

**ANALYSIS OF STRUCTURAL SHELLS WITH
TRANSVERSE SHEAR DEFORMATION**

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FOREWORD

This final technical report was prepared by Anamet Laboratories, Inc., Berkeley, California, under Air Force Contract No. AF 33(615)-70-C-1019. It was administered under the Structures Division, Air Force Flight Dynamics Laboratory, with Messrs. T. N. Bernstein (FBR) and J. L. Rudd (FBR), acting as project engineers.

This report was completed in June 1971 and covers the work performed from June 1969 to February 1971. The supervision of the project, the development of the mathematical analyses, and the programming of the modified Kalnins Static Program were carried out by Richard L. Citerley.

This technical report has been reviewed and is approved.



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ABSTRACT

This report describes the results of an investigation for the numerical analysis procedures of shells of revolution and consultation on the execution of shell problems at the Air Force Flight Dynamics Laboratory Facilities. A computer program for the analysis of the thick shells of revolution, the analysis of thin shells subjected to torsion and the theory upon which it is based is presented. This computer program is tailored to an existing program previously written by Dr. Kalnins under Air Force Contract No. 33 (615)-3870. Detailed input instructions and three example problems are included to illustrate the added capabilities of the computer program. In addition, numerical procedures and subroutines that effect an improvement in numerical integration procedures and efficient eigenvalue search techniques are discussed.

TABLE OF CONTENTS

SECTION		PAGE
I	INTRODUCTION	1
II	THIN SHELL COMPUTER PROGRAM	3
	1. Basic Assumptions and Geometry	3
	2. Derivation of the Governing Set of Equations	14
	3. Program Modification	18
	4. Development of Torsion	26
	5. Program Size	26
III	THE EIGENSOLUTION	28
	1. Modification to AEP	31
IV	NUMERICAL INTEGRATION PACKAGES	41
	1. ADMIP USAGE	42
	2. GLAUZ	53
	3. DZ BKY ZAM	59
	4. AL ADAM	64
	5. DZ C STR	74
	6. D2 GDC CSPDIF	81
V	SOLUTION OF EXAMPLE PROBLEMS	86
	REFERENCES	90
	APPENDIX A - TKSHL COMPUTER PROGRAM LISTING	92
	APPENDIX B - EXAMPLE PROBLEMS - COMPUTER OUTPUT	161
	1. Torsional Problem Results - Case I	162
	2. Internal Pressure Loading - Case II	168
	3. External Pressure Loading - Case III	180

ILLUSTRATIONS

FIGURE		PAGE
1.	ELEMENT OF A SHELL OF REVOLUTION	5
2.	MULTILAYERED SHELL ELEMENT	9
3.	FLOW CHART OF AEP	33
4.	TORSIONAL PROBLEM	86
5.	INTERNAL PRESSURE LOADING	86
6.	EXTERNAL PRESSURE LOADING	86
7.	HOOP STRESS VERSUS POSITION ACROSS THICKNESS (INTERNAL PRESSURE LOADING)	87
8.	HOOP STRESS VERSUS POSITION ACROSS THICKNESS (EXTERNAL PRESSURE LOADING)	88

TABLES

TABLE		PAGE
I	RESULTS FOR CANTILEVERED CONICAL SHELL	30

SYMBOLS

u, v, w, β	Displacement and Rotation Components
N, N_n, Q	Membrane Stress Resultants
n	Fourier Harmonic Index
$M_\phi, M_\theta, M_{\phi\theta}$	Moment Resultants
m	Surface Moments
P	Surface Pressure
T	Temperature
Z	Coordinate of Layers
R	Radius of Curvature
K, C	Membrane Stiffness
D	Bending Stiffness
B, a, G, H, E, F	Material Elastic Properties
h	Thickness of Shell
θ, s, r, ϕ	Coordinates for a Shell of Revolution
$\vec{t}_\phi, \vec{t}_\theta, \vec{n}$	Coordinate Vectors
ω	Circular Frequency
σ	Stress
ϵ	Strain
ν	Poisson's Ratio
Δ	FORTRAN Blank
λ	Eigenvalue
Ω	Characteristic Frequency
c	Velocity of Sound

SECTION I
INTRODUCTION

The following report covers the consulting effort Anamet has performed for the Air Force Flight Dynamics Laboratory (AFFDL/FBR) over the period from June 1969 to February 1971. The primary purpose of the consultation is to provide a computer program for the analysis of shells of revolution to include the effects of transverse shear deformation. The second purpose is to investigate numerical procedures that may effect an improvement in computer efficiency and cost effectiveness for numerical integration procedures and for eigenvalue search techniques. The third purpose is to provide consultation on the execution of shell problems at the AFFDL computer facilities.

The presentation of a thick shell program is made on the basis of developing the governing system of first order linear differential equations using an existing shell theory. The development is tailored to an existing equation solver. The analysis of multi-layered orthotropic shells of revolution subjected to nonaxisymmetric surface loads is provided by the computer program written by Dr. Kalnins under USAF Contract No. 33(615)-3870⁽¹⁾.

The numerical analysis of the governing system of equations based upon classical shell theory was partially developed during past association with and during the course of research by the personnel of Anamet and Dr. Kalnins⁽²⁾. The basis of the basic computational method has been well documented in both private⁽³⁾⁽⁴⁾⁽⁵⁾ and archive sources⁽⁶⁾⁽⁷⁾. Other authors have also independently developed the basic numerical analysis technique to other classes of shells

and structures. This has been adequately brought out during early discussions⁽⁸⁾ as well as excellent presentations by Kraus⁽⁹⁾ and by Anderson et al⁽¹⁰⁾.

The formulation for the governing equations for the deformation of layered orthotropic shells of revolution including the effects of transverse shear deformation and normal stress has been presented by Hildebrand, Reissner and Thomas⁽¹¹⁾. The presentation of an early version of a computer program based upon Reissner's equations was supplied to FBR at the onset of this consulting effort. FBR found it desirable to be capable of analyzing thick or thin shells of revolution subjected to nonsymmetric surface loads with the same basic shell code. On the assumption that transverse shear deformation can account for the static behavior of moderately thick shells, the same program could be used as previously developed. In this way, minimum changes in the input parameters could be attained. The user would not have to drastically revamp his input data format. Thus, his utilization of the basic shell code could continue.

SECTION II THIN SHELL COMPUTER PROGRAM

The development of the fundamental equations for shell analysis has been presented in Chapter III, Reference 1. In addition the procedure for generating the terms from a particular shell theory to a set of linear first order differential equations has been presented. In order to make this report complete, the derivation of the system of equations is repeated. The equations are again as they appear in Reference 11, pages 48-53.

1. BASIC ASSUMPTIONS AND GEOMETRY

The stress-strain relations for each layer of a layered orthotropic shell of revolution where effects of normal stress have been ignored are given by the following:⁽¹²⁾

$$\begin{bmatrix} \epsilon_{11} \\ \epsilon_{22} \\ \epsilon_{12} \\ \epsilon_{13} \\ \epsilon_{23} \end{bmatrix} = \begin{bmatrix} a_{11} & a_{12} & 0 & 0 & 0 \\ a_{21} & a_{22} & 0 & 0 & 0 \\ 0 & 0 & a_{44} & 0 & 0 \\ 0 & 0 & 0 & a_{55} & 0 \\ 0 & 0 & 0 & 0 & a_{66} \end{bmatrix} \begin{bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{12} \\ \sigma_{13} \\ \sigma_{23} \end{bmatrix} \quad (1)$$

where:

$$a_{11} = \frac{1}{E_1} ; a_{12} = -\frac{\nu_{12}}{E_2} ; a_{22} = \frac{1}{E_2}$$
$$a_{44} = \frac{1}{G_{12}} ; a_{55} = \frac{1}{G_{13}} ; a_{66} = \frac{1}{G_{23}}$$

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In describing the shell's properties, and deformations, it is convenient to choose directions depicting the shape of the shell by specifying the two principal radii of curvature; therefore, the indexed directions in Equation (1) would replace 1 by ϕ , 2 by θ and 3 by z . If the material lines of orthotropy do not coincide with the principal directions of the radii of curvature and differ by an angle α , the new coefficients are expressed as:

$$a_{11}^* = a_{11} \cos^4 \alpha + (2a_{12} + a_{66}) \sin^2 \alpha \cos^2 \alpha + a_{22} \sin^4 \alpha$$

$$a_{12}^* = (a_{11} + a_{22} - 2a_{12} - a_{66}) \sin^2 \alpha \cos^2 \alpha + a_{12}$$

$$a_{16}^* = [a_{22} \sin^2 \alpha - a_{11} \cos^2 \alpha + 1/2 (2a_{12} + a_{66}) \cos 2\alpha] \sin 2\alpha$$

$$a_{22}^* = a_{11} \sin^4 \alpha + (2a_{12} + a_{66}) \sin^2 \alpha \cos^2 \alpha + a_{22} \cos^4 \alpha$$

$$a_{26}^* = [a_{22} \cos^2 \alpha - a_{11} \sin^2 \alpha - 1/2 (2a_{12} + a_{66}) \cos 2\alpha] \sin 2\alpha$$

$$a_{66}^* = 4(a_{11} + a_{22} - 2a_{12} - a_{66}) \sin^2 \alpha \cos^2 \alpha + a_{66}$$

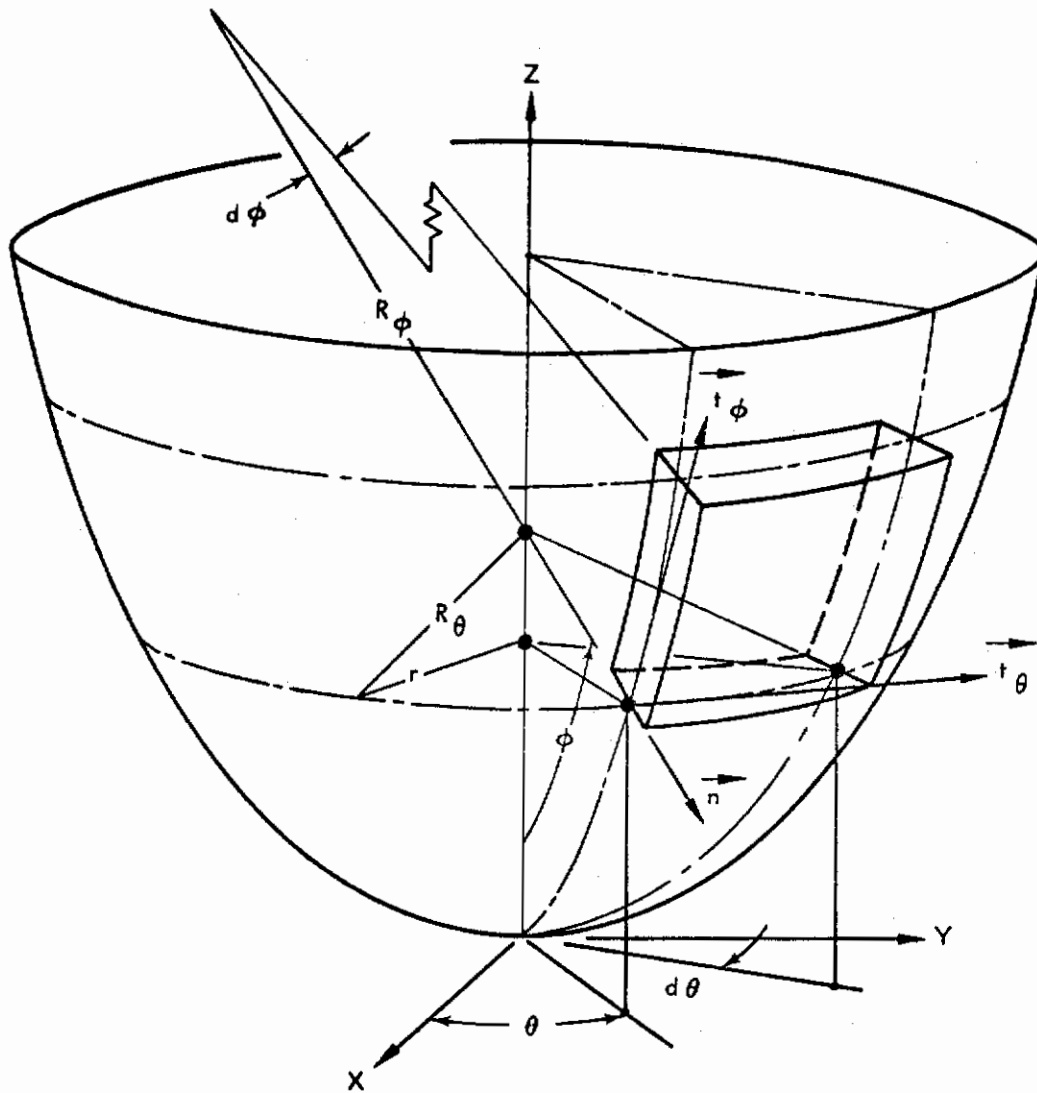


Figure 1. Element of a Shell of Revolution

If the lines of orthotropy do not coincide with the lines of principal curvature for the shell, the shell's deformations and stress distribution will no longer follow the laws governing orthotropic shells of revolution and the procedures developed in this report would not be applicable.

The remaining assumption restricting the behavior of the orthotropic shell is the displacement vector (referenced to triad, Figure 1) given by the relation:

$$\vec{u} = \vec{t}_\phi (u_\phi + z\beta_\phi) + \vec{t}_\theta (u_\theta + a\beta_\theta) + \vec{n}w \quad (2)$$

where all deformed parameters are only functions of ϕ and θ . Again, it should be pointed out (Reference 6) that once \vec{t}_ϕ and \vec{t}_θ have been chosen, the direction for the normal vector is dictated by:

$$\vec{n} = \vec{t}_\phi \times \vec{t}_\theta$$

Using the above assumed relation for the displacement vector, the strain-displacement relations are taken to be:

$$\left(1 + \frac{z}{R_\phi}\right) \epsilon_\phi = \bar{\epsilon}_\phi + z\kappa_\phi \quad (3.a)$$

$$\left(1 + \frac{z}{R_\theta}\right) \epsilon_\theta = \bar{\epsilon}_\theta + z\kappa_\theta \quad (3.b)$$

$$2\left(1 + \frac{z}{R_\phi}\right) \left(1 + \frac{z}{R_\theta}\right) \epsilon_{\phi\theta} = \left(1 + \frac{z}{R_\theta}\right) (\gamma_\phi + z\delta_\phi) + \left(1 + \frac{z}{R_\phi}\right) (\gamma_\theta + z\delta_\theta) \quad (3.c)$$

$$2\left(1 + \frac{z}{R_\phi}\right) \epsilon_{\phi z} = \gamma_{\phi z} \quad (3.d)$$

$$2\left(1 + \frac{z}{R_\theta}\right) \epsilon_{\theta z} = \gamma_{\theta z} \quad (3.e)$$

where:

$$\bar{\epsilon}_\phi = \frac{1}{R_\theta} (u_{\phi,\phi} + w) \quad (3.f)$$

$$\bar{\epsilon}_\theta = \frac{1}{r} u_{\theta,\theta} + \frac{\cos \phi}{r} u_\phi + \frac{\sin \phi}{r} w \quad (3.g)$$

$$\kappa_\phi = \frac{\beta_{\phi,\phi}}{R_\phi} \quad ; \quad \delta_\phi = \frac{\beta_{\theta,\phi}}{R_\phi} \quad (3.h-3.i)$$

$$\kappa_\theta = \frac{1}{r} \beta_{\theta,\theta} + \frac{\cos \phi}{r} \beta_\phi \quad ; \quad (3.j)$$

$$\delta_\theta = \frac{1}{r} \beta_{\phi,\theta} - \frac{\cos \phi}{r} \beta_\theta \quad (3.k)$$

$$\gamma_\phi = \frac{u_{\theta,\phi}}{R_\phi} \quad ; \quad \gamma_\theta = \frac{1}{r} u_{\phi,\theta} - \frac{\cos \phi}{r} u_\theta \quad (3.l-3.m)$$

$$\gamma_{\phi z} = \frac{1}{R_\phi} (w_{,\phi} - u_{,\phi}) + \beta_\phi \quad (3.n)$$

$$\gamma_{\theta z} = \frac{1}{r} (w_{,\theta} - \sin \phi u_\theta) + \beta_\theta \quad (3.o)$$

For a multilayered orthotropic shell containing m -layers (Figure 2) with a total thickness h , where z_i denotes the distance from the bottom surface of the shell to the top surface of the i -th layer, and where z_r is the reference distance from the bottom of the shell surface to the reference surface, the stress resultants are defined by:

$$\begin{aligned}
 \begin{bmatrix} N_\phi \\ N_{\theta\phi} \\ N_{\phi\theta} \\ N_\theta \end{bmatrix} &= \sum_{i=1}^m \int_{z_i - z_r}^{z_{i+1} - z_r} \begin{bmatrix} \sigma_\phi \left(1 + \frac{z}{R_\theta}\right) \\ \sigma_{\phi\theta} \left(1 + \frac{z}{R_\theta}\right) \\ \sigma_{\phi\theta} \left(1 + \frac{z}{R_\phi}\right) \\ \sigma_\theta \left(1 + \frac{z}{R_\phi}\right) \end{bmatrix} dz \\
 \begin{bmatrix} M_\phi \\ M_{\phi\theta} \\ M_{\theta\phi} \\ M_\theta \end{bmatrix} &= \sum_{i=1}^m \int_{z_i - z_r}^{z_{i+1} - z_r} \begin{bmatrix} \sigma_\phi \left(1 + \frac{z}{R_\theta}\right) \\ \sigma_{\phi\theta} \left(1 + \frac{z}{R_\theta}\right) \\ \sigma_{\phi\theta} \left(1 + \frac{z}{R_\phi}\right) \\ \sigma_\theta \left(1 + \frac{z}{R_\phi}\right) \end{bmatrix} z dz \\
 \begin{bmatrix} Q_\phi \\ Q_\theta \end{bmatrix} &= \sum_{i=1}^m \int_{z_i - z_r}^{z_{i+1} - z_r} \begin{bmatrix} \sigma_{\phi z} \left(1 + \frac{z}{R_\theta}\right) \\ \sigma_{\theta z} \left(1 + \frac{z}{R_\phi}\right) \end{bmatrix} dz
 \end{aligned} \tag{4}$$

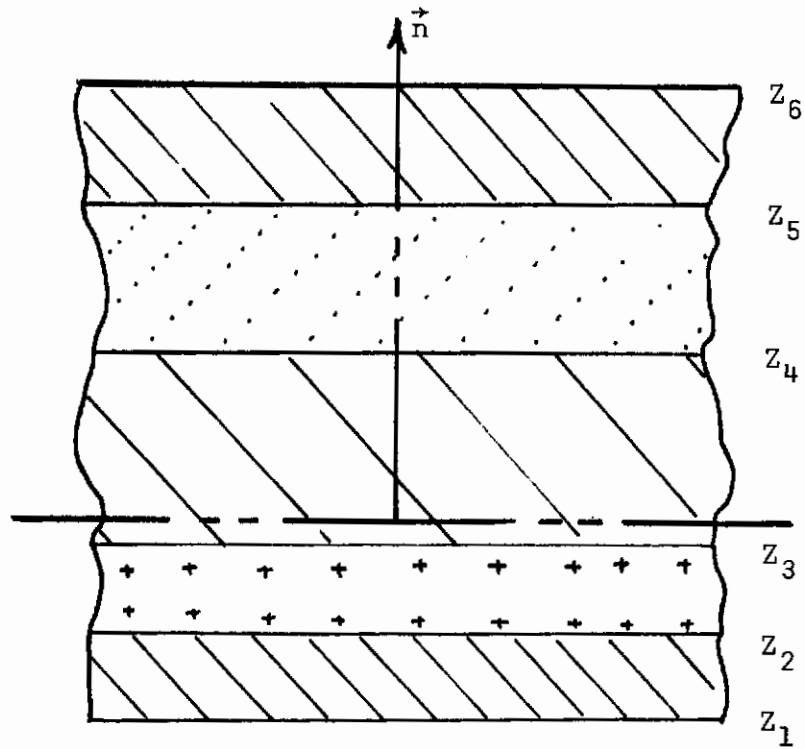


Figure 2. Multilayered Shell Element

Substituting Equations (3) and (4) into the inverse of Equation (1) with the usual assumptions that:

$$\frac{h^2}{R^2} \ll 1$$

the resulting stress resultant-strain relations are obtained:

$$N_{\phi} = C_{11} \epsilon_{\phi} + C_{12} \epsilon_{\theta} + K_{11} \gamma_{\phi} + K_{12} \gamma_{\theta} \quad (5.a)$$

$$N_{\phi\theta} = E_{11} \gamma_{\phi} + E_{12} \gamma_{\theta} + F_{11} \delta_{\phi} + F_{12} \delta_{\theta} \quad (5.b)$$

$$N_{\theta\phi} = E_{12} \gamma_{\phi} + E_{22} \gamma_{\theta} + F_{12} \delta_{\phi} + F_{22} \delta_{\theta} \quad (5.c)$$

$$N_{\theta} = C_{12} \epsilon_{\phi} + C_{22} \epsilon_{\theta} + K_{12} \gamma_{\phi} + K_{22} \gamma_{\theta} \quad (5.d)$$

$$M_{\phi} = K_{11} \epsilon_{\phi} + K_{12} \epsilon_{\theta} + D_{11} \gamma_{\phi} + D_{12} \gamma_{\theta} \quad (5.e)$$

$$M_{\phi\theta} = F_{11} \gamma_{\phi} + F_{12} \gamma_{\theta} + H_{11} \delta_{\phi} + H_{12} \delta_{\theta} \quad (5.f)$$

$$M_{\theta\phi} = F_{12} \gamma_{\phi} + F_{22} \gamma_{\theta} + H_{12} \delta_{\phi} + H_{22} \delta_{\theta} \quad (5.g)$$

$$M_{\theta} = K_{12} \epsilon_{\phi} + K_{22} \epsilon_{\theta} + D_{12} \gamma_{\phi} + D_{22} \gamma_{\theta} \quad (5.h)$$

$$Q_{\phi} = H_{55} \gamma_{\phi z} \quad (5.i)$$

$$Q_{\theta} = H_{66} \gamma_{\theta z} \quad (5.j)$$

where the elastic parameter is given for a layered shell:

$$\begin{Bmatrix} C_{11} \\ C_{12} \\ C_{22} \end{Bmatrix} = \sum_{i=1}^m \begin{Bmatrix} B_{11} \\ B_{12} \\ B_{22} \end{Bmatrix} \left\{ \left(z_{i+1} - z_i \right) + \left(1 - \frac{R_2}{R_1} \right) \sum_{n=2}^m \left(\frac{1}{R_2} \right)^{n-1} z_n \right\} \quad (6)$$

$$\begin{Bmatrix} K_{11} \\ K_{12} \\ K_{22} \end{Bmatrix} = \frac{1}{2} \sum_{i=1}^m \begin{Bmatrix} B_{11} \\ B_{12} \\ B_{22} \end{Bmatrix} \left\{ \left[\left(z_{i+1} - z_r \right) - \left(z_i - z_r \right) \right] + \left(1 - \frac{R_2}{R_1} \right) \sum_{n=3}^m \left(\frac{1}{R_2} \right)^{n-2} z_n \right\} \quad (7)$$

$$\begin{Bmatrix} D_{11} \\ D_{12} \\ D_{22} \end{Bmatrix} = \frac{1}{3} \sum_{i=1}^m \begin{Bmatrix} B_{11} \\ B_{12} \\ B_{22} \end{Bmatrix} \left\{ \left[\left(z_{i+1} - z_r \right) - \left(z_i - z_r \right) \right] + \left(1 - \frac{R_2}{R_1} \right) \sum_{n=4}^m \left(\frac{1}{R_2} \right)^{n-3} z_n \right\} \quad (8)$$

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$$E_{11} = E_{12} = E_{22} = \sum_{i=1}^m B_{44}^i (z_{i+1} - z_i) + \left(1 - \frac{R_2}{R_1}\right) \sum_{n=2}^m \left(\frac{1}{R_2}\right)^{n-1} z_n \quad (9)$$

$$F_{11} = F_{12} = F_{22} = \frac{1}{2} \sum_{i=1}^m B_{44}^i \left[(z_{i+1} - z_r)^2 - (z_i - z_r)^2 \right] + \left(1 - \frac{R_2}{R_1}\right) \sum_{n=3}^m \left(\frac{1}{R_2}\right)^{n-2} z_n \quad (10)$$

$$H_{11} = H_{12} = H_{22} = \frac{1}{3} \sum_{i=1}^m B_{44}^i \left[(z_{i+1} - z_r)^3 - (z_i - z_r)^3 \right] + \left(1 - \frac{R_2}{R_1}\right) \sum_{n=4}^m \left(\frac{1}{R_2}\right)^{n-3} z_n \quad (11)$$

where $z_n = (z_{i+1}^n - z_i^n) / n$

$$H_{55} = \sum_{i=1}^m B_{55}^i (z_{i+1} - z_i) \quad (12)$$

$$H_{66} = \sum_{i=1}^m B_{66}^i (z_{i+1} - z_i) \quad (13)$$

and where:

$$B_{11}^i = \frac{E_{\phi}^i}{1 - v_{\phi}^i v_{\theta}^i} ; \quad B_{12}^i = \frac{v_{\theta}^i E_{\phi}^i}{1 - v_{\phi}^i v_{\theta}^i} ; \quad B_{22}^i = \frac{E_{\theta}^i}{1 - v_{\phi}^i v_{\theta}^i} \quad (14)$$

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$$B_{44}^i = G_{\phi\theta} \quad ; \quad B_{55}^i = G_{\phi z} \quad ; \quad B_{66}^i = G_{\theta z} \quad (15)$$

Finally basic equations of equilibrium for a shell element and in terms of the stress resultants defined in Equation (4) are required:

$$N_{\theta,\theta} + \frac{r}{R_\phi} N_{\phi\theta,\phi} + 2 \cos \phi N_{\theta\phi} + Q_\theta \sin \phi + r p_\theta = 0 \quad (16a)$$

$$N_{\theta\phi,\theta} + \frac{r}{R_\phi} N_{\phi,\phi} + (n_\phi - n_\theta) \cos \phi + \frac{r}{R_\phi} Q_\phi + r p_\phi = 0 \quad (16b)$$

$$Q_{\theta,\theta} + \frac{r}{R_\phi} Q_{\phi,\phi} + Q_\phi \cos \phi - \sin \phi N_\theta - \frac{r}{R_\phi} N_\phi + r p_n = 0 \quad (16c)$$

$$M_{\theta,\theta} + \frac{r}{R_\phi} M_{\phi\theta,\theta} + 2 \cos \phi M_{\theta\phi} - r Q_\theta + r m_\theta = 0 \quad (17a)$$

$$M_{\theta\phi,\theta} + \frac{r}{R_\phi} M_{\phi,\phi} + (M_\phi - M_\theta) \cos \phi - r Q_\phi + r m_\phi = 0 \quad (17b)$$

As established for the isotropic case, Equations (3) through (17) have the order of 10 when transverse shear deformation is included, and the order of eight when transverse shear is neglected. Since the above defined displacements and stress resultants enter into natural boundary conditions at the edge of a medium plane where $\phi = \text{constant}$, then the two point boundary problem for rotationally symmetric shells can be solved.

2. DERIVATION OF THE GOVERNING SET OF EQUATIONS

The object of the following procedure is to develop a set of first order ordinary differential equations. The fundamental set of generalized displacements $w, u_\phi, u_\theta, \beta_\phi, \beta_\theta$, and generalized stress resultants $Q_\phi, N_\phi, N_{\theta\phi}, M_\phi, M_{\theta\phi}$ can be expressed in such a manner as to be consistent with the thin shell theory of either Reissner or Naghdi. Also it is convenient to reference all derivatives to the intrinsic coordinate system along the surface meridional length. In this way, continuous shells having a geometry where more than one radial station lies at one latitude plane of the shell can be analyzed (e.g. a circular plate). In addition, it is assumed that the variables are separable in the form:

$$\begin{cases} w, u_\phi, \beta_\phi \\ Q_\phi, N_\phi, M_\phi, N_\theta, M_\theta \end{cases} = \begin{cases} w_n, u_{\phi n}, \beta_{\phi n} \\ Q_{\phi n}, N_{\phi n}, M_{\phi n}, N_{\theta n}, M_{\theta n} \end{cases} \cos n \theta$$

$$\begin{cases} u_\theta, \beta_\theta, Q_\theta, M_{\theta\phi} \\ M_{\phi\theta}, N_{\theta\phi}, N_{\phi\theta} \end{cases} = \begin{cases} u_{\theta n}, \beta_{\theta n}, Q_{\theta n}, M_{\theta\phi n} \\ M_{\phi\theta n}, N_{\theta\phi n}, N_{\phi\theta n} \end{cases} \sin n \theta$$

To set up displacement derivatives it is necessary to use Equation (5). Considering Equations (5a) and (5c), we obtain:

$$\epsilon_{\phi n} = \frac{1}{[C_{11}D_{11} - K_{11}^2]} \left\{ \begin{aligned} &D_{11}(N_{\phi n} - C_{12}\epsilon_{\theta n} - K_{12}^k\theta_n) \\ &- K_{11}(M_{\phi n} - K_{12}\epsilon_{\theta n} - D_{12}^k\theta_n) \end{aligned} \right\} \quad (18a)$$

$$\begin{aligned} \kappa_{\phi n} = \frac{-1}{[C_{11}D_{11} - K_{11}^2]} & K_{11}(N_{\phi n} - C_{12}\epsilon_{\theta n} - K_{12}\kappa_{\theta n}) & (18b) \\ & - C_{11}(M_{\phi n} - K_{12}\epsilon_{\theta n} - D_{12}\kappa_{\theta n}) \end{aligned}$$

Combining Equations (5e) and (5b) we obtain:

$$\begin{aligned} \gamma_{\phi n} = \frac{1}{[E_{11}H_{11} - F_{11}^2]} & H_{11}(N_{\phi n} - E_{12}\gamma_{\theta n} - F_{12}\delta_{\theta n}) & (19a) \\ & - F_{11}(M_{\phi n} - F_{12}\gamma_{\theta n} - H_{12}\delta_{\theta n}) \end{aligned}$$

$$\begin{aligned} \delta_{\phi n} = \frac{-1}{[E_{11}H_{11} - F_{11}^2]} & F_{11}(N_{\phi n} - E_{12}\gamma_{\theta n} - F_{12}\delta_{\theta n}) & (19b) \\ & - E_{11}(M_{\phi n} - F_{12}\gamma_{\theta n} - H_{12}\delta_{\theta n}) \end{aligned}$$

From Equations (3c), (3n), and (5c) we get:

$$w_{n,s} = \frac{u_{\phi n}}{R_{\phi}} - \beta_{\phi n} + \gamma_{\phi zn} \quad (20)$$

or using Equation (5i)

$$w_{n,s} = \frac{u_{\phi n}}{R_{\phi}} - \beta_{\phi n} + \frac{Q_{\phi n}}{H_{55}} \quad (21a)$$

From Equation (3) we obtain:

$$u_{\phi n,s} = \epsilon_{\phi n} - \frac{1}{R_{\phi}} w_n \quad (21b)$$

$$u_{\theta n,s} = \gamma_{\phi n} \quad (21c)$$

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$$\beta_{\phi n, s} = \kappa_{\phi n} \quad (21d)$$

$$\beta_{\theta n, s} = \delta_{\phi n} \quad (21e)$$

Noting that these quantities are already functions of the fundamental variables:

$$\epsilon_{\theta n} = \frac{n}{r} u_{\theta n} + \frac{\cos \phi}{r} u_{\phi n} + \frac{\sin \phi}{r} w_n \quad (22)$$

$$\kappa_{\theta n} = \frac{n}{r} \beta_{\theta n} + \frac{\cos \phi}{r} \beta_{\phi n} \quad (23)$$

$$\gamma_{\theta n} = -\frac{n}{r} u_{\phi n} - \frac{\cos \phi}{r} u_{\theta n} \quad (24)$$

$$\delta_{\theta n} = -\frac{n}{r} \beta_{\phi n} - \frac{\cos \phi}{r} \beta_{\theta n} \quad (25)$$

$$\gamma_{\theta zn} = -\left(\frac{n}{r} w_n - \frac{\sin \phi}{r} u_{\theta n}\right) \beta_{\theta n} \quad (26)$$

the following functions, therefore, can be solved using Equation (5):

$$N_{\theta n} = C_{12} \epsilon_{\phi n} + C_{22} \epsilon_{\theta n} + K_{12} \kappa_{\phi n} + K_{22} \kappa_{\theta n}$$

$$M_{\theta n} = K_{12} \epsilon_{\phi n} + K_{22} \epsilon_{\theta n} + D_{12} \kappa_{\phi n} + D_{22} \kappa_{\theta n}$$

$$N_{\theta \phi n} + E_{12} \gamma_{\phi n} + E_{22} \gamma_{\theta n} + F_{12} \delta_{\phi n} + F_{22} \delta_{\theta n}$$

$$M_{\theta \phi n} = F_{12} \gamma_{\phi n} + F_{22} \gamma_{\theta n} + H_{12} \delta_{\phi n} + H_{22} \delta_{\theta n}$$

$$Q_{\theta n} = H_{66} \gamma_{\theta zn}$$

(27)

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AFFDL-TR-71-116

Finally from the equations of equilibrium,

$$\begin{aligned}Q_{\phi n,s} &= -\frac{\cos \phi}{r} Q_{\phi n} - \frac{n}{r} Q_{\theta n} + \frac{N_{\phi n}}{R_{\phi}} + \frac{\sin \phi}{r} N_{\theta n} - P_n \\N_{\phi n,s} &= -\frac{\cos \phi}{r} N_{\phi n} - \frac{n}{r} N_{\theta \phi n} + \frac{\cos \phi}{r} N_{\theta n} - \frac{Q_{\phi n}}{R_{\phi}} - P_{\phi n} \\N_{\phi \theta n,s} &= -\frac{\cos \phi}{r} (N_{\phi \theta n} + N_{\theta \phi n}) - \frac{n}{r} N_{\theta n} - \frac{\sin \phi}{r} Q_{\theta n} - P_{\theta n} \quad (28) \\M_{\phi n,s} &= -\frac{\cos \phi}{r} (M_{\phi n} - M_{\theta n}) - \frac{n}{r} M_{\theta \phi n} + Q_{\phi n} - M_{\phi n} \\M_{\phi \theta n,s} &= -\frac{\cos \phi}{r} (M_{\phi \theta n} + M_{\theta \phi n}) - \frac{n}{r} M_{\theta n} + Q_{\theta n} - M_{\theta n}\end{aligned}$$

By calculating Equations(18)through (28) we develop the set of first order differential equations. Therefore, these equations can be solved numerically using a direct integration over a segment length where sufficient accuracy is maintained and the numerical scheme discussed earlier.

3. PROGRAM MODIFICATION

The computer program SP presented in Reference 1 had been written for the IBM 7094. Modifications to this program to accept the effects of transverse shear deformation have been forwarded to FBR. This program has modification identifiers located in column 80. This particular listing for the IBM 7094 is on file at FBR but will not be maintained. A CDC 6600 version under the UPDATE mode will be maintained at this facility and is presented in Appendix A. This version remains within a core limit of 77777_8 . However to remain within the framework of the original program and core limit, the total number of segments had to be reduced (from 100 to 65 segments) for the thick shell option. The thin option can still use 100 segments. Other versions have been submitted for the IBM 7094 utilizing overlays that permit 150 segments but again will not be maintained.

Program size and governing common block allocations will be deferred from discussion until Part 5. In order to maintain the essential input parameters and organization of the original code, the description of those modified parameter cards are presented. Since the thick shell equations are based upon an order of 10, modification to the original code is required. For axisymmetrically loaded shells the order is reduced to six. Thus every parameter defined in the original code relating to nonaxisymmetric loads had to be redefined. Further additional subroutines and control cards are employed. All other parameter cards not mentioned herein remain the same.

Contrails

AFFDL-TR-71-116

The following card is added and must precede the original data deck:

<u>Card 1</u>	<u>Analysis</u>	<u>Control</u>	<u>Format (A6,4X,A8)</u>
Col. 1-6	ITYPE	If ITYPE = THICK _Δ Thick Shell equations are to be used. If ITYPE = THIN _{ΔΔ} Thin Shell equations are to be used.	
Col.11-18	JTYPE	If JTYPE = TORSION _Δ Torsion problem to be solved. ITYPE must equal THIN _{ΔΔ} Normally JTYPE = blank	

Elastic Parameter Card

Application - SP, AEP, NEP

Number of consecutive cards = as many as the number of layers (MLY), starting with Layer No. 1. One such set per Part.

<u>Variables</u>	<u>Format</u>	<u>Column</u>	<u>Description</u>
If ITYPE = THIN _{ΔΔ}			
B11	F10.0	1-10	Isotropic - Young's modulus Orthotropic - $E_{\phi} / (1 - \nu_{\phi} \nu_{\theta})$
B12	F10.0	11-20	Isotropic - Poisson's ratio Orthotropic - $\nu_{\phi} E_{\phi} / (1 - \nu_{\phi} \nu_{\theta})$
B22	F10.0	21-30	Isotropic - Blank Orthotropic - $E_{\theta} / (1 - \nu_{\phi} \nu_{\theta})$
B66	F10.0	31-40	Isotropic - Blank Orthotropic - $G_{\phi\theta}$
AL1	F10.0	41-50	Coefficient of thermal expansion in meridional direction. Not used in AEP and NEP

<u>Variables</u>	<u>Format</u>	<u>Column</u>	<u>Description</u>
AL2	F10.0	51-60	Coefficient of thermal expansion in circumferential direction. Not used in AEP and NEP.
RHO	F10.0	61-70	Mass density of material. For SP used only in spinning shell is analyzed. Not used for stability analysis.
IL1	15	71-75	If IL1=0, then all elastic parameters are constant in this layer. For variable elastic parameters, see Comments.
If ITYPE = THICK _Δ			
B11	F10.0	1-10	Isotropic - Young's modulus Orthotropic - $E_{\phi}/(1-\nu_{\phi}\nu_{\theta})$
B12	F10.0	11-20	Isotropic - Poisson's ratio Orthotropic - $\nu_{\phi}E_{\phi}/(1-\nu_{\phi}\nu_{\theta})$
B22	F10.0	21-30	Isotropic - Blank Orthotropic - $E_{\theta}/(1-\nu_{\phi}\nu_{\theta})$
B66	F10.0	31-40	Isotropic - Blank Orthotropic - $G_{\phi\theta}$
AL1	F10.0	41-50	Coefficient of thermal expansion, in meridional direction. Not used in AEP and NEP.
AL2	F10.0	51-60	Coefficient of thermal expansion, in circumferential direction. Not used in AEP and NEP.

<u>Variables</u>	<u>Format</u>	<u>Column</u>	<u>Description</u>
RHO	F10.0	61-70	Mass density of material. For SP used only if spinning shell is analyzed. Not used for stability analysis.
IL1	15	71-75	If IL1=0, then all elastic parameters are constant in this layer. For variable elastic parameters, see Comments.

If ITYPE = THICK_Δ and B22≠0

B55	F10.0	1-10	Orthotropic $G_{\phi z}$
B44	F10.0	11-20	Orthotropic $G_{\theta z}$

Comments

- An isotropic layer is identified by setting B22=0.0.
- E_{ϕ} = Young's modulus in meridional direction.
 E_{θ} = Young's modulus in circumferential direction.
 ν_{ϕ} = Poisson's ratio (contraction) in ϕ direction produced by normal stress in θ direction.
 ν_{θ} = Poisson's ratio (contraction) in ϕ direction produced by normal stress in θ direction.
 $G_{\phi\theta}$ = Shear modulus in ϕ, θ plane.
 $G_{\phi z}$ = Shear modulus in ϕ, z plane.
 $G_{\theta z}$ = Shear modulus in θ, z plane.
 [Note that $\nu_{\phi} E_{\phi} = \nu_{\theta} E_{\theta}$]
- If IL1=+N and ITYPE=THIN_{ΔΔ} then N designates the code number of the first of six FGEN card sets, by which variables B11, B12, B22, B66, AL1, AL2 are read, in that order, immediately after the Elastic Parameter Card of an orthotropic layer.

4. If $IL1=-N$, then N designates the code number of one FGEN set, by which Young's modulus is read immediately after the Elastic Parameter Card of an isotropic layer.

Boundary Condition Card

If $ITYPE = THICK_{\Delta}$ and NX on the wave number card $\neq 0$ additional boundary conditions need to be considered. Two additional parameters need to be defined. The complete set would be:

- | | | |
|---------------------------|--------------------------|--------------------------|
| 1 for u_1 ; | 2 for Q_1 ; | 3 for u_2 ; |
| 4 for Q_2 ; | 5 for β_{ϕ} ; | 6 for M_{ϕ} ; |
| 7 for u_{θ} ; | 8 for $N_{\phi\theta}$; | 9 for β_{θ} ; |
| 10 for $M_{\phi\theta}$. | | |

Starting Edge

Application - SP, AEP, NEP

Number of consecutive cards = one per subcase.

<u>Variables</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
1A(1)	I5	1-5	Code number of first prescribed variable on Starting Edge.
GA(1)	F10.0	6-15	Its prescribed value.
IA(2)	I5	16-20	Second prescribed variable.
GA(2)	F10.0	21-30	Its prescribed value.
IA(3)	I5	31-35	Third prescribed variable.
GA(3)	F10.0	36-45	Its prescribed value.
IA(4)	I5	46-50	Fourth prescribed variable. If $NX \neq 0$
GA(4)	F10.0	51-60	Its prescribed value.
IA(5)	I5	61-65	Fifth prescribed variable. If $NX \neq 0$ and $ITYPE=THICK_{\Delta}$
GA(5)	F10.0	66-75	Its prescribed value.

Contrails

AFFDL-TR-71-116

Comments:

1. For code numbers of prescribed variables see Item 6, Section 2, Chapter I, of Part II⁽¹⁾.
2. Remember that the prescribed edge loads are actually the Fourier coefficients for a given wave number NX.

Final or Branch Edges

Application - SP, AEP, NEP

Number of consecutive cards = as many as Branches plus one. One set per subcase.

<u>Variables</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
IB(1,K)	I5	1-5	Code number of first prescribed variable on Final or Branch Edge.
GB(1,K)	F10.0	6-15	Its prescribed value.
IB(2,K)	I5	16-20	Second prescribed variable.
GB(2,K)	F10.0	21-30	Its prescribed value.
IB(3,K)	I5	31-35	Third prescribed variable.
GB(3,K)	F10.0	36-45	Its prescribed value.
IB(4,K)	I5	46-50	Fourth prescribed variable. If NX#0
GB(4,K)	F10.0	51-60	Its prescribed value.
IB(5,K)	I5	61-65	Fifth prescribed variable. If NX#0 and ITYPE=THICK _Δ
GB(5,K)	F10.0	66-75	Its prescribed value.

Comments

1. The first card of this set refers to the Final Edge of Main Shell. The second, third, etc., cards refer to Branch Edges in the order in which they are encountered, when going from the Starting to Final Edge of Main Shell.
2. For code numbers of prescribed variables, see Item 6, Section 2, Chapter I, of Part II. (Reference 1)

3. Remember that the prescribed edge loads are actually the Fourier coefficients for a given wave number NX.

Load Parameter Card

Application - SP only

Number of consecutive cards = one per Part.

<u>Variables</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
VK(I,1)	F10.0	1-10	If INORM=0, then surface load along normal, p . If INORM=1, then surface load parallel to axis of symmetry, p_1 .
VK(I,2)	F10.0	11-20	If INORM=0, then surface load along meridian, p_ϕ . If INORM=1, then surface load perpendicular to axis of symmetry, p_2 .
VK(I,3)	F10.0	21-30	Surface load along circumference, p_θ .
VK(I,4)	F10.0	31-40	Temperature on inside bounding surface of shell (in -Z direction), T_L .
VK(I,5)	F10.0	41-50	Temperature on outside bounding surface of shell (in +Z direction), T_U .
IK2	I5	51-55	If IK2=0, then all loads are constant along meridian. Otherwise, IL2=number of variable loads.
INORM	I3	56-58	INORM=0 means that surface loads are along the normal and meridional tangent of Reference Surface. INORM=1 means that they are parallel and perpendicular to axis of symmetry.

Contrails

AFFDL-TR-71-116

<u>Variables</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
IPRESS	I1	59	1 Pressure on outside specified else 0
	I1	60	1 Pressure on inside specified else 0
RPM	F10.0	61-70	Revolutions per minute for a shell spinning about its axis of symmetry.

Comments

1. For positive directions of rotated loads, P_1 and P_2 , see Figure 9. (Reference 1)
2. The values of variable loads, entered on Load Parameter Card, are used for reference only.
3. Remember that the loads entered here are actually Fourier coefficients for a given wave number NX.
4. For a spinning shell, the same RPM must be entered on Load Parameter Card for every Part.
5. IPRESS is normally set equal to zero for the pressure is stated at the reference surface. When IPRESS=10 the pressure is assumed to be along the outside surface and then calculated within the program as:

$$p = p_o \left[1 - \frac{z_{last} - z_{ref}}{R_o} \right]$$

When IPRESS=01 the pressure is assumed to be acting along the inside surface.

$$p = p_i \left[1 + \frac{z_{ref} - z_1}{R_o} \right]$$

Note: If above option is used, z_{ref} cannot be at z_{last} or z_1 .

4. DEVELOPMENT OF TORSION

Under the assumptions in the development of the thin shell equations, the Reissner equations uncouple for the Fourier harmonic $n=0$. They take the form:

$$\frac{d}{ds} \begin{Bmatrix} y_0 \\ y_n \end{Bmatrix} = \begin{bmatrix} A(s) & 0 \\ 0 & A(s) \end{bmatrix} \begin{Bmatrix} y_0 \\ y_n \end{Bmatrix} + \begin{Bmatrix} B_0 \\ B_n \end{Bmatrix}$$

If the usual initial boundary value problems are determined in the form:

$$\begin{Bmatrix} y_1 \\ y_2 \end{Bmatrix}_{i+1} = \begin{bmatrix} Y_1 & Y_2 \\ Y_3 & Y_4 \end{bmatrix} \begin{Bmatrix} y_1 \\ y_2 \end{Bmatrix}_i$$

two rows and columns of matrix Y_2 are singular for the $n=0$ torsion problem. Re-arrangement of the rows and columns of Y_4 and Y_2 will permit successful solution. If $JTYPE=TORSION_{\Delta}$ on the first control card and $ITYPE=THIN_{\Delta\Delta}$, the singular matrices are converted to nonsingular matrices.

