 10.8 9.7 1589 5.0 4.5 900 1350 40.2 1599 1599 1599 1599 1599 1599 1599 1599 1599 1600		}	1		i			
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592 5.0 3.0 6.0 28.4 21.9 19.0 22.9 19.0 22.9 19.0 22.9 22.9 22.9 23.5 29.0 26.2 26.2 26.2 26.9 26.2 26.2 26.2 26.2 26.9 26.2 26.2 26.0 27.0 27.2 27.2 27.1 27.2 27.1 27.2 26.0 27.1 27.2 26.8 27.1 27.2 26.8 <td< td=""><td></td><td>5.0</td><td>4.5</td><td></td><td>}</td><td></td><td>1350</td><td></td></td<>		5. 0	4.5		}		1350	
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593 4.0 2.0 2.5 594 19.0 595 1325 21.5 597 1350 28.5 598 29.0 600 26.9 601 29.9 602 28.0 598' 25.0 599' 23.3 600' 25.6 601' 27.1 602' 27.2 610 4.0 2.5 611 6.0 2.0 23.4 612 7.0 1.25 1.5 613 6.0 1.5 1.5 614 7.0 1.25 1.25 615 26.8 24.5 616 24.3 21.5 617 24.3 21.5 616 24.3 21.5 617 18.4 21.5		د ا	3.0	2.J	(000.	1,72,7	
594 19.0 596 1325 21.5 597 1350 28.5 598 29.0 26.9 600 26.9 28.0 598' 25.0 29.9 602 25.0 29.8 599' 29.8 25.6 600' 20.0 20.0 1350 23.4 610 4.0 2.5 1350 23.4 611 6.0 2.0 2.0 1375 26.8 612 7.0 1.25 1.5 28.0 614 7.0 1.25 1.5 28.0 615 616 24.3 21.5 26.8 616 24.3 21.5 26.8 24.5 616 24.3 21.5 21.7 21.7 616 617' 18.4 18.4 18.4		5.0			1		\	
	יוס: יוס:	4.0	2.0	2.5	1	*	į	
1325 21.5 1350 28.5 1350 28.5 1350 28.5 29.0 26.9 26.9 26.9 28.0 29.9 28.0 29.9 28.0 29.9 29.9 20.0 20.1 20.1 20.1 20.1 20.1 20.1 20.1	12 1	Į		l	1		1	
1350 28.5 998 999 26.2 900 26.2 901 29.9 98 999 28.0 98 999 28.0 999 28.0 999 28.	06	ſ	{	{	1	1	1325	
29.0 26.2 26.9 26.9 29.9 29.9 29.9 29.9 29.9	סקי	})	ļ		
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26.9 29.9 28.0 29.9 28.0 29.9 29.9 28.0 25.0 27.1 27.2 27.1 27.2 21.0 2.5 31.1 3.0 3.4 3.7 3.1 3.5 3.4 3.7 3.8 3.1 3.6 3.1 3.5 3.8 3.1 3.6 3.1 3.7 3.8 3.8 3.8 3.8 3.8 3.8 3.8 3.8 3.8 3.8		1		[i	1	}	
29.9 502 598' 599' 500' 501' 502' 510 4.0 2.5 1.5 511 6.0 2.0 2.0 1375 26.8 512 7.0 1.25 1.5 513 6.0 1.5 1.5 514 7.0 1.25 1.5 515 28.1 516 516 517 21.5 517'			Ì		1			
28.00 598' 597' 599' 500' 501' 502' 511 6.0 2.0 2.0 512 7.0 1.25 1.5 513 6.0 1.5 1.5 514 7.0 1.25 1.5 515 28.1 516 516 517 21.7 517'		}	1	}	J	\ \		
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23.3 599' 500' 501' 502' 510	502 508 !	ļ	[1			
29.8 600' 601' 602' 610		ł	1		1	}		
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502' 510		ſ	1	{	}	1		
510 4.0 2.5 1350 23.4 511 6.0 2.0 1375 26.8 512 7.0 1.25 1.5 28.1 513 6.0 1.5 1.5 28.0 514 7.0 1.25 1.25 26.8 515 24.5 516 24.3 515' 21.7 517' 18.4		l	1		J			
611 6.0 2.0 2.0 2.0 1375 26.8 612 7.0 1.25 1.5 28.1 28.0 614 7.0 1.25 1.25 26.8 24.5 615 616 615 616 617 615 71.7 18.4	610	μ 0		ء ا د	j	1	1350	23 T
612 7.0 1.25 1.5 613 6.0 1.5 1.5 614 7.0 1.25 1.25 615 616 617 617 21.5 616' 617'			2,0	2.0	1		1375	26.8
613 6.0 1.5 1.5 28.0 614 7.0 1.25 26.8 24.5 615 616 617 617 21.7 18.4	612	7.0	1.25	1.5	- 1	}	1,77	28.1
614 7.0 1.25 1.25 26.8 24.5 616 617 617 21.7 617 18.4	613		1.5	1.5	}			
615 616 617 615' 616' 617'	614	7.0	1 25	1 25	1	į		
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617 615' 616' 617'		1	1	}	Į.	ļ		
615' 616' 617'		ļ			ł			
616' 617'				1	l		}	
617'		{	{	l	1		}	
01/. 1 10.4		}	1		ļ			10 /
614' 7.0 1.25 1.25 0.0 880 1375 24.2	6141	٦١,	,	1.25	0.0	990	1270	24.2