5. PROGRAM SIZE

The present version compiled on RUN compiler of the CDC 6600 system has a program length of 65100₈ or 27100₁₀. The size of the program is based upon the number of parts, segments, materials, layers, branches etc. Common block size and allocation are described below. Their names in general follow the association or area to which each common block pertains.

<u>Common Block Name</u>	<u>Size</u>	<u>Principal Size Governing Equation</u>
GARB	8109	$[NSEG(NDE+1)+1] [NDE+1]$
RUN	64	--
MATL	720	$PARTS * NL * 9$
XM17	54	--
GEOM	1235	$(NL+10)PARTS + NBR+1$ $+ (NBR+ 2) NDE * NDE$
THICK	15	--
TRIA	130	$NDE/2 * (NBR+1)+NSEG+5$
ADAM	3	--

For the present case, the maximum values are:

NSEG = 100
NDE = 10
PARTS = 20
NL = 4
NBR = 4

Note for NDE = 10 (THICK option) NSEG \leq 65, for
 $[65(10+1)+1] [10+1] = [65 \times 11 + 1] [11] = 7876$
which is less than the limit 8109. If one wanted to
use a single variable Blank Common, the above calcula-
tion for each common length could be used to RFL the
length of the program to meet the problem being
considered.

SECTION III THE EIGENSOLUTION

The present dynamic shell program uses a "cut and try" method for an eigensolution. For a system of homogeneous equations that govern the free vibration of shells of revolution, nontrivial solutions exist when the determinant of coefficients vanish. Blum and Fulton⁽¹³⁾ have shown that the problems associated with the determinant using Potter's method (which is the essential basis for the multisegment method) are:

- 1) "changes in sign of the residual associated with the poles may be incorrectly interpreted as crossings (zeros) of the residual.

- 2) if the poles are near the zeros, two successive choices of λ (eigenvalue) ..no change in sign, and a zero may be missed in the search." The same procedure can be followed and stated with the present program's parameters. If the parameter DETB is plotted as a function of Omega, ω , the zero of the parameter DETB will be λ , the eigenvalue. The present program utilizes an interpolation procedure which is adequate when $\omega \approx \lambda$. If $\omega \gg \lambda$, one would have to review the modal pattern in order to establish which way should be applied. For some shell structures, this modal pattern behavior may not be as well behaved as one would like, especially for highly orthotropic junctured shells.

For any search method utilizing the determinant of coefficient the procedure that should be followed is:

- (1) examine the determinant relative to determinants at adjacent frequencies for an indicated zero by change in sign.

2) examine the modified determinant for change in sign.

3) examine mode shape.

When all three items are satisfied, the assumed frequencies can be utilized for an interpolation of the true eigenvalue. The pitfall in employing this procedure occurs when one or more of the items indicate a zero (and in particular the mode shape). If the step size ($\Delta\omega$) is sufficiently small, one may circumvent one of the pitfalls and the present interpolation procedure may be utilized with some partial success but at the expense of a great amount of computer time. This procedure can be followed provided the eigenvalue is known a priori.

To illustrate the possible difficulties with and to further amplify on Blum and Fulton's observations one could experience utilizing the above technique, we present results of some problems that were recently solved. In Table I a list is presented of frequencies, determinants, modified determinants and nodal (number of zero crossings) points for a cantilevered conical shell having variable extensional and bending rigidities.

From Table I the first indicated zero lies between $\Omega = \omega L/c = .004$ and $.005$. The next indicated zero (by mode shape) lies between $\Omega = .006$ and $.008$. This turns out to be a false indicated root. In fact a total of five false zeros were encountered for the entire frequency range investigated. In order to verify the existence or nonexistence of each indicated eigenvalue, three to five additional runs had to be performed.

TABLE I
RESULTS FOR CANTILEVERED CONICAL SHELL

Ω	$Z = \frac{C_m}{Y_{c+1}}$	C_M	Nodes
.002	-0.120	.34 x 10 ⁺⁹	0
.003	-0.624	.39 x 10 ⁺⁸	0
.004	-0.028	.62 x 10 ⁺⁶	0
.005	0.567	-.58 x 10 ⁺⁷	1
.006	1.350	-.53 x 10 ⁺⁷	1
.008	1.370	-.42 x 10 ⁺⁷	3
.010	0.517	-.16 x 10 ⁺⁷	1
.012	-1.210	-.10 x 10 ⁺⁸	2
.014	-2.470	-.30 x 10 ⁺⁷	4
.016	-2.200	-.17 x 10 ⁺⁷	2
.018	-0.400	-.70 x 10 ⁺⁶	2
.020	-0.200	.29 x 10 ⁺⁶	4
.022	4.200	-.22 x 10 ⁺⁷	3
.024	-0.190	.93 x 10 ⁺⁸	4
.026	0.256	.13 x 10 ⁺⁷	4
.028	0.251	.61 x 10 ⁺⁶	5
.030	-0.470	-.17 x 10 ⁺⁹	6

Thus a total of sixty-seven computer runs were required to establish the eigenvalues. Approximately one and one half hours of computing time were required to produce these results. Similar results were obtained for other classes of problems that typify the above findings. During the course of our investigations, many numerical procedures have been attempted that show the futility of the present approach for engineering problems when the eigenvalue cannot be adequately estimated. More direct methods are available such as Stodola, Rayleigh quotient, Householder or even Jacobi. These methods require an extensive modification to AEP and would require discarding the bulk of the present program.

1. MODIFICATION TO AEP

An attempt to salvage the program and develop a more automatic procedure within the framework of the program has been made. This modification is primarily concentrated on the subroutine TRIANG and in MAIN immediately after the call to TRIANG. The modification of the program's logic is set so it will follow the original execution path if $\Delta\omega$ (DOMEGA) in the program is read in as a nonzero value. Otherwise, the alternate path will be followed. Figure 3 gives the logical flow for the alternate procedure. The dotted lines signify the same logical path as the original AEP.

The method is simply to specify the range of the eigenvalue search by ω_o and ω_f . $\Delta\omega$ is set by:

$$\Delta\omega = \frac{\omega_f - \omega_o}{2}$$

A solution is obtained at three frequencies ω_o , $\omega_M = \omega_o + \Delta\omega$ and ω_f . If the parameter $C_M(\text{DET})$ is the same sign for all three values, two additional points are obtained at:

$$\omega = \omega_o + \Delta\omega/4$$

and

$$\omega = \omega_o + 3\Delta\omega/4$$

The sign between adjacent values of C_M are again checked. If no change is seen the interval of $\Delta\omega$ is halved and new values are obtained. This procedure is followed for six iterations. If unsuccessful, the procedure is stopped. If successful the usual interpolation formulae are used to obtain λ . If the solution converges the procedure is continued between λ_i and ω_f until all the desired numbers of eigenvalues are found.

Again, it must be noted that the above procedure has the same shortcomings of the original procedure. The method will, however, allow for a more systematic search.

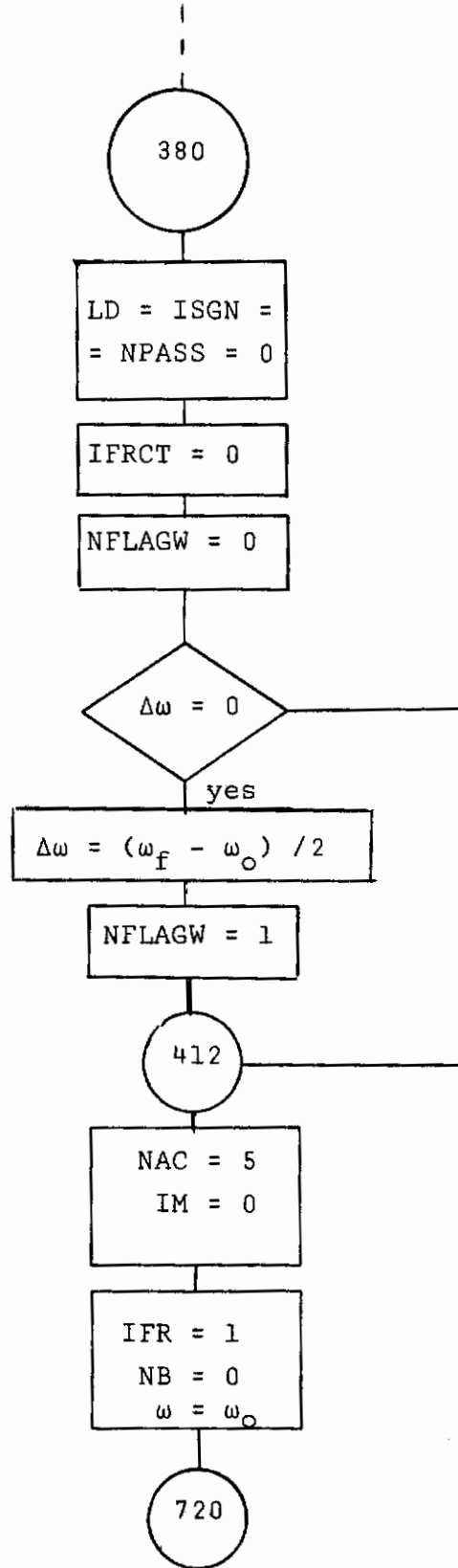


Figure 3 Flow Chart of AEP

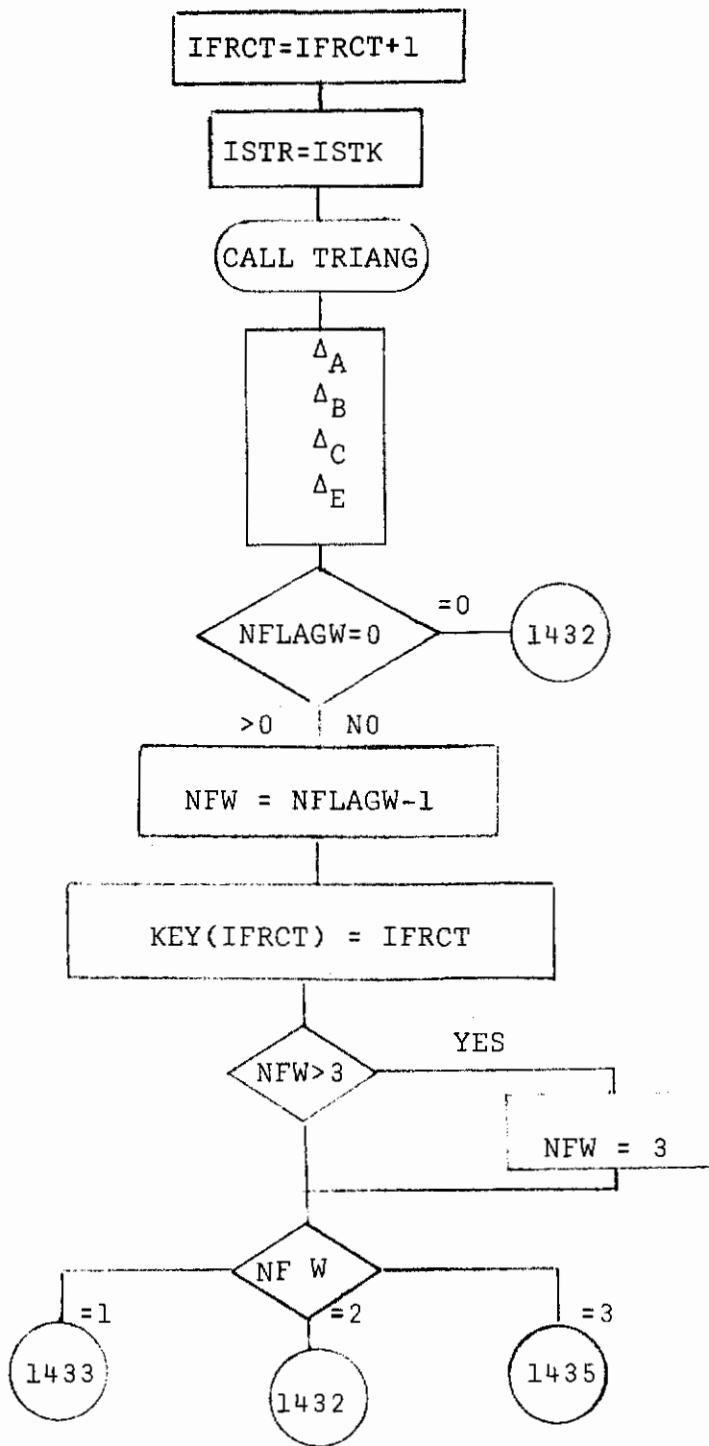


Figure 3 (con't)

Contrails

AFFDL-TR-71-116

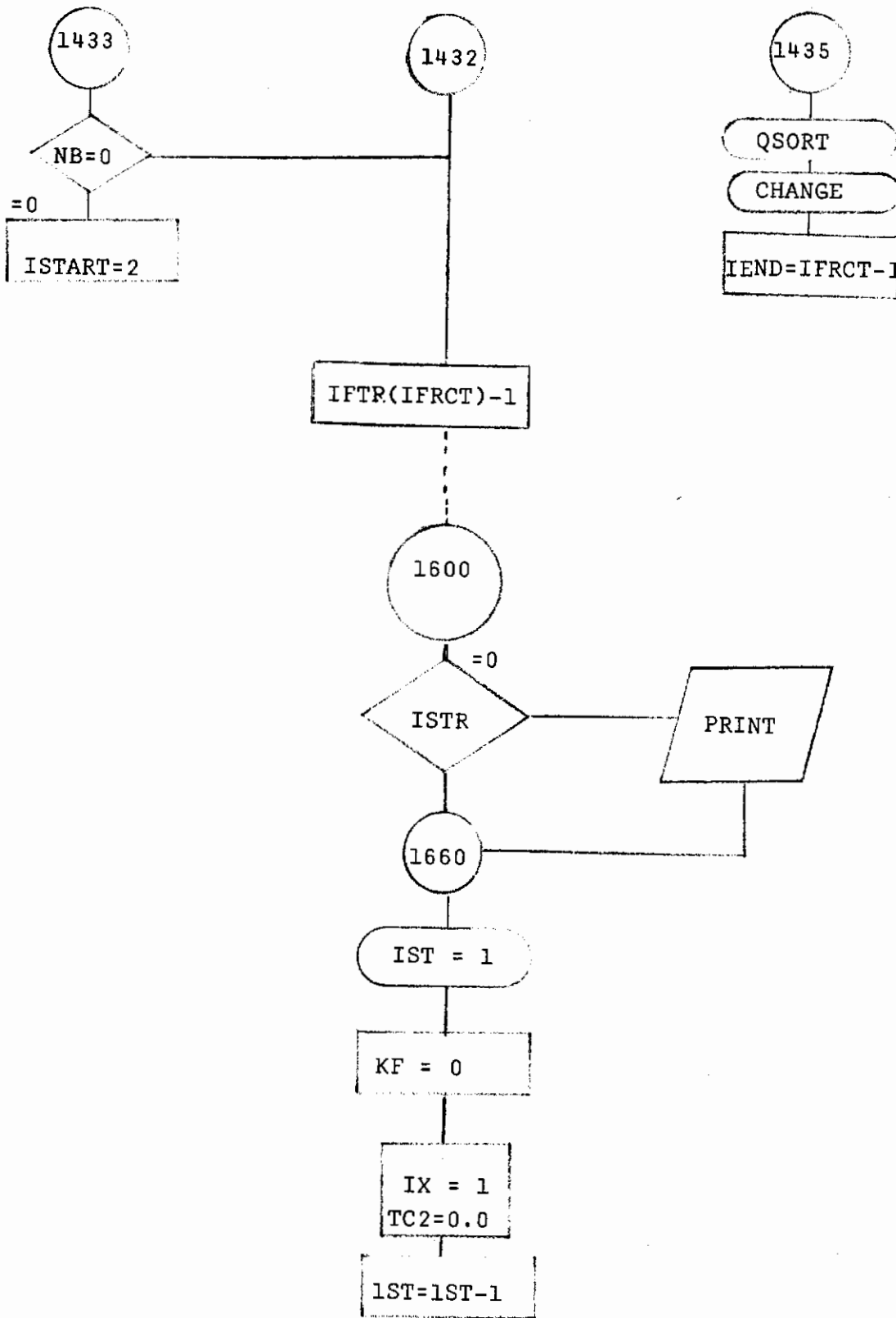


Figure 3 (con't)

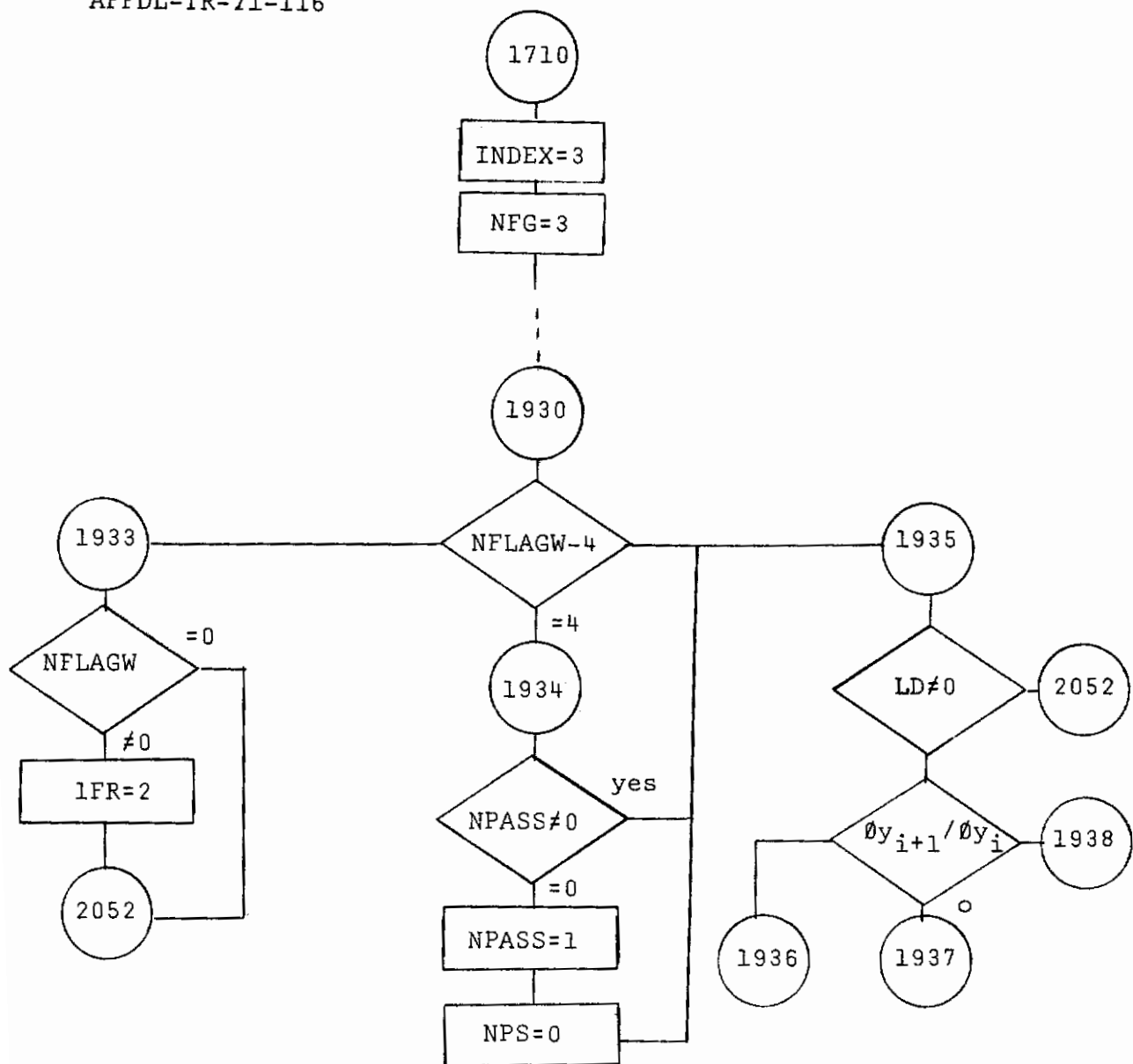


Figure 3 (con't)

Contrails

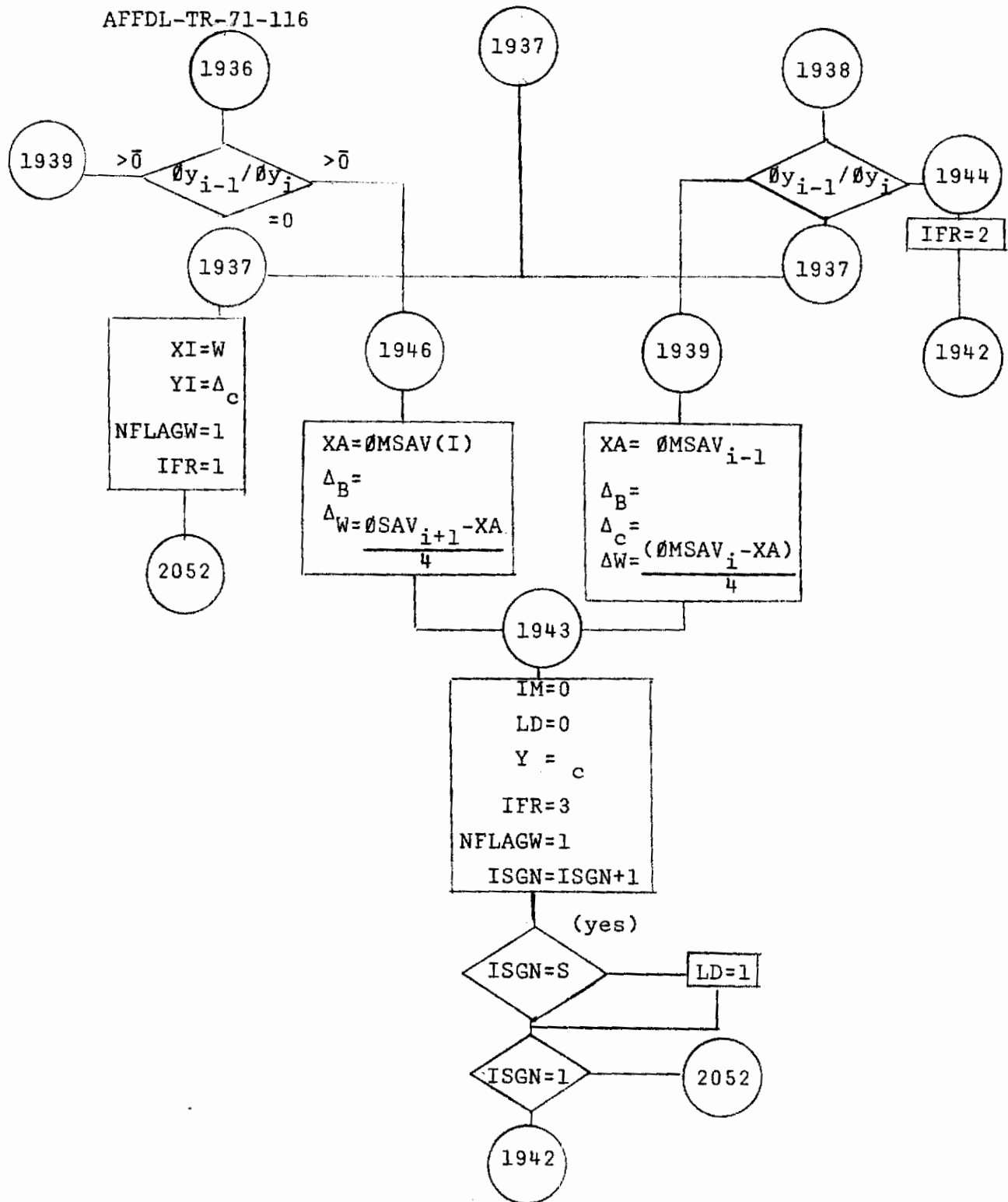


Figure 3 (con't)

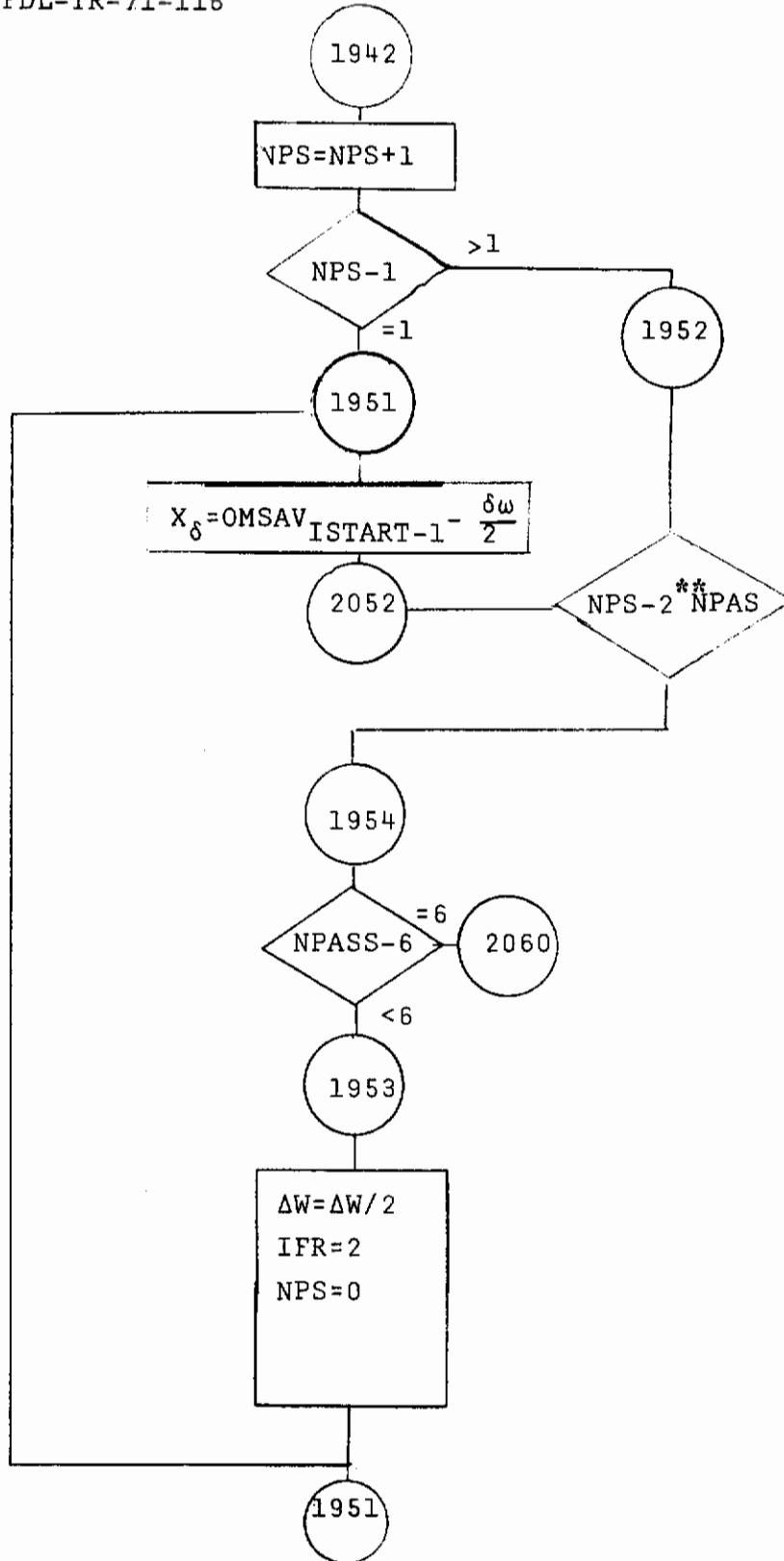


Figure 3 (con't)

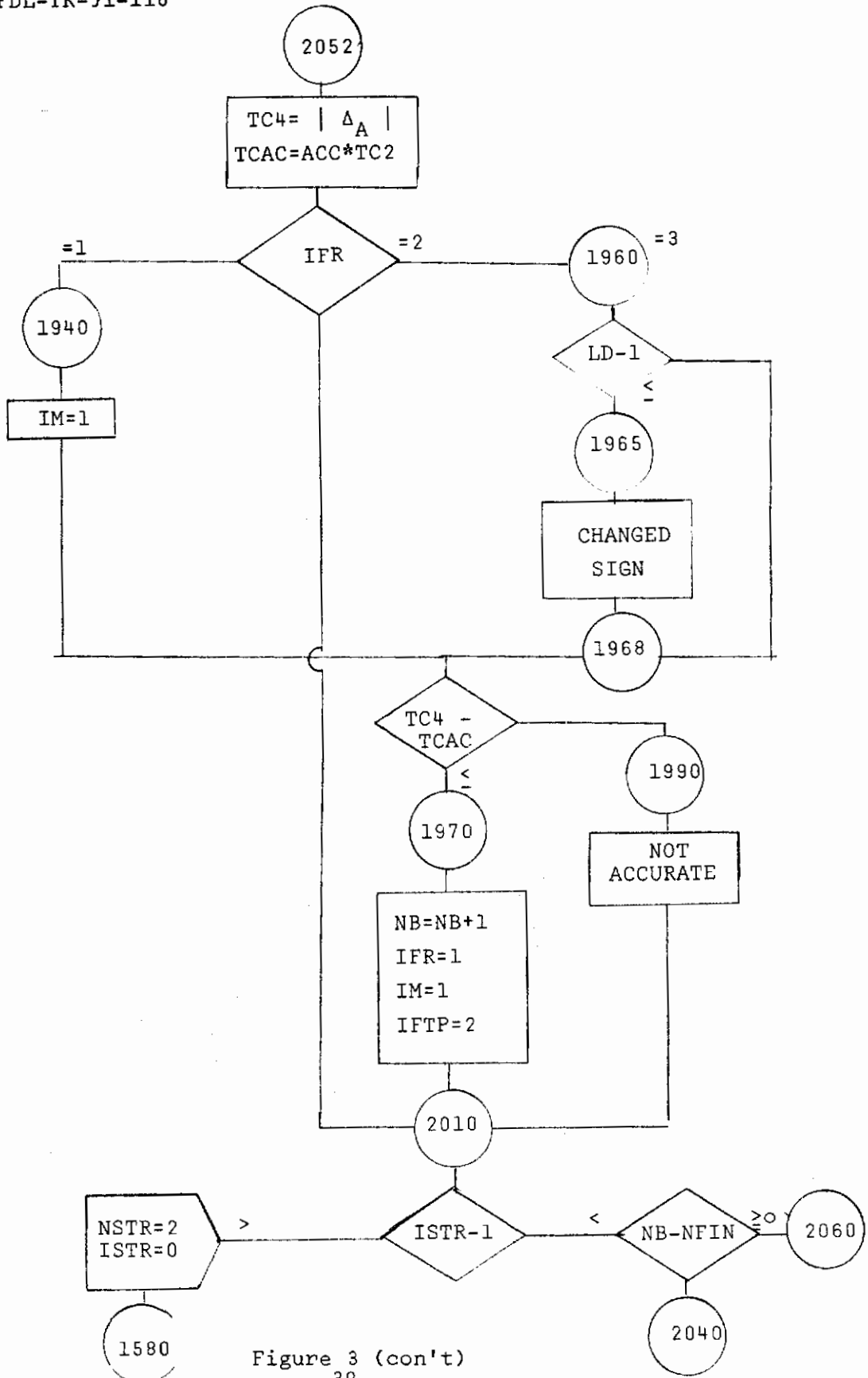


Figure 3 (con't)
39

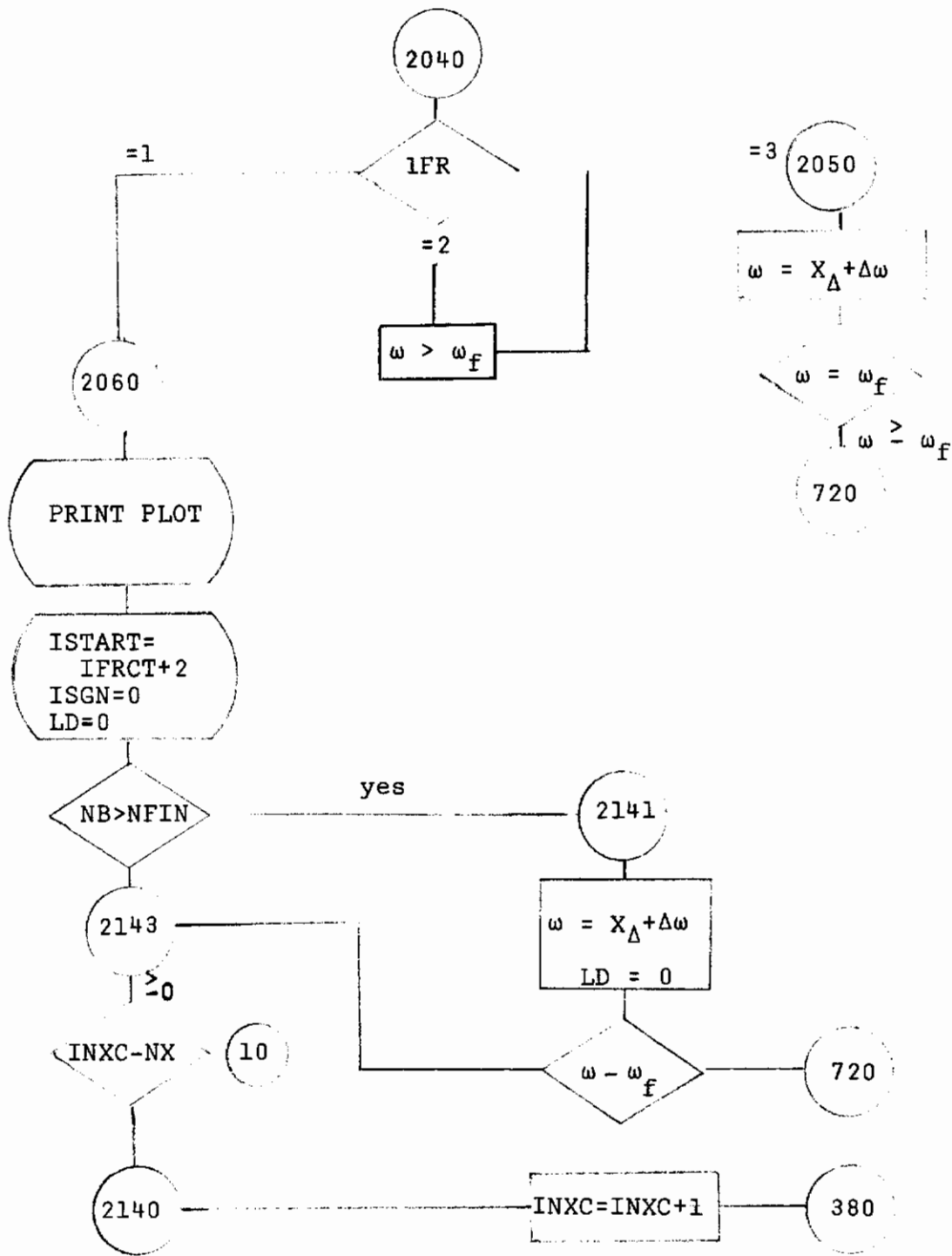


Figure 3 (con't)

SECTION IV
NUMERICAL INTEGRATION PACKAGES

One of the common problems when a researcher is investigating a particular area within his particular sphere of influence is that he must utilize tools that have been developed by others. An independent evaluation is sometimes difficult to obtain. It is not the intent to suggest a better numerical scheme on the basis of running time or solution accuracy alone. Running time, solution accuracy, ease of application and implementation are all necessary factors for evaluation. Acceptance of a method must be on the basis of the general problem solver. For example, if the problem requires a good starting estimate, one may use a Runge-Kutta method. If a considerable amount of halving or doubling is required, Adams' method is attractive.

For in-core solvers, the Adams' predictor-corrector method appears to be one of the more overall efficient techniques. The inherent difficulty with the Adams' method is that it requires a separate starter procedure.

The original shell program developed in conjunction with Dr. Kalnins used a sixth order Adams' method. This FORTRAN II program required parameter definition through Common. A listing of several differential equation solvers is presented.

AFFDL-TR-71-116

1. ADMIP

USAGE

- 1) Set up initial values in YTAB(I) and Y(I,1)
- 2) Set up Starting XO and Finishing S_{max} values for interval of integration. STA array will contain these values.
- 3) YTAB(I) will contain the corrected terms at S_{max} .

Typical Call

```

      DØ I N=1, NSEG
      XO = STA(N)
      Smax = STA(N+1)
      DØ 2 I=1,NDE
      Y(I,1)=0.0
2     YTAB(J)=0.0
      DØ 3 I=1,NDE
      J = NBCON(I)
      Y(J,1) = 1.0
      YTAB(J) = 1.0
      CALL AMIP
      DØ 3 K=1,NDE
3     D(N,I,K)=YTAB(K)
1     Continue
```

Since this routine was written for the FORTRAN II compiler on the IBM 7094 computer, double precision arithmetic is required. Further the starting procedure requires six evaluations prior to testing for halving or doubling the step size. This routine may be used on CDC machine but extensive revisions would be required.

Contrails

```
* AMIP
* LIST8
* LABEL AFFDL-TR-71-116
* DIAGOL
SUBROUTINE AMIP
```

```
CAMIP
```

```
C
C
```

```
COMMON YTAB, ZTAB, Z, ATAB, ITAB
```

```
COMMON YDP, YPD, YPDP, YCD, YCDP, Y
```

```
COMMON WANL, VANL, UANL, BANL
```

```
DIMENSION YTAB(10), ZTAB(10), Z(10,12), ATAB(20), ITAB(10)
```

```
DIMENSION YDP(10), YPD(10), YPDP(10), YCD(10), YCDP(10), Y(10,12)
```

```
EQUIVALENCE (YTAB( 1), WN), (YTAB( 2), VN), (YTAB( 3), UN),
```

```
1 (YTAB( 4), BN), (YTAB( 5), XMN), (YTAB( 6), XNP),
```

```
2 (YTAB( 7), XNN), (YTAB( 8), QN)
```

```
EQUIVALENCE (ZTAB( 1), WDOT), (ZTAB( 2), VDOT), (ZTAB( 3), UDOT),
```

```
1 (ZTAB( 4), BDOT), (ZTAB( 5), XMDOT), (ZTAB( 6), XNPDOT),
```

```
2, (ZTAB( 7), XNNDOT), (ZTAB( 8), QDOT)
```

```
EQUIVALENCE (ATAB( 1), XO), (ATAB( 2), S), (ATAB( 3), H),
```

```
1 (ATAB( 4), EPSIL), (ATAB( 5), DX), (ATAB( 6), FLAG),
```

```
2 (ATAB( 7), DELTA), (ATAB( 8), RTN), (ATAB( 9), SM),
```

```
3 (ATAB(10), HSTAR), (ATAB(11), DBLGT), (ATAB(12), HAVG),
```

```
4 (ATAB(13), GAMMA)
```

```
EQUIVALENCE (ITAB( 1), NDEQ), (ITAB( 2), L), (ITAB( 3), INDEX),
```

```
1 (ITAB( 4), K)
```

```
L = 1
```

```
RTN = 0.0
```

```
K = DX
```

```
S = XO
```

AFFDL-TR-71-116

```
10 CALL DEQNS
   DO 20 I = 1, NDEQ
20 Z(I,L) = ZTAB(I)
   IF(L - 5) 30, 30, 40
30 CALL START(1)
   GOTO 10
40 LSTAR = L
50 L = LSTAR
   S = XO + H
   INDEX = 1
   K = 2
   CALL START(2)
130 FLAG = 0.0
   CALL TEST
   IF (INDEX - 1) 170, 150, 150
150 DO 160 J = 2, 6
   DO 160 I = 1, NDEQ
D 160 Y(I,J) = Y(I, J+5)
   GOTO 50
170 S = XO
   DO 175 I = 1, NDEQ
175 YTAB(I) = Y(I,1)
C   CALL DEQNS
   CALL OUT(1)
   DO 190 J = 2, 6
   S = S + H
   DO 180 I = 1, NDEQ
180 YTAB(I) = Y(I,J)
C   CALL DEQNS
190 CALL OUT(2)
   DO 195 I = 1, NDEQ
D   YDP(I) = 0.0
D 195 YDP(I) = Y(I,6)
200 S = XO + 6.0 * H
215 CALL PRED
   CALL TEST
   IF (HAVGT) 230, 230, 220
220 CALL HAV
   GOTO 215
230 IF (DBLGT) 245, 245, 240
240 CALL DBL
245 IF(RTN) 215, 215, 250
250 RETURN
   END
```

```
*   START
*   LIST8
*   LABEL
*   DIAGOL
SUBROUTINE START(IX)
```

```
CSTART
C
C
```

Contrails

AFFDL-TR-71-116

```
COMMON YTAB, ZTAB, Z, ATAB, ITAB
COMMON YDP, YPD, YPDP, YCD, YCDP, Y
COMMON WANL, VANL, UANL, BANL
DIMENSION YTAB(10), ZTAB(10), Z(10,12), ATAB(20), ITAB(10)
D DIMENSION YDP(10), YPD(10), YPDP(10), YCD(10), YCDP(10), Y(10,12)
EQUIVALENCE (YTAB( 1), WN), (YTAB( 2), VN), (YTAB( 3), UN),
1 (YTAB(4 ), BN), (YTAB( 5), XMN), (YTAB( 6), XNPN),
2 (YTAB( 7), XNN), (YTAB( 8), QN)
EQUIVALENCE (ZTAB( 1), WDOT), (ZTAB( 2), VDOT), (ZTAB( 3), UDOT),
1 (ZTAB( 4), BDOT), (ZTAB( 5), XMDOT), (ZTAB( 6), XNPDOT)
2, (ZTAB( 7), XNNDOT), (ZTAB( 8), QDOT)
EQUIVALENCE (ATAB( 1), XO), (ATAB( 2), S), (ATAB( 3), H),
1 (ATAB( 4), EPSIL), (ATAB( 5), DX), (ATAB( 6), FLAG),
2 (ATAB( 7), DELTA), (ATAB( 8), RTN), (ATAB( 9), SMAX),
3 (ATAB(10), HSTAR), (ATAB(11), DBLGT), (ATAB(12), HAVGT),
4 (ATAB(13), GAMMA)
EQUIVALENCE (ITAB( 1), NDEQ), (ITAB( 2), L), (ITAB( 3), INDEX),
1 (ITAB( 4), K)
GOTO(10,30), IX
10 DO 20 I = 1, NDEQ
D DUM = 0.0
DUM = H * Z(I,L)
D Y(I,L+1) = Y(I,L) + DUM
20 YTAB(I) = Y(I,L+1)
S = S + H
L = L + 1
9950 RETURN
30 INDEX=INDEX
GOTO (40,50,60,70,80), INDEX
DO 41 I = 1, NDEQ
D DUM = 0.0
DUM = H / 1440.0 * (475.0 * Z(I,1) + 1427.0 * Z(I,2)
1 -798.0 * Z(I,3) + 482.0 * Z(I,4) - 173.0 * Z(I,5) +27.0 * Z(I,6))
D Y(I,L+1) = Y(I,1) + DUM
GOTO 100
DO 51 I = 1, NDEQ
D DUM = 0.0
DUM = H/90.0 * (28.0 * Z(I,1) + 129.0 * Z(I,2) + 14.0
1 * Z(I,3) + 14.0 * Z(I,4) - 6.0 * Z(I,5) + Z(I,6))
D 51 Y(I,L+1) = Y(I,1) + DUM
GOTO 100
60 DO 61 I = 1, NDEQ
D DUM = 0.0
DUM = H * 0.01875 * (17.0 * Z(I,1) + 73.0 * Z(I,2) +
1 38.0 * Z(I,3) + 38.0 * Z(I,4) - 7.0 * Z(I,5) + Z(I,6))
D 61 Y(I,L+1) = Y(I,1) + DUM
GOTO 100
70 DO 71 I = 1, NDEQ
D DUM = 0.0
DUM = 2.0 * H / 45.0 * (7.0 * Z(I,1) + 32.0 * Z(I,2) +
1 12.0 * Z(I,3) + 32.0 * Z(I,4) + 7.0 * Z(I,5))
D 71 Y(I,L+1) = Y(I,1) + DUM
GOTO 100
80 DO 81 I = 1, NDEQ
D DUM = 0.0
```

AFFDL-TR-71-116

```

      DUM      =      5.0 * H / 288.0 * (19.0 * Z(I,1) + 75.0 * Z(I,2)
      1 + 50.0 * Z(I,3) + 50.0 * Z(I,4) + 75.0 * Z(I,5) + 19.0 * Z(I,6))
D  81 Y(I,L+1) = Y(I,1) + DUM
    100 DO 101 I = 1, NDEQ
    101 YTAB(I) = Y(I,L+1)
      CALL DEQNS
      DO 110 I = 1, NDEQ
    110 Z(I,K) = ZTAB(I)
      L = L + 1
      K = K + 1
      INDEX = INDEX + 1
      S = S + H
      IF(K-6) 30,30, 9999
      END
  
```

```

*   PRED
*   LIST8
*   LABEL
*   DIAGOL
SUBROUTINE PRED
  
```

```

CPRED
C
C
  
```

```

COMMON YTAB, ZTAB, Z, ATAB, ITAB
COMMON YDP, YPD, YPDP, YCD, YCDP, Y
COMMON WANL, VANL, UANL, BANL
DIMENSION YTAB(10), ZTAB(10), Z(10,12), ATAB(20), ITAB(10)
D  DIMENSION YDP(10), YPD(10), YPDP(10), YCD(10), YCDP(10), Y(10,12)
EQUIVALENCE (YTAB( 1), WN), (YTAB( 2), VN), (YTAB( 3), UN),
1 (YTAB( 4), BN), (YTAB( 5), XMN), (YTAB( 6), XNPN),
2 (YTAB( 7), XNN), (YTAB( 8), QN)
EQUIVALENCE (ZTAB( 1), WDOT), (ZTAB( 2), VDOT), (ZTAB( 3), UDOT),
1 (ZTAB( 4), BDOT), (ZTAB( 5), XMDOT), (ZTAB( 6), XNPDOT)
2, (ZTAB( 7), XNNDOT), (ZTAB( 8), QDOT)
EQUIVALENCE (ATAB( 1), XO), (ATAB( 2), S), (ATAB( 3), H),
1 (ATAB( 4), EPSIL), (ATAB( 5), DX), (ATAB( 6), FLAG),
2 (ATAB( 7), DELTA), (ATAB( 8), RTN), (ATAB( 9), SMAX),
3 (ATAB(10), HSTAR), (ATAB(11), DBLGT), (ATAB(12), HAVGT),
4 (ATAB(13), GAMMA)
EQUIVALENCE (ITAB( 1), NDEQ), (ITAB( 2), L), (ITAB( 3), INDEX),
1 (ITAB( 4), K)
DO 10 I = 1, NDEQ
  DUM      = H / 1440.0 * (4277.0 * Z(I,K-1) - 7923.0 * Z(I,K-2) +
1 9982.0 * Z(I,K-3) - 7298.0 * Z(I,K-4) + 2877.0 * Z(I,K-5) -
2 475.0 * Z(I,K-6))
D  YPD(I)   = 0.0
  YPD(I)   = DUM
D  YPDP(I)  = 0.0
D  YPDP(I)  = YDP(I) + YPD(I)
10 YTAB(I) = YPDP(I)
  CALL DEQNS
  DO 20 I = 1, NDEQ
    Z(I,K) = ZTAB(I)
  
```

AFFDL-TR-71-116

DUM = H / 1440.0 * (475.0 * Z(I,K) + 1427.0 * Z(I,K-1) -
 1 798.0 * Z(I,K-2) + 482.0 * Z(I,K-3) - 173.0 * Z(I,K-4) + 27.0 *
 2 Z(I,K-5))

D YCD(I) = 0.0
 YCD(I) = DUM
 D YCDP(I) = 0.0
 D YCDP(I) = YDP(I) + YCD(I)
 20 YTAB(I) = YCDP(I)
 CALL DEQNS
 FLAG = 1
 RETURN
 END

* DEQNS
 * LIST8
 * LABEL
 * DIAGOL
 SUBROUTINE DEQNS

10

CDEQNS

COMMON YTAB, ZTAB, Z, ATAB, ITAB
 COMMON YDP, YPD, YDPD, YCD, YCDP, Y
 COMMON IND, NHALT

D DIMENSION YTAB(10), ZTAB(10), Z(10,12), ATAB(20), ITAB(10)
 DIMENSION YDP(10), YPD(10), YDPD(10), YCD(10), YCDP(10), Y(10,12)
 DIMENSION A(64)

EQUIVALENCE (YTAB(1), Q1), (YTAB(2), Q2), (YTAB(3), Q3),
 1 (YTAB(4), Q4), (YTAB(5), Q5), (YTAB(6), Q6),
 2 (YTAB(7), Q7), (YTAB(8), Q8)

EQUIVALENCE (ZTAB(1), P1), (ZTAB(2), P2), (ZTAB(3), P3),
 1 (ZTAB(4), P4), (ZTAB(5), P5), (ZTAB(6), P6),
 2 (ZTAB(7), P7), (ZTAB(8), P8)

EQUIVALENCE (ATAB(1), X0), (ATAB(2), S), (ATAB(3), H),
 1 (ATAB(4), EPSIL), (ATAB(5), DX), (ATAB(6), FLAG),
 2 (ATAB(7), DELTA), (ATAB(8), RTN), (ATAB(9), SMAX),
 3 (ATAB(10), HSTAR), (ATAB(11), DBLGT), (ATAB(12), HAVGT),
 4 (ATAB(13), GAMMA), (ATAB(18), XN)

EQUIVALENCE (ITAB(1), NDEQ), (ITAB(2), L), (ITAB(3), INDEX),
 1 (ITAB(4), K)

EQUIVALENCE (ITAB(6), JUMP)
 IF(IND) 10,10,20

10 READ INPUT TAPE 5, 1, NFGN, E, XMU, C, ZZ

1 FORMAT (I10, 6F10.0)

66

WRITE OUTPUT TAPE 6, 3, E, XMU, XN, C, ZZ

3 FORMAT (1H0/8X, 1HE, 16X, 3HXMU, 11X, 2HXN, 13X, 1HC, 14X, 2HZZ /1PE14.4,

1 OPF16.4, F11.0, F15.1, F15.1 //)

DO 15 I = 1,8

70

5 BUD = FGEN(I,1,S)

80

N = NFGN - 8

90

DO 16 I = 1,N

100

J = I + N + 7

16 BUD = FGEN(J,1,S)

WRITE OUTPUT TAPE 6, 4

4 FORMAT (1H1/)

Contrails

AFFDL-TR-71-116

```
100 IF(ABSF(H)-HMIN)120,120,101
101 DO 103 I=1,NDEQ
D   DIF(I) = 0.0
D   DIF(I) = ABSF(YPD(I) - YCD(I))
103 EMAX(I) = MAX1F(ABSF(YCDP(I)), GAMMA)
104 DO 105 I = 1, NDEQ
C   TEMAX(I)= 0.25 * ABSF(YCDP(I)) + 0.75 * EMAX(I)
C   EMAX(I) = MAX1F (ABSF(YCDP(I)), GAMMA)
C   IF(DIF(I) - C1 * TEMAX(I)) 105, 105, 110
   IF (DIF(I) - C1 * EMAX(I)) 105, 105, 110
105 CONTINUE
C   DO 106 I = 1, NDEQ
C 106 EMAX(I) = TEMAX(I)
   GOTO 120
110 HAVGT = 1.0
   CALL OUT(2)
   GOTO 70
120 DO 125 I = 1, NDEQ
   Z(I,K) = ZTAB(I)
D 125 YDP(I) = YCDP(I)
   IF (K - 11) 130, 140, 140
130 K = K + 1
135 CALL OUT(2)
   S = S + H
   GOTO 200
140 IF(ABSF(H)-HMAX)141,160,160
141 DO 150 I = 1, NDEQ
D   DIF(I) = 0.0
D   DIF(I) = ABSF(YPD(I) - YCD(I))
   EMAX(I) = MAX1F(ABSF(YCDP(I)), GAMMA)
   IF(DIF(I) - C2 * EMAX(I)) 150, 150, 160
150 CONTINUE
   DBLGT = 1.0
   CALL OUT(2)
   GOTO 70
160 DO 170 I = 1, NDEQ
   DO 170 J = 1, 10
170 Z(I,J) = Z(I,J+1)
   GOTO 135
200 IF (SIGNH * (S - SMAX)) 65, 220, 210
210 HSTAR = SMAX - S + H
   CALL MOD
220 S = SMAX - 0.00001
221 CALL PRED
   S = SMAX
   CALL OUT(1)
   RTN = 1.0
   GOTO 65
END
```

```
*   HAVGT
*   LISTB
*   LABEL
```

AFFDL-TR-71-116

* DIAGOL
SUBROUTINE HAV

CHAV

C
C

COMMON YTAB, ZTAB, Z, ATAB, ITAB
COMMON YDP, YPD, YDPD, YCD, YCDP, Y
COMMON WANL, VANL, UANL, BANL

DIMENSION YTAB(10), ZTAB(10), Z(10,12), ATAB(20), ITAB(10)

D DIMENSION YDP(10), YPD(10), YDPD(10), YCD(10), YCDP(10), Y(10,12)
DIMENSION ZH(10,3)

EQUIVALENCE (YTAB(1), WN), (YTAB(2), VN), (YTAB(3), UN),

1 (YTAB(4), BN), (YTAB(5), XMN), (YTAB(6), XNPN),

2 (YTAB(7), XNN), (YTAB(8), QN)

EQUIVALENCE (ZTAB(1), WDOT), (ZTAB(2), VDOT), (ZTAB(3), UDOT),

1 (ZTAB(4), BDOT), (ZTAB(5), XMDOT), (ZTAB(6), XNPDOT)

2, (ZTAB(7), XNNDOT), (ZTAB(8), QDOT)

EQUIVALENCE (ATAB(1), XO), (ATAB(2), S), (ATAB(3), H),

1 (ATAB(4), EPSIL), (ATAB(5), DX), (ATAB(6), FLAG),

2 (ATAB(7), DELTA), (ATAB(8), RTN), (ATAB(9), SMAX),

3 (ATAB(10), HSTAR), (ATAB(11), DBLGT), (ATAB(12), HAVGT),

4 (ATAB(13), GAMMA)

EQUIVALENCE (ITAB(1), NDEQ), (ITAB(3), INDEX),

1 (ITAB(4), K)

H = H / 2.0

CON = 1.0 / 256.0

IF (SENSE SWITCH 6) 1, 2

1 CALL OUT(3)

2 DO 5 I = 1, NDEQ

ZH(I,1) = CON * (7.0 * Z(I,K-6) - 45.0 * Z(I,K-5) + 126.0 *

1 Z(I,K-4) - 210.0 * Z(I,K-3) + 315.0 * Z(I,K-2) + 63.0 * Z(I,K-1))

ZH(I,2) = CON * (-3.0 * Z(I,K-6) + 21.0 * Z(I,K-5) - 70.0 *

1 Z(I,K-4) + 210.0 * Z(I,K-3) + 105.0 * Z(I,K-2) - 7.0 * Z(I,K-1))

ZH(I,3) = CON * (3.0 * Z(I,K-6) - 25.0 * Z(I,K-5) + 150.0 *

1 Z(I,K-4) + 150.0 * Z(I,K-3) - 25.0 * Z(I,K-2) + 3.0 * Z(I,K-1))

5 CONTINUE

L = - 1

M = 5

DO 10 J = 1, 4

L = L + 2

M = M - 1

N = K - M

DO 10 I = 1, NDEQ

10 Z(I,L) = Z(I,N)

L = 8

DO 20 N = 1, 3

L = L - 2

DO 20 I = 1, NDEQ

20 Z(I,L) = ZH(I,N)

K = 2

S = S - H

IF (SENSE SWITCH 6) 30, 100

30 WRITE OUTPUT TAPE 6, 3, ((Z(I,L), L = 1, 6), I = 1, 4)

3 FORMAT (28H VALUES OF PAST DERIVATIVES/(1P6E20.7//))

100 RETURN

AFFDL-TR-71-116

END

```
* DBL
* LIST8
* LABEL
* DIAGOL
SUBROUTINE DBL
```

CDBL

C
C

```
COMMON YTAB, ZTAB, Z, ATAB, ITAB
COMMON YDP, YPD, YPDP, YCD, YCDP, Y
COMMON WANL, VANL, UANL, BANL
DIMENSION YTAB(10), ZTAB(10), Z(10,12), ATAB( 20), ITAB(10)
DIMENSION SAVZ(10,12)
```

```
D DIMENSION YDP(10), YPD(10), YPDP(10), YCD(10), YCDP(10), Y(10,12)
EQUIVALENCE (YTAB( 1), WN), (YTAB( 2), VN), (YTAB( 3), UN),
1 (YTAB( 4), BN), (YTAB( 5), XMN), (YTAB( 6), XNPN),
2 (YTAB( 7), XNN), (YTAB( 8), QN)
EQUIVALENCE (ZTAB( 1), WDOT), (ZTAB( 2), VDOT), (ZTAB( 3), UDOT),
1 (ZTAB( 4), BDOT), (ZTAB( 5), XMDOT), (ZTAB( 6), XNPDOT)
2 (ZTAB( 7), XNNDOT), (ZTAB( 8), QDOT)
EQUIVALENCE (ATAB( 1), XO), (ATAB( 2), S), (ATAB( 3), H),
1 (ATAB( 4), EPSIL), (ATAB( 5), DX), (ATAB( 6), FLAG),
2 (ATAB( 7), DELTA), (ATAB( 8), RTN), (ATAB( 9), SMAX),
3 (ATAB(10), HSTAR), (ATAB(11), DBLGT), (ATAB(12), HAVGT),
4 (ATAB(13), GAMMA)
EQUIVALENCE (ITAB( 1), NDEQ), (ITAB( 2), L), (ITAB( 3), INDEX),
1 (ITAB( 4), K)
```

```
SAVS=S
SAVH=H
H = 2.0 * H
S=S+H
FLAG=2.0
CALL TEST
IF(RTN-1.0)20,100,100
20 DO 30 I=1,NDEQ
DO 30 J=1,11
30 SAVZ(I,J)=Z(I,J)
DO 40 I=1,NDEQ
DO 40 J=1,6
K=2*J-1
40 Z(I,J)=SAVZ(I,K)
K=7
CALL PRED
CALL TEST
IF(HAVGT)100,100,50
50 DO 60 I=1,NDEQ
DO 60 J=1,10
60 Z(I,J)=SAVZ(I,J+1)
K=11
HAVGT = 0.0
H=SAVH
```


AFFDL-TR-71-116

S=SAVS+H
100 RETURN
END

```
* MOD
* LIST8
* LABEL
* DIAGOL
  SUBROUTINE MOD
```

CMOD

C
C

```
COMMON YTAB, ZTAB, Z, ATAB, ITAB
COMMON YDP, YPD, YPDP, YCD, YCDP, Y
COMMON WANL, VANL, UANL, BANL
DIMENSION YTAB(10), ZTAB(10), Z(10,12), ATAB(20), ITAB(10)
D DIMENSION YDP(10), YPD(10), YPDP(10), YCD(10), YCDP(10), Y(10,12)
DIMENSION YST(10, 5)
EQUIVALENCE (YTAB( 1), WN), (YTAB( 2), VN), (YTAB( 3), UN),
1 (YTAB(4 ), BN), (YTAB( 5), XMN), (YTAB( 6), XNPN),
2 (YTAB( 7), XNN), (YTAB( 8), QN)
EQUIVALENCE (ZTAB( 1), WDOT), (ZTAB( 2), VDOT), (ZTAB( 3), UDOT),
1 (ZTAB( 4), BDOT), (ZTAB( 5), XMDOT), (ZTAB( 6), XNPDOT)
2, (ZTAB( 7), XNNDOT), (ZTAB( 8), QDOT)
EQUIVALENCE (ATAB( 1), XO), (ATAB( 2), S), (ATAB( 3), H),
1 (ATAB( 4), EPSIL), (ATAB( 5), DX), (ATAB(6), FLAG),
2 (ATAB( 7), DELTA), (ATAB( 8), RTN), (ATAB( 9), SMAX),
3 (ATAB(10), HSTAR), (ATAB(11), DBLGT), (ATAB(12), HAVGT),
4 (ATAB(13), GAMMA)
EQUIVALENCE (ITAB( 1), NDEQ), (ITAB( 2), L), (ITAB( 3), INDEX),
1 (ITAB( 4), K)
```

```
CON = 1.0 / 120.0
HI = HSTAR / H
DO 10 M = 1, 5
  XM = M
  XMH = XM * HI
  A = 5.0 - XMH
  B = A - 1.0
  C = B - 1.0
  D = C - 1.0
  E = D - 1.0
  F = E - 1.0
  DO 10 I = 1, NDEQ
10 YST(I,M) = CON * (-B * C * D * E * F * Z(I,K-6) + 5.0 * A * C * D
1 * E * F * Z(I,K-5) - 10.0 * A * B * D * E * F * Z(I,K-4) + 10.0 *
2 A * B * C * E * F * Z(I,K-3) - 5.0 * A * B * C * D * F * Z(I,K-2)
3 + A * B * C * D * E * Z(I,K-1))
  DO 5 I = 1, NDEQ
5 Z(I,6) = Z(I,K-1)
  N = 6
  DO 20 J = 1, 5
  M = M - 1
  DO 20 I = 1, NDEQ
```

Contrails

AFFDL-TR-71-116

20 Z(I,M) = YST(I,J)

H = HSTAR

K = 7

RETURN

END

2. GLAUZ

The second package used a routine obtained from SHARE written by R. Glauz. This routine was written in FAP and later converted to MAP. The program used a third order Runge-Kutta as a starter routine and then switched over to Adams' method. The nice feature about this routine is that after a full difference table has been set, halving and doubling could be accomplished without re-evaluation of the derivatives at three or four additional points. A FORTRAN subroutine has been written following the general logic.

* FAP	SUBROUTINE - ADAMS METHOD INTEGRATION	AS124001
* GLAUZ	METHOD OF INTEGRATION	AS124003
ENTRY	ADMSET	AS124005
ENTRY	ADMRES	AS124006
ENTRY	ADMINT	AS124007
ENTRY	ADMCDR	AS124008
ENTRY	ADMPAR	AS124009
ADMSET	CLA* 1,4	AS124010
ARS	18	AS124011
STA	BZ	AS124012
STO	TEMP	AS124013
ALS	2	AS124014
ADD	TEMP	AS124015
STA	CF	AS124016
STO	TEMPA	AS124017
CLA	2,4	AS124018
ADD	ADMSET	AS124019
STA	LF	AS124020
CLA	3,4	AS124021
ADD	ADMSET	AS124022
STA	LD	AS124023
CLA	4,4	AS124024
ADD	ADMSET	AS124025
STA	CP	AS124026
CLA	5,4	AS124027
ADD	ADMSET	AS124028
STA	CC	AS124029
SUB	TEMPA	AS124030
STA	CK	AS124031
SUB	TEMP	AS124032
STA	CA	AS124033
SUB	TEMP	AS124034
STA	LG	AS124035
CLA	6,4	AS124036
STA	CX	AS124037
CLA	7,4	AS124038
STA	CH	AS124039
CLA	8,4	AS124040
ADD	ADMSET	AS124041
STA	CD	AS124042
TAI	**1,4,-8	AS124043
ADMRES	STZ FLAG	AS124044
SXA	IR4A,4	AS124045
LXA	B2,4	AS124046
CAL	**4	AS124047
NZT*	CD	AS124048
CLA	TEST	AS124049
ANA	MASKT	AS124050
ADD	FACT	AS124051
STO	**4	AS124052
TIX	CD,4,1	AS124053
IRT	**4	AS124054
SXA	IRIR,1	AS124055
LXA	CP,1	AS124056
STZ*	CC	AS124057
TIX	9-1,1,1	AS124058
ART	5r,1	AS124059
SXD	GNT,1	AS124060
IRIR	**1	AS124061
TRA	1,4	AS124062

N TO ACC	
N TO ADDR	
N TO ADDR OF BZ	
N IN ADDR	
4N	
+N	
5N TO ADDR CF	
5N IN ADDR	
F FUNCTION LOC.	
1 IN ADDR	
FUNCTION VALUE LOCATION	
DERIV LOC	
+1	
DERIV VALUE LOC.	
PARTIAL STEP LOC.	
+1	
PARTIAL STEP LOCATION	
DIFF TABLE LOC.	
+1	
DIFF TABLE LOC.	
-N	
OLD FCM. LCC.	
-N	
ACCURACY EXPONENT TABLE	
-N	
TEST VALUE LOC	
K LOC.	
H LOC.	
ACCURACY INPUT TABLE	
ZERO FLAG	
N TO IR4	
ACCURACY INPUT,4	
TEST FOR ZERO ACCURACY INPUT	
ACCURACY SET EQUAL TO +152	
ACCURACY EXPONENT,4	
5N TO IRI	
DIFF TABLE,1	
LOOP TO ZERO TABLE	

ADMINT SXA	IR4,4	STOKE RETURN ADDR.	AS124063
NZT	FLAG	RETURN IF FLAG=0	AS124064
TRA	1,4	N TO IR4	AS124065
AXT	**4	FUNCTION,4	AS124066
CLAF	LF	OLD FUNCTION,4	AS124067
STO	**4	PRED TABLE LOC	AS124070
TIX	-2,4,1	X	AS124071
AXT	COEFC,4	H	AS124072
SXA	CC,4	STORE IN X	AS124073
TSX	COMP,4		AS124074
CLA	**		AS124075
FAD	**		AS124076
STO*	CX		AS124077
AXT	**4		AS124078
TRA	1,4		AS124079
SXA	IR1,1		AS124080
IR2,2			AS124081
SXA	IR4,4		AS124082
LXA	BZ,4	N TO IR4	AS124083
AXT	**1	SN TO IR1	AS124084
AXT	5,2	5 TO IR2	AS124085
STZ	TEMP	DIFF TABLE,1	AS124086
LDQ	**1	COEFC,2	AS124087
FMP	**2		AS124088
FAD	TEMP		AS124089
STO	TEMP		AS124090
TXI	**1,1,-1	NEXT DIFFERENCE	AS124091
TIX	CC,2,1	NEXT COEF	AS124092
AXT	0,2	0 TO IR2	AS124093
LDQ	**4	DERIV,4	AS124094
FMP*	CO	COEFC,2	AS124095
FAD	TEMP		AS124096
XCA	CH		AS124097
FMP*	CH	H	AS124098
STO	**4	TEST VALUE,4	AS124099
FAD*	CK	OLD FUNCTION,4	AS124100
STO*	LF	FUNCTION,4	AS124101
TIX	CB,4,1	NEXT EQUATION	AS124102
AXT	**4		AS124103
IR2	**2		AS124104
IR1	**1		AS124105
TRA	1,4		AS124106
ADHOC	CLA	ZERO FLAG GO TO FIRST	AS124107
TZE	FIRST	0 TO DECR TT	AS124108
SXA	IR4D,4	N TO IR4	AS124109
STD	TT	TEST VALUE,4	AS124110
LXA	BZ,4	PREDICTOR TEST VALUE,4	AS124111
CLA*	LG	CORRECTOR TABLE LOC	AS124112
STO	TEMPA,4		AS124113
TIX	DA,4,1		AS124114
AXT	COEFC,4		AS124115
SXA	CO,4		AS124116
TSX	COMP,4		AS124117
LXA	BZ,4		AS124118
CAL	**4	N TO IR4	AS124119
ANA	MASKT	FUNCTION,4	AS124120
SUB	FACS	KEEP EXPONENT ONLY	AS124121
UFA*	LG	SHIFT EXPONENT	AS124122
UFS	TEMPA,4	TEST VALUE,4	AS124123
SSP	CA	DEL YC - DEL YP	
UFA*	CA		

ANA	MASK	AS124124
TZE	**2	AS124125
SXD	TI,4	AS124126
ANA	MASKH	AS124127
TXZ	HALV	AS124128
DE	LF,4,1	AS124129
LXA	BZ,4	AS124130
LXL	CNT,1	AS124131
TXL	**3,1,0	AS124132
TXI	**1,1,-1	AS124133
SXD	CNT,1	AS124134
LXA	CF,1	AS124135
LP	AXT	AS124136
	S,2	AS124137
LDO*	LD	AS124138
CLS*	CC	AS124139
STO*	CC	AS124140
FAD*	CC	AS124141
CP	KCA	AS124142
CNT	**2,2,**	AS124143
LDO	COEPP	AS124144
TXI	**1,1,-1	AS124145
TXI	CPR,2,1	AS124146
TXI	LP,4,1	AS124147
TXL	IR40,4,**	AS124148
AXT	TDBL,4	AS124149
SXA	TY,4	AS124150
TSX	MAT,4	AS124151
LDO*	CH	AS124152
FMP	THAL+1	AS124153
STO*	CH	AS124154
IR40	AXT	AS124156
TRA	IR2	AS124157
HALV	AXT	AS124158
SXA	TY,4	AS124159
TSX	MAT,4	AS124160
GLA*	CX	AS124161
F8B*	CH	AS124162
STO*	CX	AS124163
LDC*	CH	AS124164
FMP	TPBL	AS124165
STO*	CH	AS124166
TSX	IR2,4	AS124167
TRA	BF	AS124168
FIRST	STL	AS124169
SXA	FLAG	AS124170
LXA	IRIF,1	AS124171
LXA	BZ,4	AS124172
LXA	CF,1	AS124173
GLA*	LD	AS124174
STO*	CC	AS124175
TXI	**1,1,-5	AS124176
TXI	**3,4,1	AS124177
AXI	**1	AS124178
TRA	ADMIT+3	AS124179
MAT	LXA	AS124180
SXA	CF,1	AS124181
TXA	IR4M,4	AS124182
AXI	4,4	AS124183
AXI	9,2	AS124184
SXD	TU,4	
TXI	**1,1,-1	
STZ	TEMP	
LDO*	CC	

DOUBLING TEST 0 DOUBLE
NO DOUBLE FLAG OVER 0
HALVING TEST 0 NO HALVE
NO ZERO HALVE
N TO IR4
REDUCE CNT BY 1
SN TO IR1
S TO IR2
DERIV,4
DIFF TABLE,1
DIFF TABLE,1
DIFF TABLE,1
0.
NEXT DIFF
** ZERO DOUBLE
H
2.0
H
HALVING TABLE
X
H X BACKED UP
H
-5
NEW H
RESTORE IR1,IR2 DESTROYED BY MAT
FLAG NOT EQUAL ZERO
STORE IR1
N TO IR4
SN TO IR1
DERIV,4
DIFF TABLE,1
MOVE TO NEXT DERIV IN DIFF TABLE
SN TO IR1
4 TO IR4
9 TO IR2
STORE IR4 IN SHIFT LOC.
MOVE PAST Y OR PAST DIFF.
DIFF TABLE,1

TY	FMP	**2	AS124185
	FAD	TEMP	AS124186
	STO	TEMP	AS124187
	TXI	**1,1,-1	AS124188
	TXI	**1,2,-1	AS124189
	TXI	TX,4,1	AS124190
	TXI	**1,1,00	AS124191
TU	STQ*	CC	AS124192
TZ	LXD	TU,4	AS124193
	TX	TV,4,1	AS124194
	TX	TV,1,1	AS124195
	TX	**1,4	AS124196
IRAH	AXT	1,4	AS124197
	TRA	IR,4	AS124198
	ADMPAR	SXA	AS124199
	SXA	IR,2,2	AS124200
	SXA	IR,1,1	AS124201
	SXA	IR,4PA,4	AS124202
	CLAP	1,4	AS124203
	FSB*	CX	AS124204
	FDP*	CH	AS124205
	STO	TEMPA-5	AS124206
	STZ	TEMPA	AS124207
	AXT	-4,4	AS124208
	FMP	TEMPA-5	AS124209
	STO	TEMPB-4,4	AS124210
	XCA	**1,4,1	AS124211
	TXI	8-4,4,0	AS124212
	TXH	5,1	AS124213
	AXT	CPP,4	AS124214
	AXT	TX,4	AS124215
	SXA	TZ,4	AS124216
	AXT	TPBL,4	AS124217
	SXA	TY,4	AS124218
	TXH	MAT+1,4	AS124219
	AXT	CC,4	AS124220
	SXA	TX,4	AS124221
	SXA	TZ,4	AS124222
	AXT	1,1	AS124223
	AXT	4,2	AS124224
CPP	CLA	TEMPB+1,1	AS124225
	STO	TEMPA,2	AS124226
	TXI	**1,1,1	AS124227
	TXI	**2,2,1	AS124228
	TXI	TEMPA,4	AS124229
	SXA	CD,4	AS124230
	AXT	LF,4	AS124231
	AXT	CCM,4	AS124232
	SXA	CP,4	AS124233
	SXA	CM,4	AS124234
	TXH	COMP+2,4	AS124235
	AXT	LF,4	AS124236
	SXA	CM,4	AS124237
	AXT	CK,4	AS124238
	AXT	CCR,4	AS124239
	SXA	**1,4	AS124240
	IRAPA	AXT	AS124241
	TRA	2,4	AS124242
	MASKH	OCT	AS124243
	MASKD	OCT	AS124244
	MASKY	OCT	AS124245
	FACS	OCT	
	FACT	OCT	

MAT COEF,2
 MOVE TO NEXT DIFF.
 MOVE TO NEXT COEF.
 GO FOR MORE TERMS
 MOVE BACK TO ORIG DIFF
 DIFF TABLE,1
 RESET IRA
 REDUCE IRA BY 1
 MOVE TO Y TERM OR ALL THROUGH

XP
 XP-X
 XP-X/H=P
 P TO TERRA-5
 GENERATE POWERS OF P

AVOIDS SN TO IRA

AFFDL-TR-71-1116

TEST	OCT	+15200000000						
FLAG	NOP	0						
DEC	DEC	1.						
DEC	DEC	.5						
OCT	OCT	+1776525253						
DEC	DEC	.375						
OCT	OCT	+17754476477						
COEFP	DEC	0.						
OCT	OCT	+20051540541						
OCT	OCT	+176466026603						
DEC	DEC	.068055556						
OCT	OCT	+173660266027						
DEC	DEC	0.						
COEFC	OCT	+177546764477						
DEC	DEC	.00833333333						
DEC	DEC	.0625						
DEC	DEC	.152777778						
DEC	DEC	.125						
DEC	DEC	.0416666667						
DEC	DEC	.166666667						
DEC	DEC	.166666667						
DEC	DEC	.166666667						
DEC	DEC	.25						
TPBL	DEC	.5						
DEC	DEC	.125						
DEC	DEC	.0625						
DEC	DEC	.0390625						
DEC	DEC	.25						
DEC	DEC	.125						
DEC	DEC	.078125						
DEC	DEC	.125						
DEC	DEC	.09375						
THAL	DEC	.0625						
DEC	DEC	2.						
DEC	DEC	-1.						
DEC	DEC	0.						
DEC	DEC	0.						
DEC	DEC	4.						
DEC	DEC	-4.						
DEC	DEC	1.						
DEC	DEC	8.						
DEC	DEC	-12.						
TOBL	DEC	16.						
TEMP	COMMON	-206						
TEMPA	COMMON							
TEMPB	SYN	TEMP-1						
TEMPC	SYN	TEMP-10						
TEMPD	END							

3. DZ BKY ZAM

A subroutine written by Meissner at University of California, Berkeley Lawrence Radiation Lab, using an Adams-Moulton predictor-corrector method, with a starting procedure based upon Zonneveld formulae. The subroutine uses a variable step size, but the steps are forced to land on multiples of the specified print interval. If change in step size is encountered, re-evaluation of the difference table is required.

Contrails

AFFDL-TR-71-116

```
FORTRAN IV SUBROUTINE INTO(NO,X,DERI ,Y,F,T,HPRO)
COMMON /INTC/ IPMX,AREF,EMAX,SSSR,HFAC,SWAM,SWEX
COMMON /INTP/ HPR,XX,N,EUB,ELE,IP,IT,NRKS,SWIN
DIMENSION Y(1),F(1),T(8,1)
LOGICAL SWAM,SWEX,SWIN
INTEGER HFAC
DOUBLE PRECISION T,HPRO,HPR,XX
DATA IPMX,AREF,EMAX,SSSR,HFAC,SWAM,SWEX
$ /1024,1.0,1.0E-6,100.0,2,.TRUE.,.TRUE./
```

C

```
HPR=HPRO
XX=DBLE(X)
N=NO
EUB=EMAX
ELB=EMAX/SSSR
IP=1
IT=0
NRKS=0
SWIN=SWEX
CALL DERI (X,Y,F)
DO 9 I=1,N
T(5,I)=DBLE(Y(I))
9 CONTINUE
RETURN
END
```

```
SUBROUTINE INT(X,DERI ,Y,F,T,SWPR
X )
```

```
COMMON /INTC/ IPMX,AREF,EMAX,SSSR,HFAC,SWAM,SWEX
COMMON /INTP/ HPR,XX,N,EUB,ELE,IP,IT,NRKS,SWIN
```

C

```
DIMENSION Y(1),F(1),T(8,1)
LOGICAL SWAM,SWEX,SWIN
LOGICAL SWPR
INTEGER HFAC
DOUBLE PRECISION T,HPR,XX
DOUBLE PRECISION D,H
```

```
6000 FORMAT (36H0 CANNOT DECREASE H BECAUSE OF HMIN. ,1PE16.8,I20)
```

C

```
1 CONTINUE
SWPR=.FALSE.
TEST=0.0
H=HPR/DBLE(FLOAT(IP*24))
IF ((NRKS .LT. 3) .OR. (.NOT. SWAM)) GO TO 200
```

Contrails

AFPDL-TR-71-116

C

C ADAMS-MOULTON STEP.

100 CONTINUE
DO 109 I=1,N
D=DBLE(F(I))
T(4,I)=D
Y(I)=SNGL(T(5,I)+H*(
X 55.000*D-59.000*T(3,I)+37.000*T(2,I)- 9.000*T(1,I)))

109 CONTINUE
X=SNGL(XX+24.000*H)
CALL DERI (X,Y,F)
DO 119 I=1,N
D=DBLE(F(I))
D=(T(5,I)+H*(
X 9.000*D+19.000*T(4,I)- 5.000*T(3,I)+ T(2,I)))
T(6,I)=D
E=ABS(SNGL(D)-Y(I))/14.0
TEST=AMAX1(E/AMAX1(AREF,ABS(SNGL(D))),TEST)

119 CONTINUE

C

GO TO 300

C

C ZONNEVELD STEP.

200 CONTINUE
DO 209 I=1,N
D=DBLE(F(I))
T(4,I)=D

C

1
Y(I)=SNGL(T(5,I)+H*(
X 12.000*D))

209 CONTINUE
X=SNGL(XX+12.000*H)
CALL DERI (X,Y,F)
DO 219 I=1,N
D=DBLE(F(I))
T(6,I)=D

C

2
Y(I)=SNGL(T(5,I)+H*(
X 12.000*D))

219 CONTINUE
CALL DERI (X,Y,F)
DO 229 I=1,N
D=DBLE(F(I))
T(7,I)=D

C

3
Y(I)=SNGL(T(5,I)+H*(
X 24.000*D))

229 CONTINUE
X=SNGL(XX+24.000*H)
CALL DERI (X,Y,F)
DO 239 I=1,N
D=DBLE(F(I))
T(8,I)=D

C

4
Y(I)=SNGL(T(5,I)+H*(
X 3.7500*T(4,I)+5.2500*T(6,I)+9.7500*T(7,I)-0.7500*D))

239 CONTINUE
X=SNGL(XX+18.000*H)
CALL DERI (X,Y,F)
DO 249 I=1,N

Contrails

AFFDL-TR-71-116

```
D=DBLE(F(I))
E=ABS(SNGL(H*(
X -16.000*T(4,I)+48.000*T(6,I)+48.000*T(7,I)+48.000*T(8,I)
X -128.000*D )))
C
5
D=( T(5,I)+H*(
X 4.000*T(4,I)+ 8.000*T(6,I)+ 8.000*T(7,I)+ 4.000*T(8,I)
X ))
T(6,I)=0
TEST=AMAX1(E/AMAX1(AREF,ABS(SNGL(D))),TEST)
249 CONTINUE
C
C BOTH ADAMS-MOULTON AND ZONNEVELD METHODS CONTINUE FROM HERE.
300 CONTINUE
X=SNGL(XX+24.000*H)
IF (TEST .LE. EUB) GO TO 310
IF (IP*HFAC .GT. IPMX) GO TO 309
C
C REPEAT STEP WITH SMALLER H.
NRKS=0
IP=IP*HFAC
IT=IT*HFAC
DO 305 I=1,N
Y(I)=SNGL(T(5,I))
F(I)=SNGL(T(4,I))
305 CONTINUE
GO TO 1
C
C CANNOT DECREASE H BECAUSE OF HMIN.
309 CONTINUE
IF (.NOT. SWIN) GO TO 310
PRINT 6000, X,IPMX
SWIN=.FALSE.
C
310 CONTINUE
C
C
C ACCEPT CURRENT STEP.
C
C XX STILL HAS NOT BEEN CHANGED SINCE ENTRY.
C YY(XX) IS STILL IN T(5, ).
C F(YY) IS IN T(4, ).
C
IT=IT+1
XX=XX+HPR/DBLE(FLOAT(IP))
NRKS=MIN0(NRKS+1,4)
DO 319 I=1,N
D=T(6,I)
T(5,I)=D
Y(I)=SNGL(D)
319 CONTINUE
X=SNGL(XX)
CALL DERI (X,Y,F)
IF (IT .LT. IP) GO TO 320
C
C X IS A MULTIPLE OF HPRINT.
SWPR=.TRUE.
IT=IT-IP
C
320 CONTINUE
```

Contrails

AFFDL-TR-71-116

```
----- 2
IF (TEST .GE. ELB) GO TO 330                2
IF (MOD(IP,HFAC)+MOD(IT,HFAC) .NE. 0) GO TO 330 2
C                                           2
C PROCEED TO NEXT STEP WITH LARGER H, USING ZONNEVELD METHOD. 2
NRKS=0                                       2
IP=IP/HFAC                                   2
IT=IT/HFAC                                   2
RETURN                                       2
C                                           2
C                                           2
C PROCEED TO NEXT STEP WITH SAME H.         2
330 CONTINUE                                2
DO 339 I=1,N                                2
T(1,I)=T(2,I)                                2
T(2,I)=T(3,I)                                2
T(3,I)=T(4,I)                                2
339 CONTINUE                                2
RETURN                                       2
END                                           2
```

AFFDL-TR-71-116

4. AL ADAM

This subroutine written by Jeske at NASA-Ames uses a fourth order Adams' method with an iterated fourth order Adams' first sum method with a Mersman starter included. This routine uses a fixed step.

```

SUBROUTINE ADAMS(T,A,DERIV,H,K,N)
DIMENSION A(9,N)
C
C DO LOOPS ON THE SECOND SUBSCRIPT IN THE A ARRAY ARE INDICATED BY
C J=1(1)N IN THE COMMENTS. THE SUBSCRIPT J DENOTES THE JTH COMPONENT
C OF WHATEVER VECTOR IS INDICATED (I.E., THE JTH COMPONENT OF X,F,
C ETC.). SECOND SUBSCRIPT IS OMITTED IN THE COMMENTS (I.E.,A1=A(1,J)
C A2=A(2,J),ETC.).
C
C NOTATION--
C T=INDEPENDENT VARIABLE
C A=ARRAY OF DEPENDENT VARIABLES,DERIVATIVES,BACKWARD DIFFERENCES,
C AND FIRST SUMS
C DERIV=NAME OF THE SUBROUTINE THAT CALCULATES THE DERIVATIVES
C F(X,T) AND STORES THEM IN A2 FOR J=1(1)N.
C H=INCREMENT OF INTEGRATION (I.E.,INTEGRATION STEP,STEP-SIZE,OR
C DELTA T)
C K=ENTRY CODE (DEFINED BELOW)
C N=NUMBER OF FIRST ORDER DIFFERENTIAL EQUATIONS
C
C *** FOR A DISCUSSION OF THE METHOD SEE NASA TN D-2936, SELF-STARTING
C MULTISTEP METHODS FOR THE NUMERICAL INTEGRATION OF ORDINARY
C*** DIFFERENTIAL EQUATIONS, BY WILLIAM A. MERSMAN, JULY, 1965.
C
C ENTRY POINT--BEGIN SUBROUTINE--
C
C IF (H.EQ.0.0) CALL EXIT
DT=H
B=T
M=N
C
C SELECT MODE--
C A. FORWARD STARTER (K=0)
C B. BACKWARD STARTER (K=-1)
C C. INTEGRATE ONE STEP WITH PREDICTOR - CORRECTOR (K GE 1)
C
C IF (K.LT.0) GO TO 2
C IF (K.GT.0) GO TO 49
C - - F O R W A R D - B A C K W A R D S T A R T E R - - - - -
C
C H IS SET TO -H IF THE FORWARD STARTER IS REQUESTED.
C THE CODING IS WRITTEN FOR THE BACKWARD STARTER.
C

```

Contrails

AFFDL-TR-71-116

```
C BEFORE THE FORWARD OR BACKWARD STARTERS CAN BE EXECUTED THE A
C ARRAY MUST CONTAIN --
C A1=X0, THE INITIAL VALUES OF X FOR J=1(1)N, ALSO SET T=T0, THE
C INITIAL VALUE OF T.
C
3 DT=-DT
2 D=DT/1440.
CALL DERIV(B,A)
DO 24 J=1,M
A(9,J)=A(1,J)
A(8,J)=A(1,J)-0.5*DT*A(2,J)
DO 26 I=3,7
26 A(I,J)=A(2,J)
24 CONTINUE
C
C STATEMENTS 2 - 24 CALCULATES F0, SAVES X0, SETS F0=F1=F2=F3=F4,
C AND COMPUTES FS4 (THE FIRST SUM AT T=T0+4H). THIS COMMENT, AND ALL
C FURTHER COMMENTS FOR THE FORWARD STARTER, APPLIES TO THE BACKWARD
C STARTER, TOO, IN WHICH CASE REPLACE F1 BY F-1, F2 BY F-2, F3 BY
C F-3, F4 BY F-4, FS4 BY FS0 AND X4 BY X-1 TO X-4.
C
C THE A ARRAY NOW CONTAINS FOR J=1(1)N
C A1=X0
C A2=F0
C A3=F0
C A4=F0
C A5=F0
C A6=F0
C A7=F0
C A8=FS4 OR FS0
C A9=X0
C
DO 38 II=1,8
28 B=B-DT
C
C SET T=T0+H OR T=T0-H
C
DO 27 J=1,M
27 A(1,J)=A(8,J)+D*(11.0*A(7,J)-66.0*A(6,J)+192.0*A(5,J)-830.0*A(4,J)
1-1467.0*A(3,J))
CALL DERIV(B,A)
DO 29 J=1,M
29 A(4,J)=A(2,J)
C
C A NOW CONTAINS FOR J=1(1)N
C A1=X1 OR X-1
C A2=F1 OR F-1
C A3=F0
C A4=F1 OR F-1
C A5=F0
C A6=F0
C A7=F0
C A8=FS4 OR FS0
C A9=X0
C
C B=B-DT
C
C SET T=T0+2H OR T=T0-2H
C
DO 30 J=1,M
```


Contrails

AFFDL-TR-71-116

```
30 A(1,J)=A(8,J)-D*(11.0*A(7,J)-82.0*A(6,J)+720.0*A(5,J)+1522.0*A(4,J)
1)+1429.0*A(3,J))
CALL DERIV(B,A)
DO 31 J=1,M
31 A(5,J)=A(2,J)
C
C A NOW CONTAINS FOR J=1(1)N
C A1=X2 OR X-2
C A2=F2 OR X-2
C A3=F0
C A4=F1 OR F-1
C A5=F2 OR F-2
C A6=F0
C A7=F0
C A8=FS4 OR FS0
C A9=X0
C
C B=B-DT
C
C SET T=T0+3H OR T=T0-3H
C
C DO 32 J=1,M
32 A(1,J)=A(8,J)+D*(27.0*A(7,J)-610.0*A(6,J)-1632.0*A(5,J)-1374.0*A(4
1,J)-1451.0*A(3,J))
CALL DERIV(B,A)
DO 33 J=1,M
33 A(6,J)=A(2,J)
C
C A NOW CONTAINS FOR J=1(1)N
C A1=X3 OR X-3
C A2=F3 OR F-3
C A3=F0
C A4=F1 OR F-1
C A5=F2 OR F-2
C A6=F3 OR F-3
C A7=F0
C A8=FS4 OR FS0
C A9=X0
C
C B=B-DT
C
C SET T=T0+4H OR T=T0-4H
C
C DO 34 J=1,M
34 A(1,J)=A(8,J)-D*(475.0*A(7,J)+1902.0*A(6,J)+1104.0*A(5,J)+1586.0*A
1(4,J)+1413.0*A(3,J))
CALL DERIV(B,A)
DO 35 J=1,M
35 A(7,J)=A(2,J)
C
C A NOW CONTAINS FOR J=1(1)N
C A1=X4 OR X-4
C A2=F4 OR F-4
C A3=F0
C A4=F1 OR F-1
C A5=F2 OR F-2
C A6=F3 OR F-3
C A7=F4 OR F-4
C A8=FS4 OR FS0
C A9=X0
```

Contrails

AFFDL-TR-71-116

```
C
DO 36 J=1,M
36 A(8,J)=A(9,J)+D*(27.0*A(7,J)-146.0*A(6,J)+336.0*A(5,J)-462.0*A(4,J)
1)+965.0*A(3,J))
B=T
38 CONTINUE

C
C STATEMENTS 28 THROUGH 38 CONSTITUTE AN ITERATION LOOP (WITH EIGHT
C ITERATIONS) FOR THE FORWARD OR BACKWARD STARTER.
C
C DURING THE ITERATION THE A ARRAY CONTAINS FOR J=1(1)N
C A1=X1,X2,X3, OR X4 -- OR X-1,X-2,X-3, OR X-4
C A2=F1,F2,F3, OR F4 -- OR F-1,F-2,F-3, OR F-4
C A3=F0 (ORIGINAL VALUE)
C A4=F1 OR F-1 (CURRENT VALUE)
C A5=F2 OR F-2 (CURRENT VALUE)
C A6=F3 OR F-3 (CURRENT VALUE)
C A7=F4 OR F-4 (CURRENT VALUE)
C A8=FS4 OR FS0 (CURRENT VALUE)
C A9=X0 (ORIGINAL VALUE)
C
C BEGIN CALCULATION OF BACKWARD DIFFERENCES --
C
C 37 IF(K.LT.0) GO TO 39
C
C DIFFERENCES FOR FORWARD STARTER --
C
C THE A ARRAY CONTAINS FOR J=1(1)N
C A1=X4
C A2=F4
C A3=F0
C A4=F1
C A5=F2
C A6=F3
C A7=F4
C A8=FS4
C A9=X0
C
C DO 40 J=1,M
C
C STEP ONE
C
C A(1,J)=A(9,J)
C A(2,J)=A(3,J)
C DO 41 I=3,6
41 A(I,J)=A(I+1,J)-A(I,J)
C
C STEP TWO
C
C DO 42 I=1,3
C L=7-I
42 A(L,J)=A(L,J)-A(L-1,J)
C
C STEP THREE
C
C A(6,J)=A(6,J)-A(5,J)
C A(5,J)=A(5,J)-A(4,J)
C
C STEP FOUR
C
```

Contrails

AFFDL-TR-71-116

A(6,J)=A(6,J)-A(5,J)

DO 43 I=1,3

L=6-I

43 A(L,J)=A(L,J)-A(L+1,J)

C

C STEP FIVE

C

A(7,J)=A(5,J)-A(6,J)

A(4,J)=A(4,J)-A(7,J)

40 A(5,J)=A(7,J)-A(6,J)

C

C THE DIFFERENCES OF F0 HAVE BEEN CONSTRUCTED ACCORDING TO THE
C FOLLOWING TABLE. AN EXAMPLE OF THE NOTATION IS 2D2F3, WHICH MEAN
C THAT THE SECOND ITEM STORED IN THIS COLUMN (INDICATED BY THE FIRST
C DIGIT 2) IS THE SECOND DIFFERENCE (INDICATED BY D2) OF F3. THE
C SECOND COLUMN IS THE DATA THAT WAS IN THE A ARRAY AT THE TIME OF
C COMPLETION OF THE ITERATION. ALL TABLE VALUES ARE FOR J=1(1)N.