Parameters and Results of Silicon Depositions

			Dilution	Bubbler	Reactor	
Run	Cubatana	H ₂ Flow	Flow	Flow	Temperature	Δ Wt.
No.	Substrate	(½/min)	(l/min)	(l/min)	(°C)	(mg)
1405-1 2	1202-1 1202	7.0	7.0	0.15	1 090	22.1
3	1188			1	1 090 1 090	40.6 18.6
1406-1	1399	ſ		1	1 080	8.1
2 3	1392	·	{	}	1 080	9.2
1405'-1	1071 1202-1	1		-	1 080 1 090	6.0 11.5
2	1202	}		\	1090	
3 1406'-1	1188				1 090	4.7
2	1399 1392	j	}		1 080 1 080	6.9 12.4
3	1071			{	1080	5.7
1413-1	1311				1 090	18.7
2 3	1310 1302	}		}		24.6 13.1
14131-1	1311					9.5
2	1310	Ì		Ì		11.9
3 1414-1	1302 1185			{		10.4 26.4
2	1186					30.9
3	1184	1		{	1090	54.3
1418-1 2	1391 1393				1085	28.4 21.0
3	1394	1	1			27.3
4	1390	İ		J		23.2
5 6	1395 1389	Ì	ì	1		39.7 23.5
14181-1	1391	{	(10.6
2	1393					12.5
3 4	1394 1390	}	1			16.4 13.3
5	1395					12.1
. 6	1389)	1085	13.5
1419-1 2	1400 1404				1 08 0	32.4 40.5
3 4	1398	ļ				37.3
	1403		(}		37.3
5 6	1402 1401	ļ				35.9 29.8
1419'-1	1400	}		ĺ		13.1
2	1404				1	17.0
3 4 5 6	1398 1403	ĺ				16.1 15.2
5	1402	1				4.4
	1401					5.2
1420-1 2	1408 1410					32.7 33.7
3	1407		1			29.9
1421-1	1409		1	Ì		40.1
2 3	1412 1411		1	.		31.1 28.9
1420'-1	1408					17.7
2	1410	}	-	. {		16.6
3 1421'-1	1407 1409		1		J	13.2 17.9
2	1412		}		İ	19.5
3	1411		1		{	19.3
1422-1 2	1105 1072	}]	[29.1 27.3
3	1066				1080	22.0
1456-1	1130		}		1090	27.0
2 3	1142 1141		}			10.0 6.7
145611	1130	7.0	7,0	o.15	1090	12.6
-	-	• •	•	-	_	

Run No.	Substrate	H ₂ Flow (£/min)	Dilution Flow (l/min)	Bubbler Flow (1/min)	Reactor Temperature (°C)	4 Wt. (mg)
1456'-2	1142	7.0	7.0	0,15	1 090	2.2
3 1457-1	1141 1182	1	}			2.1 24.8
2 3	1175 1181		}		ł	18.2 15.1
1458-1 2	1431 1418					14.4 16.7
3 1459-1	1416 1370					12.9 18.0
2 3	1368 1371				{	16.6 17.3
1460-1 2	1434 1428				{	13.5 16.5
3 1461-1	1433 1427				1	16.1 20.7
2 3	1425 1426				Ì	18.2 14.9
1460'-1 2	1433 1428				Ì	18.1 9.4
3 1461'-1	1434 1427					2.4 2.3
2 3	1425 1426				1 090	10.6 22.1
1462-1 2	1172 1174				1 085 1 085	24.9
3 1463-1	1173 1437		Ì		1 08 5 1 0 9 0	29.1 20.9
2 3	1440 1463				Ī	17.9 19.0
1470-1 2	1153 1154					23.7 17.8
3 1471 - 1	1155 1450				İ	13.6 22.4
2 3	1108 1447					22.3 17.7
1472-1 2	1448 1442					27.9 22.8
3 1473-1	1444 1445					15.3 27.9
2	1446 1443					19.0 13.2
1472'-1 2	1448 1442					15.0 11.5
3 1473 '-1	1 <i>444</i> 1445				{	7.5 14.2
2	1446 1443		j			12.4
1474-1 2	1449 1453					5.8 29.2 26.4
3 1475-1	1454 1451	}				21.4 32.7
2 3	1452 1455					25.4 20.7
1474'-1 2	1449 1453					7.7 7.8
3 1475'-1	1454 1451					4.6 22.5
2	1452 1455					20.2
1491-1 2	1 046 1 047					18.2
3 1492-2	1 048 982					10.7
3 1493-1	976 1128				1 090 1 080	11.9 20.0
2 3	1142 1142	7.0	7 . 0	0.15	1 08 0 1 08 0	3.8 2.4