C

C LOC STEP ONE STEP TWO STEP THREE STEP FOUR STEP FIVE

C

C A1 X4 1X0

C A2 F4 2F0

C A3 F0 3D1F1 4D1F0

C A4 F1 4D1F2 3D2F2 3D2F1 2D2F0

C A5 F2 5D1F3 2D2F3 2D3F3 2D3F2 3D3F0

C A6 F3 6D1F4 1D2F4 1D3F4 1D4F0

C A7 F4 1D3F1

C A8 FS4

C A9 X0

C

C NOTE -- IT IS ASSUMED THAT D4F4=D4F3=D4F2=D4F1=D4F0=CONSTANT

C

C THE A ARRAY NOW CONTAINS FOR J=1(1)N

C A1=X0

C A2=F0

C A3=D1F0

C A4=D2F0

C A5=D3F0

C A6=D4F0

C A7=D3F1

C A8=FS4

C A9=X0

C

C FS0 WILL NOW BE CALCULATED AND STORED IN A8

C

D=-0

93 DO 44 J=1,M

44 A(8,J)=A(1,J)+D*(27.0*A(6,J)+38.0*A(5,J)+60.0*A(4,J)+120.0*A(3,J)+
1720.0*A(2,J))

C

C END OF FORWARD STARTER

C

CALL DERIV(T,A)

RETURN

C

C DIFFERENCES FOR BACKWARD STARTER

C

39 DO 45 J=1,M

C

C STEP ONE

Contrails

```

C      AFFDL-TR-71-116
C      A(1,J)=A(9,J)
C      A(2,J)=A(3,J)
C      DO 46 I=3,6
C      46 A(I,J)=A(I,J)-A(I+1,J)
C
C      STEPS TWO, THREE AND FOUR
C
C      DO 47 I=1,3
C      L=6
C      LL=I+2
C      48 A(L,J)=A(L-1,J)-A(L,J)
C      L=L-1
C      IF (L.NE.LL) GO TO 48
C      47 CONTINUE
C      45 CONTINUE
C
C      THE DIFFERENCES OF F0 HAVE BEEN COMPUTED ACCORDING TO THE
C      FOLLOWING TABLE. AN EXAMPLE OF THE NOTATION IS 2D2F-1, WHICH
C      MEANS THAT THE SECOND ITEM STORED IN THIS COLUMN (INDICATED BY THE
C      FIRST DIGIT 2) IS THE SECOND DIFFERENCE (INDICATED BY D2) OF F-1.
C      THE SECOND COLUMN IS THE DATA THAT WAS IN THE A ARRAY AT THE TIME
C      OF COMPLETION OF THE ITERATION. ALL VALUES ARE FOR J=1(1)N.
C
C      LOC.      STEP ONE      STEP TWO      STEP THREE      STEP FOUR
C
C      A1  X-4      1X0
C      A2  F-4      2F0
C      A3  F0       3D1F0
C      A4  F-1      4D1F-1      3D2F0
C      A5  F-2      5D1F-2      2D2F-1      2D3F0
C      A6  F-3      6D1F-3      1D2F-2      1D3F-1      1D4F0
C      A7  F-4
C      A8  F50
C      A9  X0
C
C      THE A ARRAY NOW CONTAINS FOR J=1(1)N
C      A1=X0
C      A2=F0
C      A3=D1F0
C      A4=D2F0
C      A5=D3F0
C      A6=D4F0
C      A7=F-4
C      A8=F50
C      A9=X0
C
C      END OF BACKWARD STARTER
C
C      GO TO 93
C
C      BEGIN INTEGRATION WITH THE ADAMS-BASHFORTH PREDICTOR AND THE
C      ADAMS-MOULTON CORRECTOR. AT THIS ENTRY POINT THE A ARRAY CONTAINS
C      FOR J=1(1)N
C      A1=XI
C      A2=FI
C      A3=D1FI
C      A4=D2FI
C      A5=D3FI
C      A6=D4FI

```

Contrails

AFEDL-TR-71-116

```
C A7=D5FI
C A8=FSI
C A9=PREDICTED VALUE OF FI FROM THE PREVIOUS INTEGRATION STEP
C
C WHERE FI=F(X(T0+IH),T0+IH), I=1,2,3,...
C
C D5FI SHOULD BE NEARLY ZERO AND IS AN INDICATION OF THE ACCURACY
C OF THE INTEGRATION.
C
C - - A D A M S - B A S H F O R T H   P R E D I C T O R   - - - - -
C
C BEGIN PREDICTOR
C
C 49 DO 570 KK=1,K
C     B=B+DT
C     DO 92 J=1,M
C       I=7
C     50 A(I,J)=A(I-1,J)
C       I=I-1
C       IF (I.NE.2) GO TO 50
C     92 CONTINUE
C       D=DT/1440.0
C       DO 51 J=1,M
C     51 A(1,J)=A(8,J)+D*(475.0*A(7,J)+502.0*A(6,J)+540.0*A(5,J)+600.0*A(4,
C       1J)+720.0*A(3,J))
C       CALL DERIV(B,A)
C       DO 52 J=1,M
C       DO 53 L=3,7
C     53 A(L,J)=A(L-1,J)-A(L,J)
C     52 CONTINUE
C
C THE SOLUTIONS X(I+1)=X(T0+(I+1)H), DERIVATIVES AND BACKWARD
C DIFFERENCES HAVE BEEN COMPUTED IN THE ORDER INDICATED IN THE
C FOLLOWING TABLE
C
C LOC.          ONE    TWO    THREE
C
C A1  XI                X(I+1)
C A2  FI                F(I+1)
C A3  D1FI    FI                D1F(I+1)
C A4  D2FI    D1FI                D2F(I+1)
C A5  D3FI    D2FI                D3F(I+1)
C A6  D4FI    D3FI                D4F(I+1)
C A7  D5FI    D4FI                D5F(I+1)
C A8  FSI
C
C END PREDICTOR
C
C - - A D A M S - M O U L T O N   C O R R E C T O R   - - - - -
C
C BEGIN CORRECTOR
C
C THE A ARRAY CONTAINS THE FOLLOWING FROM THE PREDICTOR FOR J=1(1)N
C (THE LEADING P IN THE ENTRIES MEANS PREDICTED VALUE)
C
C A1=PX(I+1)
C A2=PF(I+1)
C A3=PD1F(I+1)
C A4=PD2F(I+1)
C A5=PD3F(I+1)
```

Contrails

AFFDL-TR-71-116

C A6=PD4F(I+1)

C A7=PD5F(I+1)

C A8=FSI

C

556 DO 557 J=1,N

A(1,J)=A(1,J)+475.0*D*A(7,J)

557 A(9,J)=A(2,J)

CALL DERIV(B,A)

559 DO 565 J=1,N

DELTA=A(2,J)-A(9,J)

DO 566 I=3,7

566 A(I,J)=A(I,J)+DELTA

565 CONTINUE

C

C THE CORRECTOR FORMULA HAS NOW BEEN APPLIED TO THE PREDICTED
C VALUES, AND THE PREDICTED BACKWARD DIFFERENCES HAVE BEEN ADJUSTED
C IN TERMS OF THE CORRECTED VALUES. THE ORDER IN WHICH ALL THIS WAS
C DONE IS SHOWN IN THE FOLLOWING TABLE. (THE LEADING P INDICATES
C PREDICTED VALUES AND THE LEADING C INDICATES CORRECTED VALUES).
C ALL ENTRIES IN THE TABLE ARE FOR J=1(1)N.

C

C LOC. ONE TWO THREE

C

C A1 PX(I+1) 1CX(I+1)

C A2 PF(I+1) 2CF(I+1)

C A3 PD1F(I+1) 1CD1F(I+1)=PD1F(I+1)+DELTA

C A4 PD2F(I+1) 2CD2F(I+1)=PD2F(I+1)+DELTA

C A5 PD3F(I+1) 3CD3F(I+1)=PD3F(I+1)+DELTA

C A6 PD4F(I+1) 4CD4F(I+1)=PD4F(I+1)+DELTA

C A7 PD5F(I+1) 5CD5F(I+1)=PD5F(I+1)+DELTA

C A8 FSI

C A9 PF(I+1)

C

C THE LEADING DIGIT IN COLUMNS TWO AND THREE INDICATES THE ORDER IN
C WHICH THE ENTRIES WERE STORED.

C

C THE A ARRAY NOW CONTAINS FOR J=1(1)N

C A1=CX(I+1)=X

C A2=CF(I+1)=F

C A3=CD1F(I+1)=D1F

C A4=CD2F(I+1)=D2F

C A5=CD3F(I+1)=D3F

C A6=CD4F(I+1)=D4F

C A7=CD5F(I+1)=D5F

C A8=FSI

C

C END OF PREDICTOR CORRECTOR

C

C COMPUTE FIRST SUM

54 DO 59 J=1,M

59 A(8,J)=A(8,J)+H*A(2,J)

570 CONTINUE

T=B

RETURN

C ON RETURN- A ARRAY CONTAINS FOR J=1(1)N--

C

C A1=X

C A2=F

C A3=D1F

C A4=D2F

C A5=D3F

Contrails

AFFDL-TR-71-116

C	A6=D4F
C	A7=05F
C	A8=FS
C	END

AFFDL-TR-71-116

5. DZ C STR

This routine was written by Hudson from Commonwealth Scientific and Industrial Research Organization, Canberra, Australia. Two FORTRAN statements require a CDC 3600 configuration. A trivial modification can be made and this program can be executed on the CDC 6600. The basic integration is a self-starting variation of the Adams' method incorporating automatic step size control.


```

SUBROUTINE NORDSET (K,T,H,N,Y,F,DELTA,Y,B,N,T,L,NPL,PL)
C K CONTROL INTEGER FOR USER STATEMENTS INTEGER
C T INDEPENDENT VARIABLE REAL
C H INTEGRATION STEP SIZE REAL
C N NUMBER OF FIRST ORDER EQUATIONS INTEGER
C Y DEPENDENT VARIABLES REAL
C F DERIVATIVES REAL
C DELAY ERROR CONTROL VECTOR REAL
C B TEMPORARY STORAGE, DIMENSION 10*N REAL
C NTL NUMBER OF ENTRIES IN TL INTEGER
C TL LIST OF INTERRUPT TIMES REAL
C NPL NUMBER OF ENTRIES IN PL REAL
C PL LIST OF INTERRUPT FUNCTIONS REAL
C B(1,I) EQUIVALENT
C B(2,I) OF
C B(3,I) ADAMS
C B(4,I) DIFFERENCES
C B(5,I) PREDICTED DERIVATIVES
C B(6,I) Y AT START OF INTEGRATION STEP
C B(7,I) SECOND PRECISION PART OF Y ABOVE
C B(8,I) F AT START OF INTEGRATION STEP
C B(9,I) HOLE FOR INITIAL Y WHILE STARTING
C B(10,I) SECOND PRECISION PART OF Y
C DIMENSION Y(1),F(1),B(10,N),TL(1),PL(1),DPTA(2),TEST(2)
C IFND(10),PLEFT(10),PRITE(10)
C EQUIVALENCE (DPTA,DPTEMA)
C TYPE INTEGER STEP
C TYPE DOUBLE OPTEMA
C TYPE LOGICAL FIND,HALVE,DOUBLE
C COMMON/NORDCON/IDER,IEOS,ITL,IPL,STEP,MHAX,MHIN,MHIG,HL
C DATA (MHIG=0),(HL=0)
C DELY(1)=H*(B(8,I)+(B(1,I)+(B(2,I)+(B(3,I)+(B(4,I)+Z))))
C RETURN
C ENTRY NORDINT
C IF (K) GO TO KFLIP
C TEST FOR CALLING SEQUENCE ERROR
C IF (H=LE,0.OR.N=LE,0.OR.N.GT,2000.OR.NTL.LT,0.OR.NTL.GT,500.OR.NPL
C XLT,0.OR.NPL.GT,500.OR.DELTA.LE,0.OR.I.LT,0)
C XCALL QBERROR (0,3)+ERROR IN NORDSET CALLING SEQUENCE.)
C SET SUBROUTINE COUNTERS AND STEP SIZE DATA
C MHAX=MHIN=IDER=IEOS=ITL=IPL=STEP=0
C CONTROL SECTION FOR STARTING INTEGRATION
C ASSIGN 3000 TO KFLIP
C GO TO 1001
C 3000 MH=AND,3777400000000000
C DO 3002 J=1,NTL

```

```

IF (T.EC.TL(J)) 3001,3002
3001 ASSIGN 3002 TO KFLIP
GO TO 1002
3002 CONTINUE
T LEFT=
DO 3004 J=1,NPL
PLEFT(J)=PL(J)
IF(PL(J).EQ.0.)3003,3004
3003 ASSIGN 3004 TO KFLIP
GO TO 1003
3004 CONTINUE
DO 3010 I=1,N
3010 B(I)=Y(I)
D1=1.
ASSIGN 3100 TO ISFOUR
GO TO 1400
3020 I=STEP,AND,3
IF (I) GO TO 2000
I=STEP/4
3030 D1=1.
GO TO (3030,3050,3030,3080,3030,3040) I
ASSIGN 2000 TO ISFOUR
GO TO 1400
3040 D1=2.
HMAX=HMIN=H
ASSIGN 3050 TO ISFOUR
GO TO 1400
3050 DO 3060 I=1,N
Y(I)=B(I)
3060 B(I)=Y(I)+0.0
3070 ASSIGN 3030 TO KFLIP
GO TO 1000
3080 D1=5
ASSIGN 3090 TO ISFOUR
GO TO 1400
3090 IF (HALVE)3100,3050
3100 STEP=0
3110 DO 3110 I=1,N
B(I)=B(I)+B(I)*B(I)+B(I)*B(I)+0.0
GO TO 3050
C
C CONTROL SECTION FOR TIME INTERRUPTS DURING NORMAL INTEGRATION
C STATEMENT 1700 INTEGRATES FORWARD-RETURNING TO 1701
C
1700 GO TO 1600
1701 DO 1702 I=1,N
B(I)=Y(I)
1702 B(I)=F(I)
TSAVE=T
1703 Z=Z+TSAVE
DO 1705 I=1,NL
IF (YL(I).LT,Z) 1704,1705
1704 Z=YL(I)
1705 CONTINUE
IF (Z.GE,TSAVE) GO TO 1707

```

```

NORD0560
NORD0570
NORD0580
NORD0590
NORD0600
NORD0610
NORD0620
NORD0630
NORD0640
NORD0650
NORD0660
NORD0670
NORD0680
NORD0690
NORD0700
NORD0710
NORD0720
NORD0730
NORD0740
NORD0750
NORD0760
NORD0770
NORD0780
NORD0790
NORD0800
NORD0810
NORD0820
NORD0830
NORD0840
NORD0850
NORD0860
NORD0870
NORD0880
NORD0890
NORD0900
NORD0910
NORD0920
NORD0930
NORD0940
NORD0950
NORD0960
NORD0970
NORD0980
NORD0990
NORD1000
NORD1010
NORD1020
NORD1030
NORD1040
NORD1050
NORD1060
NORD1070
NORD1080
NORD1090
NORD1100
NORD1110

```

```

ASSIGN 1706 TOKFLIP
RTEST=TSAVE/Z
RTEST=RTEST.AND..NOT.3
IF (RTEST.EQ.1.0) 17051,17053
17051 DO 17052 I=1,N
17052 Y(I)=B(6,I)
T=TSAVE
GO TO 1001
17053 T=Z-TSAVE
ASSIGN 1001 TO ISTWO
GO TO 1200
1706 ASSIGN 1703 TO KFLIP
ASSIGN 1002 TO ISTHREE
GO TO 1300
1707 DO 1708 I=1,N
F(I)=B(6,I)
1708 Y(I)=B(6,I)
T=TSAVE
ASSIGN 1300 TO KFLIP
ASSIGN 1709 TO ISTHREE
GO TO 1001
1709 RTEST=Z
RTEST=RTEST.AND..NOT.3
IF (RTEST.EQ.1.0) 1710,1711
1710 ASSIGN 1711 TO KFLIP
GO TO 1002
1711 DO 1712 I=1,NPL
1712 F(I)=B(10,I)
PIND(I)=FALSE
GO TO 1700
C INTEGRATE ONE STEP
C SAVE CONDITIONS AT START OF STEP
2000 DO 2010 I=1,N
S(6,I)=Y(I)
S(7,I)=B(10,I)
2010 S(8,I)=F(I)
T=START
C ENTRY FOR HALVED STEP
C
2020 T=TAH
DO 2030 I=1,N
Z=0
Y(I)=B(6,I)+DELY(I)
2030 S(5,I)=F(I)+12.*B(11,I)+13.*B(12,I)+14.*B(13,I)+S.(8(4+I))
C ITERATE TWICE, DEVELOP TEST PARAMETERS
C
HALVE=.FALSE.
DOUBLE=.TRUE.
TEST(1)=TEST(2)=0.
DO 2070 J=1,2
ASSIGN 2040 TO KFLIP
GO TO 1000

```

```

NORD1120
NORD1130
NORD1140
NORD1150
NORD1151
NORD1152
NORD1153
NORD1154
NORD1160
NORD1170
NORD1180
NORD1190
NORD1200
NORD1210
NORD1220
NORD1230
NORD1240
NORD1250
NORD1260
NORD1270
NORD1280
NORD1290
NORD1300
NORD1310
NORD1320
NORD1330
NORD1340
NORD1350
NORD1360
NORD1370
NORD1380
NORD1390
NORD1400
NORD1410
NORD1420
NORD1430
NORD1440
NORD1450
NORD1460
NORD1470
NORD1480
NORD1490
NORD1500
NORD1510
NORD1520
NORD1530
NORD1540
NORD1550
NORD1560
NORD1570
NORD1580
NORD1590
NORD1600
NORD1610
NORD1620
NORD1630

```

```

2040 DO 2070 I=1,N
      ZP(I)=B(5,I)
      IF (J.FQ.2) 2050,2060
2050 Z=ABS(Z+M)
      RTEST=DELTA*ABS(Y(I))
      IF (ZZ.GT.RTEST) HALVE=.TRUE.
      IF (ZZ.GT.RTEST*.015625) DOUBLE=.FALSE.
2060 OPTA(I)=B(6,I)
      OPTA(2)=B(7,I)
      Z=Z+.329861111111
      DPTEMA=OPTEMA*DELY(I)
      Z=ABS(DPTA(I)+Y(I))
      IF (ZZ.GT.TEST(J)) TEST(J)=Z
      Y(I)=OPTA(I)
      B(10,I)=OPTA(2)
2070 CONTINUE
C
C CHECK TEST PARAMETERS,BUMP COUNT OF INTEGRATION STEPS
C
STEP=STEP+1
IF (STEP.GT.I.AND.STEP.LT.25) GO TO 1100
IF (B.TEST(2).GT.TEST(1).AND.NOT.DOUBLE) GO TO 1500
IF (B.TEST(2).GT.TEST(1)) DOUBLE=.FALSE.
IF (STEP.EQ.1) GO TO 1100
IF (HALVE) GO TO 1500,1100
C
C UPDATE ROUTINE,RETURNS TO 3020 IF STARTING = 1701 OTHERWISE
C
1100 DO 1101 I=1,N
      ZP(I)=B(5,I)
      B(1,I)=B(1,I)+13.*B(2,I)+16.*B(3,I)+110.*B(4,I)+2/.96))
      B(2,I)=B(2,I)+14.*B(3,I)+110.*B(4,I)+2*0.4861111111)
      B(3,I)=B(3,I)+5.*B(4,I)+2/9.6)
1101 B(4,I)=B(4,I)+Z/120.
      IF (STEP.LE.24) GO TO 3020
      IF (H.GT.HMAX) HMAX=H
      IF (H.LT.HMIN) HMIN=H
      GO TO 170)
C
C ROUTINE TESTPHI, FALSE EXIT IS S1300, TRUE EXIT IS 1800
C
1300 DO 1301 I=1,NPL
      IF (FIND(I)) GO TO 1301
      IF (PL(I)*LEFT(I).LT.0) GO TO 1303
1301 CONTINUE
1302 LEFT(I)=PL(I)
      TLEFT=T
      GO TO 13THREE
1303 DO 1304 I=1,NPL
1304 PRITE(I)=PL(I)
      TRITE=T
      GO TO 1800
C
C DEPENDENT VARIABLE SEARCH PROCEDURE, ENTERED IF PL(I) CHANGES SIGN
C

```

NORD1640
 NORD1650
 NORD1660
 NORD1670
 NORD1680
 NORD1690
 NORD1700
 NORD1710
 NORD1720
 NORD1730
 NORD1740
 NORD1750
 NORD1760
 NORD1770
 NORD1780
 NORD1790
 NORD1800
 NORD1810
 NORD1820
 NORD1830
 NORD1840
 NORD1850
 NORD1860
 NORD1870
 NORD1880
 NORD1890
 NORD1900
 NORD1910
 NORD1920
 NORD1930
 NORD1940
 NORD1950
 NORD1960
 NORD1970
 NORD1980
 NORD1990
 NORD2000
 NORD2010
 NORD2020
 NORD2030
 NORD2040
 NORD2050
 NORD2060
 NORD2070
 NORD2080
 NORD2090
 NORD2100
 NORD2110
 NORD2120
 NORD2130
 NORD2140
 NORD2150
 NORD2160
 NORD2170
 NORD2180
 NORD2190

```

1800 Z=0.0
DO 1802 I=1,NPL
IF (FIND(I)) GO TO 1802
IF (PRITE(I).EQ.0) PLEFT(I)=0
Z=PLEFT(I)/PRITE(I)
IF (Z.LE.Z) 1801,1802
1801 Z=Z
J=J+1
1802 CONTINUE
MPS=(TRITE-TSAVE)-(TRITE-TLEFT)/(I.-Z)
IF ((TSAVE-MP).EQ.T.OR.Z.EQ.0) 1803, 604
1803 ASSIGN 1703 TO KFLIP
FIND(J)=.TRUE.
GO TO 1003
1804 ASSIGN 1001 TO ISTWO
ASSIGN 1300 TO KFLIP
ASSIGN 1800 TO IS THREE
GO TO 1200
C
C CHECK FOR DOUBLE OF STEP SIZE
C
1600 IF (DOUBLE.AND.(.NOT.HBIG.OR.(H*H).LE.HBIG)) 1601,2000
1601 D1=2.
GO TO 1400
C
C SUBROUTINE CALLS. ASSUMES KFLIP SET PRIOR TO ENTRY
C
1100 K=1
IDER=IDER+1
RETURN
1001 K=2
IFOS=IFOS+1
RETURN
1002 K=J+2
ITL=ITL+1
RETURN
1003 K=J+NTL+2
IPL=IPL+1
RETURN
C
C SUBROUTINE TO CHANGE STEP SIZE
C
1400 H=H*D1
D2=D1+D1
D3=D2+D1
D4=D3+D1
DO 1401 I=1,N
B(1,I)=B(1,I)*D1
B(2,I)=B(2,I)*D2
B(3,I)=B(3,I)*D3
B(4,I)=B(4,I)*D4
GO TO ISFOUR
C
C ROUTINE TO PREDICT INTERMEDIATE VALUES OF Y(I)
C

```

```

NORD2200
NORD2210
NORD2220
NORD2230
NORD2240
NORD2250
NORD2260
NORD2270
NORD2280
NORD2290
NORD2300
NORD2310
NORD2320
NORD2330
NORD2340
NORD2350
NORD2360
NORD2370
NORD2380
NORD2390
NORD2400
NORD2410
NORD2420
NORD2430
NORD2440
NORD2450
NORD2460
NORD2470
NORD2480
NORD2490
NORD2500
NORD2510
NORD2520
NORD2530
NORD2540
NORD2550
NORD2560
NORD2570
NORD2580
NORD2590
NORD2600
NORD2610
NORD2620
NORD2630
NORD2640
NORD2650
NORD2660
NORD2670
NORD2680
NORD2690
NORD2700
NORD2710
NORD2720
NORD2730
NORD2740
NORD2750

```

```

1200 T=TSAVE*HP
D1=HP/H
D2=01*01
D3=02*01
D4=03*01
D0 1201 I=1:N
1201 Y(I)=B(6,I)*HP*(B(8,I)+D1)*D2*(B(2,I)+D3*(B(3,I)+D4*(B(4,I)
I))))
GO TO 1210
C
RESTORE T,Y,F. HALVE STEP SIZE, TRY STEP AGAIN
C
1500 RTST=H/2
IF (HL.AND.(RTST.LI=HL)) GO TO 2020
IF (T.EQ.(T*RTST)) CALL QBEROR (0+23MH LESS THAN 20*(-36)ST.)
STEP=STEP-I
T=START
D0 1501 I=1:N
Y(I)=B(6,I)
B(I(0,I)+B(7,I))
B(I(0,I)+B(8,I))
1501 F(I)=B(8,I)
D1=.5
ASSIGN 2020 TO 15FOUR
GO TO 1400
END

```

```

NORD2760
NORD2770
NORD2780
NORD2790
NORD2800
NORD2810
NORD2820
NORD2830
NORD2840
NORD2850
NORD2860
NORD2870
NORD2880
NORD2890
NORD2900
NORD2910
NORD2920
NORD2930
NORD2940
NORD2950
NORD2960
NORD2970
NORD2980
NORD2990
NORD3000

```

AFFDL-TR-71-116

6. D2 GDC CSPDIF

This routine is a single step integration procedure utilizing a pseudo-Runge-Kutta method. This program is a catalog routine from Data Centers Division - CDC.

Contrails

D2 GDC CSPDIF AFFDL-TR-71-116

D2 CS PDIFE1, DIFE11 PSEUDO-RUNGE-KUTTA METHOD
PDIFE1 PERFORMS A SINGLE STEP OF INTEGRATION BY A PSEUDO-RUNGE-KUTTA METHOD. IT REQUIRES A #STARTER# ROUTINE FOR THE FIRST STEP. THE ROUTINE USED IS CALLED DIFE11 WHICH USES THE RUNGE-KUTTA-MERSON METHOD.

SUBROUTINE PDIFE1(DERIW,NDUMMY,HDUMMY,X,Y,YZ)

C

C

C THIS SUBROUTINE PERFORMS ONE STEP OF INTEGRATION BY THE
C PSEUDO-RUNGE-KUTTA METHOD. IT USES THREE EVALUATIONS
C OF THE DERIVATIVES PER INTEGRATION STEP.

C

C DERIW =EXTERNAL NAME OF SUBROUTINE WHICH EVALUATES THE DERIVATI

C NDUMMY =NUMBER OF DIFFERENTIAL EQUATIONS IN THE SYSTEM

C HDUMMY =INTEGRATION STEP SIZE

C X =VALUE OF INDEPENDENT VARIABLE

C Y =VECTOR OF VALUES OF THE INTEGRATION VARIABLES

C YZ =VECTOR OF VALUES OF THE DERIVATIVES OF THE INTEGRATION
C VARIABLES. THESE ARE INCLUDED FOR THE PURPOSE OF OUTPUT

C

EXTERNAL DERIW

COMMON / DIFE00 / YY(78),P1(78),P2(78),P3(78),P4(78),P5(78),P6(78)

* ,YZ(78),XLDP,DUDP,HDP

DOUBLE PRECISION XLDP,YLDP, DUDP,HDP

DIMENSION Y(1)

COMMON / YLDP / YLDP(78)

H= HDUMMY

N= NDUMMY

IF (H.GT.0.0) GO TO 100

CALL DIFE11(DERIW,N,0.0,X,Y,P5,YZ)

ASSIGN 200 TO IP

RETURN

100 GO TO IP ,(200,550)

200 ASSIGN 550 TO IP

DO 250 I=1,N

250 YY(I)=Y(I)

X1=X

CALL DIFE11(DERIW,N,H,X,Y,P5,YZ)

U1=.541

U2=.763

U3= 2.0*U2*(U2-U1)/(U1*(4.0-5.0*U1))

C1= (5.0*(U1+U2)-6.0*U1*U2-4.0)/(12.0*U1*U2)

C2= (4.0-5.0*U2)/(12.0*U1*(U1-U2))

C3= (5.0*U1-4.0)/(12.0*U2*(U1-U2))

U2MU3=U2-U3

300 CALL DERIW(X1,YY,P1)

X2=X1+U1*H

X3=X1+U2*H

DO 400 I=1,N

P1(I)=H*P1(I)

400 YZ(I)= YY(I)+ U1*P1(I)

CALL DERIW(X2,YZ,P2)

AFFDL-TR-71-116

```

DO 500 I=1,N
P2(I)=P2(I)*H
500 YZ(I)= YY(I)+U2MU3*P1(I)+U3*P2(I)
CALL DERIW(X3,YZ,P3)
DO 600 I=1,N
YZ(I)=P5(I)
600 P3(I)=P3(I)*H
RETURN
550 CONTINUE
X4=X+U1*H
X5=X+U2*H
CALL DERIW(X,Y,P4)
DO 700 I=1,N
P4(I)=P4(I)*H
700 YZ(I)= Y(I)+U1*P4(I)
CALL DERIW(X4,YZ,P5)
DO 800 I=1,N
P5(I)=P5(I)*H
800 YZ(I)= Y(I)+U2MU3*P4(I)+U3*P5(I)
CALL DERIW(X5,YZ,P6)
DO 900 I=1,N
YZ(I)=P6(I)
P6(I)=P6(I)*H
DU= P4(I) -C1*(P4(I)-P1(I)) + C2*(P5(I)-P2(I))
* +C3*(P6(I)-P3(I))
DUOP=DBLE(DU)
YLDP(I)=YLDP(I)+DUOP
Y(I)= SNGL(YLDP(I))
P1(I)=P4(I)
P2(I)=P5(I)
900 P3(I)=P6(I)
XLDP= XLDP+HDP
X= SNGL(XLDP)
RETURN
END
SUBROUTINE DIFE11(DERIW,NDUMMY,HDDUMMY,X,Y,P5,YY)
C
C
C
C THIS SUBROUTINE PERFORMS ONE STEP OF INTEGRATION BY THE
C MERSON'S MODIFIED RUNGE-KUTTA METHOD. IT USES FIVE SUBSTITUTIONS
C INTO THE DIFFERENTIAL-EQUATIONS AND GIVES AN ESTIMATE OF THE
C LOCAL TRUNCATION ERROR, WHICH CAN BE USED FOR THE ADJUSTMENT
C OF THE STEP-SIZE
C
C DERIW = DUMMY-NAME FOR SUBROUTINE FOR THE EVALUATION
C OF THE RIGHT HAND SIDES OF THE DIFF.-EQUATIONS
C N = NUMBER OF THE EQUATIONS IN THE SYSTEM
C H = STEPSIZE
C X = INDEPENDENT VARIABLE
C Y(I) = DEPENDENT VARIABLE
C ERR(I) = LOCAL TRUNCATION ERROR
C
C THE COMPUTED VALUES OF X AND Y ARE SAVED IN DOUBLE-PRECISION
C FOR THE USE AS INITIAL-VALUES IN THE NEXT STEP
C
C EACH TIME WHEN A NEW INTEGRATION SEQUENCE IS STARTED, THE
C SUBROUTINE MUST BE CALLED WITH H=0 AND THE INITIAL-VALUES OF X AND
C Y(I).
C

```



```
300 YY(I) = Y(I) + (P1(I)-P3(I)+P4(I))*H2
    XX = X+H
    CALL DERIW (XX,YY,P5)
    DO 320 I=1,N
C *** DU = SINGLE PRECISION INCREMENT IN DEPENDENT VARIABLE
    DU = (P1(I) + P4(I) + P5(I))*H6
    DUDP = DBLE (DU)
    YLDP(I) = YLDP(I) + DUDP
    Y(I) = SNGL (YLDP(I))
320 CONTINUE
    XLDP = XLOP + HDP
    X = SNGL (XLDP)
    RETURN
    END
```

Contrails

SECTION V SOLUTION OF EXAMPLE PROBLEMS

The following three problems typify the extended use of the SP computer program. The first case illustrates the program's capability to accept torsion. The second and third cases illustrate the effects of pressure on the inside and outside surfaces of a cylindrical shell respectively.

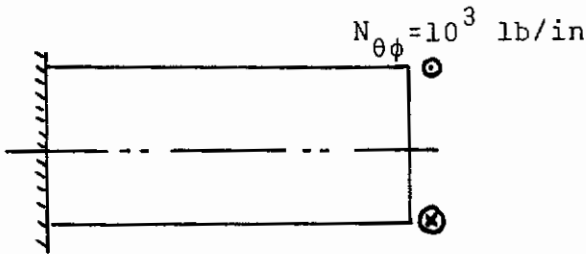


Figure 4
Torsional Problem

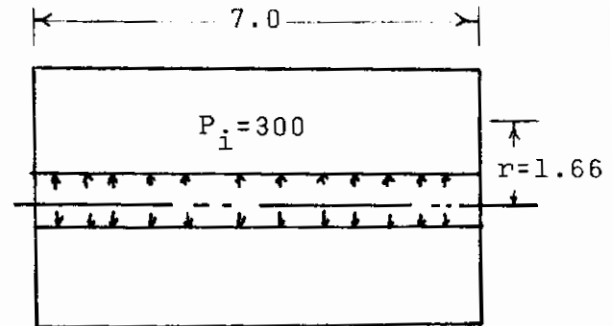


Figure 5
Internal Pressure Loading

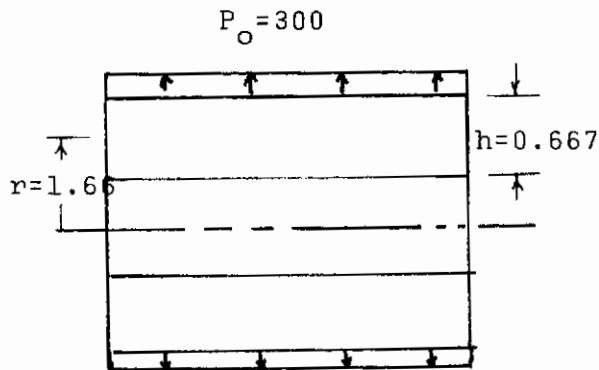


Figure 6
External Pressure Loading

AFFDL-TR-71-116

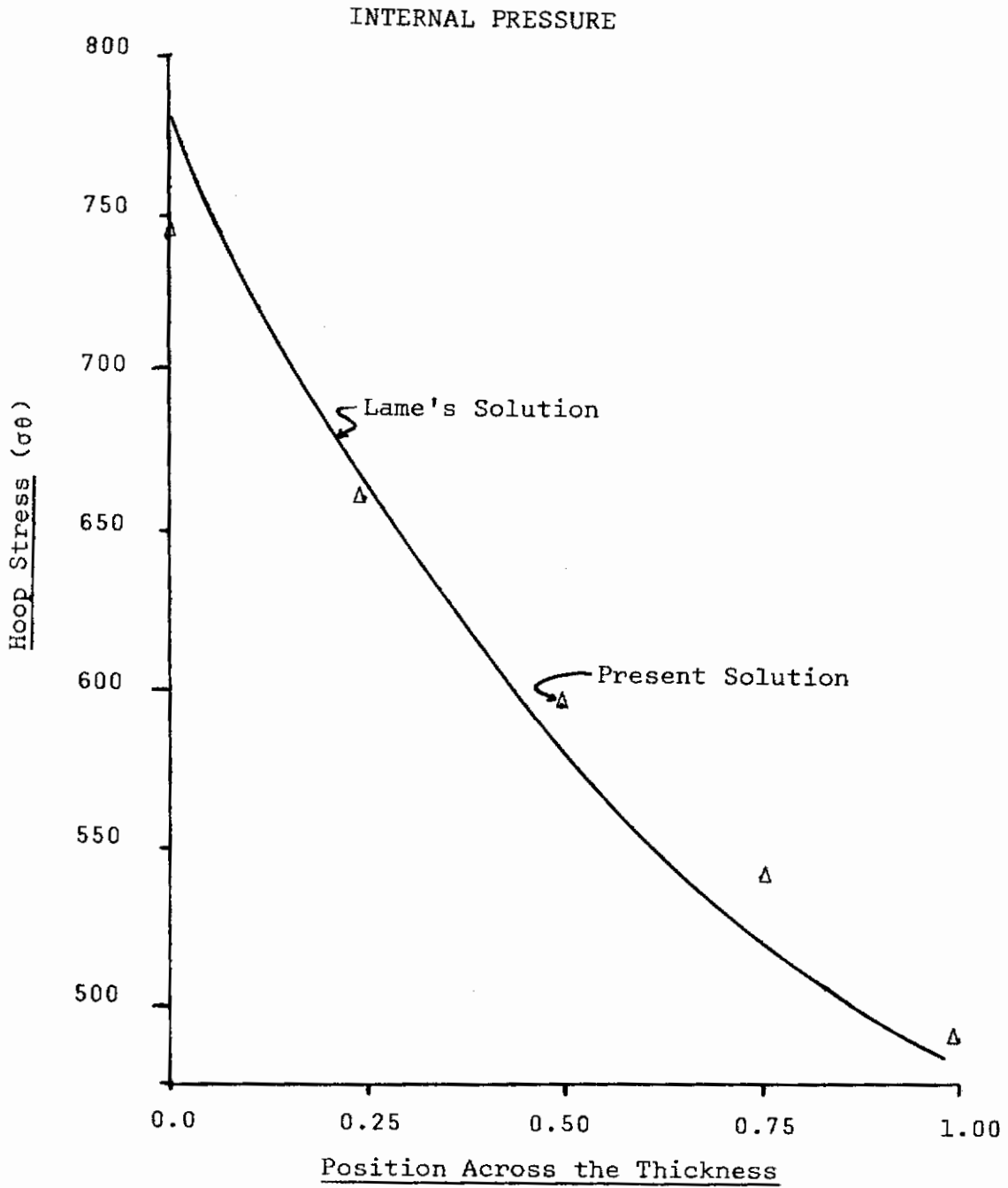


Figure 7
Hoop Stress Versus Position Across Thickness
(Internal Pressure Loading)

EXTERNAL PRESSURE

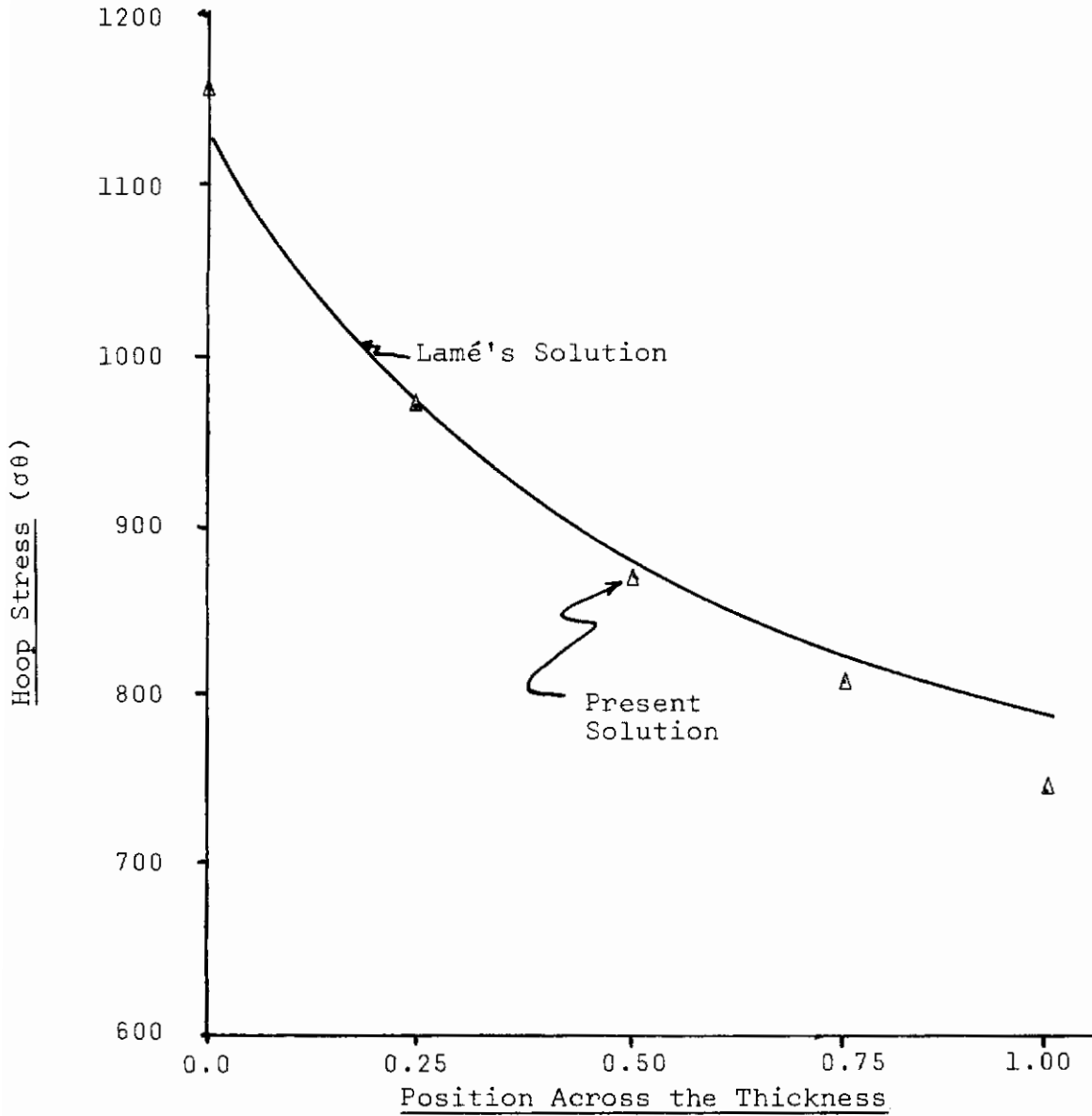


Figure 8

Hoop Stress Versus Position Across Thickness
(External Pressure Loading)

The analysis of a circular cylinder subjected to a torsional end moment is known from elementary beam theory. The angular twist is linear with the meridional length and the shear stress is constant. Problem one illustrates this solution and is trivial.

Problems two and three illustrate the effects of a thick shell solution. For a free circular cylinder, the Lamé solution away from the ends is given by:

for internal pressure:

$$\sigma_{\theta} = P_i \frac{a^2(b^2+r^2)}{r^2(b^2-a^2)}$$

for external pressure:

$$\sigma_{\theta} = -P_o \frac{b^2(a^2+r^2)}{r^2(b^2-a^2)}$$

The Lamé solution assumes $\sigma_r + \sigma_{\theta}$ is invariant through the thickness. Since the shell equations are based on the assumption that $\sigma_r = 0$, a discrepancy in the two theories are known at the onset. Even so, the numerical deviation is less than ten percent as indicated in the two solutions.

AFFDL-TR-71-116

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AFFDL-TR-71-116

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AFFDL-TR-71-116

APPENDIX A
TKSHL COMPUTER PROGRAM LISTING

AFFDL-TR-71-116

PROGRAM	TKSHL	THICK	DATE
05	PROGRAM TKSHL *(INPUT,OUTPUT,PUNCH,TABE6=INPUT,TABE7=OUTPUT, TABE7=PLUNCH) COMMON/RUN/NDE,S,Y(10),YD(10),HR,J9,JMAX,M9,XOUT,IFREQ,DUMM(33), LIBCL,KORTN,ERP COMMON/GARR/ D(7200),DM(909) COMMON/XDUM/JTYPE,IBRM 5 READ(5,100) ITYPE,JTYPE 100 FURMAT(A6,4X,AB)	THICK FEB6 THICK THICK THICK FEB6 FEB6	2 1 4 5 2 3 4
10	C REMOVE THIS CARD AND THE NEXT THREE CARDS IF RUNNING ON IBM EQUIP. IF (EOF(S)) 3333,3334 3333 CALL EXIT 3334 CONTINUE	THICK THICK JUN16 JUN16	9 1 11 12
15	999 FURMAT(5X,46,5X,AB) IF (ITYPE.EQ.6HSTOP) CALL EXIT KORTN=0 IF (ITYPE.EQ.6HTHICK) KORTN=1 IF (KORTN) 2,1,2	THICK THICK THICK THICK	3 13 14 15 16
20	1 CALL MAIN(D,100,8,9,DM,101,9) GO TO 5 2 CALL MAIN(D,65,10,11,DM,66,11) GO TO 5 END	THICK THICK THICK THICK	17 18 19 20 21

SYMBOLIC REFERENCE MAP

ENTRY POINTS
6076 TKSHL

AFFDL-TR-71-116

VARIABLES	SN	TYPE	RELOCATION	DM	ERP	REAL	ARRAY	GARB
0	0	REAL	ARRAY	16040	77	REAL	ARRAY	GARB
34	DUMM	REAL	ARRAY			REAL		RUN
26	HH	REAL	RUN	75	IACL	INTEGER		RUN
1	IURM	INTEGER	XDUM	33	IFREQ	INTEGER		RUN
6206	I	INTEGER	XDUM	30	JMAX	INTEGER		RUN
0	JTYPE	INTEGER	RUN	27	J9	INTEGER		RUN
76	KORTN	INTEGER	RUN	31	M9	INTEGER		RUN
0	NDE	INTEGER	RUN	1	S	REAL		RUN
32	XOUT	REAL	RUN	2	Y	REAL	ARRAY	RUN
14	YO	REAL	ARRAY			REAL		RUN

FILE NAMES	MODE	2022	OUTPUT	4044	PUNCH	0	TAPES	FMT
0	INPUT	FMT	2022	OUTPUT	4044	PUNCH		
2022	TAPE6	FMT	4044	TAPE7				

EXTERNALS	TYPE	ARGS
EOF	REAL	1
MAIN	REAL	7
		EXIT
		0

STATEMENT LABELS

0	1	INACTIVE	6137	2	6077	5
6165	100	FMT	6167	999	FMT	
6112	3334					0 3333
						INACTIVE

COMMON BLOCKS	LENGTH
RUN	64
GARB	8109
XDUM	2

STATISTICS

PROGRAM LENGTH	1218	81
BUFFER LENGTH	60668	3126
COMMON LENGTH	177578	8175

AFFDL-TR-71-116

05	C	ARE DENOTED BY AN * IN COLUMN 80. ADDED CARDS ARE DENOTED BY A SEQUENCE NUMBER IN COLUMN 80.	THICK	26
05	C	SEQUENCE NUMBER IN COLUMN 80.	THICK	27
05	C	STATIC PROGRAM FOR STRESS ANALYSIS OF SHELLS OF REVOLUTION	THICK	28
05	C	DIMENSION D(IJC,JRC,KRC),DM(IJSY,JSY)	THICK	29
10	C	COMMON/RUN/ NDE(5),Y(10),HHS,J9,JMAX,J9,XOUT,IFREQ,DUMM(33),	FEB6	31
10	C	COMMON/RUN/ JTYPE,IRRM	THICK	32
10	C	COMMON/MAIL/ B11(20,4),B12(20,4),B22(20,4),B66(20,4),B44(20,4),	THICK	33
15	C	COMMON/XMIT/ IRR,ISH,ANXLD,OMSG,HIIT,RI,R2,R3,SKN,CXS,INDEX,PNA,PL,	THICK	34
15	C	IPC,T0,T1,MH1,RH2,RH3,NBR,C11,C12,C22,C66,E11,E12,E22,E66,D11,D12,	THICK	35
15	C	2022,066,H44,H55,H66,K11,K12,XK22,XE66,F11,F12,F22,MH11,MH12,MH22	THICK	37
15	C	3,EMF,EMTA,MH11,MH12,MH22,M21,G12,G21	THICK	38
20	C	COMMON/GEOM/ SI(20),SX(20),XPR(20),ING(20),IPAR(20),Z(20),ISS(20),	THICK	39
20	C	ITRI(10,10),TR2(10,10),NTP(20),DSC(20,10),MLY(20),ZLY(20,5),IA(10),	THICK	40
20	C	ZIB(10,4),ALFL ,IR(10,10,4),IL(10,10),ALFR(4)	THICK	41
20	C	COMMON/THICK/ XNT,XMT,XNTP,XMTP,GPZ,GTZ,QT,CPP,EPP,CPT,DLP,EPT,	THICK	42
20	C	IGMP,DLT,GMT	THICK	43
25	C	COMMON/TRIA/ GA(5),GB(5,4),ITP(100),NH,NF,NPL ,NFP,NPRT	THICK	44
25	C	COMMON/ADAM/ HDMAX,INTG,XLAST	THICK	45
25	C	DIMENSION I0(33)	THICK	46
25	C	EQUIVALENCE (I0,DUMM)	THICK	47
30	C	DATA IN/3H IN,IOR/BHTORSION /,IOUT/3HOUT/	JUN16	4
30	C	10 READ (5,19) IHRM,ISTK,NBR,NKI,IV8,NPRT,ERP	THICK	48
30	C	REMOVE THIS CARD AND THE NEXT THREE CARDS IF RUNNING ON IBM EQUIP.	THICK	49
30	C	IF(EOF(5)) 3333,3334	THICK	5
35	C	3333 RETURN	FEB6	7
35	C	3334 CONTINUE	THICK	52
35	C	19 FORMAT(6I5,E10.2)	THICK	53
35	C	20 FORMAT (16I5)	THICK	54
35	C	IF(IHRM) 30,30,50	THICK	55
40	C	30 WRITE (9,40) IHRM	THICK	56
40	C	40 FORMAT (1H0,50H1BRM#0 WHICH INDICATES END OF JOB, EXIT IS CALLED)	THICK	57
40	C	CALL EXIT	THICK	58
40	C	50 CONTINUE	THICK	59
40	C	IF(ERP.LE.0.0) ERP=1.0E-05	THICK	60
40	C	K100=100	THICK	61
40	C	K4=4	THICK	62
45	C	K8=8	THICK	63
45	C	IF(KORTN) 51,52,51	THICK	64
45	C	51 K4=5	THICK	65
45	C	K8=10	THICK	66
50	C	221 K100=65	THICK	67
50	C	52 XLD=0.0	THICK	68
50	C	IHL=0	THICK	69
50	C	INXC=1	THICK	70
55	C	WRITE (6,60)	THICK	71
55	C	60 FORMAT (1H1, 26X, 67HSTRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUT	THICK	72
55	C	ION UNDER STATIC LOADS	THICK	73
55	C	220X,70HWRIGHT PATTERSON AIR FORCE BASE FLIGHT DYNAMICS LABORATORY	THICK	74
55	C	3VERSION, 22 JULY 1968/ 15X,	THICK	75

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4      88H MODIFIED PROGRAM TO ACCOUNT FOR TRANSVERSE SHEAR DEF THICK 76
      INFORMATION --WRITTEN BY R.L. CIERLEY/ 66HANAMET LABORATORIES INC. THICK 77
5      27X, 1 OCT.1969) THICK 78
6      BERKELEY, CALIFORNIA THICK 79
      WRITE (6,70) IRR,NBR,NXT THICK 80
70     FORMAT (1H0,20X,18MSTATIC ANALYSIS ,1,8H PARIS=,13, THICK 81
      11H BRANCHES=,13,21H NUMBER OF SUBCASES=,13) THICK 82
      IF(NBR=3) 100,100,80 THICK 83
65     80 WRITE (6,90) NBR THICK 84
      90 FORMAT (1H0,13,38H BRANCHES EXCEED ALLOWED MAXIMUM OF 3) THICK 85
      CALL EXIT THICK 86
      100 CONTINUE THICK 87
      IF(11HM=20) 130,130,110 THICK 88
      110 WRITE (6,120) IRRM THICK 89
70     120 FORMAT (1H0,13,36H PARTS EXCEED ALLOWED MAXIMUM OF 20) THICK 90
      CALL EXIT THICK 91
      130 CONTINUE THICK 92
      NBR=NBR+1 THICK 93
75     READ (5,140) ALFL, (ALFR(I),I=1,NBR) THICK 94
      140 FORMAT (5F10.5) THICK 95
      WRITE (6,150) ALFL, (ALFR(I),I=1,NBR) THICK 96
      150 FORMAT (1H0,51HANGLES OF ROTATION OF BOUNDARY CONDITIONS ARE ALFL THICK 97
      1, , 412.5, 4H ALFRS=, 4E12.5) THICK 98
      IGBT=0 THICK 99
80     UV 220 I=1,IBRM THICK 100
      IBNEI THICK 101
      WRITE (6,160) I THICK 102
      160 FORMAT (1H0,10X,7HPART NO,13) THICK 103
      READ (5,170) THICK 104
85     1 S(I),SX(I),IPAR(I),ING(I),ISS(I),NTP(I),MLY(I) THICK 105
      170 FORMAT (2F10.5,5I5) THICK 106
      1 S(I),SX(I),IPAR(I),ING(I),ISS(I),NTP(I),MLY(I) THICK 108
      180 FORMAT (1H0,3HSI=,E12.5,6H SX=,E12.5, 8H IPAR=,13,7H ING=,13 THICK 109
      1,14H SHELL TYPE,12, 7H NTP=,12, 14H LAYERS MLY=,12) THICK 110
      IGBT=IGBT+IPAR(I) THICK 111
      ISH=ISS(I) THICK 112
      INDEA=1 THICK 113
      CALL INPUT THICK 114
      CALL ORIH0 THICK 115
      IF(MLY(I)-4) 210,210,190 THICK 116
      190 WRITE (6,200) MLY(I),I THICK 117
      200 FORMAT(1H0,13,19H LAYERS IN PART NO,13,22H EXCEED ALLOWED MAX 4) THICK 118
      CALL EXIT THICK 119
      210 CONTINUE THICK 120
      220 WRITE (6,1830) THICK 121
      222 IF(IGCT-K100) 250,250,230 THICK 122
      230 WRITE (6,240) IGCT ,K100 THICK 123
      240 FORMAT (1H0, 13, 37H SEGMENTS EXCEED ALLOWED MAXIMUM OF ,14) THICK 124
      CALL EXIT THICK 125
      250 CONTINUE THICK 126
      260 LU=0 THICK 127
      HEAD (5,20) NX,NPCH THICK 128
      NPCH=0 MEANS PRESTRESS IS NOT PUNCHED, NPCH=1 MEANS IT IS PUNCHED THICK 129
      NIN=0 THICK 130

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AFFDL-TR-71-116

PAGE 3

CDC 6600 FIN V3.0-P243 OPT#1 06/19/71 13.01.44

AFFDL-TR-71-116

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SUBROUTINE MAIN
WRITE (6,60)
IF(NK) 290,270,270
270 WRITE (6,280) INAC,NK
280 FORMAT (1H0,30X,10HSUBCASE NO,13,27H FOR FOURIER HARMONIC COS,
115 112, 6H THETA)
GO TO 310
290 IS=NA
WRITE (6,300) INXC,I
300 FORMAT (1H0,30X,10HSUBCASE NO,13,27H FOR FOURIER HARMONIC SIN,
120 112, 5H ITHETA)
310 CONTINUE
ANENK
IF (NK.EQ.0.AND.JTYPE.NE.IOR) K=3
HEAD (5,320) (IA(I),GA(I),I=1,K4)
125 320 FORMAT (5(I5,F10.5))
WRITE (6,330)
1((IA(I),GA(I),I=1,K4)
330 FORMAT(1H0,39HBOUNDARY CONDITIONS AT STARTING EDGE ,5(I5,E12.5))
DO 340 K=1,NBR
130 READ (5,320)
1(I(I), ,GB(I,K),I=1,K4)
DO 339 I=1,K4
339 1B(I,K+K)=I(I)
IF(K-1) 340,340,360
135 340 WRITE (6,350) (I(I) ,GB(I,K), I=1,K4)
350 FORMAT(1H0,39HBOUNDARY CONDITIONS AT FINAL EDGE ,5(I5,E12.5))
GO TO 380
360 JK=1
WRITE (6,370) J,
140 1(JU(I) ,GB(I,K),I=1,K4)
370 FORMAT(1H0,36HBOUNDARY CONDITION AT BRANCH EDGE NO,13,5(I5,E12.5))
380 CONTINUE
IF (NK.NE.0.OR.JTYPE.EQ.IOR) GO TO 410
390 NDE=5
145 DO 400 L=7,10
IA(L)=0
DO 400 K=1,NBR
400 IB(L,K)=0
150 GO TO 420
410 NDE=10
IF (KURT) 420,411,420
411 NDE=R
DO 412 L=9,10
IA(L)=0
DO 412 K=1,NBR
155 412 IB(L,K)=0
420 N=NDE/2
M=NH+1
L=1
160 DO 460 IK=M,NDE
DO 440 N=L,NDE
DO 430 J=L,NH
IF (IA(J)=N) 430,440,430
430 CONTINUE
GO TO 450
165 THICK 131
THICK 132
THICK 133
THICK 134
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THICK 136
THICK 137
THICK 138
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THICK 142
FEB6 8
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FEB6 9
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AFFDL-TR-71-116

440	CONTINUE	THICK	186
450	I=I+1	THICK	187
460	IA(I,K)=N	THICK	188
	DO 500 K=1,NBR	THICK	189
	L=1	THICK	190
	DO 520 JK=1,NH	THICK	191
	DO 490 N=L,NDE	THICK	192
	DO 470 J=M,NDE	THICK	193
	IF (IB(J,K)-N) 470,480,470	THICK	194
175	470 CONTINUE	THICK	195
	GO TO 490	THICK	196
480	CONTINUE	THICK	197
490	L=L+1	THICK	198
500	IA(I,K)=N	THICK	199
	DO 510 J=1,20	THICK	200
	DO 510 J=1,Kd	THICK	201
510	OSC(I,J)=0.0	THICK	202
	DO 520 I=1,K8	THICK	203
	Y(I)=0.0	THICK	204
	Z(I)=0.0	THICK	205
185	DO 520 L=1,K3	THICK	206
	TR1(I,J)=0.0	THICK	207
520	TR2(I,L)=0.0	THICK	208
	CALL BCOND	THICK	209
190	DO 530 J=1,NDE	THICK	210
	K=IAL(J)	THICK	211
	DO 530 I=1,NDE	THICK	212
530	DM(I,J)=TL(I,K)	THICK	213
	DO 540 I=1,NDE	THICK	214
	DO 540 J=1,NDE	THICK	215
540	TL(I,J)=DM(I,J)	THICK	216
	DO 560 K=1,NBR	THICK	217
	DO 550 J=1,NDE	THICK	218
	DO 550 J=1,NDE	THICK	219
200	L=IB(I,K)	THICK	220
	550 DM(I,J)=TL(L,J,K)	THICK	221
	DO 560 J=1,NDE	THICK	222
	DO 560 J=1,NDE	THICK	223
560	TR(I,J,K)=DM(I,J)	THICK	224
205	IF(NDE=8) 570,580,581	THICK	225
570	TR1(3,5)=1.0	THICK	226
	TR1(5,6)=1.0	THICK	227
	TR2(5,3)=1.0	THICK	228
	TR2(6,6)=1.0	THICK	229
210	GO TO 590	THICK	230
580	TR1(3,7)=1.0	THICK	231
	TR1(4,5)=1.0	THICK	232
	TR1(7,8)=1.0	THICK	233
	TR1(8,6)=1.0	THICK	234
215	TR2(5,4)=1.0	THICK	235
	TR2(6,8)=1.0	THICK	236
	TR2(7,3)=1.0	THICK	237
	TR2(8,7)=1.0	THICK	238
	GO TO 590	THICK	239
220	581 TR1(3,7)=1.0	THICK	240

AFFDL-TR-71-116

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TR1(4,5)=1.0 THICK 241
TR1(5,9)=1.0 THICK 242
TR1(8,8)=1.0 THICK 243
TR1(9,6)=1.0 THICK 244
225 TR1(10,10)=1.0 THICK 245
TR2(5,4)=1.0 THICK 246
TR2(6,9)=1.0 THICK 247
TR2(7,3)=1.0 THICK 248
230 TR2(8,8)=1.0 THICK 249
TR2(9,5)=1.0 THICK 250
TR2(10,10)=1.0 THICK 251
590 NPL=NDP+1 THICK 252
600 I8=1 THICK 253
ISIR=1STK THICK 254
NSIR=0 THICK 255
NPF=0 THICK 256
NFP=2 THICK 257
NFI=1 THICK 258
240 610 I=IBR THICK 259
WRITE (6,1830) IBR,INXC THICK 260
WRITE (6,620) IBR,INXC THICK 261
620 FORMAT (1H,20X,17HLOADS FOR PART NO,13,12H SUBCASE NO,13) THICK 262
ICYL=0 THICK 263
640 245 ISH=ISS(I) THICK 265
READ (5,140) (DSC(I,J),J=2, K8,2) THICK 266
IF (KURT) 629,628,629 THICK 267
629 WRITE(6,631)(DSC(I,J),J=2,K8,2) THICK 268
631 FORMAT(1H,39HRING LOADS AT END OF THIS PART ARE Q,E12.5, THICK 269
17H MPH1=E12.5, 7H MPH1=E12.5, 7H MPH1=E12.5, 7H MPH1=E12.5, THICK 270
2 E12.5) THICK 271
60 IO 627 THICK 272
628 WRITE (6,630) (DSC(I,J),J=2,K8,2) THICK 273
630 FORMAT (1H,39HRING LOADS AT END OF THIS PART ARE Q,E12.5, THICK 274
17H MPH1=E12.5, 7H MPH1=E12.5, 4H N,E12.5) THICK 275
627 S=SI(I) THICK 276
INDEX=2 THICK 277
CALL INPUT THICK 278
INDEX=3 THICK 279
CALL INPUT THICK 280
CALL ORTHO THICK 281
INDEX=4 THICK 282
IF (INPT) /30,730,640 THICK 283
640 CONTINUE THICK 284
JJ=JLY(IBR)+1 THICK 285
WRITE (6,650) IBR THICK 286
650 FORMAT (1H, 20X, 28HAT BEGINNING AND END OF PART,13, THICK 287
115H PARAMETERS ARE ) THICK 288
WRITE (6,1830) THICK 289
DO 720 K=1,2 THICK 290
WRITE (6,650) R1,R2,R3,SKN,CXS THICK 291
WRITE (6,670) PN,PL, PC,T0,T1 THICK 292
WRITE (6,680) C11,C12,C22,C56,E11,E12 THICK 293
WRITE (6,690) E22,E66,D11,D12,D22,D66 THICK 294
WRITE (6,700) H11,H12,H22,H21,G12,G21 THICK 295

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IF (KORTM) 651,652,651
651 WRITE (6,701) XH1,XH12,XH22,F11,F12,F22
   1 ,XK11,XK12,XK22,H55,H66
652 WHITE (6,710) (ZLY(1BR,J), J=1,JJ)
701 FORMAT (1H,5X)H1= E12.4,5X)H12= E12.4,5X)H22= E12.4,
   1 5H F11= E12.4, 5H F12= E12.4, 5H F22= E12.4,5H)H55= E12.4,5H)H66= E12.4,
   2 5H)K12= E12.4,5X)K22= E12.4,5H)H55= E12.4,5H)H66= E12.4)
WHITE (6,1330)
660 FORMAT (1H,5H H1=,E12.4,5H H2=, E12.4,5H H3=,E12.4,
   15H)SX=, E12.4, 5H)CX=, E12.4)
670 FORMAT (1H,5H PN=, E12.4,5H PL=, E12.4,5H PC=,E12.4,
   15H)IO=,E12.4,5H)II=, E12.4)
680 FORMAT (1H, 5H C11=,E12.4, 5H C12=,E12.4, 5H C22=, E12.4,
   15H)C6=, E12.4, 5H)E12.4, 5H)E12.4, 5H)E12.4, 5H)E12.4, 5H)E12.4,
   15H)O12=, E12.4, 5H)O22=, E12.4, 5H)O66=, E12.4)
700 FORMAT (1H, 5H H11=, E12.4, 5H H12=, E12.4, 5H H22=, E12.4,
   15H)H21=, E12.4, 5H)G12=, E12.4, 5H)G21=, E12.4)
710 FORMAT (1H0, 7HZ) ARE=,10E12.4)
295 SSSX(I)
   CALL INPUT
   CALL ORTHO
720 CONTINUE
730 CONTINUE
300 K=IPAR(I)+NF-1
   DO 740 J=NF,K
740 IIP(J)=0
   IF (NTP(1BR)) 780,780,750
750 DO 760 J=NF,K
760 IIP(J)=1
770 IIP(K)=2
780 CONTINUE
   PARTS=IPAR(1BR)
   SMAX=(SX(1BR) - SI(1BR))/PARTS
   SZERO=SI(1BR)
   XINT=ING(1BR)
   XPR(1BR)=SMAX/XINT
   NFF=NF-IPAR(1BR)
   SMAX=SZERO+SMAX
315 XOUT=SMAX
   IF (NPT-1) 810,810,790
790 WRITE (6,800) IBR,IPAR(1BR)
800 FORMAT(1H0,62H)INITIAL VALUE INTEGRATIONS (COLUMNS OF Y(X) MATRIX)
   1 OF PART NO,13,5H OVER,13,16H SEGMENTS FOLLOW)
810 CONTINUE
820 J=0
325 XLD=0.0
   XLD=J.1
   DO 840 I=1,K8
840 Y(I)=0.0
850 Y(J)=1.0
860 IF (NPT) 880,880,870

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AFFDL-TR-71-116

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SUBROUTINE	MAIN	CDC 6600 FTN V3.0-P243 OPT=1	06/19/71	13.01.44:	PAGE	7
335	870 WRITE (6,950) 880 CONTINUE S=SZERO HH=0.01*SMXX INIG=0 IBCL=0 IF(LCYL) 910,910,890 890 DO 900 I=1,NDE K=L+K8 340 900 U(MF,I,J)=OM(K,I,J) GO TO 930 910 CALL RUNGE DO 920 J=1,NDE K=L+K8 DM(K,J)=Y(I) 345 920 U(MF,I,J)=Y(I) 930 IF(NPRT=1) 970,970,960 940 WRITE (6,950) 350 950 JSMAX, (U(MF,I,J),I=1,NDE) 950 FOMAT (1H, F9.3,10E12.5) WRITE (6,960) IBCL 960 FOMAT (1H, 7HPPOINTS=,I5) 970 CONTINUE IF (J=NDE) 830,980,990 355 980 XLDEL=0 ICYL=0 GO TO 830 360 990 IF(NF=NF) 1010,1000,1000 1000 IF(IPAR(1BR)-1) 1010,1010,1110 1010 IF(IRS) 1040,1080,1020 1020 IAS=0 S=SI(1HR) CALL INPUT 365 TR2(1,1)=CXS TR2(1,2)=SXN TR2(3,1)=-SXN TR2(3,2)=CXS 370 1030 TR2(2,4)=CXS TR2(2,5)=SXN TR2(4,4)=-SXN TR2(4,5)=CXS 375 1040 TR2(2,5)=CXS TR2(2,6)=SXN TR2(4,5)=-SXN TR2(4,6)=CXS GO TO 1050 380 1041 TR2(2,6)=CXS TR2(2,7)=SXN TR2(4,6)=-SXN TR2(4,7)=CXS 385 1050 DO 1060 I=1,NDE DO 1060 K=L+K8 THICK 351 THICK 352 THICK 353 THICK 354 THICK 355 THICK 356 THICK 357 THICK 358 THICK 359 THICK 360 THICK 361 THICK 362 THICK 363 THICK 364 THICK 365 THICK 366 THICK 367 THICK 368 THICK 369 THICK 370 THICK 371 THICK 372 THICK 373 THICK 374 THICK 375 THICK 376 THICK 377 THICK 378 THICK 379 THICK 380 THICK 381 THICK 382 THICK 383 THICK 384 THICK 385 THICK 386 THICK 387 THICK 388 THICK 389 THICK 390 THICK 391 THICK 392 THICK 393 THICK 394 THICK 395 THICK 396 THICK 397 THICK 398 THICK 399 THICK 400 THICK 401 THICK 402 THICK 403 THICK 404 THICK 405					

AFFDL-TR-71-116

AFFDL-TR-71-116

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DM(I,K)=0.0
DO 1050 I=L,NDE
1050 DM(I,K)=DM(I,K)+D(NF,I,L)*TR2(L,K)
DO 1070 I=L,NDE
DO 1070 K=L,NDE
1070 D(NF,I,K)=DM(I,K)
1080 IF(IPAR(IHR)-1) 1110,1110,1090
1090 NF=NF+1
NF=NF+1
SZERO=SMAX
SMAX=SMAX+SZERO
XOUT=SMAX
IF(IISH+2) 820,1100,820
1100 ICYL=1
GO TO 820
1110 DO 1120 I=L,NDE
1120 D(NF,I,NPL)=D(NF,I,NPL)-DSC(IHR,I)
1130 IF(IHR-IHRM) 1140,1220,1220
1140 IF(IIP(NF)-2) 1150,1210,1210
1150 SSK(IHR)
CALL INPUT
TR1(1,1)=CXS
TR1(1,3)=SXN
TR1(2,1)=SXN
TR1(2,3)=CXS
IF(INDE=8) 1160,1170,1171
1160 TR1(4,2)=CXS
TR1(4,4)=SXN
TR1(5,2)=SXN
TR1(5,4)=CXS
GO TO 1180
1170 TR1(5,2)=CXS
TR1(5,4)=SXN
TR1(6,2)=SXN
TR1(6,4)=CXS
GO TO 1180
1171 TR1(6,2)=CXS
TR1(6,4)=SXN
TR1(7,2)=SXN
TR1(7,4)=CXS
1180 DO 1190 I=L,NDE
DO 1190 K=L,NPL
DM(I,K)=0.0
DO 1190 L=L,NDE
DM(I,K)=DM(I,K)+TR1(I,L)*D(NF,L,K)
DO 1200 I=L,NDE
DO 1200 K=L,NPL
1200 D(NF,I,K)=DM(I,K)
1210 IHR=IHR+1
NF=NF+1
NF=NF+1
IHR=I
GO TO 610
1220 DO 1230 I=L,NDE
DO 1230 J=L,NDE

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THICK 406
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SUBROUTINE	MAIN	CDC 6600 FTN V3.0-P2*3 OPT*1	06/19/71	13.01.44	PAGE	9
	DM(I,J)=0.0			THICK	461	
	DO 1230 L=1,NDE			THICK	462	
	1230 DM(I,J)=DM(I,J)+D(1,I,L)*TL1(L,J)			THICK	463	
	DO 1240 J=1,NDE			THICK	464	
445	DO 1240 J=1,NDE			THICK	465	
	1240 D(I,I,J)=DM(I,J)			THICK	466	
	KF=1			THICK	467	
	IA=NF			THICK	468	
	DO 1290 K=1,NHR			THICK	469	
450	DO 1250 I=1,NDE			THICK	470	
	DO 1250 J=1,NPL			THICK	471	
	DM(I,J)=0.0			THICK	472	
	DO 1250 L=1,NDE			THICK	473	
	1250 DM(I,J)=DM(I,J)+TR(I,L,K)*D(IX,I,J)			THICK	474	
455	DO 1260 I=1,NDE			THICK	475	
	DO 1260 J=1,NPL			THICK	476	
	1260 D(IX,I,J)=DM(I,J)			THICK	477	
	DO 1270 IX=KF,NF			THICK	478	
	JJ=IX			THICK	479	
460	IF(IIP(IX)-2) 1270,1280,1280			THICK	480	
	1270 CONTINUE			THICK	481	
	1280 IX=JJ			THICK	482	
	KJ=IX+1			THICK	483	
	1290 CONTINUE			THICK	484	
465	IF(INPT) 1320,1320,1300			THICK	485	
	1300 CONTINUE			THICK	486	
	1310 FORMAT(1H,5O12)			THICK	487	
	WRITE (6,1310) (ITP(I),I=1,NF)			THICK	488	
470	WRITE (6,1330)			THICK	489	
	1320 CONTINUE			THICK	490	
	CALL TRIANG(U,IRC,JRC,KRC,DM,ISY,JSY)			THICK	491	
	IF(INPT) 1370,1370,1330			THICK	492	
475	WRITE (6,1340)			THICK	493	
	1330 CONTINUE			THICK	494	
	REF,NFP,KN, ERP			THICK	495	
	1340 FORMAT(1H0,5X,4HNEXT, 13, 41H ROWS ARE VARIABLES AT ENDPOINTS			THICK	496	
	1X,....X, 12, 12H OF SEGMENTS,			THICK	497	
	22X, 3HX=, F5, 172X,			THICK	498	
	WRITE (6,1430)			THICK	499	
480	DO 1350 I=1,NFP			THICK	500	10H ACCURACY=,E6.1)
	1350 WRITE (6,1360)			THICK	501	
	1360 FORMAT(1H 13HX=X,12,2X,10E12.5)			THICK	502	
485	1370 CONTINUE			THICK	503	
	IST=0			THICK	504	
	DO 1440 K=1,NF			THICK	505	
	IF(ITP(K)-1) 1440,1380,1400			THICK	506	
490	IF(IST) 1390,1390,1440			THICK	507	
	1390 IST=2			THICK	508	
	IFK			THICK	509	
	GO TO 1440			THICK	510	
	1400 IF(IST) 1410,1410,1420			THICK	511	
	1410 IFK			THICK	512	
495	1420 CONTINUE			THICK	513	
				THICK	514	
				THICK	515	

AFFDL-TR-71-116

SUBROUTINE	MAIN	CDC 6600 FIN V3.0-P243 OPT=1	06/19/71	13:01:44	PAGE	10
	DO 1430 J=1,NH		THICK	516		
	1430 DM(K+1,J)=DM(I,J)		THICK	517		
	1430 IS1=0		THICK	518		
500	1440 CONTINUE		THICK	519		
	DO 1450 I=1,NDE		THICK	520		
	Y(I)=0.0		THICK	521		
	DO 1450 J=1,NDE		THICK	522		
	1450 Y(I)=Y(I)+[L(I,J)+DM(I,J)]		THICK	523		
	DO 1460 I=1,NDE		THICK	524		
505	1460 DM(I,J)=Y(I)		THICK	525		
	1470 CONTINUE		THICK	526		
	WRITE (6,1480)		THICK	527		
	WRITE (6,1480) NX		THICK	528		
510	1480 FORMAT (1H0,20X,50HSIATIC SOLUTION AT POINTS ALONG MERIDIAN FOR WA		THICK	529		
	1VE NUMBER NX=,I3)		THICK	530		
	1480 IS1=1		THICK	531		
	KF=0		THICK	532		
	1480		THICK	533		
	JPA=IPAR(I)+1		THICK	534		
515	DO 1450 N=1,IBRM		THICK	535		
	ISM=ISS(N)		THICK	536		
	IBM=N		THICK	537		
	L=IBM(N)+1		THICK	538		
	S=S1(N)		THICK	539		
520	KF=IPAR(N)+KF		THICK	540		
	IF (IIP(KF)-1) 1540,1550,1550		THICK	541		
	1540 IF(N=1) 1562,1562,1563		THICK	542		
	1562 WRITE(6,1560)N		THICK	543		
	GO TO 1564		THICK	544		
525	1563 WRITE(6,1561)N		THICK	545		
	1564 IF(KORTN) 1488,1489,1488		THICK	546		
	1488 IF (ISTR=1) 1491,1511,1511		THICK	547		
	1491 WRITE(6,1501)		THICK	548		
530	1501 FORMAT(1H0,3A,1HS,8X,1HW,8X,5MU PHI,5X,7HU THETA, 8X,8HBETA PHI,2X		THICK	549		
	1,10HBETA THETA,9H LAYER NO,2X,7HSIG PHI,3X,9HSIG THETA,3X,7HSIG T		THICK	550		
	2P,3X,9HTAU PHI Z,12H 1AU THETA Z)		THICK	551		
	GO TO 1530		THICK	552		
	1511 WRITE(6,1502)		THICK	553		
535	1502 FORMAT(1H0,3A,1HS,9X,1HW,9X,5MU PHI,6X,7HU THETA,5X,8HBETA PHI,3X,		THICK	554		
	1,10HBETA THETA,3X,		THICK	555		
	110M PHI/UPHI,3X,16M N THETA/Q THETA,16M M PHI / N T PHI ,		THICK	556		
	216M * THETA/M T PHI)		THICK	557		
	GO TO 1530		THICK	558		
540	1489 IF (ISTR=1) 1510,1490,1490		THICK	559		
	1490 WRITE (6,1500)		THICK	560		
	1500 FORMAT (1H0,9H S ,2X,12H W , 12H , 12H		THICK	561		
	112H UPHI , 12H NPHI , 12H BPHI , 12H MPHI		THICK	562		
	2, 12H UTHETA , 12H N , 12H NTHETA , 10H MTHETA)		THICK	563		
545	GO TO 1530		THICK	564		
	1510 WRITE (6,1520)		THICK	565		
550	1520 FORMAT (1H0,9H S ,2X,12H W , 12H , 12H		THICK	566		
	112H UPHI , 12H NPHI , 12H BPHI , 12H MPHI		THICK	567		
	2, 12H UTHETA IN , 12H STHETA OUT , 12H SFITH IN , 12H SFITH OUT)		THICK	568		
555	1530 CONTINUE		THICK	569		
	GO TO 1580		THICK	570		

AFFDL-TR-71-116

SUBROUTINE	MAIN	CDC 6600 FTN V3.0-P243 OPT=1	06/19/71	13.01.44.	PAGE	11
555	1540 WHITE (6,1570) N IF (KORIN) 1541,1542,1541 1541 IF (ISTR=1) 1543,1544,1544 1544 WHITE (6,1502) GO TO 1559 1543 WRITE (6,1501) GO TO 1559	THICK 571 THICK 572 THICK 573 THICK 574 THICK 575 THICK 576 THICK 577				
560	1542 IF (ISTR=1) 1547,1546,1546 1546 WRITE (6,1500) GO TO 1559 1547 WHITE (6,1520) 1559 ISI=ISI+1 1560 FORMAT (1M0, 20H MAIN SHELL PART NO,13) 1561 FORMAT (1M1, 20H MAIN SHELL PART NO,13) 1570 FORMAT (1M1, 20H BRANCH SHELL PART NO,13) 1580 INDEX=3	THICK 578 THICK 579 THICK 580 THICK 581 THICK 582 THICK 583 THICK 584 THICK 585 THICK 586				AFFDL-TR-71-116
570	CALL INPUT CALL ORTHO IF (N=1) 1660,1660,1590 1590 IF (INST) 1600,1600,1660 1600 TR2(1,1)=CX5 TR2(1,2)=SXN TR2(3,1)=SXN	THICK 587 THICK 588 THICK 589 THICK 590 THICK 591 THICK 592 THICK 593				
575	1610 TR2(3,2)=CX5 IF (NDE=H) 1610,1620,1621 TR2(2,4)=CX5 TR2(2,5)=SXN TR2(4,4)=SXN TR2(4,5)=CX5	THICK 594 THICK 595 THICK 596 THICK 597 THICK 598 THICK 599				
580	GO TO 1630 1620 TR2(2,5)=CX5 TR2(2,6)=SXN TR2(4,5)=SXN TR2(4,6)=CX5	THICK 600 THICK 601 THICK 602 THICK 603 THICK 604				
585	GO TO 1630 1621 TR2(2,6)=CX5 TR2(2,7)=SXN TR2(4,6)=SXN TR2(4,7)=CX5	THICK 605 THICK 606 THICK 607 THICK 608 THICK 609				
590	1630 DO 1640 I=1,NDE Y(I)=0 DO 1640 J=1,NDE 1640 Y(I)=Y(I)+TR2(I,J)*DM(JPA,J) DO 1650 I=1,NDE 1650 DM(JPA,I)=Y(I) JPA=JPA+IPAR(N) 1660 XX=0	THICK 610 THICK 611 THICK 612 THICK 613 THICK 614 THICK 615 THICK 616 THICK 617				
600	INDEX=4 S=IPAR(N) DELS=0.1*(SX(N)-SI(N))/S SZERO=SI(N) IF (ISTR=1) 1700,1670,1670 1670 IF (NPK) 1700,1700,1680 1680 NTH=IPAR(N)+ING(N)+1 I=1	THICK 618 THICK 619 THICK 620 THICK 621 THICK 622 THICK 623 THICK 624 THICK 625				


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C FOLLOWING STATEMENT IS INTENDED TO PUNCH CARDS FOR PRESTRESS STATE THICK 626
WRITE (7,1690) N,NR,I THICK 627
1690 FORMAT (3I5) THICK 628
1700 CONTINUE THICK 629
DO 1740 K=IX,KF THICK 630
Z(1)=0.0 THICK 631
Z(6)=0.0 THICK 632
Z(7)=0.0 THICK 633
Z(11)=0.0 THICK 634
DO 1710 I=1,NDE THICK 635
Z(I)=DM(K*I) THICK 636
1710 Y(I)=DM(K*I) THICK 637
HM=DELS THICK 638
INIG=0 THICK 639
DO 1720 M=1,XL THICK 640
SZERO=SZERO+XPR(N)*XX THICK 641
S=SZERO THICK 642
XA=1.0 THICK 643
CALL INPUT THICK 644
CALL ORTHO THICK 645
XNR=XNR*2 THICK 646
SAR=SAR*X2 THICK 647
CARE=CX*S*2 THICK 648
ELH=H1-SAR THICK 649
106 ELJ=K1+S*H THICK 650
IF(KURIN) 1728,1711,1728 THICK 651
1711 ETH=XNR*Z(7)+CX*Z(3)+SAR*Z(1) THICK 652
BH=XNR*Z(1)+SAR*Z(7) THICK 653
HM=XNR*BH+CX*Z(5) THICK 654
EPS=XNR*Z(3)-CX*Z(7) THICK 655
HKA=-2.0*XNR*Z(5)-2.0*XNR*CX*Z(1)+CX*(ELH-SAR)*Z(7)+XNR*R1*Z(3) THICK 656
DEL=C11*D11-E11*E11 THICK 657
EN=Z(4)-C12*ETH-E12*HTH-(H11*10*H12*11) THICK 658
EM=Z(6)-E12*ETH-D12*HTH-(H12*10*G12*11) THICK 659
EF1=(EN*D11-EM*E11)/DEL THICK 660
EF1=(EN*D11-EM*E11)/DEL THICK 661
IF(1STR=1) 1720,1750,1750 THICK 662
1720 ZD=(Z(8)-(C66+SAR*E66)*EPS-(E66+SAR*D66)*HKA) THICK 663
E1F=EPS*ZU THICK 664
H1F=HKA+SAR*ZD THICK 665
Z(2)=Z(3) THICK 666
Z(3)=Z(7) THICK 667
Z(4)=Z(5) THICK 668
J=EMLY(I*BR) THICK 669
DO 1740 I=1,J THICK 670
J=I+1 THICK 671
Z(5)=B11(N,I)*(EF1+ZLY(N,I)*HFI)+B12(N,I)*(ETH+ZLY(N,I)*H11) THICK 672
1=B11(N,I)*A11(N,I)+B12(N,I)*A12(N,I)*(10+ZLY(N,I)*11) THICK 673
Z(6)=B11(N,I)*(EF1+ZLY(N,I)*HFI)+B12(N,I)*(ETH+ZLY(N,I)*H11) THICK 674
1=B11(N,I)*A11(N,I)+B12(N,I)*A12(N,I)*(10+ZLY(N,I)*11) THICK 675
Z(7)=B12(N,I)*(EF1+ZLY(N,I)*HFI)+B22(N,I)*(ETH+ZLY(N,I)*H11) THICK 676
1=B12(N,I)*A11(N,I)+B22(N,I)*A12(N,I)*(10+ZLY(N,I)*11) THICK 677
Z(8)=B12(N,I)*(EF1+ZLY(N,I)*HFI)+B22(N,I)*(ETH+ZLY(N,I)*H11) THICK 678
1=B12(N,I)*A11(N,I)+B22(N,I)*A12(N,I)*(10+ZLY(N,I)*11) THICK 679
1=B12(N,I)*A11(N,I)+B22(N,I)*A12(N,I)*(10+ZLY(N,I)*11) THICK 680

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AFFDL-TR-71-116

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Z(9) =B66(N,I)*(EIF+ZLY(N,I)*HIF) THICK 681
Z(10) =B66(N,I)*(EIF+ZLY(N,I)*HIF) THICK 682
WRITE (6,1730) S, (Z(1),Z(10)) THICK 683
1730 FORMAT (1A,1H,7F8.5,2X,10E12.5) THICK 684
665 1740 CONTINUE THICK 685
C THE FOLLOWING CARD CAUSES STRESSES TO BE DOUBLE SPACED THICK 686
C WHITE (6,1830) THICK 687
GO TO 1790 THICK 688
1750 Z(9)=C12*EFI+C22*ETH+E12*HFI+E22*HTH+(H22*TO+H21*T1) THICK 689
Z(10)= THICK 690
1 D12*HFI+D22*HTH+E12*EFI+E22*ETH+(H21*TO+D21*T1) THICK 691
WRITE (6,1730) S, (Z(1),Z(10)) THICK 692
IF (NPH) 1790,1790,1760 THICK 693
1760 IF (M-L) 1770,1790,1790 THICK 694
675 1770 WRITE (7,1780) S, Z(4),Z(9),Z(8) THICK 695
GO TO 1790 THICK 696
1728 XLU=1.0 THICK 697
PN=0.0 THICK 698
PC=0.0 THICK 699
CALL COEFF THICK 701
Z(2)=Z(3) THICK 702
Z(3)=Z(7) THICK 703
Z(4)=Z(5) THICK 704
Z(5)=Z(9) THICK 705
IF (ISTR=1) 1722,1751,1751 THICK 706
1722 JJ=MLY(18) THICK 707
DO 1721 I=1,JJ THICK 708
J=I+1 THICK 709
H=ZLY(N,I) THICK 710
HP=ZLY(N,J) THICK 711
S11=1.0+H*H1 THICK 712
S12=1.0+H*SXR THICK 713
S22=1.0+H*P*1 THICK 714
S21=1.0+H*P*SXR THICK 715
Z(6)=B11(N,I)*(EPP+H*CPP)/S11+B12(N,I)*(EPI+H*CPI)/S12 THICK 716
1-(B11(N,I)*A11(N,I)/S11 THICK 717
+B12(N,I)*A12(N,I)/S12 THICK 718
3 THICK 719
Z(11)= THICK 720
1 B11(N,I)*(EPP+H*CPP)/S22+B12(N,I)*(EPI+H*CPT)/S21 THICK 721
1-(B11(N,I)*A11(N,I)/S22 THICK 722
3 THICK 723
2 B12(N,I)*A12(N,I)/S21 THICK 724
3 THICK 725
Z(7)=B12(N,I)*(EPP+H*CPP)/S11+B22(N,I)*(EPI+H*CPT)/S12 THICK 725
1-(B12(N,I)*A11(N,I)/S11 THICK 726
3 THICK 727
2 B22(N,I)*A12(N,I)/S12 THICK 728
3 THICK 729
Z(12)=B12(N,I)*(EPP+H*CPP)/S22+B22(N,I)*(EPI+H*CPT)/S21 THICK 729
1-(B12(N,I)*A11(N,I)/S22 THICK 730
3 THICK 731
2 B22(N,I)*A12(N,I)/S21 THICK 731
3 THICK 732
Z(8)=B66(N,I)*S12*(GMP+H*OLP)/S11+B66(N,I)*S11*(GMT+H*OLT)/S12 THICK 733
Z(13)=B66(N,I)*S21*(GMP+H*OLP)/S22+S22*(GMT+H*OLT)/S21 THICK 734
Z(9)=B55(N,I)*CPZ/S11 THICK 735

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Z(10)=B*(N,I)*GTZ/S12 THICK 736
Z(14)=B55(N,I)*GPZ/S22 THICK 737
Z(15)=B*(N,I)*GTZ/S21 THICK 738
IF(I.EQ.1) WRITE(6,1723)S,(Z(KK),KK=1,5),I,INA(Z(1L),LL=6,10) THICK 739
IF(I.GT.1) WRITE(6,1782) I,N,(Z(1L),LL=6,10) FEB6 10
WRITE(6,1724) IOUT,(Z(1L),LL=1,15) THICK 741
1721 CONTINUE THICK 742
GO TO 1790 THICK 743
1751 Z(6)=Y(4) THICK 744
Z(7)=XNT THICK 745
Z(8)=Y(6) THICK 746
Z(9)=XMT THICK 747
Z(10)=Y(2) THICK 748
Z(11)=QT THICK 749
1730 Z(12)=Y(8) THICK 750
Z(13)=Y(10) THICK 751
WRITE(6,1725) S,(Z(KK),KK=1, 13) THICK 752
1782 FORMAT(63X,13,2X,A3,1X,5E11.4) FEB6 11
1723 FORMAT(F8.4,5E11.4,13,2X,A3,1X,5E11.4) THICK 754
1724 FORMAT( 68X ,A3,1X,5E11.4) THICK 755
1725 FORMAT(F8.4,5E12.5,4(E14.5,2X)/68X,4(E14.5,2X)) THICK 756
1780 FORMAT(4F20.8) THICK 757
1790 IF(N=L) 1800,1820,1820 THICK 758
1800 XOUT=XPR(18R)*SZERO THICK 759
IBCL=0 THICK 760
CALL RUNGE THICK 761
INIG=1 THICK 762
Z(7)=0.0 THICK 763
Z(8)=0.0 THICK 764
Z(9)=0.0 THICK 765
Z(10)=0.0 THICK 766
DO 1410 I=1,NDE THICK 767
1810 Z(I)=Y(I) THICK 768
1820 CONTINUE THICK 769
1750 XX=0.0 THICK 770
WRITE (6,1830) THICK 771
1830 FORMAT (1H ) THICK 772
1840 CONTINUE THICK 773
IX=IX+IPAR(N) THICK 774
1850 CONTINUE THICK 775
1860 ISIR=0 THICK 776
NNN=2 THICK 777
NSIR=2 THICK 778
GO TO 1470 THICK 779
1870 IF(I*XC=NAT) 1880,1890,1890 THICK 780
1880 INAC=INAC+1 THICK 781
GO TO 260 THICK 782
1890 RETURN THICK 783
765 THICK 784
END THICK 785