Run		H ₂ Flow	Dilution Flow	Bubbler Flow	Reactor Temperature	۷ Wt.
No.	Substrate	(l/min)	(l/min)	(l/min)	(°C)	(mg)
1494-1 2	1231 1230	7.0	7 . 0	0.15	1090	
3 1495-1	1202			j	}	12.0
2	1266 1265					16.5 15.6
3 1496-1	1267 1467		1		1090	13.5
2	1466	1	}	1	1 095 1 095	29.8 33.6
3 1497 - 1	1465 1469				1 095 1 080	33.3 27.8
2	1468				1080	27.8
3 1496 - 1	1464 1467				1 08 0 1 09 5	11.3 11.5
2 3	1466 1465	-			1095	11.8
14971-1	1469	,			1 095 1 0 8 0	10.4 11.6
2	1468 1464			}	1 08 0 1 08 0	12.7 14.1
1504-1	1479	{			1090	24.2
2 3	1480 1476					25.8 13.7
1505 - 1 2	1481 1482	}				29.1
3	1478	}				32.9 19.0
15041-1 2	1479 1480					8.6 8.9
3	1476					2.8
1506 - 1 2	1229 1268					15.0 17.4
3 1505'-1	1246 1481		}		}	15.5
2	1482					9.2 6.0
3 1507 - 1	1478 1489					4.2 32.0
2	1488				ť	30.2
3 1508-1	1485 1487					33.0 29.5
2 3	1486 1484					29.4 38.0
1507'-1	1489	}				26.5
2 3	1488 1485				}	18.8 13.3
15081-1	1487 1486					20.1
2 3	1484)		19.4 18.6
1516-1 2	1386 1388					7.8 6.8
3	1387			}		4.7
1517-1 2	1490 1005					9.5 20.0
3	1483 822					
1537-1 2	832					17.0 16.7
3 1538-1	831 849					10.1 11.6
2	851				1	11.2
3 1552 - 1	850 846				1	7•7 [.] 3 . 8
2	844 847				1	5.4 2.4
1553-1	1439		Ì			16.5
2 3	1436 1509		1			16.6 15.0
1554-1	1367 1369					18.6 16.9
2 3	1366	7.0	7.0	0.15	1090	9.5

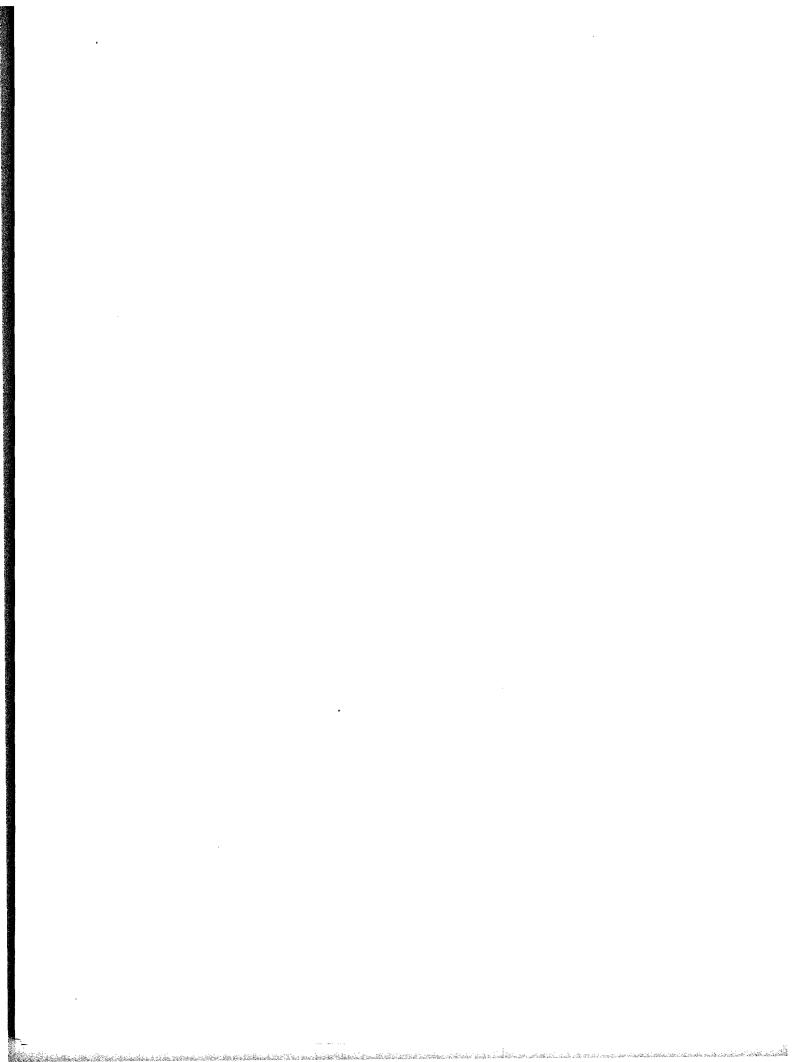
Run		H ₂ Flow	Dilution Flow	Bubbler Flow	Reactor Temperature	Δ Wt.
No.	Substrate	(l/min)	(l/min)	(1/min)	(°C)	(mg)
1555-1	1503	7.0	7.0	0,15	1090	9.5
2 3	1499 1498					16.1 15.2
1556-1	1501 1500			1		19.9
2 3	1502					18.1 12.4
1555'-1 2	1503 1499					3.1 4.9
3	1498			Ì		6.2
1556'-1 2	1501 1500					6.8 7.7
3	1502					5.3
1557 - 1 2	1300 1361			ĺ		19.5
3	1360					16.7
155 8- 1 2	1514 1513					24.4 30.1
3	1510	ĺ		ĺ	1	18.7
1559-1 2	1511 1515					25.5 24.3
3	1512	ļ				22.0
1558 '-1 2	1514 1513	ĺ		Í	ľ	7.2 8.7
3	1510					9.3
1559'-1 2	1511 1515					7.7 16.2
3	1512			l	{	12.5
1560 -1 2	1523 1518					19.0 16.0
3 1561-1	1519		1			5.3
2	1522 1520					17.2 3.6
3 1560'-1	1521 1523					11.3 10.9
2	1518					3.0
3 1561 ¹ – 1	1519 1522		1			3.9 11.5
2	1520					7.2
3 1562-1	1521 841					6.3
2	781		ĺ	ĺ	1	14.4
3 1563-1	771 1530					19.7 36.2
2	1527			Ì		29.4
3 15 6 4-1	1524 1528				1	7.1 21.7
2	1526			l		21.8
3 1563'-1	1525 1530					17.1 18.3
2	1527			ĺ		25.5
15641 - 1	1524 1528					20.6 24.2
2	1526					13.0
3 1572 - 1	1525 787			{	[18.4 17.0
2 3	733 734					15.6 11.5
1573-1	1536		J			30.8
2	1531 1533					18.1 18.2
1574-1	1532					17.4
2 3	1534 1535			Ì		26.0 24.2
3 1573'-1	1536				Í	20.3
2	1531 1533			\		20.3 13.5
15741-1	1532	7.0	7.0	0.15	1090	19.7