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MORE MEMORY WOULD HAVE RESULTED IN BETTER OPTIMIZATION

SYMBOLIC REFERENCE MAP

ENTRY POINTS
2 MAIN

VARIABLES	SN	TYPE	RELOCATION	2317	ALFR	REAL	ARRAY	GEOM	GEOM
1332	ALFL	REAL	GEOM	2317	ALFR	REAL	ARRAY	MAIL	MAIL
740	AL1	REAL	MAIL	1060	AL2	REAL	ARRAY	MAIL	MAIL
4446	BTH	REAL	MAIL	0	B11	REAL	ARRAY	MAIL	MAIL
120	B12	REAL	MAIL	240	B22	REAL	ARRAY	MAIL	MAIL
500	B44	REAL	MAIL	620	B55	REAL	ARRAY	THICK	THICK
360	B66	REAL	MAIL	7	CPP	REAL	ARRAY	MAIL	MAIL
11	CPT	REAL	THICK	4442	CXR	REAL	ARRAY	THICK	THICK
12	CXS	REAL	THICK	25	C11	REAL	ARRAY	THICK	THICK
26	C12	REAL	THICK	27	C22	REAL	ARRAY	THICK	THICK
30	C66	REAL	THICK	0	D	REAL	ARRAY	THICK	THICK
4452	DEL	REAL	THICK	4436	DELS	REAL	ARRAY	THICK	THICK
12	DLP	REAL	THICK	15	DLT	REAL	ARRAY	THICK	THICK
0	DM	REAL	THICK	550	DSC	REAL	ARRAY	THICK	THICK
34	DUMM	REAL	THICK	35	D11	REAL	ARRAY	THICK	THICK
36	D12	REAL	THICK	37	D22	REAL	ARRAY	THICK	THICK
40	D66	REAL	THICK	4455	EFI	REAL	ARRAY	THICK	THICK
4443	ELM	REAL	THICK	4444	ELJ	REAL	ARRAY	THICK	THICK
4454	EM	REAL	THICK	56	EMFI	REAL	ARRAY	THICK	THICK
57	EMTH	REAL	THICK	4453	EN	REAL	ARRAY	THICK	THICK
10	EPP	REAL	THICK	4450	EPS	REAL	ARRAY	THICK	THICK
13	EPT	REAL	THICK	77	ERP	REAL	ARRAY	THICK	THICK
4460	ETF	REAL	THICK	4445	ETH	REAL	ARRAY	THICK	THICK
31	E11	REAL	THICK	32	E12	REAL	ARRAY	THICK	THICK
33	E22	REAL	THICK	34	E66	REAL	ARRAY	THICK	THICK
50	F11	REAL	THICK	51	F12	REAL	ARRAY	THICK	THICK
52	F22	REAL	THICK	0	GA	REAL	ARRAY	THICK	THICK
5	GB	REAL	THICK	14	GMP	REAL	ARRAY	THICK	THICK
16	GMT	REAL	THICK	4	GPZ	REAL	ARRAY	THICK	THICK
5	G12	REAL	THICK	64	G12	REAL	ARRAY	THICK	THICK
65	G21	REAL	THICK	4463	H	REAL	ARRAY	THICK	THICK
0	HDMAX	REAL	THICK	4456	HFI	REAL	ARRAY	THICK	THICK
26	HM	REAL	THICK	4451	HKA	REAL	ARRAY	THICK	THICK
4464	HP	REAL	THICK	4461	HTF	REAL	ARRAY	THICK	THICK
4447	HTH	REAL	THICK	5	HTT	REAL	ARRAY	THICK	THICK
60	H11	REAL	THICK	61	H12	REAL	ARRAY	THICK	THICK
63	H21	REAL	THICK	62	H22	REAL	ARRAY	THICK	THICK
41	H44	REAL	THICK	42	H55	REAL	ARRAY	THICK	THICK
43	H66	REAL	THICK	4402	I	REAL	ARRAY	THICK	THICK
1250	IA	INTEGER	THICK	1262	IB	INTEGER	ARRAY	THICK	THICK
75	IBCL	INTEGER	THICK	0	IBR	INTEGER	ARRAY	THICK	THICK
1	IBRM	INTEGER	THICK	4422	ICYL	INTEGER	ARRAY	THICK	THICK
34	ID	INTEGER	THICK	33	IFREQ	INTEGER	ARRAY	THICK	THICK
4403	IGCT	INTEGER	THICK	4414	IK	INTEGER	ARRAY	THICK	THICK
3536	IN	INTEGER	THICK	13	INDEX	INTEGER	ARRAY	THICK	THICK
74	ING	INTEGER	THICK	1	INTG	INTEGER	ARRAY	THICK	THICK
4401	INXC	INTEGER	THICK	3537	IOR	INTEGER	ARRAY	THICK	THICK
3540	IOUT	INTEGER	THICK	120	IPAR	INTEGER	ARRAY	THICK	THICK
0	IRC	INTEGER	THICK	4421	IRS	INTEGER	ARRAY	THICK	THICK

AFHDL-TR-71-116

SUBROUTINE MAIN

VARIABLES	SN	TYPE	RELOCATION	170	ISS	INTEGER	ARRAY	GEOM
1 ISH		INTEGER	XMIT			INTEGER		
4433 IST		INTEGER		4373	ISTK	INTEGER		
4416 ISTR		INTEGER		0	ISV	INTEGER		F.P.
31 ITP		ARRAY	TRIA	4375	IVB	INTEGER		
4432 IX		INTEGER		4411	J	INTEGER		
4423 JJ		INTEGER		30	JMAX	INTEGER		RUN
4434 JPA		INTEGER		0	JRC	INTEGER		F.P.
0 JSY		INTEGER	F.P.	0	JTYPE	INTEGER		XOUM
27 J9		INTEGER	RUN	4410	K	INTEGER		
4431 KF		INTEGER		4473	KK	INTEGER		
4471 KKK		INTEGER		76	KORIN	INTEGER		RUN
0 KKC		INTEGER	F.P.	4376	K100	INTEGER		
4377 K4		INTEGER		4400	K8	INTEGER		
4412 L		INTEGER		4404	LD	INTEGER		
4472 LL		INTEGER		4462	LLL	INTEGER		
4413 M		INTEGER		1060	MLY	INTEGER		GEOM
31 M9		INTEGER	RUN	4415	N	INTEGER		
24 NBR		INTEGER	XMIT	0	NDE	INTEGER		RUN
176 NF		INTEGER	TRIA	4420	NFF	INTEGER		
200 NFP		INTEGER	TRIA	175	NH	INTEGER		TRIA
4407 NNN		INTEGER		4406	NPCH	INTEGER		
177 NPL		INTEGER	TRIA	201	NPRT	INTEGER		TRIA
4417 NSTR		INTEGER	TRIA	524	NTP	INTEGER		GEOM
4437 NTR		INTEGER		4405	NX	INTEGER		
4374 NXT		INTEGER		4	OMSQ	REAL		XMIT
4424 PARTS		REAL		16	PC	REAL		XMIT
15 PL		REAL	XMIT	14	PN	REAL		XMIT
6 PT		REAL	THICK	1200	RHO	REAL		ARRAY
21 RH1		REAL	XMIT	22	RH2	REAL		XMIT
23 RH3		REAL	XMIT	6	R1	REAL		XMIT
7 R2		REAL	XMIT	10	R3	REAL		XMIT
1 S		REAL	RUN	0	S1	REAL		GEOM
4430 SMAK		REAL		4425	SMAK	REAL		ARRAY
24 SX		REAL	ARRAY	11	SXN	REAL		XMIT
4441 SXR		REAL		4426	SZERO	REAL		
4465 S11		REAL		4466	S12	REAL		
4470 S21		REAL		4467	S22	REAL		
2153 T11		REAL	ARRAY	1333	TR	REAL		GEOM
214 TR1		REAL	ARRAY	360	TR2	REAL		GEOM
17 T0		REAL	XMIT	20	T1	REAL		XMIT
47 XE66		REAL	XMIT	53	XH11	REAL		XMIT
54 XH12		REAL	XMIT	55	XH22	REAL		XMIT
4427 XINT		REAL		44	XK11	REAL		XMIT
45 XK12		REAL	XMIT	46	XK22	REAL		XMIT
2 XLAST		REAL	ADAM	3	XLD	REAL		XMIT
1 XMT		REAL	THICK	3	AMTP	REAL		THICK
2 XN		REAL	XMIT	4440	XNR	REAL		XMIT
0 XNT		REAL	THICK	2	XNTP	REAL		THICK
32 XOUT		REAL	RUN	50	XPR	REAL		ARRAY
4435 XX		REAL		2	Y	REAL		ARRAY
14 YD		REAL	ARRAY	144	Z	REAL		ARRAY
4457 Z0		REAL	RUN	1104	Z1Y	REAL		GEOM

AFDDL-TR-71-116

SUBROUTINE MAIN

STATEMENT LABELS

1741	1180	0	1190	0	1200	
2012	1210	2017	1220	0	1230	
0	1240	0	1250	0	1260	
0	1270	2161	1280	0	1290	
0	1300	4162	1310	2280	1320	AFFDL-TR-71-16
0	1330	4164	1340	0	1350	INACTIVE
4200	1360	2256	1370	0	1380	INACTIVE
0	1390	2270	1400	0	1410	INACTIVE
2273	1420	0	1430	2305	1440	INACTIVE
0	1450	0	1460	2333	1470	
4204	1480	0	1488	2411	1489	INACTIVE
0	1490	0	1491	4253	1500	INACTIVE
4215	1501	4234	1502	2417	1510	FMT
2405	1511	4275	1520	2422	1530	FMT
0	1540	0	1541	2443	1542	FMT
2437	1543	0	1544	0	1546	INACTIVE
2451	1547	2423	1550	2454	1559	INACTIVE
4317	1560	4323	1561	0	1562	INACTIVE
2371	1563	2376	1564	4327	1570	FMT
2456	1580	0	1590	0	1600	INACTIVE
0	1610	2502	1620	2507	1621	INACTIVE
2513	1630	0	1640	0	1650	INACTIVE
2541	1660	0	1670	0	1680	INACTIVE
4333	1690	2572	1700	0	1710	INACTIVE
0	1711	0	1720	0	1721	INACTIVE
0	1722	4345	1723	4351	1724	FMT
4354	1725	3137	1728	4335	1730	FMT
0	1740	3067	1750	3413	1751	FMT
0	1760	0	1770	4361	1780	FMT
4341	1782	3435	1790	0	1800	INACTIVE
0	1810	3455	1820	4363	1830	FMT
0	1840	0	1850	0	1860	INACTIVE
3500	1870	0	1880	3505	1890	INACTIVE
0	3333	102	3334			

COMMON BLOCKS LENGTH

ADUM	2
RUN	64
MAIL	720
XMIT	54
GEOM	1235
THICK	15
TRIA	130
ADAM	3

STATISTICS

PROGRAM LENGTH	46118	2441
COMMON LENGTH	42578	2223

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SUBROUTINE ORTHO
SUBROUTINE ORTHO
COMMON/RTUN/ INDE,S,X,Y(10),YD(10),HH,J9,JMAX,M9,XOUT,IFREQ,DUMMH(33),
1IBC,KORTN,EXP
COMMON/MAL/ B11(20,4),B12(20,4),B22(20,4),B66(20,4),B44(20,4),
1B55(20,4),A11(20,4),A12(20,4),RHO(20,4)
COMMON/AMIT/ IBR,ISH,AN,XLO,OMSQ,HI,I,R1,R2,R3,SAN,CXS,INDEX,PN,PL,
1PC,T0,T1,RH1,RM2,RH3,NBR,C11,C12,C22,C66,E11,E12,E22,E66,D11,D12,
2022,D66,H44,H55,H66,XE11,XE12,XE22,XE66,E11,F12,F22,XH11,XH12,XH22
3,EMFI,EMTH,H11,H12,H22,H21,G12,G21
COMMON/GEOM/ SI(20),SX(20),XPR(20),ING(20),IPAR(20),Z(20),ISS(20),
1TR(10,10),IR2(10,10),NTP(20),OSC(20,10),MLY(20),ZLY(20,5),IA(10),
2IB(10,4),ALEL,TR(10,10,4),TLI(10,10),XALFR(4)
DIMENSION IL1(20,4),IL2(20),PSR(20,4)
60 TO (10,10),220,190),INDEX
15 10 MK=MLY(1BR)
DO 60 I=1,MK
READ (5,20) B11(1BR,I),B12(1BR,I),B22(1BR,I),B66(1BR,I),
14LI(1BR,I),A12(1BR,I),AHO(1BR,I),LI(1BR,I)
20 PSR(1BR,I)=B12(1BR,I)
IF (LI(1BR,I)) 30,60,*0
30 L3=LI(1BR,I)
DY=FGEN (L3,I)
60 TO 60
40 L1=LI(1BR,I)
K5=5
IF (KORTN.NE.0) K5=7
L2=LI(1BR,I),K5
DO 50 J=L1,L2
50 DY=FGEN (J,I)
60 CONTINUE
70 READ (5,70) (ZLY(1BR,I),I=1,5),JL2(1BR)
70 FORMAT (5F10.6,15)
IF (IL2(1BR,I)) 100,100,*0
35 50 L4=IL2(1BR,I)*MK
L1=IL2(1BR)
DO 90 I=L1,L2
90 DY=FGEN (I,I)
100 CONTINUE
MK=MLY(1BR)
DO 180 I=1,MK
J=I+1
WRITE (6,110) I, ZLY(1BR,I),ZLY(1BR,J)
110 FORMAT (1H0, 8H LAYER NO. 13, 9H FROM Z=,F12.5,7H TO Z=,F12.5)
45 IF (B22(1BR,I)-1.0E-8) 120,120,140
120 E=B11(1BR,I)
P=B12(1BR,I)
P1=1.0-P*P
B11(1BR,I)=E/P1
B12(1BR,I)=P*E/P1
B66(1BR,I)=B11(1BR,I)
B66(1BR,I)=E*0.5/(1.0+P)
B44(1BR,I)=B66(1BR,I)
B55(1BR,I)=B66(1BR,I)
55 WRITE (6,130) E,P
THICK 786
THICK 787
THICK 788
THICK 789
THICK 790
THICK 791
THICK 792
THICK 793
THICK 794
THICK 795
THICK 796
THICK 797
THICK 798
THICK 799
THICK 800
THICK 801
THICK 802
THICK 803
THICK 804
THICK 805
THICK 806
THICK 807
THICK 808
THICK 809
THICK 810
THICK 811
THICK 812
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THICK 816
THICK 817
THICK 818
THICK 819
THICK 820
THICK 821
THICK 822
THICK 823
THICK 824
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THICK 826
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THICK 828
THICK 829
THICK 830
THICK 831
THICK 832
THICK 833
THICK 834
THICK 835
THICK 836
THICK 837
THICK 838
THICK 839
THICK 840

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AFFDL-TR-71-116

AFFDL-TR-71-116

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130 FORMAT(1H,5X,49HCONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= THICK 841
1, E12.5, 20H POISSONS RATIO NU=E12.5) THICK 842
90 TO 160 THICK 843
140 WHITE (6,150) B11(1BR,I),B12(1BR,I),B22(1BR,I),B66(1BR,I) THICK 844
150 FORMAT(1H,5X,37HCONSISTS OF ORTHOTROPIC MATERIAL WITH, THICK 845
16H, B11=E12.5,16H B12=E12.5,16H B22=E12.5,16H B66=E12.5) THICK 846
160 WHITE (6,170) A11(1BR,I),A12(1BR,I),RHO(1BR,I) THICK 847
170 FORMAT(1H,5X,33HCOEFFICIENTS OF THERMAL EXPANSION, 4H, A E12. THICK 848
1E12.5, 9H A1ETA=E12.5, 19H MASS DENSITY RHO=E12.5) THICK 849
180 CONTINUE THICK 850
RETURN THICK 851
190 IF(1L2(1BR)) 200,200,230 THICK 852
200 DO 210 I=1,MK THICK 853
IF(1L1(1BR,I)) 250,210,250 THICK 854
210 CONTINUE THICK 855
RETURN THICK 856
220 LI=1L2(1BR) THICK 857
MK=MLY(1BR) THICK 858
L2=1L2(1BR)+MK THICK 859
IF(1L2(1BR)) 290,250,230 THICK 860
230 J=0 THICK 861
DO 240 I=L1,L2 THICK 862
J=J+1 THICK 863
240 ZLY(1BR,J)=FGEN(I,2) THICK 864
250 DO 280 I=1,MK THICK 865
IF(1L1(1BR,I)) 260,280,270 THICK 866
260 L3=1L1(1BR,I) THICK 867
F=EGEN(L3,2) THICK 868
P=PSR(1PH,I) THICK 869
P1=1.0-P*P THICK 870
B11(1BR,I)=E/P1 THICK 871
B22(1BR,I)=B11(1BR,I) THICK 872
B12(1BR,I)=P*B11(1BR,I) THICK 873
B55(1BR,I)=E*0.5/(1.0*P) THICK 874
B44(1BR,I)=B66(1BR,I) THICK 875
B55(1BR,I)=B66(1BR,I) THICK 876
GO TO 280 THICK 877
270 J=1L1(1BR,I) THICK 878
B11(1BR,I)= FGEN (J ,2) THICK 879
B12(1BR,I)= FGEN (J,1,2) THICK 880
B22(1BR,I)= FGEN (J,2,2) THICK 881
B55(1BR,I)= FGEN (J,3,2) THICK 882
A11(1BR,I)= FGEN (J,4,2) THICK 883
A12(1BR,I)= FGEN (J,5,2) THICK 884
IF(KORTN) 271,280,271 THICK 885
271 B55(1BR,I)=FGEN(J+6,2) THICK 886
B44(1BR,I)=FGEN(J+7,2) THICK 887
280 CONTINUE THICK 888
290 C11=0.0 THICK 889
S11=1.0 THICK 890
R11=1.0 THICK 891
M55=0.0 THICK 892
H66=0.0 THICK 893
F11=0.0 THICK 894
F12=0.0 THICK 895

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AFDLD-TR-71-116

F22=0.0	THICK	896
RHM=0.0	THICK	897
XE11=0.0	THICK	898
XE12=0.0	THICK	899
XE22=0.0	THICK	900
XH11=0.0	THICK	901
XH12=0.0	THICK	902
XH22=0.0	THICK	903
C12=0.0	THICK	904
C22=0.0	THICK	905
C66=0.0	THICK	906
E11=0.0	THICK	907
E12=0.0	THICK	908
E22=0.0	THICK	909
E66=0.0	THICK	910
D11=0.0	THICK	911
D12=0.0	THICK	912
D22=0.0	THICK	913
D66=0.0	THICK	914
H11=0.0	THICK	915
H12=0.0	THICK	916
H22=0.0	THICK	917
H21=0.0	THICK	918
G12=0.0	THICK	919
G21=0.0	THICK	920
KH1=0.0	THICK	921
KH2=0.0	THICK	922
KH3=0.0	THICK	923
KH4=0.0	THICK	924
KH5=0.0	THICK	925
KH6=0.0	THICK	926
KH7=0.0	THICK	927
KH8=0.0	THICK	928
KH9=0.0	THICK	929
KH10=0.0	THICK	930
KH11=0.0	THICK	931
KH12=0.0	THICK	932
KH13=0.0	THICK	933
KH14=0.0	THICK	934
KH15=0.0	THICK	935
KH16=0.0	THICK	936
KH17=0.0	THICK	937
KH18=0.0	THICK	938
KH19=0.0	THICK	939
KH20=0.0	THICK	940
KH21=0.0	THICK	941
KH22=0.0	THICK	942
KH23=0.0	THICK	943
KH24=0.0	THICK	944
KH25=0.0	THICK	945
KH26=0.0	THICK	946
KH27=0.0	THICK	947
KH28=0.0	THICK	948
KH29=0.0	THICK	949
KH30=0.0	THICK	950

AFFDL-TR-71-116

VARIABLES	SN	TYPE	HELOCATION	2153	ILI	REAL	ARRAY	GEOM
1030 S12		REAL		214	TR1	REAL	ARRAY	GEOM
1333 TR		REAL	ARRAY	17	I0	REAL	ARRAY	GEOM
360 TR2		REAL	ARRAY	44	XE11	REAL		XMIT
20 T1		REAL	XMIT	46	XE22	REAL		XMIT
45 XE12		REAL	XMIT	53	XH11	REAL		XMIT
47 XE66		REAL	XMIT	55	AH22	REAL		XMIT
54 XHIP		REAL	XMIT	2	XN	REAL		XMIT
3 XLD		REAL	XMIT	50	XPR	REAL	ARRAY	GEOM
32 XOUT		REAL	RUN	14	YD	REAL	ARRAY	RUN
2 Y		REAL	ARRAY	1104	ZLY	REAL	ARRAY	GEOM
144 Z		REAL	ARRAY	1026	ZZ	REAL		
1025 Z1		REAL						
1027 Z3		REAL						

FILE NAMES	MODE	TAPE5	TAPE6	FMT

EXTERNALS	TYPE	ARGS
FGEN	REAL	2

STATEMENT LABELS	721	20	FMT	0	30	INACTIVE
12 10	721	20	FMT	0	30	INACTIVE
74 40	0	50		117	60	
724 70	0	80	INACTIVE	0	90	
153 100	727	110	FMT	0	120	INACTIVE
735 130	233	140		750	150	FMT
256 160	764	170	FMT	0	180	
300 190	0	200	INACTIVE	0	210	
314 220	321	230		0	240	
334 250	0	260	INACTIVE	370	270	
0 271	455	280		460	290	
0 291	601	292		0	294	INACTIVE
0 300						

COMMON BLOCKS	LENGTH
RUN	64
MATL	720
XMIT1	54
GEOM	1235

STATISTICS	PROGRAM LENGTH	1320B	720
	COMMON LENGTH	4031B	2073

SUBROUTINE COEFF
CDC 6600 FIN V3.0-P243 OPT=1 06/19/71 13:01.44: THICK 1033
PAGE 2

AFFDL-TR-71-116

SYMBOLIC REFERENCE MAP

ENTRY POINTS
1 COEFF

AFFDL-TR-71-116

VARIABLES	SN	TYPE	RELOCATION
740 AL1	REAL	ARRAY	MATL
0 B11	REAL	ARRAY	MATL
307 B2	REAL	ARRAY	MATL
310 B4	REAL	ARRAY	MATL
620 B55	REAL	ARRAY	MATL
311 B8	REAL	ARRAY	MATL
11 CPT	REAL	THICK	THICK
304 CXS	REAL	THICK	THICK
26 C12	REAL	THICK	THICK
30 C66	REAL	THICK	THICK
15 DLT	REAL	THICK	THICK
35 D11	REAL	THICK	THICK
37 D22	REAL	THICK	THICK
56 EMI	REAL	THICK	THICK
302 ENQ	REAL	THICK	THICK
13 EPT	REAL	THICK	THICK
44 E11	REAL	THICK	THICK
46 E22	REAL	THICK	THICK
50 F11	REAL	THICK	THICK
52 F22	REAL	THICK	THICK
16 GMT	REAL	THICK	THICK
5 G12	REAL	THICK	THICK
65 G21	REAL	THICK	THICK
5 H11	REAL	THICK	THICK
61 H12	REAL	THICK	THICK
62 H22	REAL	THICK	THICK
42 H55	REAL	THICK	THICK
75 IBCL	REAL	THICK	THICK
33 IFREQ	INTEGER	THICK	THICK
1 ISH	INTEGER	THICK	THICK
27 J9	INTEGER	THICK	THICK
31 M9	INTEGER	THICK	THICK
0 NDE	INTEGER	THICK	THICK
16 PC	INTEGER	THICK	THICK
14 PN	INTEGER	THICK	THICK
1200 RHO	REAL	THICK	THICK
22 RH2	REAL	THICK	THICK
303 R50	REAL	THICK	THICK
7 R2	REAL	THICK	THICK
1 S	REAL	THICK	THICK
305 SXNR	REAL	THICK	THICK
312 THN	REAL	THICK	THICK
20 T1	REAL	THICK	THICK
53 XH11	REAL	THICK	THICK
55 XH22	REAL	THICK	THICK
32 XK12	REAL	THICK	THICK
3 XLD	REAL	THICK	THICK
3 XNTP	REAL	THICK	THICK

AFFDL-TR-71-116

SUBROUTINE COEFF. CDC 6600 FIN V3.0-P243 OPT=1 06/19/71 1320144. PAGE 4

RELOCATION

VARIABLES	SN	TYPE	RELOCATION
306 XNB		REAL	
2 XNTP		REAL	THICK RUN
2 Y		REAL	ARRAY RUN

0	XNT	REAL	THICK
32	XOUT	REAL	RUN
14	YD	REAL	ARRAY RUN

COMMON BLOCKS LENGTH

THICK	15
RUN	64
MATL	720
XMIT	54

STATISTICS

PROGRAM LENGTH	3148	204
COMMON LENGTH	15258	853

SUBROUTINE	DIFFEQ	CDC 6600 F7N V3.0-P243 OPT=1	06/19/71	13.01.44.	PAGE	1
05	1034	THICK				
05	1035	THICK				
05	1036	THICK				
05	1037	THICK				
05	1038	THICK				
05	1039	THICK				
05	1040	THICK				
05	1041	THICK				
05	1042	THICK				
05	1043	THICK				
05	1044	THICK				
05	1045	THICK				
05	1046	THICK				
05	1047	THICK				
05	1048	THICK				
05	1049	THICK				
05	1050	THICK				
05	1051	THICK				
05	1052	THICK				
05	1053	THICK				
05	1054	THICK				
05	1055	THICK				
05	1056	THICK				
05	1057	THICK				
05	1058	THICK				
05	1059	THICK				
05	1060	THICK				
05	1061	THICK				
05	1062	THICK				
05	1063	THICK				
05	1064	THICK				
05	1065	THICK				
05	1066	THICK				
05	1067	THICK				
05	1068	THICK				
05	1069	THICK				
05	1070	THICK				
05	1071	THICK				
05	1072	THICK				
05	1073	THICK				
05	1074	THICK				
05	1075	THICK				
05	1076	THICK				
05	1077	THICK				
05	1078	THICK				
05	1079	THICK				
05	1080	THICK				
05	1081	THICK				
05	1082	THICK				
05	1083	THICK				
05	1084	THICK				
05	1085	THICK				
05	1086	THICK				
05	1087	THICK				
05	1088	THICK				

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DEL=C11*D11-E11*E11          THICK 1089
EN= Y(4)-C12*ETH-E12*HIH-(H11*I0+H12*I1)*XLD THICK 1090
EM= Y(6)-E12*ETH-D12*HTH-(H12*I0+G12*I1)*XLD THICK 1091
EFI= (EN*D11-E0*E11)/DEL THICK 1092
HF1= (EM*D11-E0*E11)/DEL THICK 1093
YD(1)=R1*Y(3)-Y(5) THICK 1094
YD(3)= EFI-H1*Y(1) THICK 1095
YD(5)=HF1 THICK 1096
THN= C12*EFI+C22*ETH+E12*HFI+E22*HTH*(H22*I0+H21*I1)*XLD THICK 1097
THM= D12*HFI+D22*HTH+E12*EFI+E22*ETH*(H21*I0+G21*I1)*XLD THICK 1098
YD(2)=-CXRY(2)+SXR*THN*RI*Y(4)-PN*XLD THICK 1099
YD(4)=CXRY(THN-Y(4))-RI*Y(2)-PL*XLD THICK 1100
YD(6)= CXR*(THN-Y(6))+Y(2)-EMFI*XLD THICK 1101
GO TO 160 THICK 1102
150 XNR=XNR2 THICK 1103
ELM=RI-SXR THICK 1104
ELJ=RI+SXR THICK 1105
ETH=XNR*Y(7)+ CXR*Y(3) *SKQ*Y(1) THICK 1106
G1H=XNR*Y(1) +SXR*Y(7) THICK 1107
H1H=XNR*B1H+CXRY(5) THICK 1108
EPS=-XNK*Y(3)-CXK*Y(7) THICK 1109
HKA=-2.0*XNR*Y(5)-2.0*XNR*CXRY(1)+CXRY(ELH-SXR)*Y(7)+XNR*RI*Y(3) THICK 1110
DEL=C11*D11-E11*E11 THICK 1111
EM= Y(4)-C12*ETH-E12*HTH-(H11*I0+H12*I1)*XLD THICK 1112
EM= Y(6)-E12*ETH-D12*HTH-(H12*I0+G12*I1)*XLD THICK 1113
EFI= (EM*D11-E0*E11)/DEL THICK 1114
HF1= (EM*C11-E0*E11)/DEL THICK 1115
YD(1)=R1*Y(3)-Y(5) THICK 1116
YD(3)= EFI-R1*Y(1) THICK 1117
YD(5)=HF1 THICK 1118
YD(7)= ( Y(8)-(C66+SXR*E66)*EPS-(E66+SXR*D66)*HKA) THICK 1119
1/(C66+2.0*E66*SXR+D66*SXR*SXR) THICK 1120
THN= C12*EFI+C22*ETH+E12*HFI+E22*HTH*(H22*I0+H21*I1)*XLD THICK 1121
THM= D12*HFI+D22*HTH+E12*EFI+E22*ETH*(H21*I0+G21*I1)*XLD THICK 1122
TFM= E66*(EPS*YD(7))+ D66*(HKA+SXR*YD(7)) THICK 1123
YD(2)= -2.0*XNR*CXRY(2)+SXR*THN* RI*Y(4)+ XNR*XNR*THM THICK 1124
IPN*XLD THICK 1125
YD(4)= -XNR*Y(4)+XNR*ELJ*TFM+CXRY(THN-Y(4)) -RI*Y(2)-PL*XLD THICK 1126
YD(6)= -2.0*XNR*TFM+CXRY(THM-Y(6)) +Y(2) THICK 1127
YD(8)= ELM*CXRY(8)-2.0*CXRY(8) *XNR*THN+ XNR*SXR*THM-PC*XLD THICK 1128
GO TO 160 THICK 1129
141 CALL COEFF THICK 1130
160 IF(R3-1.0) 170,190,170 THICK 1131
170 GO 160 I=I+1 THICK 1132
180 YD(1)=YD(1)/R3 THICK 1133
190 HELD=RN THICK 1134
END THICK 1135

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AFFDL-TR-71-116

SYMBOLIC REFERENCE MAP

ENTRY POINTS
1 DIFFEQ

AFFDL-TR-71-1116

VARIABLES	SN	TYPE	RELOCATION	2317	ALFR	REAL	ARRAY	GEOM
1332 ALFL		REAL	GEOM	1060	AL2	REAL	ARRAY	GEOM
740 AL1		REAL	MATL			REAL	ARRAY	MATL
714 B1H		REAL		0	B11	REAL	ARRAY	MATL
120 B12		REAL	MATL	240	B22	REAL	ARRAY	MATL
500 B44		REAL	MATL	620	B55	REAL	ARRAY	MATL
360 B66		REAL	MATL	7	CPP	REAL	ARRAY	THICK
11 CPT		REAL	THICK	676	CXR	REAL		XMIT
12 CAS		REAL	XMIT	25	C11	REAL		XMIT
26 C12		REAL	XMIT	27	C22	REAL		XMIT
30 C66		REAL	XMIT	702	DEL	REAL		XMIT
12 DLP		REAL	THICK	15	DLT	REAL		THICK
550 DSC		REAL	GEOM	34	DUMM	REAL	ARRAY	RUN
35 D11		REAL	XMIT	36	D12	REAL		XMIT
37 D22		REAL	XMIT	40	D66	REAL		XMIT
705 EF1		REAL		712	ELH	REAL		XMIT
713 ELJ		REAL		704	EM	REAL		XMIT
56 EMF1		REAL	XMIT	57	EMTH	REAL		XMIT
703 EN		REAL		10	EPP	REAL		THICK
715 EPS		REAL		13	EPT	REAL		THICK
77 ERP		REAL	RUN	700	ETH	REAL		XMIT
31 E11		REAL	XMIT	32	E12	REAL		XMIT
33 E22		REAL	XMIT	34	E66	REAL		XMIT
50 F11		REAL	XMIT	51	F12	REAL		XMIT
52 F22		REAL	XMIT	14	GMP	REAL		THICK
16 GMT		REAL	THICK	4	GPZ	REAL		THICK
5 GI7		REAL	THICK	64	G12	REAL		XMIT
65 G21		REAL	XMIT	706	HFI	REAL		XMIT
26 HH		REAL	RUN	716	HKA	REAL		XMIT
701 H1H		REAL	XMIT	5	H1I	REAL		XMIT
60 H1I		REAL	XMIT	61	H12	REAL		XMIT
63 H21		REAL	XMIT	62	H22	REAL		XMIT
41 H44		REAL	XMIT	42	H55	REAL		XMIT
43 H66		REAL	XMIT	673	I	REAL		XMIT
1250 IA		INTEGER	ARRAY	1262	IB	INTEGER	ARRAY	GEOM
75 IBCL		INTEGER	RUN	0	IBR	INTEGER		XMIT
33 IFREQ		INTEGER	RUN	13	INDEX	INTEGER		XMIT
74 ING		INTEGER	GEOM	120	IPAR	INTEGER	ARRAY	GEOM
1 ISH		INTEGER	XMIT	170	ISS	INTEGER	ARRAY	GEOM
675 J		INTEGER		674	JJ	INTEGER		RUN
30 JMAX		INTEGER	RUN	27	J9	INTEGER		GEOM
76 KORTN		INTEGER	RUN	1060	MLY	INTEGER	ARRAY	XMIT
31 M9		INTEGER	RUN	24	NBR	INTEGER		GEOM
0 NUE		INTEGER	RUN	524	NTP	INTEGER	ARRAY	GEOM
4 OMSQ		REAL	XMIT	16	PC	REAL		XMIT
15 PL		REAL	XMIT	14	PN	REAL		XMIT
6 QT		REAL	THICK	1200	RHO	REAL	ARRAY	MATL
21 RH1		REAL	XMIT	22	RH2	REAL		XMIT
23 RH3		REAL	XMIT	6	R1	REAL		XMIT

AFFDL-TR-71-116

SUBROUTINE DIFFEQ

VARIABLES	SN	TYPE	RELOCATION	10	R3	REAL	XMIT	GEOM	ARRAY	XMIT
7 R2		REAL	XMIT	0	SI	REAL	GEOM			
1 S		REAL	RUN			REAL	GEOM		ARRAY	XMIT
24 SX		REAL	GEOM	11	SXN	REAL	XMIT			
677 SAR		REAL	ARRAY	717	TFM	REAL	XMIT			
710 THM		REAL		707	JHN	REAL	XMIT			
2153 TLI		REAL	GEOM	1333	TR	REAL	GEOM		ARRAY	GEOM
214 TRI		REAL	GEOM	360	TR2	REAL	GEOM		ARRAY	GEOM
17 T0		REAL	XMIT	20	T1	REAL	XMIT			
47 XE66		REAL	XMIT	53	XH11	REAL	XMIT			
54 XH12		REAL	XMIT	55	XH22	REAL	XMIT			
44 XK11		REAL	XMIT	45	XK12	REAL	XMIT			
46 XK22		REAL	XMIT	3	XLD	REAL	XMIT			
1 XMT		REAL	THICK	3	XMTF	REAL	THICK			
2 AN		REAL	XMIT	711	XNR	REAL	THICK			
0 XNT		REAL	THICK	2	XNTP	REAL	THICK			
32 XOUT		REAL	RUN	50	XPR	REAL	GEOM		ARRAY	GEOM
2 Y		REAL	RUN	14	YD	REAL	GEOM		ARRAY	RUN
144 Z		REAL	GEOM	1104	ZLY	REAL	GEOM		ARRAY	GEOM

FILE NAMES

FILE NAMES	MODE
TAPE6	FMT

EXTERNALS	TYPE	ARGS
COEFF		0
INPUT		0
EXIT		0
ORTHO		0

STATEMENT LABELS

STATEMENT LABELS	TYPE	ARGS	551	20	FMT	565	30	FMT
0 10	INACTIVE		551	20	FMT	565	30	FMT
572 40	FMT		600	50	FMT	603	60	FMT
606 70	FMT		616	80	FMT	626	90	FMT
640 100	FMT		652	110	FMT	664	120	FMT
165 130			0	139	INACTIVE	0	140	INACTIVE
537 141			321	150		541	160	
0 170	INACTIVE		0	180		550	190	

COMMON BLOCKS

COMMON BLOCKS	LENGTH
THICK	15
RUN	64
MATL	720
XMIT	54
GEOM	1235

STATISTICS

STATISTICS	PROGRAM LENGTH	720H	464
COMMON LENGTH <td>4050B <td>2088 <td></td> </td></td>	4050B <td>2088 <td></td> </td>	2088 <td></td>	

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SUBROUTINE BCOND
COMMON/RUN/ NDE,SAY(10),YD(10),HH,J9,JMAX,M9,XOUT,IFREQ,DUMM(33),
1IBCL,KORTN,ERP
COMMON/MATL/ B11(20,4),B12(20,4),B22(20,4),RHO(20,4),
1B55(20,4),AL1(20,4),AL2(20,4),XLD,OMSQ,H11,H1,R2,AR3,SKN,CAS,INDEX,PN,PL,
COMMON/XMIT/ IGR,ISH,XN,XLD,OMSQ,H11,H1,R2,AR3,SKN,CAS,INDEX,PN,PL,
1PC,10,T1,KH1,PH2,RH3,NBR,C11,C12,C22,C66,E11,E12,E22,E66,O11,O12,
2D22,D66,H44,H55,H66,XK11,XK12,XK22,XL66,F11,F12,F22,XH11,XH12,XH22
3EMFI,EMTH,H11,H12,H22,H21,G12,G21
COMMON/GEOM/ SI(20),SX(20),XPR(20),ING(20),IPAR(20),Z(20),ISS(20),
1TRI(10,10),TR2(10,10),NIP(20),OSC(20,10),PLY(20),ZLY(20,5),IA(10),
2IB(10,4),ALFL,TR(10,10,4),TLI(10,10),ALFR(4)
C
C NOTE THAT MATRIX TLI HERE IS Y(A)=TLI*(U(A)
C TR MATRIX IS (UB)=TR*(Y(B)
15 DO 10 I=1,NDE
DO 10 J=1,NDE
TLI(I,J)=0.0
DO 10 K=1,NBR
DO 20 L=1,NDE
TLI(I,L)=1.0
DO 20 K=1,NBR
20 TR (I,I,N)=1.0
ALFA=ALFL*1.74533E-02
SIL=SIN (ALFA)
COS=COS (ALFA)
TLI(1,1)=CUL
TLI(1,3)=SIL
TLI(2,2)=CUL
TLI(3,1)=-SIL
TLI(3,3)=CUL
TLI(4,2)=-SIL
TLI(4,4)=CUL
DO 30 K=1,NBR
ALFA=ALFR(K)*1.74533E-02
SIR=SIN (ALFA)
COS=COS (ALFA)
TR(1,1,K)=COR
TR(1,3,K)=-SIR
TR(2,2,K)=COR
TR(2,4,K)=-SIR
TR(3,1,K)=SIR
TR(3,3,K)=COR
TR(4,2,K)=SIR
TR(4,4,K)=COR
30 RETURN
40 END
THICK 1136
THICK 1137
THICK 1138
THICK 1139
THICK 1140
THICK 1141
THICK 1142
THICK 1143
THICK 1144
THICK 1145
THICK 1146
THICK 1147
THICK 1148
THICK 1149
THICK 1150
THICK 1151
THICK 1152
THICK 1153
THICK 1154
THICK 1155
THICK 1156
THICK 1157
THICK 1158
THICK 1159
THICK 1160
THICK 1161
THICK 1162
THICK 1163
THICK 1164
THICK 1165
THICK 1166
THICK 1167
THICK 1168
THICK 1169
THICK 1170
THICK 1171
THICK 1172
THICK 1173
THICK 1174
THICK 1175
THICK 1176
THICK 1177
THICK 1178
THICK 1179
THICK 1180
THICK 1181
THICK 1182
THICK 1183

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AFFDL-TR-71-116

AFFDL-TR-71-116

SYMBOLIC REFERENCE MAP

ENTRY POINTS
1 BCOND

VARIABLES	SN	TYPE	RELOCATION
101 ALFA	1332	REAL	ALFL
2317 ALFH	740	REAL	ALI
1090 AL2	0	REAL	B11
120 B12	240	REAL	B22
500 B44	620	REAL	B55
360 B66	103	REAL	COL
105 C0H	12	REAL	CXS
25 C11	26	REAL	C12
27 C22	30	REAL	C66
550 DSC	34	REAL	DUMM
35 D11	36	REAL	D12
37 D22	40	REAL	D66
56 EMP1	57	REAL	EMTH
77 EXP	31	REAL	E11
32 E12	33	REAL	E22
34 E66	50	REAL	F11
51 F12	52	REAL	F22
64 G12	65	REAL	G21
26 HH	5	REAL	H11
60 H11	61	REAL	H12
63 H21	62	REAL	H22
41 H44	42	REAL	H55
43 H66	76	REAL	I
1250 IA	1262	REAL	Ib
75 IBCL	0	REAL	IbR
33 IFREQ	13	REAL	INDEX
74 ING	120	REAL	IPAR
1 ISH	170	REAL	ISS
77 J	30	REAL	JMAX
27 J9	100	REAL	K
76 KORTN	1060	REAL	MLY
31 M9	24	REAL	NBR
0 N0E	924	REAL	NTP
4 OMSQ	16	REAL	PC
15 PL	14	REAL	PN
1200 R40	21	REAL	RH1
22 RH2	23	REAL	RH3
6 R1	7	REAL	R2
10 R3	1	REAL	S
0 S1	102	REAL	SIL
104 SIR	24	REAL	SX
11 SKN	2153	REAL	ILI
1333 TR	214	REAL	TR1
360 TR2	17	REAL	T0
20 T1	47	REAL	XE66
53 XM11	54	REAL	XH12
55 XM22	44	REAL	XK11
45 XK12	46	REAL	XK22

VARIABLES	SN	TYPE	RELOCATION
3 XLD	REAL	XMIT	
32 XOUT	REAL	RUN	
2 Y	REAL	ARRAY	2 XN
144 Z	REAL	ARRAY	50 XPR
		ARRAY	14 YD
		ARRAY	1104 ZLY

AFFDL-TR-71-116

EXTERNALS	TYPE	ARGS	SIN	LIBRARY
COS	REAL			J LIBRARY

STATEMENT LABELS	INACTIVE
0 10	0 20
0 40	0 30

COMMON BLOCKS	LENGTH
RUN	64
MATL	720
XMIT	54
GEOM	1235

STATISTICS	PROGRAM LENGTH	COMMON LENGTH
	1069	70
	40318	2073

AFFDL-TR-71-116

SUBROUTINE	INPUT	CDC 6600 FIN V3.0-P243 OPT=1	06/19/71	13.01.44.	PAGE	1
05	SUBROUTINE INPUT COMMON/MATL/ B11(20,4),B12(20,4),B22(20,4),B66(20,4),B44(20,4), 1B55(20,4),A11(20,4),A12(20,4),R40(20,4) COMMON/KRUM/ INDE,SY(10),YD(10),HH,J9,JMAX,M9,XOUT,IFREQ,DUMM(33), 1IBCL,KORTN,ERP COMMON/AMII/ IRR,ISH,XN,XLD,OMSG,HI,R1,R2,R3,AXN,CXS,INDEX,PM,PL, 1PC,10,T1,RH1,RH2,RH3,NBR,C11,C12,C22,C66,E11,E12,E22,E66,D11,D12, 2D22,D66,H44,H55,H66,XK11,XK12,XK22,XE66,E11,E12,E22,XH11,XH12,XH22 3EMFI,EMTM,HI1,HI2,H22,H21,G12,G21 10 COMMON/GEOM/ SI(20),SA(20),XPR(20),ING(20),IPAR(20),Z(20),ISS(20), 1FRI(10,10),TH2(10,10),NTP(20),OSC(20,10),ZLY(20),ZLY(20),IA(10), 2IBL(10,4),ALEL ,IR(10,10,4),TL(10,10),ALER(4) DIMENSION VN(10,4),IL1(10),IL2(10),IFG(10,4) DIMENSION VK(10,5),IK2(10),IKL(10),IKG(10,5),INORM(10) 15 GO TO (10,20,30,40),INDEX 10 READ (5,20) (VN(1BR,1), I=1,4), IL2(1BR) 20 FORMAT (4F10.5,15) IF IL2(1BR) 70,70,30 30 READ (5,40) (IFG(1BR,I),I=1,3), IL1(1BR) 40 FORMAT (10I5) WRITE (6,50) (IFG(1BR,I),I=1,3) 50 FORMAT (1H0, 29HVARIABLE SHELL PARAMETERS ARE, 4IS) 1FIL1(1BR) L2=IL2(1BR),L1(1BR)-1 60 DO 60 I=L1,L2 70 GO TO (80,100,120,140,160,200,220,240,180), ISH 80 WRITE (6,190) 90 FORMAT (1H0, 30HNO SHELL NO. 1 IN THIS PROGRAM) CALL EXIT 30 100 WRITE (6,110) ISH,VN(1BR,1),VN(1BR,2), VN(1BR,3) 110 FORMAT (1H0, 20HCYLINDRICAL SHELL NO,13,2X, 1 4H H=E12.5,2X,4H H=E12.5, 2X, 4HPHI=F10.3, 9H DEGREES) RETURN 35 120 WRITE (6,130) ISH,VN(1BR,1),VN(1BR,2), VN(1BR,4) 130 FORMAT (1H0, 18HSPHERICAL SHELL NO,13,4X, 1 4H H=E12.5,2X,4H H=E12.5, 12H DIRECTIONS, F3.0) RETURN 40 140 WRITE (6,150) ISH,VN(1BR,1),VN(1BR,2), VN(1BR,4) 150 FORMAT (1H0, 21HPARABOLOIDAL SHELL NO,13,3X, 1 4H H=E12.5,2X,5H 2P=E12.5, 12H DIRECTIONS, F3.0) RETURN 45 160 WRITE (6,170) ISH,VN(1BR,1),VN(1BR,2),VN(1BR,3),VN(1BR,4) 170 FORMAT (1H0, 20HELLIPSOIDAL SHELL NO,13,3X, 1 4H H=E12.5,2X,4H A=E12.5, 2X,4H B=E12.5,10H DIRECTIONS,F3.0) RETURN 50 180 WRITE (6,190) ISH,VN(1BR,1),VN(1BR,2),VN(1BR,3),VN(1BR,4) 190 FORMAT (1H0, 20HYPERBOLIC SHELL NO,13,3X, 1 4H H=E12.5,2X,4H A=E12.5, 2X,4H B=E12.5,10H DIRECTIONS,F3.0) RETURN 200 WRITE (6,210) ISH,VN(1BR,1),VN(1BR,2),VN(1BR,3) 210 FORMAT (1H0, 16HCONICAL SHELL NO,13,2X, 4H H=F10.6, 2X, 14HPI=F10.3, 8H DEGREES, 2X,4H A=E12.5) RETURN 55 220 WRITE (6,230) ISH,VN(1BR,1),VN(1BR,2),VN(1BR,3),VN(1BR,4) RETURN					

AFFDL-TR-71-116

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230 FORMAT (1H0, 17HTOROIDAL SHELL NO.,I3,3X, THICK 1239
1 4H H=E12.5,2X,4H A=E12.5, 2X,4H B=E12.5,10H DIRECTION,F3.0) THICK 1240
RETURN THICK 1241
240 WRITE (6,250) ISH, (VN(1BR,I),I=1,4) THICK 1242
250 FORMAT (1H0, 16HGENERAL SHELL NO., I3, 4H H=E12.5, THICK 1243
17H IZREFI=E12.5, 5H RE, E12.5, 4H F10.3, 4H DEG) THICK 1244
RETURN THICK 1245
260 READ (5,270) (VK(1BR,I),I=1,5), IK2(1BR),INORM(1BR),RPM THICK 1246
270 FORMAT (5F10.5,2I5,F10.5) THICK 1247
RPS=6.28318*KPM/60.0 THICK 1248
EMHEO=0 THICK 1249
EMP I=0.0 THICK 1250
IF (INORM(1BR)/100) 300,280,300 THICK 1251
280 WRITE (6,290) (VK(1BR,I),I=1,5) FE86 12
290 FORMAT (1H0,30HSURFACE AND TEMP LOADS ARE P=E12.5,6H PF1,E12.5 THICK 1253
1,9H PTHETA=E12.5,5H TL=E12.5,5H TU=E12.5) THICK 1254
GO TO 340 THICK 1255
300 WRITE (6,310) (VK(1BR,I),I=1,5) THICK 1256
310 FORMAT (1H0,30HSURFACE AND TEMP LOADS ARE P1=E12.5,6H P2=E12.5 THICK 1257
1,9H PTHETA=E12.5,5H TL=E12.5,5H TU=E12.5) THICK 1258
IF (RPM) 340,340,320 THICK 1259
320 WRITE (6,330) RPM THICK 1260
330 FORMAT (1H0, 45HSHELL IS SPINNING ABOUT AXIS OF SYMMETRY WITH, THICK 1261
1E12.5, 5H RPM) THICK 1262
340 CONTINUE THICK 1263
IF (IK2(1BR)) 3H0,340,350 THICK 1264
350 READ (6,40) (IKG(1BR,I),I=1,5), IK1(1BR) THICK 1265
WRITE (6,360) (IKG(1BR,I),I=1,5) THICK 1266
360 FORMAT (1H0, 28HVARIBLE LOAD PARAMETERS ARE, 5I5) THICK 1267
K1=IK1(1BR) THICK 1268
K2=IK2(1BR)+IK1(1BR)-1 THICK 1269
UU 370 I=K1,K2 THICK 1270
370 UY= FGEN(I,1) THICK 1271
380 RETURN THICK 1272
390 CONTINUE THICK 1273
L1=IL1(1BR) THICK 1274
L2=IL2(1BR)+IL1(1BR)-1 THICK 1275
K1=IK1(1BR) THICK 1276
K2=IK2(1BR)+IK1(1BR)-1 THICK 1277
MK=MLY(1BR) THICK 1278
GO TO(80,400,410,420,430,450,460,470,440), ISH THICK 1279
400 K2=APS (1.0/VN(1BR,2)) THICK 1280
R1=VN(1BR,2) FE86 13
R1=0.0 THICK 1281
R3=1.0 THICK 1282
CASE=0.0 THICK 1283
ALFA=VN(1BR,3)*1.745329E-02 THICK 1284
SAME SIN (ALFA) THICK 1285
GO TO 480 THICK 1286
410 R1=1.0/VN(1BR,2) THICK 1287
R1=VN(1BR,2) FE86 14
R3=ABS (R1) THICK 1288
420 GO TO 480 THICK 1289
430 R5U= (VN(1BR,3)/VN(1BR,2))*(VN(1BR,3)/VN(1BR,2)) THICK 1291

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AFFDL-TR-71-116

115	GO TO 480	THICK 1292
	440 BSQ= (VN(IHR,3)/VN(IHR,2))*VN(IHR,3)/VN(IHR,2))	THICK 1293
	GO TO 480	THICK 1294
	450 KI=0.0	THICK 1295
	R3=1.0	THICK 1296
	ALFA = VN(IHR,2)*0.1745329E=01	THICK 1297
	SX= SIN (ALFA)	THICK 1298
	CX= COS (ALFA)	THICK 1299
	GO TO 480	THICK 1300
120	460 RI=1.0/VN(IHR,3)	THICK 1301
	KI=VN(IHR,3)	FEB6 15
	R3=ABS (KI)	THICK 1302
	GO TO 480	THICK 1303
	470 R3=1.0	THICK 1304
125	RT=VN(IHR,3)	JUN16 6
	480 IF (IL2(IHR)) 510,510,490	THICK 1305
	490 IK=0	THICK 1306
	DO 500 I=L1,L2	THICK 1307
	IK=IK+1	THICK 1308
	J=IF3(IHR,IK)	THICK 1309
130	500 VN(IHR,J)=FGEN (I+2)	THICK 1310
	510 IF (IK2(IHR)) 540,540,520	THICK 1311
	520 IK=0	THICK 1312
	DO 530 I=K1,K2	THICK 1313
135	I=IK+1	THICK 1314
	J=IK3(IHR,IK)	THICK 1315
	530 VK(IHR,J)=FGEN (I+2)	THICK 1316
	540 PN=VK(IHR,1)	THICK 1317
140	PL=VK(IHR,2)	THICK 1318
	PC=VK(IHR,3)	THICK 1319
	HT=ZLY(IHR,KK+1)-ZLY(IHR,1)	FEB6 16
	T0= (VK(IHR,4)*ZLY(IHR,KK+1) -VK(IHR,5)*ZLY(IHR,1))/HT	THICK 1320
	T1= (VK(IHR,5)-VK(IHR,4))/HT	THICK 1321
	GO TO(80,620,550,560,570,590,600,610,590), ISH	THICK 1322
145	550 ARG=VN(IHR,4)	THICK 1323
	SX= SIN (ARG)	THICK 1324
	CX= COS (ARG)	THICK 1325
	R2=ABS (KI/SXN)	THICK 1326
	GO TO 620	THICK 1327
150	560 ARG=VN(IHR,4)	THICK 1328
	SX= SIN (ARG)	THICK 1329
	CX= COS (ARG)	THICK 1330
	R2=ABS (CX/(SXN*VN(IHR,2)))	THICK 1331
	R1= ABS(CX*CX/SXN)/VN(IHR,2)	THICK 1332
155	R3=ABS (R1)	THICK 1333
	KI=1.0/R2	FEB6 18
	GO TO 620	THICK 1334
570	ARG=VN(IHR,4)	THICK 1335
	SX= SIN (ARG)	THICK 1336
	CX= COS (ARG)	THICK 1337
160	K= SORT (BSQ*(1.-BSQ)*SXN*SXN)	THICK 1338
	R1=HR*KR/(HSQ*VN(IHR,2))	THICK 1339
	R2=ABS (R / (SXN*VN(IHR,2)))	THICK 1340
	R3=ABS (R1)	THICK 1341
165	KI=1.0/R2	FEB6 19

AFFDL-TR-71-116

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170      GO TO 620
171      580 ARG=SVN(1BR,4)
172      SXS=SIN (ARG)
173      CAS=COS (ARG)
174      R=SQRT (SXS*SXS-BSQ*CAS*CAS)
175      R1= -VN(1BR,2)*R/R/(VN(1BR,3)*VN(1BR,3))
176      R2=ABS (R / (SXS*VN(1BR,2)))
177      R3=ABS (R1)
178      R1=1.0/R2
179      GO TO 620
180      590 R2= ABS (1.0/((VN(1BR,3)+S)*CAS))
181      R1=1.0/R2
182      GO TO 620
183      600 ARG=S*VN(1BR,4)
184      SXS=SIN (ARG)
185      CAS=COS (ARG)
186      R2=ABS (1.0/(VN(1BR,2)+VN(1BR,3)*SXS))
187      GO TO 620
188      610 R1=VN(1BR,2)
189      R2=ABS (1.0/VN(1BR,3))
190      ARG= VN(1BR,4)*0.1745329E-01
191      CAS=COS(ARG)
192      SXS=SIN(ARG)
193      620 IF (ALD=0.5) 650,650,630
194      630 IF (1*ORM(1BR)/100) 650,650,640
195      640 PH= VK(1BR,2)*R2/R2*(RH1/R2+ SXS*RH2)
196      PH=VK(1BR,1)*CAS*PHR*SXS
197      PL=VK(1BR,1)*SXS*PHR*SXS
198      IF (MOD((1*ORM(1BR)+100),EQ,0)) RETURN
199      IF (MOD((1*ORM(1BR)+10),EQ,0)) GO TO 651
200      BNE(HT,ZLY(1BR,1))/RT
201      652 PN=BN*PN
202      PL=V*PL
203      PC=BN*PC
204      RETURN
205      651 BNE(HT,ZLY(1BR,MK+1))/RT
206      GO TO 652
207      END

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THICK 1342

THICK 1343

THICK 1344

THICK 1345

THICK 1346

THICK 1347

THICK 1348

THICK 1349

FEB6 20

THICK 1350

THICK 1351

FEB6 21

THICK 1352

THICK 1353

THICK 1354

THICK 1355

THICK 1356

THICK 1357

THICK 1358

THICK 1359

THICK 1360

THICK 1361

THICK 1362

THICK 1363

FEB6 22

THICK 1365

THICK 1366

THICK 1367

THICK 1368

FEB6 23

FEB6 24

FEB6 25

FEB6 26

FEB6 27

FEB6 28

FEB6 29

FEB6 30

FEB6 31

THICK 1370

AFEDI--TR-71-116

SUBROUTINE INPUT

SYMBOLIC REFERENCE MAP

ENTRY POINTS

VARIABLES	SN	TYPE	RELOCATION	1332	ALFL	REAL	GEOM	ARRAY	MATL	GEOM	MATL
1271 ALFA		REAL		740	AL1	REAL		ARRAY		GEOM	MATL
2317 ALFR		REAL		1275	ARG	REAL					
1060 AL2		REAL		1272	MSJ	REAL					
1300 HN		REAL		120	H12	REAL		ARRAY	MATL		
0 H11		REAL		500	H44	REAL		ARRAY	MATL		
240 H22		REAL		360	B66	REAL		ARRAY	MATL		
620 B55		REAL		25	C11	REAL		ARRAY	MATL		
12 CAS		REAL		27	C22	REAL		ARRAY	XMIT		
26 C12		REAL		550	D5C	REAL				GEOM	
30 C66		REAL		1261	DY	REAL					
34 DUMM		REAL		36	D12	REAL				XMIT	
35 D11		REAL		40	D66	REAL				XMIT	
37 D22		REAL		57	EMTH	REAL				XMIT	
56 EMFI		REAL		31	E11	REAL				XMIT	
77 ERP		REAL		33	E22	REAL				XMIT	
32 E12		REAL		50	F11	REAL				XMIT	
34 F66		REAL		52	F22	REAL				XMIT	
51 F12		REAL		65	G21	REAL				XMIT	
64 G12		REAL		5	H11	REAL				XMIT	
26 H6		REAL		61	H12	REAL				XMIT	
60 H11		REAL		62	H22	REAL				XMIT	
63 H21		REAL		42	H55	REAL				XMIT	
41 H44		REAL		1256	I	REAL				XMIT	
43 H66		REAL		1262	IB	REAL				XMIT	
1250 IA		INTEGER		0	IBR	INTEGER		ARRAY		GEOM	
75 IBCL		INTEGER		33	IFREQ	INTEGER				XMIT	
1375 IFG		INTEGER		1593	IKG	INTEGER		ARRAY		RUN	
1273 IK		INTEGER		1527	IN2	INTEGER		ARRAY			
1541 IK1		INTEGER		1363	IL2	INTEGER		ARRAY			
1351 IL1		INTEGER		74	ING	INTEGER		ARRAY		GEOM	
13 INDEX		INTEGER		120	IPAR	INTEGER		ARRAY		GEOM	
1635 INDRM		INTEGER		170	ISS	INTEGER		ARRAY		GEOM	
1 ISH		INTEGER		30	JMAX	INTEGER		ARRAY		RUN	
1274 J		INTEGER		76	KORTN	INTEGER				RUN	
27 J9		INTEGER		1266	K2	INTEGER					
1285 K1		INTEGER		1260	L2	INTEGER					
1257 L1		INTEGER		1060	MLY	INTEGER		ARRAY		GEOM	
1267 MK		INTEGER		24	NBR	INTEGER		ARRAY		XMIT	
31 M9		INTEGER		524	NTP	INTEGER		ARRAY		GEOM	
0 NDE		INTEGER		16	PC	REAL				XMIT	
4 OMSQ.		REAL		15	PL	REAL				XMIT	
1277 PHR		REAL		1276	R	REAL				XMIT	
14 PN		REAL		21	RH1	REAL		ARRAY		XMIT	
1200 RMO		REAL		23	RH3	REAL				XMIT	
22 RM2		REAL		1264	RPO	REAL				XMIT	
1262 RPN		REAL		1270	RT	REAL				XMIT	
1263 RPS		REAL		7	R2	REAL				XMIT	
6 R1		REAL									

AFFDL-TR-71-116

SUBROUTINE INPUT

VARIABLES	SN	TYPE	RELOCATION	1	S	REAL	ARRAY	GEOM	RUN
10 H3		REAL	XMIT						
0 S1		REAL	GEOM	24	SX		ARRAY	GEOM	
11 SAN		REAL	XMIT	2153	ILI		ARRAY	GEOM	
1333 TR		REAL	GEOM	214	TRI		ARRAY	GEOM	
300 TR2		REAL	GEOM	17	TD			XMIT	
20 T1		REAL	XMIT	1445	VK		ARRAY		
1301 VN		REAL	ARRAY	47	XE66				
53 XH11		REAL	XMIT	54	XH12			XMIT	
55 XH22		REAL	XMIT	44	XK11			XMIT	
45 XK12		REAL	XMIT	46	XK22			XMIT	
3 XLD		REAL	XMIT	2	XN			XMIT	
32 XOUT		REAL	RUN	50	APR		ARRAY	GEOM	
2 Y		REAL	RUN	14	YD		ARRAY	RUN	
144 Z		REAL	GEOM	1104	ZLY		ARRAY	GEOM	

FILE NAMES

MODE	TAPES	TAPES	FMT

EXTERNALS

TYPE	ARGS	EXIT	LIBRARY
HEAL	1	SIN	0
HEAL	2		1
REAL	1		LIBRARY

INLINE FUNCTIONS

TYPE	ARGS	MOD	INIEGER	2	INIRIN
REAL	1				

STATEMENT LABELS

12	10	1044	20	FMT	0	30	INACTIVE
1047	40	1051	50	FMT	0	60	
74	70	111	80		1056	90	FMT
116	100	1063	110	FMT	132	120	
1075	130	146	140		1106	150	FMT
142	160	1117	170	FMT	200	180	
1131	190	216	200		1143	210	FMT
232	220	1154	230	FMT	250	240	
1166	250	266	260		1200	270	FMT
0	280	1203	290	FMT	334	300	
1216	310	0	320		INACTIVE		FMT
358	340	0	350		INACTIVE		FMT
0	370	423	380		424	390	
451	400	463	410		470	420	
471	430	474	440		477	450	
510	460	515	470		520	480	
0	490	0	500		542	510	
0	520	0	530		564	540	
622	550	634	560		653	570	
676	580	722	590		730	600	
744	610	756	620				INACTIVE
0	640	1004	650		1030	651	
1024	652						

COMMON BLOCKS

LENGTH	MAIL	HUN
720		
64		

SUBROUTINE INPUT CDC 6600 FIN V3.0-P243 OPT=1 06/19/71 13:01.44: PAGE 7

COMMON BLOCKS LENGTH
XMIT 54
GEOM 1235

STATISTICS

PROGRAM LENGTH 16478 935
COMMON LENGTH 40318 2073

AFEDL-TR-71-116

```

FUNCTION FGEN                                CDC 6600 FTN V3.0-P243 OPT=1 06/19/71 13:01:44. PAGE 1
      FUNCTION FGEN (NM,N1)
      DIMENSION XP(30,20),YP(30,20),SL(30,20),M(30)
      COMMON/RUN/ NDE,S,Y(10),YD(10),HH,J9,JMAX,M9,XOUT,IFREQ,DUMM(33),
      1 IBL,KORIN,ERP
      IF(NM=30) 30,30,10
      10 WRITE (6,20)
      20 FORMAT(1H0,50HMAXIMUM NUMBER OF 30 FGEN SETS HAVE BEEN EXCEEDED )
      CALL EXIT
      30 CONTINUE
      40 GO TO (40,120),N1
      40 READ (5,50) NAMEG, M(NM)
      50 FORMAT (A5,I5)
      WRITE (6,60) NAMEG, NM,M(NM)
      60 FORMAT (1H0, 10X, A5, 32H LINEAR FUNCTION GENERATOR NO.,
      13, 5H FROM, I4, 7H POINTS)
      MK=M(NM)
      READ (5,70) (XP(NM,I), YP(NM,I), I=1,MK)
      MU=0
      70 FORMAT (B,F10.5)
      MZ=10
      MX= (M(NM)-1)/10+1
      DO 100 J=1,MX
      LE=MINO (MK, MZ)
      LE=MAX(1, LE)
      25 WRITE (6,80) (YP(NM,I), I=1,LE)
      WRITE (6,90) (XP(NM,I), I=1,LE)
      80 FORMAT (1H0, 13HY COORDINATES,3X,10F10.5)
      90 FORMAT (1H0, 13HX COORDINATES,3X,10F10.5)
      MZ=MZ+10
      100 MU=MU+10
      MM=M(NM)-1
      DO 110 I=1,MM
      SL(NM,I)=(YP(NM,I+1)-YP(NM,I))/(XP(NM,I+1)-XP(NM,I))
      110 YP(NM,I)=YP(NM,I) -SL(NM,I)*XP(NM,I)
      RETURN
      120 MK=M(NM)
      DO 130 I=1,MK
      J=I
      IF(XP(NM,I)-S) 130,130,140
      130 CONTINUE
      140 IF(J=1) 150,150,160
      150 J=2
      160 FGEN= SL(NM,J-1)*S +YP(NM,J-1)
      RETURN
      45 END

```

AFDDL-TR-71-116

137

SYMBOLIC REFERENCE MAP

ENTRY POINTS
2 FGEN

AFFDL-TR-71-116

VARIABLES SN TYPE RELOCATION

VARIABLES	SN	TYPE	RELOCATION
34 DUMM		REAL	ARRAY
252 FGEN		REAL	ARRAY
255 I	77	INTEGER	ERP
33 JFREQ	26	INTEGER	HH
30 JMAX	75	INTEGER	IBCL
76 KORTN	261	INTEGER	J
263 L1	27	INTEGER	J9
254 MK	262	INTEGER	L
256 MU	3675	INTEGER	M
257 MZ	260	INTEGER	MM
253 NAMEG	31	INTEGER	M9
0 NM	0	INTEGER	NDE
1 S	0	REAL	N1
32 XOUT	2545	REAL	SL
2 Y	265	REAL	XP
1415 YP	14	REAL	YD

FILE NAMES MODE

FILE NAMES	MODE
TAPE5	FMT
TAPE6	FMT

EXTERNALS TYPE ARGS

EXTERNALS	TYPE	ARGS
EXIT		0

INLINE FUNCTIONS TYPE ARGS

INLINE FUNCTIONS	TYPE	ARGS
AMINO	REAL	0
INTRIN		0

STATEMENT LABELS

STATEMENT LABELS	TYPE	ARGS
0 10	INACTIVE	216 20
33 40	FMT	225 50
237 70	FMT	241 80
0 100		0 110
0 130		202 140
205 160		

COMMON BLOCKS LENGTH

COMMON BLOCKS	LENGTH
RUN	64

STATISTICS

STATISTICS	PROGRAM LENGTH	COMMON LENGTH
	3742B	2018
	100B	64

SUBROUTINE RUNGE CDC 6600 FTN V3.0-P245 OPT=1 06/19/71 13:01.44. PAGE 1
 SUBROUTINE RUNGE THICK 1416
 COMMON/RUN/N,X,Y(10),DY(10),HM,J,JMAX,M,XOUT,IFREQ,X1,X2,X3, THICK 1417
 IY1(10),Y2(10),Y3(10),ISCL,KORTN,ERP THICK 1418
 J=1 THICK 1419
 JMAX=1 THICK 1420
 IFREQ=3 THICK 1421
 N=1 THICK 1422
 CALL HUNKUT THICK 1423
 RETURN THICK 1424
 END THICK 1425

AFFDL-TR-71-116

SUBROUTINE RUNGE

SYMBOLIC REFERENCE MAP

AFFDL-TR-71-116

ENTRY POINTS

VARIABLES	SN	TYPE	RELOCATION
14 DY		REAL	RUN
26 MH		REAL	ARRAY
33 IPREU		INTEGER	RUN
30 JMAX		INTEGER	RUN
31 M		INTEGER	RUN
1 X		REAL	RUN
34 X1		REAL	RUN
36 X3		REAL	RUN
37 Y1		REAL	ARRAY
63 Y3		REAL	ARRAY
77 ERP		REAL	RUN
75 JBCL		INTEGER	RUN
27 J		INTEGER	RUN
76 KORIN		INTEGER	RUN
0 N		INTEGER	RUN
32 XOUT		REAL	RUN
35 X2		REAL	RUN
2 Y		REAL	ARRAY
51 Y2		REAL	ARRAY

EXTERNALS TYPE ARGS
RUNKUT 0

COMMON BLOCKS LENGTH
RUN 64

STATISTICS

PROGRAM LENGTH 118 9
COMMON LENGTH 1008 64

```

SUBROUTINE HUNKU1
COMMON/HUN/N,X,Y(10),DY(10),HH,J,JMAX,N,XOUT,I,FREQ,X1,X2,X3,
1Y(10),Y2(10),Y3(10),IBCL,KORIN,ERP
INDE9 = 0
05 CALL ADJSTP
IF(J=JMAX) 10,10,50
10 INDE9 = INDE9 + 1
CALL INTPUL
IF(J=JMAX) 20,20,50
10 CALL STEP
X1 = X2
X2 = X3
X3 = X
00 30 I = 1,N
Y1(I) = Y2(I)
Y2(I) = Y3(I)
30 Y3(I) = Y(I)
IF (INDE9 = IFREQ) 10,40,40
40 INDE9 = 0
CALL ADJSTP
IF(J=JMAX) 10,10,50
50 RETURN
END

```

AFFDL-TR-71-116

THICK 1426
THICK 1427
THICK 1428
THICK 1429
THICK 1430
THICK 1431
THICK 1432
THICK 1433
THICK 1434
THICK 1435
THICK 1436
THICK 1437
THICK 1438
THICK 1439
THICK 1440
THICK 1441
THICK 1442
THICK 1443
THICK 1444
THICK 1445
THICK 1446
THICK 1447
THICK 1448

SUBROUTINE RUNKUT

SYMBOLIC REFERENCE MAP

ENTRY POINTS

AAFFDL-TR-71-116

VARIABLES	SN	TYPE	RELOCATION	77 ERP	REAL	INTEGER	RUN
14 OY		REAL	ARRAY	44			RUN
26 HH		REAL	RUN				RUN
75 IBCL		INTEGER	RUN	33	IFREQ		RUN
43 INDE9		INTEGER	RUN	27	J		RUN
30 JMAX		INTEGER	RUN	76	KORTN		RUN
31 M		INTEGER	RUN	0	N		RUN
1 X		REAL	RUN	32	XOUT		RUN
34 A1		REAL	RUN	35	X2		RUN
36 A3		REAL	RUN	2	Y		ARRAY
37 Y1		REAL	ARRAY	51	Y2		ARRAY
63 Y3		REAL	ARRAY				RUN

EXTERNALS	TYPE	ARGS	INIPOL
ADJUST		0	0
STEP		0	

STATEMENT LABELS

1	10	10	0	20	INACTIVE	0	30
2	0	40	41	50			

COMMON BLOCKS

COMMON BLOCKS	LENGTH
RUN	64

STATISTICS

PROGRAM LENGTH	45H	37
COMMON LENGTH	100B	64

SUBROUTINE	ADJUSTP	CDC 6600 FTN V3.0-P243 OPT=1	06/19/71	13.01.44.	PAGE	1
05	SUBROUTINE ADJUSTP COMMON/RUN/N,X,Y(10),DY(10),MH,JJMAX,M,XOU,IFREQ,X1,X2,X3, Y1(10),Y2(10),Y3(10),IBCL,KORIN,ERP KSL=0 HFACT = 1.0 E+31 HFACT1 = 1.0 E+30 GO TO (30,10), M 10 H1 = MH MH = 2.0 * MH X = X1 20 Y(1) = 1, N 30 KSL=0 40 H1 = MH XXX = X 50 Y(1) = Y(1) X1 = X CALL INPOL IF (J=JMAX) 60,60,250 60 CALL STEP 70 Y2(1) = 1, N X2 = X CALL INPOL IF (J=JMAX) 80,80,250 80 CALL STEP 90 Y(1) = Y(1) X3 = X MH = 2.0 * MH 100 CALL STEP DO 150 I = 1, N UELY = ABS (Y(1)-Y3(1)) / 30.0 IF (DELY - ABS (Y2(1)) * ERP) 120,110,110 110 IF (ABS (Y2(1)) - ERP) 120,130,130 120 HFIRST = 1.0 E+30 GO TO 140 130 HFIRST = (ABS (Y2(1)) * ERP / DELY) ** 0.2 140 CONTINUE 150 HFACT=MINI (HFACT, HFIRST) IF (HFACT1 = 2.0 * H1 160 MH = H1 GO TO (40,230), M 170 MH = H1 * HFACT GO TO (180,230), M 180 IF (KSL) 220,220,190 190 KSL=0 IF (ABS (MH) = ABS (H1)) 200,220,220 200 DO 210 I = 1, N 210 Y(1) = Y(1) X = XXX	THICK 1449 THICK 1450 THICK 1451 THICK 1452 THICK 1453 THICK 1454 THICK 1455 THICK 1456 THICK 1457 THICK 1458 THICK 1459 THICK 1460 THICK 1461 THICK 1462 THICK 1463 THICK 1464 THICK 1465 THICK 1466 THICK 1467 THICK 1468 THICK 1469 THICK 1470 THICK 1471 THICK 1472 THICK 1473 THICK 1474 THICK 1475 THICK 1476 THICK 1477 THICK 1478 THICK 1479 THICK 1480 THICK 1481 THICK 1482 THICK 1483 THICK 1484 THICK 1485 THICK 1486 THICK 1487 THICK 1488 THICK 1489 THICK 1490 THICK 1491 THICK 1492 THICK 1493 THICK 1494 THICK 1495 THICK 1496 THICK 1497 THICK 1498 THICK 1499 THICK 1500 THICK 1501 THICK 1502 THICK 1503				
15						
20						
25						
30						
35						
40						
45						
50						
55						

AFTDL-TR-71-116

SUBROUTINE ADJSTP CDC 6600 FIN V3.0-P243 OPT=1 06/19/71 13.01.44: PAGE 2

60 GO TO 40
220 KSL=0
M = 2
230 DO 240 I = 1, N
240 Y(I) = Y3(I)
250 RETURN
END

AFEDL-TR-71-116

THICK 1504
THICK 1505
THICK 1506
THICK 1507
THICK 1508
THICK 1509
THICK 1510

AFFDL-TR-71-116

SUBROUTINE ADJSTP

SYMBOLIC REFERENCE MAP

ENTRY POINTS

1 ADJSTP

VARIABLES SN TYPE RELOCATION

VARIABLES	SN	TYPE	RELOCATION
211 DELY	14	REAL	REAL
77 EMP	204	REAL	REAL
205 HFACT1	212	REAL	REAL
26 HH	206	REAL	REAL
207 I	75	INTEGER	INTEGER
33 JFREQ	27	INTEGER	INTEGER
30 JMAX	76	INTEGER	INTEGER
203 KSL	31	INTEGER	INTEGER
0 N	1	INTEGER	REAL
32 KOUT	210	REAL	REAL
34 X1	35	REAL	REAL
36 X3	2	REAL	REAL
37 Y1	51	REAL	REAL
53 Y3		REAL	REAL

1 EXTERNALS TYPE ARGS

+5 INTPOL 0

INLINE FUNCTIONS TYPE ARGS

ABS REAL I INTRIN

STATEMENT LABELS

ABS	REAL	I	INTRIN	AMINI	REAL	INIRIN
14	10	0	20	25	30	INACTIVE
26	40	0	50	0	60	INACTIVE
0	70	0	80	INACTIVE	0	90
72	100	0	110	INACTIVE	110	120
112	130	120	140	INACTIVE	0	150
0	160	140	170	INACTIVE	150	180
0	190	0	200	INACTIVE	0	210
165	220	167	230	INACTIVE	0	240
174	250					

COMMON BLOCKS LENGTH

RUN 64

STATISTICS

PROGRAM LENGTH 2138 139
COMMON LENGTH 1008 64


```

SUBROUTINE STEP
CDC 6600 FIN V3.0-PZ43 OPT=1 06/19/71 13:01.44. PAGE 1
SUBROUTINE STEP
COMMON/RUN/N,X,Y(10),DY(10),HH,J,JMAX,M,XOUI,JFREQ,X1,X2,X3.
Y1(10),Y2(10),Y3(10),IBCL,KORIN,ERP
DIMENSION Y0(10),P1(10)
05 DO 10 I = 1, N
   Y0(I) = Y(I)
   X0 = X
   CALL DIFFEQ
10 DO 20 I = 1, N
   P1(I) = DY(I) * HH
   Y(I) = Y0(I) + P1(I)*0.5
   X = X0 + HH*0.5
   CALL DIFFEQ
15 DO 30 I = 1, N
   P1(I) = P1(I)+2.0*HH*DY(I)
   Y(I) = Y0(I) + 0.5*HH*DY(I)
   CALL DIFFEQ
20 DO 40 I = 1, N
   P1(I) = P1(I)+2.0*HH*DY(I)
   Y(I) = Y0(I) + HH*DY(I)
   X = X0 + HH
   CALL DIFFEQ
40 DO 50 I = 1, N
   Y(I) = Y0(I) + (P1(I)+HH*DY(I))*0.1666667
RETURN
END
THICK 1511
THICK 1512
THICK 1513
THICK 1514
THICK 1515
THICK 1516
THICK 1517
THICK 1518
THICK 1519
THICK 1520
THICK 1521
THICK 1522
THICK 1523
THICK 1524
THICK 1525
THICK 1526
THICK 1527
THICK 1528
THICK 1529
THICK 1530
THICK 1531
THICK 1532
THICK 1533
THICK 1534
THICK 1535
THICK 1536

```

AFFDL-TR-71-116

SYMBOLIC REFERENCE MAP

ENTRY POINTS
1 STEP

AFFDL-TR-71-116

VARIABLES SN TYPE RELOCATION

14	OY	REAL	ARRAY	RUN	77	ERP	REAL	RUN
26	HH	REAL		RUN	74	I	INTEGER	RUN
75	IBCL	INTEGER		RUN	33	IFREQ	INTEGER	RUN
27	J	INTEGER		RUN	30	JMAX	INTEGER	RUN
76	KORTN	INTEGER		RUN	31	M	INTEGER	RUN
0	N	INTEGER		RUN	110	P1	REAL	ARRAY
1	X	REAL		RUN	32	XOUT	REAL	RUN
75	X0	REAL		RUN	34	X1	REAL	RUN
35	X2	REAL		RUN	36	X3	REAL	RUN
2	Y	REAL	ARRAY	RUN	76	Y0	REAL	ARRAY
37	Y1	REAL	ARRAY	RUN	51	Y2	REAL	ARRAY
63	Y3	REAL	ARRAY	RUN				RUN

EXTERNALS TYPE ARGS
DIFREQ 0

STATEMENT LABELS

0	10	0	20	
0	40	0	50	
			0	30

COMMON BLOCKS LENGTH
RUN 64

STATISTICS

PROGRAM LENGTH	1226	82
COMMON LENGTH	1008	64

SUBROUTINE INTPOL COC 6600 FTN V3.0-P243 OPT=1 06/19/71 13.01.44. PAGE 1

```

SUBROUTINE INTPOL
COMMON/RUN/N,X,Y(10),DY(10),HH,J,JMAX,M,XOUT,IFREQ,XL,X2,X3,
1Y(10),Y2(10),Y3(10),ISCL,KORIN,ERP
IF IABS (XOUT - X) = ABS (HH) / 10, 10, 20
05 10 MH=XOUT-X
CALL STEP
J = J + 1
20 RETURN
END

```

AFFDL-TR-71-116

THICK 1537
THICK 1538
THICK 1539
THICK 1540
THICK 1541
THICK 1542
THICK 1543
THICK 1544
THICK 1545

SYMBOLIC REFERENCE MAP

AFFDL-TR-71-116

ENTRY POINTS

1 INTPOL

VARIABLES SN TYPE RELOCATION

14	DY	REAL	ARRAY	RUN	RUN	77	ERP	REAL	RUN
26	HH	REAL	ARRAY	RUN	RUN	75	IBCL	INTEGER	RUN
33	IFREQ	INTEGER	ARRAY	RUN	RUN	27	J	INTEGER	RUN
30	JMAX	INTEGER	ARRAY	RUN	RUN	76	KORTN	INTEGER	RUN
31	M	INTEGER	ARRAY	RUN	RUN	0	N	INTEGER	RUN
1	X	REAL	ARRAY	RUN	RUN	32	XOUT	REAL	RUN
34	X1	REAL	ARRAY	RUN	RUN	35	X2	REAL	RUN
36	X3	REAL	ARRAY	RUN	RUN	2	Y	REAL	RUN
37	Y1	REAL	ARRAY	RUN	RUN	51	Y2	REAL	RUN
63	Y3	REAL	ARRAY	RUN	RUN			REAL	RUN

EXTERNALS TYPE ARGS

STEP 0

INLINE FUNCTIONS TYPE ARGS

ABS REAL 1 INIRIN

STATEMENT LABELS

0 10 INACTIVE 15 20

COMMON BLOCKS LENGTH

RUN 64

STATISTICS

PROGRAM LENGTH 178 15
COMMON LENGTH 1008 64

***** CDC 6600 FTM V3.0-P243 OPT=0 06/23/71 02.12.12. PAGE 1

SURROUTINE	TRIANG	TRACE	CDC 6600 FTM V3.0-P243 OPT=0 06/23/71 02.12.12. PAGE 1
05		SUBROUTINE TRIANG(D,IRC,JRC,KRC,DM,ISY,JSY) DIMENSION D(IRC,JRC,KRC),DM(15,4,JSY) COMMON/RUN/ NDE,S,Y(10),YD(10),HH,J9,JMAX,M9,XOUT,IFREQ,DUMM(33), IIBCL,KORTN,ERP COMMON/MATL/ B11(20,4),B12(20,4),B22(20,4),B66(20,4),B44(20,4), B55(20,4),AL1(20,4),AL2(20,4),RH0(20,4) COMMON/XMIT/ IRR,ISH,XN,XLD,OMSQ,MTT,R1,R2,R3,SKN,CKS,INDEX,PN,PL, LRC,IO,I1,RH1,RH2,RH3,NBR,C11,C12,C22,C66,F11,E12,E22,E66,D11,D12, D22,D66,H44,H55,H66,K11,XK12,XK22,XE66,F11,F12,F22,XM11,XM12,XM22 3,EMFI,EMTH,H11,H12,H22,H21,G12,G21 COMMON/GEOM/ SI(20),SX(20),XPR(20),ING(20),IPAR(20),Z(20),ISS(20), LTR1(10,10),IR2(10,10),NTP(20),DSC(20,10),MLY(20),ZLY(20,5),IA(10), ZIR(10,4),ALFL *TR(10,10,4),TLI(10,10),ALFR(4) COMMON/IRIA/GA151,GB(5,4),ITP(100),NH,NE,NPL,NED,NPRT 32 COMMON/XDUM/JTYPE,IBRM DIMENSION DL1(5,5),D2(5,5),U1(5,5),U2(5,5) DATA IOR/BHTORSION / IF(JTYPE,NE,IOR).GO TO 4 JF=1 DO 2 K=1,IBRM DO 7 L=JF,IX IX=IPAR(K)+JF-2 DO 3 I=JE,IX DO 3 I=1,NPL TEMP=D(L+4,I) D(L+4,I)=D(L,7,I) 3 D(L,7,I)=TEMP IX=IX+1 JF=JF+1 DO 7 L=JF,IX DO 7 I=1,NDE TEMP=D(L,I,4) D(L,I,4)=D(L,I,7) 7 D(L,I,7)=TEMP 2 JF=IX+1 4 CONTINUE ISI=0 IRR=1 DO 10 I=1,NH DO 10 J=1,NH JJ=J+NH 10 U1(I,J) = D(I,I,JJ) DO 20 I=1,NH II=I+NH DO 20 J=1,NH U2(I,J) = 0 DO 20 L=1,NH LL=L+NH 20 U2(I,J) = U2(I,J) + D(I,II,LL) * U1(L,J) DM(1,1)=DET DO 30 I=1,NH II=I+NH D1(I,I) = - D(1,I,NPL) D1(2,I) = - D(1,II,NPL) DO 30 J=1,NH	THICK 1546 THICK 1547 THICK 1548 THICK 1549 THICK 1550 THICK 1551 THICK 1552 THICK 1553 THICK 1554 THICK 1555 THICK 1556 THICK 1557 FEB6 1558 FEB6 32 FEB6 35 JUN16 1 JUN16 2 FEB6 36 FEB6 37 FEB6 38 FEB6 39 FEB6 40 FEB6 41 FEB6 42 FEB6 43 FEB6 44 FEB6 45 FEB6 46 FEB6 47 FEB6 48 FEB6 49 FEB6 50 FEB6 51 FEB6 52 FEB6 53 THICK 1560 THICK 1561 THICK 1562 THICK 1563 THICK 1564 THICK 1565 THICK 1566 THICK 1567 THICK 1568 THICK 1569 THICK 1570 THICK 1571 THICK 1572 THICK 1573 THICK 1574 THICK 1575 THICK 1576 THICK 1577 THICK 1578 THICK 1579
10			
15			
20			
25			
30			
35			
40			
45			
50			
55			