No. Substrate (L/min) (L/min) (L/min) (°C) 1574'-2 1534 7.0 7.0 0.15 1090 3 1535 1575-1 1429 2 1430 3 1313 1576-1 1441 2 1517 3 1397 1577-1 1124 2 1126 3 1125	
3 1535 1575-1 1429 2 1430 3 1313 1576-1 1441 2 1517 3 1397 1577-1 1124 2 1126 3 1125	(mg) 28.5
2 1430 3 1313 1576-1 1441 2 1517 3 1397 1577-1 1124 2 1126 3 1125	17.6
1576-1 1441 2 1517 3 1397 1577-1 1124 2 1126 3 1125	13.8 14.1
2 1517 3 1397 1577-1 1124 2 1126 3 1125	17.8 16.0
1577-1 1124 2 1126 3 1125	11.0
2 1126 3 1125	12.2
3 1125	16.0 16.9
1578-1 1543	9.5
2 1541	27.7 31.5
3 1545 1579-1 1542	23.9 28.8
2 1540	32.5
3 1544 1578'-1 1543	32.8 26.3
2 1541	17.3
3 1545 1579'-1 1542	14.6 23.0
2 1540	7.2
3 1544 1580-1 722	4.7 6.9
2 720	14.6
3 729 1581-1 1548	10.8 14.8
2 1546	14.9
3 1550 1582-1 1549	8.8 19.6
2 1547	16.1
3 1551 1581'-1 1548	9.6 14.8
2 1546 3 1550	13.6
15821-1 1549	8.7 14.8
2 1547 3 1551	13.1 7.2
1583-1 1569	12.9
2 1567 3 1571	15.7 4.6
1584-1 1568	14.6
2 1570 3 1566	14.5 10.8
1583'-1 1569	9.0
3 1571	6.3 7.6 7.2
1584'-1 1568 2 1570	7.2
3 1566	9.4 6.2
1603-1 755 2 756	15.8
3 749	18.7 13.0
1604-1 704 2 728	6.4 14.2
3 727	16.5
1605-1 792 2 793	16.9 16.9
3 786	12.0
1606-1 782 2 783	12.2 12.2
3 785	9.8
2 758	14.5 14.9
3 779 1608-1 824	9.6
2 827 7.0 7.0 0.15 1090	16.3 17.3

Run No.	Substrate	H ₂ Flow (l/min)	Dilution Flow (l/min)	Bubbler Flow (L/min)	Reactor Temperature 4 Wt. (°C) (mg)
3	825	7.0	7.0	0.15	1090 11.6
1609-1	823	1		1	13.4
2	829				13.4
3	828				8.8
1624-1	1593		Ì	ł	15.6
2	1592				12.0
3	1591				11.4
1625-1	1594				18.7
2	1596				15.3
3	1612				23.6
1626-1	1590		ł		10.6
2	1584	ĺ	[ĺ	7.2
. 3	1580				
1627-1	1599				20.0
2	1600				17.0
3	1601				6.0
1628-1	1602				20.8
2	1597		J		18.9
3	1598				17.8
1627'-1					7.6
2]			6.3
3					11.0
1628'-1					7.9
2					6.0
3		1			10.2
1629-1	1613				8.2
2	1611				2.3
1630-1	1615				8.3
2	1616				8.8
3	1614				6.1
4	1617		l		5.2
1630'-1	1615	1		1	3.9
2					4.6
3	1614	_ _	_ _		2.0
4	1617	7:0	7.'0	0.15	1090 1.5

Parameters and Results of Chromium-Titanium-Silicon Depositions

			He Flows	(l/min)				
Run No.	H ₂ Flow (<i>l</i> /min)	TiCl4 Bubbler	TiCl ₄ Dilutant	SiC14 Bubbler	SiCl ₄ Dilutant	Cr & Ti Temperature (°C)	Reactor Temperature (°C)	≟ Wt. (mg)
1618 1619 1620 1621 1622 1623 1622 1633 1634 1635 1634 1636 1637 1637 1638 1639	7.0	1.25 4.0 4.0 1.5 0.0	1.25 2.5 2.5 4.0 5.0 4.0	0.10 0.05 0.05 0.03	1.25	910 910 880	1350 1350 1250	52.8 54.0 52.8 53.2 51.1 43.2 50.2 31.0 47.5 44.3 57.4 49.0 49.0 47.5 41.5 41.5 41.5 42.1 43.2
16381	7.0	1.5	4!0	0.03	1 . 25	880	1250	35.5



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13 ABSTRACT

The basic processes necessary to manufacture refractory metal coatings using chemical vapor deposition technology were investigated. The processes developed were demonstrated on a typical oxidation resistant coating of titanium-chromium-silicon but would have general applicability to other similar coatings. The processes utilized both two-stage deposition (metal alloy coating followed by silicon deposition) and a new method developed for the rapid (10 minute) deposition of all three elements in a single step. The processes were used for deposition of metallic coatings and for controlled variation of the composition of titanium and chromium alloys. Substrate property degradation and interaction were minimal for all types of coatings.

Performance of the coatings was evaluated by oxidation testing in air and was found to be nominally comparable to coatings by other processes. Reliability and reproducibility, indicated by statistical analysis of results, were high for this initial effort program.

The processes have potentials for utilization in rapid and economical manufacturing of a variety of coatings on parts with complex shapes.

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