AFFDL-TR-71-116

```

D1(I,I) = D1(I,I) - D(I,I,J) * GA(J)
30 D1(2,I) = D1(2,I) - D(I,I,J) * GA(J)
   DO 40 I=1,NH
   DO 40 J=1,NH
60 D1(2,I) = D1(2,I) - U2(I,J) * D1(I,J)
   II = I+NH
   D(I,I,NBL) = D(I,I,I)
65 D(I,I,NPL) = D(I,I,I)
   DO 50 J=1,NH
50 D(I,I,J) = U1(I,J)
   IF(NF-1) 520,520,60
60 CALL INVERT (U2,NH,DET)
   DM(I,2) = DET
70 DO 70 I=1,NH
   J=I+NH
   DO 70 J=1,NH
   JJ=J+NH
75 DO 510 K=2,NF
   DO 80 I=1,NH
   DO 80 J=1,NH
   JJ=J+NH
   D2(I,J) = 0.0
   DO 80 L=1,NH
   LL=L+NH
80 D2(I,J) = D2(I,J) + D(K,I,L) * D(K=L+1,J)
   IF(ITP(K-1)=2) 110,90,90
90 DO 100 I=1,NH
   DO 100 J=1,NH
   JJ=J+NH
   U1(I,J) = D(K,I,J)
   U1(I,J) = D(K,I,J)
100 U1(I,J) = U1(I,J) - D2(I,L) * D(K=L+1,J)
110 DO 120 I=1,NH
   DO 120 J=1,NH
   JJ=J+NH
95 U1(I,J) = D(K,I,J) + D2(I,J)
130 IF (ITP(K-1) 200,140,140)
140 IF (IST -J) 150,150,160
150 IRR=IRR+1
   IST=K
100 DO 170 I=1,NH
   II=I+NH
   DO 160 J=1,NH
   JJ=J+NH
   D(K,I,J) = 0.0
160 D(K=1,I,J) = 0.0
170 D(K=1,I,I) = 1.0
180 DO 190 I=1,NH
   DO 190 J=1,NH
   JJ=J+NH
110 D(K,I,J) = 0.0
   DO 190 L=1,NH

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AFFDL-TR-71-118

THICK 1580
 THICK 1581
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 THICK 1632
 THICK 1633
 THICK 1634

151

AFFDL-TR-71-116

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190 D(K,I,J) = D(K,I,J) - D2(I,L) * D(K-I,L,JJ)
200 CALL INVERT (UL,NH,DET)
DM(K,1)=DET
DO 210 I=1,NH
  D1(I,I) = - D(K,I,NPL)
DO 210 J=1,NH
  JJ=J+NH
  D2(I,I) = D1(I,I) - D2(I,J) * D(K-I,J,NPL)
DO 220 I=1,NH
  II=I+NH
  DO 220 J=1,NH
    JJ=J+NH
    D2(I,J) = 0.0
    DO 220 L=1,NH
      LL = L + NH
      D2(I,J) = D2(I,J) - D(K,I,LL) * D(K-I,LL,JJ)
IF ( ITP(K)-1) 250,230,230
230 DO 240 I=1,NH
  DO 240 J=1,NH
    JJ=J+NH
    D(K,I,J)=0.0
    DO 240 L=1,NH
      D(K,I,JJ) = D(K,I,JJ) - D2(I,L) * D(K-I,L,JJ)
240 D(K,I,JJ) = D(K,I,JJ)
250 DO 260 I=1,NH
  II=I+NH
  D1(2,1) = -D(K,I,NPL)
  DO 260 J=1,NH
    JJ=J+NH
    D2(I,J) = D1(2,I) - D2(I,J) * D(K-I,JJ,NPL)
IF ( ITP(K)-2) 300,270,270
270 DO 280 I=1,NH
  DO 280 J=1,NH
    JJ=J+NH
    U2(I,J) = 0.0
    DO 280 L=1,NH
      U2(I,J) = U2(I,J) - D2(I,L) * D(K-I,L,JJ)
280 U2(I,J) = U2(I,J)
DO 290 I=1,NH
  DO 290 J=1,NH
    U2(I,J) = U2(I,J)
300 DO 310 I=1,NH
  II=I+NH
  DO 310 J=1,NH
    JJ=J+NH
    D2(I,J) = D(K,I,I) + D2(I,J)
DO 320 I=1,NH
  DO 320 J=1,NH
    U2(I,J) = 0.0
    DO 320 L=1,NH
      U2(I,J) = U2(I,J) - D2(I,L) * U1(L,J)
320 U2(I,J) = U2(I,J)
IF ( ITP(K)-1) 350,330,330
330 DO 340 I=1,NH
  DO 340 J=1,NH
    JJ=J+NH
    D(K,I,JJ) = D(K,I,JJ) - U2(I,L) * D(K-I,L,J)
340 D(K,I,JJ) = D(K,I,JJ)

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AFFDL-TR-71-116

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350 DO 360 I=1,NH
   DO 360 J=1,NH
360 D1(2,I) = D1(2,I) + U2(I,J) * D1(I,J)
   DO 370 I=1,NH
   II=I+NH
   D(K,I,NPL) = D1(1,I)
   D(K,II,NPL) = D1(2,I)
   DO 370 J=1,NH
370 D(K,I,J) = U1(I,J)
   IF(K=NF1-380,520,520)
380 CALL INVERT (U2,NH,DET)
   DM(K,2) = DET
   DO 390 I=1,NH
   II=I+NH
   DO 390 J=1,NH
   JJ=J+NH
390 D(K,II,JJ) = U2(I,J)
   IF(.NOT.(K=1).AND.(510,510,400))
400 DO 410 I=1,NH
   II=I+NH
410 D(K,II,NPL) = D(K,II,NPL) + GB(I,IRR)
   DO 420 I=1,NH
   II=I+NH
   DO 420 J=1,NH
   JJ=J+NH
   U1(I,J) = D(K,I,I,J)
420 D1(I,J) = D(K,I,I,J)
   DO 490 N=1,ST,K
   DO 430 I=1,NH
   JJ=J+NH
430 D2(I,J) = U2(I,J) + U1(I,I) * D1(I,I) + D1(I,I) * D1(I,I)
   DO 440 I=1,NH
   II=I+NH
   DO 440 J=1,NH
   JJ=J+NH
   D(K,I,I,JJ) = D(K,I,I,J) + U2(I,I) * D1(I,I) + D1(I,I) * D1(I,I)
   DO 450 I=1,NH
   II=I+NH
   DO 450 J=1,NH
   JJ=J+NH
450 D(K,I,II,NPL) = D(K,I,I,NPL) + U2(I,J) * D1(I,I) + D1(I,I) * D1(I,I)
   DO 460 J=1,NH
   JJ=J+NH
460 D(K,I,II,NPL) = D(K,I,II,NPL) + D2(I,J) * D1(I,I) + D1(I,I) * D1(I,I)
   DO 460 J=1,NH
   JJ=J+NH
   U1(I,J) = 0
   DO 460 L=1,NH
   LL=L+NH
460 D(K,I,II,NPL) = D(K,I,II,NPL) + D2(I,J) * D1(I,I) + D1(I,I) * D1(I,I)
   DO 460 L=1,NH
   LL=L+NH

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THICK 1690
THICK 1691
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THICK 1743
THICK 1744

153


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    U(I,I,J) = U(I,I,J) + U2(I,I,J) * D(N,I,L) + D(N,I,L,I,J)
    460 D(I,I,I,J) = D(I,I,I,J) + D2(I,I,I,J) * D(N,I,L,I,J)
    DO 470 I=1,NH
    II=I+NH
    DO 470 J=1,NH
    JJ=J+NH
    DO 470 L=1,NH
    D(K,I,I,J) = D(K,I,I,J) + D(I,I,I,J) * D(N,I,L,I,J)
    470 D(K+1,I,I,J) = D(K+1,I,I,J) - D(I,I,L) * D(N,I,L,I,J)
    230 DO 480 I=1,NH
    II=I+NH
    DO 480 J=1,NH
    D(K+1,I,I,NPL) = D(K+1,I,I,NPL) + U(I,I,J) * D(N,I,NPL)
    480 D(K+1,I,I,NPL) = D(K+1,I,I,NPL) + D(I,I,I,J) * D(N,I,I,NPL)
    490 CONTINUE
    DO 500 I=1,NH
    II=I+NH
    DO 500 J=1,NH
    D(K+1,I,I,J) = U(I,I,J)
    500 D(K+1,I,I,J) = D(I,I,J)
    240 I=I+NH
    510 CONTINUE
    520 DO 530 I=1,NH
    II=I+NH
    530 D(NF,II,NPL) = D(NF,II,NPL) + 08(I,I)
    CALL INVERT(U2,NH,DEI)
    IF(NPT) 610,610,540
    540 CONTINUE
    WRITE (6,550)
    550 FORMAT(1H0,40X,45H SOLUTION IS BASED ON FOLLOWING C-SUB-M MATRIX)
    DO 561 I=1,NH
    561 WRITE(6,560)(U2(I,I),J=1,NH)
    560 FORMAT(1H , 22X,5E10.0)
    255 I
    (DM(I,I),I=1,NF)
    J=NF+1
    570 FORMAT(1H ,8H DET(U2),=,8E15.7)
    WRITE (6,580)
    I
    (DM(I,2),I=1,J)
    580 FORMAT(1H ,8H DET(UC),=,9E15.7)
    .DIAG=1.0
    DO 590 I=1,NH
    590 DIAG=DIAG*U2(I,I)
    WRITE (6,600) DEF,DIAG
    600 FORMAT(1H0, 12HD DETERMINANT=,E16.8,7H DIAG=,E14.7)
    610 CONTINUE
    DO 620 I=1,NH
    II=I+NH
    DM(NF,II)=0.0
    DO 620 J=1,NH
    JJ=J+NH
    620 D(NE,II,J) = U2(I,I,J)
    I=I+NH
    270 DO 760 N=1,NF
    K=NFP-N
    275 THICK 1745
    THICK 1746
    THICK 1747
    THICK 1748
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    THICK 1751
    THICK 1752
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    THICK 1799

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AFFFDL-TR-71-116

AFHDL-TR-71-116

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SUBROUTINE TRIANG TRACE CDC 6600 FTN V3.0-P243 OPT=0 06/23/71 02.12.12. PAGE 6
IF ( ITP(K)=1) 630,660,650 THICK 1800
630 IF (IST=1) 680,680,640 THICK 1801
640 IST=0 THICK 1802
GO TO 660 THICK 1803
650 IK=K+1 THICK 1804
ISTA2 THICK 1805
660 DO 670 I=1,NH THICK 1806
I=I+NH THICK 1807
DO 670 J=1,NH THICK 1808
JJ=J+NH THICK 1809
670 D(K,I,NPL) = D(K,I,NPL) + D(K,I,JJ) * DM(IK,JJ) THICK 1810
IF (ITP(K)=1) 680,680,700 THICK 1811
680 DO 690 I=1,NH THICK 1812
I=I+NH THICK 1813
690 D(K,I,NPL) = D(K,I,NPL) + DM(K,I,II) THICK 1814
700 DO 710 I=1,NH THICK 1815
II=I+NH THICK 1816
DM(K,I,II) = 0.0 THICK 1817
DO 710 L=1,NH THICK 1818
LL=L+NH THICK 1819
710 DM(K,I,II) = DM(K,I,II) + D(K,II,LL) * D(K,LL,NPL) THICK 1820
IF (ITP(K)=1) 740,740,720 THICK 1821
720 DO 730 I=1,NH THICK 1822
DO 730 J=1,NH THICK 1823
JJ=J+NH THICK 1824
730 D(K,I,NPL) = D(K,I,NPL) + D(K,I,JJ) * DM(IK,JJ) THICK 1825
740 DO 750 I=1,NH THICK 1826
II=I+NH THICK 1827
750 D(K,I,NPL) = D(K,I,NPL) + DM(K,I,II) THICK 1828
DO 760 I=1,NH THICK 1829
I=I+NH THICK 1830
DM(K,II) = 0.0 THICK 1831
DO 760 L=1,NH THICK 1832
760 DM(K,II) = DM(K,II) + D(K,II,L) * D(K,L,NPL) THICK 1833
770 DM(I,J)=GA(I) THICK 1834
IF (JTYPE=NE,IOR) RETURN FE86 54
JF=2 FE86 55
DO 771 K=1,IBRM FE86 56
IX=IPAR(K)+JF-2 FE86 57
DO 772 L=JFIX FE86 58
TEMP=DM(L,4) FE86 59
DM(L,4)=DM(L,7) FE86 60
772 DM(L,7)=TEMP FE86 61
771 JFIX+2 FE86 62
RETURN THICK 1835
END THICK 1836

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AFFDL-TR-71-116

ENTRY POINTS
2 TRIANG

VARIABLES	SN	TYPE	RELOCATION	GEOM	MAIL	ARRAY	REAL	ALFR	2317	ALFR	REAL	ARRAY	GEOM
1332 ALFL		REAL		GEOM					2317	ALFR	REAL	ARRAY	GEOM
740 AL1		REAL		MAIL				1060	AL2	REAL	ARRAY	MAIL	
0 B11		REAL		ARRAY							REAL	ARRAY	MAIL
240 B22		REAL		ARRAY				500	B44	REAL	ARRAY	MAIL	
620 B55		REAL		ARRAY				360	B66	REAL	ARRAY	MAIL	
12 CXS		REAL		ARRAY				25	C11	REAL	ARRAY	MAIL	
26 C12		REAL		XMIT				27	C22	REAL	XMIT	XMIT	
30 C86		REAL		XMIT				0	D	REAL	ARRAY	F.P.	
3462 DET		REAL						3466	DIAG	REAL			
0 DM		REAL		ARRAY				550	DSC	REAL	ARRAY	GEOM	
34 DUMM		REAL		ARRAY				3470	D1	REAL	ARRAY		
35 D11		REAL		XMIT				36	D12	REAL			XMIT
3521 D2		REAL		ARRAY				37	D22	REAL			XMIT
40 D86		REAL		XMIT				56	EME1	REAL			XMIT
57 EMTH		REAL		XMIT				77	ERP	REAL			RUN
31 E11		REAL		XMIT				32	E12	REAL			XMIT
33 E22		REAL		XMIT				34	E66	REAL			XMIT
50 F11		REAL		XMIT				51	E12	REAL			XMIT
52 F22		REAL		XMIT				0	GA	REAL	ARRAY	TRIA	
5 GB		REAL		ARRAY				64	G12	REAL			XMIT
65 G21		REAL		XMIT				26	HH	REAL			RUN
5 HTT		REAL		XMIT				60	H11	REAL			XMIT
61 H12		REAL		XMIT				63	H21	REAL			XMIT
62 H22		REAL		XMIT				41	H44	REAL			XMIT
42 H55		REAL		XMIT				43	H66	REAL			XMIT
3454 I		INTEGER						1250	IA	INTEGER	ARRAY	GEOM	
1262 I9		INTEGER		ARRAY				75	IBCL	INTEGER			RUN
0 IBR		INTEGER		XMIT				1	IBRM	INTEGER			XOUM
33 IFREG		INTEGER		RUN				3463	II	INTEGER			
3467 IK		INTEGER						13	INDEX	INTEGER			XMIT
74 ING		INTEGER		ARRAY				3416	IOR	INTEGER			
120 IPAR		INTEGER		ARRAY				0	IRC	INTEGER			F.P.
3457 IRR		INTEGER						1	ISH	INTEGER			XMIT
170 ISS		INTEGER		ARRAY				3456	ISI	INTEGER			
0 ISY		INTEGER						31	ITP	INTEGER	ARRAY	TRIA	
3452 JA		INTEGER		F.P.				3460	J	INTEGER			
3450 JF		INTEGER						3461	JJ	INTEGER			
30 JMAX		INTEGER		RUN				0	JRC	INTEGER			F.P.
0 JSY		INTEGER		F.P.				0	JTYPE	INTEGER			XOUM
27 J9		INTEGER		RUN				3451	K	INTEGER			F.P.
76 KORTN		INTEGER		RUN				0	KRC	INTEGER			
3453 L		INTEGER						3464	LL	INTEGER			
1060 MLY		INTEGER		ARRAY				31	M9	INTEGER			RUN
3465 N		INTEGER						24	NBR	INTEGER			XMIT
0 NDE		INTEGER		RUN				176	NF	INTEGER			TRIA
200 NFP		INTEGER		TRIA				175	NH	INTEGER			TRIA
177 NPL		INTEGER		TRIA				201	NPRT	INTEGER			TRIA
524 NTP		INTEGER		ARRAY				4	OMS9	REAL			XMIT

MAFFDL-TR-71-116

VARIABLES	SN	TYPE	RELOCATION
1A PC		REAL	XMIT
14 PN		REAL	XMIT
21 RH1	15 PL	REAL	MATL
23 RH3	1200 RHO	REAL	ARRAY
7 R2	22 RH2	REAL	XMIT
1 S	6 R1	REAL	XMIT
24 SA	10 R3	REAL	XMIT
	0 SI	REAL	GEOM
	11 SXN	REAL	ARRAY
3455 TEMP		REAL	XMIT
1333 TR	2153 TLI	REAL	GEOM
360 TR2	21A TR1	REAL	GEOM
20 T1	17 T0	REAL	ARRAY
	3552 UJ	REAL	XMIT
3603 U2		REAL	ARRAY
53 XH11	47 XE66	REAL	XMIT
55 XH22	54 XH12	REAL	XMIT
45 XK12	44 XK11	REAL	XMIT
3 XLD	46 XK22	REAL	XMIT
32 XOUT	2 XN	REAL	XMIT
2 Y	50 XPR	REAL	GEOM
14A Z	14 YD	REAL	ARRAY
	110A ZLY	REAL	ARRAY

FILE NAMES	MODE
TAPEA	FMT

EXTERNALS	TYPE	ARGS
INVERT		3

STATEMENT LABELS	MODE	TYPE	ARGS	INVERT
0 2			0 3	162 4
0 7			0 10	0 20
0 30			0 40	0 50
0 60	INACTIVE		0 70	0 80
0 90	INACTIVE		0 100	703 110
0 120			736 130	0 140
0 150	INACTIVE		0 160	0 170
1017 180			0 190	1067 200
0 210			0 220	0 230
0 240			1276 250	0 260
0 270	INACTIVE		0 280	0 290
1444 300			0 310	0 320
0 330	INACTIVE		0 340	1603 350
0 360			0 370	0 380
0 390			0 400	0 410
0 420			0 430	0 440
0 450			0 460	0 470
0 480			0 490	0 500
2607 510			2612 520	0 530
0 540	INACTIVE		3417 550	FMT
0 561			3431 570	FMT
0 590			3441 600	FMT
0 620			0 630	2744 610
3014 650			3017 660	0 640
3054 680			0 690	0 670
0 710			0 720	3107 700
3214 740			0 750	0 730
				0 760

SUBROUTINE TRIANG TRACE CDC 6600 FTN V3.0-P243 OPT=0 06/23/71 02.12.12. PAGE 9

STATEMENT LABELS
0 770 0 771 0 772

COMMON BLOCKS	LENGTH
RUN	64
MATL	720
XMTT	54
GEOM	1235
TRIA	130
.XDUM	2

STATISTICS

PROGRAM LENGTH	40308	2072
COMMON LENGTH	42358	2205

AFFDL-TR-71-116

SUBROUTINE INVERT		CDC 6600 FTN V3.0-P243 OPT=1	06/19/71	13:01.44.	PAGE 1
		SUBROUTINE INVERT (DP,MAX,DETERM)			THICK 1837
		DIMENSION DP(5,5),M(5),C(5)			THICK 1838
		DETERM = 1.			THICK 1839
		DO 10 I = 1, MAX			THICK 1840
05		M(I) = - I			THICK 1841
		10 CONTINUE			THICK 1842
		DO 140 II = 1, MAX			THICK 1843
		O = 0.0			THICK 1844
10		DO 60 K = 1, MAX			THICK 1845
		IF (M(K)) 20,20,60			THICK 1846
		20 DO 50 L = 1, MAX			THICK 1847
		IF (M(L)) 30,30,50			THICK 1848
		30 IF (ABS(D) = ABS(DP(K,L))) 40,40,50			THICK 1849
		40 LD = L			THICK 1850
15		KD = K			THICK 1851
		O = DP(K,L)			THICK 1852
		50 CONTINUE			THICK 1853
		60 CONTINUE			THICK 1854
20		IF (KD=LD) 70,80,70			THICK 1855
		70 DETERM = -DETERM			THICK 1856
		80 IF (D.NE.0.0) DETERM=DETERM*D			THICK 1857
		NEMP = -M(LD)			FEB6 63
		M(LD)=M(KD)			THICK 1858
		M(KD)= NEMP			THICK 1859
25		DO 90 I = 1, MAX			THICK 1860
		C(I) = DP(I,LD)			THICK 1861
		DP(I,LD) = DP(I,KD)			THICK 1862
		DP(I,KD) = 0.0			THICK 1863
		90 CONTINUE			THICK 1864
30		DP(KD,KD) = 1.			THICK 1865
		DO 100 J = 1, MAX			THICK 1866
		DR=0.0			THICK 1867
		IF (D.NE.0.0) DR=DP(KD,J)/D			FEB6 65
		DP(KD,J)=DR			FEB6 66
35		100 CONTINUE			THICK 1869
		DO 130 I = 1, MAX			THICK 1870
		IF (I=KD) 110,130,110			THICK 1871
		110 DO 120 J = 1, MAX			THICK 1872
		DP(I,J) = DP(I,J) - C(I)*DP(KD,J)			THICK 1873
40		120 CONTINUE			THICK 1874
		130 CONTINUE			THICK 1875
		140 CONTINUE			THICK 1876
		DO 170 I = 1, MAX			THICK 1877
		L=0			THICK 1878
45		150 L=L+1			THICK 1879
		IF (M(L)-I) 150,160,150			THICK 1880
		160 M(L)=M(I)			THICK 1881
		M(I)=I			THICK 1882
		DO 170 J = 1, MAX			THICK 1883
		TEMP = DP(L,J)			THICK 1884
50		DP(I,J) = DP(I,J)			THICK 1885
		DP(I,J) = TEMP			THICK 1886
		170 CONTINUE			THICK 1887
		RETURN			THICK 1888
55		END			THICK 1889

AFEDL-TR-71-116

AFFDL-TR-71-116

SYMBOLIC REFERENCE MAP

ENTRY POINTS
2 INVERT

VARIABLES SN TYPE RELOCATION

VARIABLES	SN	TYPE	RELOCATION
203 C	0	REAL	165 D
174 DR	0	REAL	DP
164 II	163	INTEGER	I
166 K	173	INTEGER	J
167 L	171	INTEGER	KD
176 M	170	INTEGER	LD
172 NEMP	0	INTEGER	MAX
	175	TEMP	

INLINE FUNCTIONS TYPE ARGS
ABS REAL 1 INTRIN

STATEMENT LABELS

STATEMENT LABELS	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150
	0	40	INACTIVE	47	50	60	80	0	110	INACTIVE	0	140	137	150	0	160
	0	70	INACTIVE	60	80	0	110	INACTIVE	0	140	137	150	0	160	0	170
	0	100	INACTIVE	0	110	INACTIVE	0	140	137	150	0	160	0	170	0	180
	0	130	INACTIVE	0	140	137	150	0	160	0	170	0	180	0	190	200
	0	160	INACTIVE	0	170	0	180	0	190	200	0	210	220	230	240	250

STATISTICS

PROGRAM LENGTH 243H 163

Contrails

AFFDL-TR-71-116

APPENDIX B
EXAMPLE PROBLEMS
COMPUTER OUTPUT

AFFDL-TR-71-116

TORSIONAL PROBLEM RESULTS
CASE I

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
WRIGHT PATTERSON AIR FORCE BASE FLIGHT DYNAMICS LABORATORY VERSION 22 JULY 1968
MODIFIED PROGRAM TO ACCOUNT FOR TRANSVERSE SHEAR DEFORMATION --WRITTEN BY R.L. CITRLEY
ANALYT. LABORATORIES INC., BERKELEY, CALIFORNIA 1 OCT 1968

STATIC ANALYSIS PARTS= 2 BRANCHES= 0 NUMBER OF SURFACES= 0

ANGLES OF ROTATION OF BOUNDARY CONDITIONS ARE ALFA=0. ALFRS=0.

PART NO 1

I= 0. SK= 1.0000E-01 IPAR= 4 INQ= 1 SHELL TYPE 2 NTP= 0 LAYERS MLY= 4

CYLINDRICAL SHELL NO 2 H= 6.65000E-01 R= 1.66750E+00 PHI= 90.000 DEGREES

AYER NO 1 FROM Z=3.3000E-03 TO Z=1.6000E-03

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.5000E-01 POISSONS RATIO NU= 3.5000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATFA=0. MASS DENSITY RHO=0.

AYER NO 2 FROM Z=1.6000E-03 TO Z= 0.

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.5000E-01 POISSONS RATIO NU= 3.5000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATFA=0. MASS DENSITY RHO=0.

AYER NO 3 FROM Z= 0.

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.5000E-01 POISSONS RATIO NU= 3.5000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATFA=0. MASS DENSITY RHO=0.

AYER NO 4 FROM Z= 1.6000E-03 TO Z= 3.3000E-03

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.5000E-01 POISSONS RATIO NU= 3.5000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATFA=0. MASS DENSITY RHO=0.

PART NO 2

I= 0. SK= 1.0000E-01 IPAR= 4 INQ= 1 SHELL TYPE 2 NTP= 0 LAYERS MLY= 4

CYLINDRICAL SHELL NO 2 H= 6.65000E-01 R= 1.66750E+00 PHI= 90.000 DEGREES

AYER NO 1 FROM Z=3.3000E-03 TO Z=1.6000E-03

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.5000E-01 POISSONS RATIO NU= 3.5000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATFA=0. MASS DENSITY RHO=0.

AYER NO 2 FROM Z=1.6000E-03 TO Z= 0.

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.5000E-01 POISSONS RATIO NU= 3.5000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATFA=0. MASS DENSITY RHO=0.

AYER NO 3 FROM Z= 0.

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.5000E-01 POISSONS RATIO NU= 3.5000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATFA=0. MASS DENSITY RHO=0.

AYER NO 4 FROM Z= 1.6000E-03 TO Z= 3.3000E-03

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.5000E-01 POISSONS RATIO NU= 3.5000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATFA=0. MASS DENSITY RHO=0.

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
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ANALYT LABORATORIES INC. BERKELEY, CALIFORNIA 1 OCT 1969

SURFACE NO 1 FOR FOURIER HARMONIC COS=0 THETA

BOUNDARY CONDITIONS AT STARTING EDGE 1=0. 3=0. 5=0. 7=0.
BOUNDARY CONDITIONS AT FINAL EDGE 2=0. 4=0. 6=0. 8 1.00000E+03

LOADS FOR PART NO 1 SURFACE NO 1

LOADS AT END OF THIS PART ARE Q=0. NPHI=0. PTHETA=0. TL=0. TU=0.
SURFACE AND TEMP LOADS ARE P=0. DF=0. MPH=0. N=0.

AT BEGINNING AND END OF PART 1 PARAMETERS ARE

R1= 0. P2= 5.9970E-01 P3= 1.0000E+00 SXN= 1.0000E+00 CKS= 0.
PNE= 0. P12= 0. PC= 0. P10= 0. P11= 0.
C11= 3.3846E-03 C12= 1.1844E-03 C22= 3.3846E-03 C66= 1.1000E-03 E11= 1.3553E-20 E12= 1.0164E-20
E22= 1.3553E-20 E66= 3.3846E-03 D12= 1.2286E-08 D18= 4.3001E-09 D22= 1.2286E-08 D46= 3.9230E-09
H11= 0. H12= 0. H22= 0. H21= 0. G12= 0. G21= 0.

YS ARE= -3.1000E-03 -1.6000E-03 0. 1.6000E-03 3.1000E-03

R1= 0. P2= 5.9970E-01 P3= 1.0000E+00 SXN= 1.0000E+00 CKS= 0.
PNE= 0. P12= 0. PC= 0. P10= 0. P11= 0.
C11= 3.3846E-03 C12= 1.1844E-03 C22= 3.3846E-03 C66= 1.1000E-03 E11= 1.3553E-20 E12= 1.0164E-20
E22= 1.3553E-20 E66= 3.3846E-03 D12= 1.2286E-08 D18= 4.3001E-09 D22= 1.2286E-08 D46= 3.9230E-09
H11= 0. H12= 0. H22= 0. H21= 0. G12= 0. G21= 0.

YS ARE= -3.1000E-03 -1.6000E-03 0. 1.6000E-03 3.1000E-03

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
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AERONAUT LABORATORIES, INC. BERKELEY, CALIFORNIA 1 OCT 1968

STATIC SOLUTION AT POINTS ALONG MERIDIAN FOR HAVE NUMBER NX = 0

MAIN SHELL PART NO 1

S	W	0	UPHI	MPHI	OPHI	MPHY	UITHETA	N	MTHETA	MTHETA
0.0000	0.	0.	0.	0.	0.	0.	0.	1.0000E+03	0.	0.
.02500	0.	0.	0.	0.	0.	0.	2.27272E+04	1.0000E+03	0.	0.
.02500	0.	0.	0.	0.	0.	0.	2.27272E+04	1.0000E+03	0.	0.
.05000	0.	0.	0.	0.	0.	0.	4.54545E+04	1.0000E+03	0.	0.
.05000	0.	0.	0.	0.	0.	0.	4.54545E+04	1.0000E+03	0.	0.
.07500	0.	0.	0.	0.	0.	0.	6.81817E+04	1.0000E+03	0.	0.
.07500	0.	0.	0.	0.	0.	0.	6.81817E+04	1.0000E+03	0.	0.
.10000	0.	0.	0.	0.	0.	0.	9.09090E+04	1.0000E+03	0.	0.

AFFDL-TR-71-116

MAIN SHELL PART NO 2

S	W	Q	UPHI	NPHI	APHI	MPHI	UTHETA	N	MTHETA	MTHETA
0.0000	0.	0.	0.	0.	0.	0.	9.020900E+04	1.00000E+03	0.	0.
.02500	0.	0.	0.	0.	0.	0.	1.13636E+05	1.00000E+03	0.	0.
.02500	0.	0.	0.	0.	0.	0.	1.13636E+05	1.00000E+03	0.	0.
.05000	0.	0.	0.	0.	0.	0.	1.36363E+05	1.00000E+03	0.	0.
.05000	0.	0.	0.	0.	0.	0.	1.36363E+05	1.00000E+03	0.	0.
.07500	0.	0.	0.	0.	0.	0.	1.59091E+05	1.00000E+03	0.	0.
.07500	0.	0.	0.	0.	0.	0.	1.59091E+05	1.00000E+03	0.	0.
.10000	0.	0.	0.	0.	0.	0.	1.81818E+05	1.00000E+03	0.	0.
.10000	0.	0.	0.	0.	0.	0.	1.81818E+05	1.00000E+03	0.	0.

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
MIGHT PATTERSON AIR FORCE BASE FLIGHT DYNAMICS LABORATORY VERSION 22 JULY 1968
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ANALYT. LABORATORIES, INC. BERKELEY, CALIFORNIA 1 OCT. 1969

STATIC SOLUTION AT POINTS ALONG MERIDIAN FOR WAVE NUMBER $N_m = 0$

MAIN SHELL PART NO 1

S	W	UPHI	UTHEA	RPHI	SPHI IN	SPHI OUT	SITHA IN	SITHA OUT	SFIH IN	SFIH OUT
0.0000	0.	0.	0.	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.0000	0.	0.	0.	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.0000	0.	0.	0.	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.0000	0.	0.	0.	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
0.2500	0.	0.	2.27272E+04	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.2500	0.	0.	2.27272E+04	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.2500	0.	0.	2.27272E+04	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.2500	0.	0.	2.27272E+04	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
0.5000	0.	0.	4.54545E+04	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.5000	0.	0.	4.54545E+04	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.5000	0.	0.	4.54545E+04	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.5000	0.	0.	4.54545E+04	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
0.7500	0.	0.	6.81817E+04	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.7500	0.	0.	6.81817E+04	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.7500	0.	0.	6.81817E+04	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.7500	0.	0.	6.81817E+04	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
1.0000	0.	0.	9.09090E+04	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
1.0000	0.	0.	9.09090E+04	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
1.0000	0.	0.	9.09090E+04	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
1.0000	0.	0.	9.09090E+04	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05

MAIN SHELL PART NO 2

S	W	UPHI	UTHTA	RPMI	SPMI IN	SPMT OUT	STHETA IN	STHETA OUT	SPITH IN	SPITH OUT
0.0000	0.	0.	9.09090E+04	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.0000	0.	0.	9.09090E+04	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.0000	0.	0.	9.09090E+04	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.0000	0.	0.	9.09090E+04	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
0.2500	0.	0.	1.13636E+05	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.2500	0.	0.	1.13636E+05	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.2500	0.	0.	1.13636E+05	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.2500	0.	0.	1.13636E+05	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
0.2500	0.	0.	1.13636E+05	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.2500	0.	0.	1.13636E+05	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.2500	0.	0.	1.13636E+05	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.2500	0.	0.	1.13636E+05	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
0.5000	0.	0.	1.36363E+05	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.5000	0.	0.	1.36363E+05	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.5000	0.	0.	1.36363E+05	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.5000	0.	0.	1.36363E+05	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
0.5000	0.	0.	1.36363E+05	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.5000	0.	0.	1.36363E+05	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.5000	0.	0.	1.36363E+05	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.5000	0.	0.	1.36363E+05	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
0.7500	0.	0.	1.59091E+05	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.7500	0.	0.	1.59091E+05	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.7500	0.	0.	1.59091E+05	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.7500	0.	0.	1.59091E+05	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05
0.7500	0.	0.	1.59091E+05	0.	0.	0.	0.	0.	1.51215E+05	1.51370E+05
0.7500	0.	0.	1.59091E+05	0.	0.	0.	0.	0.	1.51370E+05	1.51515E+05
0.7500	0.	0.	1.59091E+05	0.	0.	0.	0.	0.	1.51515E+05	1.51660E+05
0.7500	0.	0.	1.59091E+05	0.	0.	0.	0.	0.	1.51660E+05	1.51815E+05

AFFDL-TR-71-116

AFFDL-TR-71-116

INTERNAL PRESSURE LOADING
CASE II

REFERENCES

---UNSATISFIED EXTERNALS----

THICK

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
WRIGHT PATTERSON AIR FORCE BASE FLIGHT DYNAMICS LABORATORY VERSION, 22 JULY 1968
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ANALYT LABORATORIES INC. BERKELEY, CALIFORNIA 1 OCT, 1969

STATIC ANALYSIS PARTS= 1 BRANCHES= 0 NUMBER OF SUBCASES= 0

ANGLES OF ROTATION OF BOUNDARY CONDITIONS ARE ALFL=0. ALFRS=0.

PART NO 1

SK= 7.00000E+00 IPR= 28 INW= 1 SHELL TYPE 2 NTP= 0 LAYERS MLY= 4

CYLINDRICAL SHELL NO 2 R= 5.65000E+01 R= 1.86750E+00 PHIS= 90:000 DEGREES

LAYER NO 1 FROM Z= 3.12500E+01 TO Z= 1.86250E+01

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.50000E+05 POISSONS RATIO NUM 3.50000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATHTA=0. MASS DENSITY RHO=0.

LAYER NO 2 FROM Z= 1.86250E+01 TO Z= 0.

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.50000E+05 POISSONS RATIO NUM 3.50000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATHTA=0. MASS DENSITY RHO=0.

LAYER NO 3 FROM Z= 0.

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.50000E+05 POISSONS RATIO NUM 3.50000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATHTA=0. MASS DENSITY RHO=0.

LAYER NO 4 FROM Z= 1.86250E+01 TO Z= 3.12500E+01

CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.50000E+05 POISSONS RATIO NUM 3.50000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI=0. ATHTA=0. MASS DENSITY RHO=0.

AFFDL-TR-71-116

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
WRIGHT PATTERSON AIR FORCE BASE FLIGHT DYNAMICS LABORATORY VERSION, 22 JULY 1968
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ANAMEY LABORATORIES INC. BERKELEY, CALIFORNIA 1 OCT 1969

SUBCASE NO 1 FOR FOURIER HARMONIC COS=0 THETA

BOUNDARY CONDITIONS AT STARTING EDGE 2 0. 4 0. 6 0.

BOUNDARY CONDITIONS AT FINAL EDGE 2 0. 3 0. 5 0.

LOADS FOR PART NO 1 SUBCASE NO 1

RING LOADS AT END OF THIS PART ARE Q=0.0 NPHI=0.0 MPMI=0.0 NT P =0.0 M TPE=0.0

SURFACE AND TEMP LOADS ARE P= 3.00000E+02 PFI=0.0 PTHETA=0.0 TLE=0.0 TUS=0.0

- AFFDL TR 71-916

67763

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
WRIGHT PATTERSON AIR FORCE BASE FLIGHT DYNAMICS LABORATORY VERSION 22 JULY 1968
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ANAMEY LABORATORIES INC. BERKELEY, CALIFORNIA 1 OCT 1969

STATIC SOLUTION AT POINTS ALONG MERIDIAN FOR WAVE NUMBER MX= 0

MAIN SHELL PART NO 1

WAVE NUMBER	U	V	U THETA	BETA PHI	BETA THETA	LAYER NO	STO PHI	STO THETA	STO T P	TAU PHI	TAU THETA	
0.0000	2.0701E-03	1.0930E-04	0.0	7.5096E-05	0.0	1	IN	7.4766E+00	7.0041E+02	0.0	0.0	
						OUT	-6.3108E-01	6.2023E+02	0.0	0.0		
						2	IN	-8.3108E-01	6.2023E+02	0.0	0.0	
						OUT	-2.9929E+00	5.5717E+02	0.0	0.0		
	.2500	2.0390E-03	-2.1613E-18	0.0	1.3751E-04	0.0	1	IN	4.9906E+00	4.6753E+02	0.0	0.0
							OUT	2.9144E+00	6.1231E+02	0.0	-5.9749E+00	
							2	IN	2.9144E+00	6.1231E+02	0.0	-5.9749E+00
							OUT	-2.5099E+00	5.4938E+02	0.0	-5.9749E+00	
.4500	2.0001E-03	4.0552E-06	0.0	1.0375E-04	0.0	1	IN	2.6780E+02	7.6097E+02	0.0	-5.9749E+00	
						OUT	2.3507E+02	6.9347E+02	0.0	-5.9749E+00		
						2	IN	2.3507E+02	6.9347E+02	0.0	-5.9749E+00	
						OUT	2.0866E+02	6.2377E+02	0.0	-5.9749E+00		
.6500	2.0001E-03	4.0552E-06	0.0	1.0227E-04	0.0	1	IN	2.7733E+02	7.1766E+02	0.0	-5.9749E+00	
						OUT	2.4033E+02	6.8365E+02	0.0	-2.3888E+00		
						2	IN	2.4033E+02	6.8365E+02	0.0	-2.3888E+00	
						OUT	2.0927E+02	6.1300E+02	0.0	-2.3888E+00		
.8500	1.9706E-03	-6.1040E-06	0.0	9.8445E-05	0.0	1	IN	2.8012E+02	7.6280E+02	0.0	-2.3888E+00	
						OUT	2.4188E+02	6.7534E+02	0.0	-1.8659E-01		
						2	IN	2.4188E+02	6.7534E+02	0.0	-1.8659E-01	
						OUT	2.0949E+02	6.0511E+02	0.0	-1.8659E-01		
.7500	1.9706E-03	-6.1040E-06	0.0	9.8445E-05	0.0	1	IN	2.8012E+02	7.6280E+02	0.0	-2.3888E+00	
						OUT	2.4188E+02	6.7534E+02	0.0	-1.8659E-01		
						2	IN	2.4188E+02	6.7534E+02	0.0	-1.8659E-01	
						OUT	2.0949E+02	6.0511E+02	0.0	-1.8659E-01		
.7500	1.9706E-03	-6.1040E-06	0.0	9.8445E-05	0.0	1	IN	2.8012E+02	7.6280E+02	0.0	-2.3888E+00	
						OUT	2.4188E+02	6.7534E+02	0.0	-1.8659E-01		
						2	IN	2.4188E+02	6.7534E+02	0.0	-1.8659E-01	
						OUT	2.0949E+02	6.0511E+02	0.0	-1.8659E-01		

AFFDL-TR-71-116

1.7500	1.6257E-03	5.9908E-06	1.6257E-03	0.	1	IN	2.6839E+02	7.4305E+02	0.	1.3566E+00	0.
						OUT	2.3575E+02	6.5975E+02	0.	1.3566E+00	0.
2					2	IN	2.3575E+02	6.5975E+02	0.	1.3566E+00	0.
						OUT	2.0884E+02	5.9278E+02	0.	1.3566E+00	0.
3					3	IN	2.0884E+02	5.9278E+02	0.	1.3566E+00	0.
						OUT	1.8608E+02	5.3770E+02	0.	1.3566E+00	0.
4					4	IN	1.8608E+02	5.3770E+02	0.	1.3566E+00	0.
						OUT	1.6645E+02	4.9155E+02	0.	1.3566E+00	0.
2.0000	1.9248E-03	4.6446E-06	6.7441E-06	0.	1	IN	2.6540E+02	7.4169E+02	0.	1.0810E+00	0.
						OUT	2.3417E+02	6.5891E+02	0.	1.0810E+00	0.
2					2	IN	2.3417E+02	6.5891E+02	0.	1.0810E+00	0.
						OUT	2.0865E+02	5.9245E+02	0.	1.0810E+00	0.
3					3	IN	2.0865E+02	5.9245E+02	0.	1.0810E+00	0.
						OUT	1.8730E+02	5.3789E+02	0.	1.0810E+00	0.
4					4	IN	1.8730E+02	5.3789E+02	0.	1.0810E+00	0.
						OUT	1.6906E+02	4.9224E+02	0.	1.0810E+00	0.
2.2500	1.9253E-03	3.9711E-06	8.4869E-07	0.	1	IN	2.6313E+02	7.4106E+02	0.	1.0810E+00	0.
						OUT	2.3296E+02	6.5664E+02	0.	1.0810E+00	0.
2					2	IN	2.3296E+02	6.5664E+02	0.	1.0810E+00	0.
						OUT	2.0851E+02	5.9254E+02	0.	1.0810E+00	0.
3					3	IN	2.0851E+02	5.9254E+02	0.	1.0810E+00	0.
						OUT	1.8822E+02	5.3834E+02	0.	1.0810E+00	0.
4					4	IN	1.8822E+02	5.3834E+02	0.	1.0810E+00	0.
						OUT	1.7106E+02	4.9308E+02	0.	1.0810E+00	0.
2.5000	1.9258E-03	3.9711E-06	8.4869E-07	0.	1	IN	2.6153E+02	7.4092E+02	0.	1.0810E+00	0.
						OUT	2.3211E+02	6.5871E+02	0.	1.0810E+00	0.
2					2	IN	2.3211E+02	6.5871E+02	0.	1.0810E+00	0.
						OUT	2.0841E+02	5.9284E+02	0.	1.0810E+00	0.
3					3	IN	2.0841E+02	5.9284E+02	0.	1.0810E+00	0.
						OUT	1.8887E+02	5.3877E+02	0.	1.0810E+00	0.
4					4	IN	1.8887E+02	5.3877E+02	0.	1.0810E+00	0.
						OUT	1.7246E+02	4.9383E+02	0.	1.0810E+00	0.
2.7500	1.9279E-03	2.9517E-06	-3.7215E-06	0.	1	IN	2.6050E+02	7.4102E+02	0.	1.0810E+00	0.
						OUT	2.3158E+02	6.5894E+02	0.	1.0810E+00	0.
2					2	IN	2.3158E+02	6.5894E+02	0.	1.0810E+00	0.
						OUT	2.0834E+02	5.9320E+02	0.	1.0810E+00	0.
3					3	IN	2.0834E+02	5.9320E+02	0.	1.0810E+00	0.
						OUT	1.8929E+02	5.3936E+02	0.	1.0810E+00	0.
4					4	IN	1.8929E+02	5.3936E+02	0.	1.0810E+00	0.

AFFDL-TR-71-116

2.7500	1.9279E-03	2.79917E-06	0.	3.2507E-01	0.	3.2507E-01	0.
	OUT	1.7337E+02	4.8446E+02				
	1	IN	2.8050E+02	7.4104E+02	0.	3.2507E-01	0.
		OUT	2.3156E+02	6.5894E+02	0.	3.2507E-01	0.
	2	IN	2.3156E+02	6.5894E+02	0.	3.2507E-01	0.
		OUT	2.0834E+02	5.9320E+02	0.	3.2507E-01	0.
	3	IN	2.0834E+02	5.9320E+02	0.	3.2507E-01	0.
		OUT	1.8929E+02	5.3936E+02	0.	3.2507E-01	0.
	4	IN	1.8929E+02	5.3936E+02	0.	3.2507E-01	0.
3.2500	1.9293E-03	2.7968E-06	0.	1.7263E-01	0.	1.7263E-01	0.
	1	IN	2.5991E+02	7.4128E+02	0.	1.7263E-01	0.
		OUT	2.3124E+02	6.5923E+02	0.	1.7263E-01	0.
	2	IN	2.3124E+02	6.5923E+02	0.	1.7263E-01	0.
		OUT	2.0830E+02	5.9355E+02	0.	1.7263E-01	0.
	3	IN	2.0830E+02	5.9355E+02	0.	1.7263E-01	0.
		OUT	1.8954E+02	5.3978E+02	0.	1.7263E-01	0.
	4	IN	1.8954E+02	5.3978E+02	0.	1.7263E-01	0.
3.2500	1.9304E-03	2.7320E-06	0.	6.9445E-02	0.	6.9445E-02	0.
	1	IN	2.5991E+02	7.4128E+02	0.	6.9445E-02	0.
		OUT	2.3109E+02	6.5911E+02	0.	6.9445E-02	0.
	2	IN	2.3109E+02	6.5911E+02	0.	6.9445E-02	0.
		OUT	2.0828E+02	5.9344E+02	0.	6.9445E-02	0.
	3	IN	2.0828E+02	5.9344E+02	0.	6.9445E-02	0.
		OUT	1.8965E+02	5.4009E+02	0.	6.9445E-02	0.
	4	IN	1.8965E+02	5.4009E+02	0.	6.9445E-02	0.
3.2500	1.9304E-03	2.7320E-06	0.	6.9445E-02	0.	6.9445E-02	0.
	1	IN	2.5991E+02	7.4128E+02	0.	6.9445E-02	0.
		OUT	2.3109E+02	6.5911E+02	0.	6.9445E-02	0.
	2	IN	2.3109E+02	6.5911E+02	0.	6.9445E-02	0.
		OUT	2.0828E+02	5.9344E+02	0.	6.9445E-02	0.
	3	IN	2.0828E+02	5.9344E+02	0.	6.9445E-02	0.
		OUT	1.8965E+02	5.4009E+02	0.	6.9445E-02	0.
	4	IN	1.8965E+02	5.4009E+02	0.	6.9445E-02	0.
3.5000	1.9312E-03	2.71024E-06	0.	5.9898E-03	0.	5.9898E-03	0.
	1	IN	2.5954E+02	7.4142E+02	0.	5.9898E-03	0.
		OUT	2.3104E+02	6.5975E+02	0.	5.9898E-03	0.
	2	IN	2.3104E+02	6.5975E+02	0.	5.9898E-03	0.
		OUT	2.0828E+02	5.9407E+02	0.	5.9898E-03	0.
	3	IN	2.0828E+02	5.9407E+02	0.	5.9898E-03	0.
		OUT	1.8968E+02	5.4031E+02	0.	5.9898E-03	0.
	4	IN	1.8968E+02	5.4031E+02	0.	5.9898E-03	0.
3.5000	1.9312E-03	2.71024E-06	0.	5.9898E-03	0.	5.9898E-03	0.
	1	IN	2.5954E+02	7.4142E+02	0.	5.9898E-03	0.
		OUT	2.3104E+02	6.5975E+02	0.	5.9898E-03	0.
	2	IN	2.3104E+02	6.5975E+02	0.	5.9898E-03	0.
		OUT	2.0828E+02	5.9407E+02	0.	5.9898E-03	0.
	3	IN	2.0828E+02	5.9407E+02	0.	5.9898E-03	0.
		OUT	1.8968E+02	5.4031E+02	0.	5.9898E-03	0.
	4	IN	1.8968E+02	5.4031E+02	0.	5.9898E-03	0.
3.7500	1.9318E-03	2.7229E-06	0.	2.8089E-02	0.	2.8089E-02	0.
	1	IN	2.5954E+02	7.4142E+02	0.	2.8089E-02	0.
		OUT	2.3106E+02	6.5944E+02	0.	2.8089E-02	0.
	2	IN	2.3106E+02	6.5944E+02	0.	2.8089E-02	0.
		OUT	2.0828E+02	5.9423E+02	0.	2.8089E-02	0.
	3	IN	2.0828E+02	5.9423E+02	0.	2.8089E-02	0.
		OUT	1.8967E+02	5.4045E+02	0.	2.8089E-02	0.
	4	IN	1.8967E+02	5.4045E+02	0.	2.8089E-02	0.
3.7500	1.9318E-03	2.7229E-06	0.	2.8089E-02	0.	2.8089E-02	0.
	1	IN	2.5954E+02	7.4142E+02	0.	2.8089E-02	0.
		OUT	2.3106E+02	6.5944E+02	0.	2.8089E-02	0.
	2	IN	2.3106E+02	6.5944E+02	0.	2.8089E-02	0.
		OUT	2.0828E+02	5.9423E+02	0.	2.8089E-02	0.
	3	IN	2.0828E+02	5.9423E+02	0.	2.8089E-02	0.
		OUT	1.8967E+02	5.4045E+02	0.	2.8089E-02	0.
	4	IN	1.8967E+02	5.4045E+02	0.	2.8089E-02	0.

AFFDL-TR-71-116

37500	1.9322E-03	1.7229E-06	0.	2.0825E-06	0.	1.8967E+02	5.4045E+02	0.	-2.8089E-02	0.
						1.8967E+02	5.4045E+02	0.	-2.8089E-02	0.
						1.7420E+02	4.9563E+02	0.	-2.8089E-02	0.
1	IN	2.5986E+02	7.4203E+02	0.		2.5986E+02	7.4203E+02	0.	-2.8089E-02	0.
						2.3106E+02	6.5994E+02	0.	-2.8089E-02	0.
2	IN	2.3106E+02	6.5994E+02	0.		2.3106E+02	6.5994E+02	0.	-2.8089E-02	0.
						2.0828E+02	5.9423E+02	0.	-2.8089E-02	0.
3	IN	2.0828E+02	5.9423E+02	0.		2.0828E+02	5.9423E+02	0.	-2.8089E-02	0.
						1.8967E+02	5.4045E+02	0.	-2.8089E-02	0.
4	IN	1.8967E+02	5.4045E+02	0.		1.8967E+02	5.4045E+02	0.	-2.8089E-02	0.
						1.7420E+02	4.9563E+02	0.	-2.8089E-02	0.
1	OUT	1.7420E+02	4.9563E+02	0.		1.7420E+02	4.9563E+02	0.	-2.8089E-02	0.
4.0800	1.9322E-03	1.7670E-06	0.	-1.4120E-06	0.	2.5986E+02	7.4219E+02	0.	-4.2174E-02	0.
						2.5986E+02	7.4219E+02	0.	-4.2174E-02	0.
						2.3111E+02	6.6007E+02	0.	-4.2174E-02	0.
1	IN	2.3111E+02	6.6007E+02	0.		2.3111E+02	6.6007E+02	0.	-4.2174E-02	0.
						2.0828E+02	5.9433E+02	0.	-4.2174E-02	0.
2	IN	2.0828E+02	5.9433E+02	0.		2.0828E+02	5.9433E+02	0.	-4.2174E-02	0.
						1.8964E+02	5.4053E+02	0.	-4.2174E-02	0.
3	IN	1.8964E+02	5.4053E+02	0.		1.8964E+02	5.4053E+02	0.	-4.2174E-02	0.
						1.7412E+02	4.9569E+02	0.	-4.2174E-02	0.
4	IN	1.7412E+02	4.9569E+02	0.		1.7412E+02	4.9569E+02	0.	-4.2174E-02	0.
						2.5977E+02	7.4219E+02	0.	-4.2174E-02	0.
1	IN	2.5977E+02	7.4219E+02	0.		2.5977E+02	7.4219E+02	0.	-4.2174E-02	0.
						2.3116E+02	6.6015E+02	0.	-4.2174E-02	0.
2	IN	2.3116E+02	6.6015E+02	0.		2.3116E+02	6.6015E+02	0.	-4.2174E-02	0.
						2.0829E+02	5.9440E+02	0.	-4.2174E-02	0.
3	IN	2.0829E+02	5.9440E+02	0.		2.0829E+02	5.9440E+02	0.	-4.2174E-02	0.
						1.8959E+02	5.4057E+02	0.	-4.2174E-02	0.
4	IN	1.8959E+02	5.4057E+02	0.		1.8959E+02	5.4057E+02	0.	-4.2174E-02	0.
						1.8959E+02	5.4057E+02	0.	-4.2174E-02	0.
4.2500	1.9324E-03	1.6238E-06	0.	-8.7000E-07	0.	2.5977E+02	7.4219E+02	0.	-4.3880E-02	0.
						2.5977E+02	7.4219E+02	0.	-4.3880E-02	0.
						2.3116E+02	6.6015E+02	0.	-4.3880E-02	0.
1	IN	2.3116E+02	6.6015E+02	0.		2.3116E+02	6.6015E+02	0.	-4.3880E-02	0.
						2.0829E+02	5.9440E+02	0.	-4.3880E-02	0.
2	IN	2.0829E+02	5.9440E+02	0.		2.0829E+02	5.9440E+02	0.	-4.3880E-02	0.
						1.8959E+02	5.4057E+02	0.	-4.3880E-02	0.
3	IN	1.8959E+02	5.4057E+02	0.		1.8959E+02	5.4057E+02	0.	-4.3880E-02	0.
						1.7403E+02	4.9571E+02	0.	-4.3880E-02	0.
4	IN	1.7403E+02	4.9571E+02	0.		1.7403E+02	4.9571E+02	0.	-4.3880E-02	0.
						2.5986E+02	7.4237E+02	0.	-3.8867E-02	0.
4.5000	1.9325E-03	1.58858E-06	0.	-6.6831E-07	0.	2.5986E+02	7.4237E+02	0.	-3.8867E-02	0.
						2.5986E+02	7.4237E+02	0.	-3.8867E-02	0.
						2.3122E+02	6.6020E+02	0.	-3.8867E-02	0.
1	IN	2.3122E+02	6.6020E+02	0.		2.3122E+02	6.6020E+02	0.	-3.8867E-02	0.
						2.0830E+02	5.9443E+02	0.	-3.8867E-02	0.
2	IN	2.0830E+02	5.9443E+02	0.		2.0830E+02	5.9443E+02	0.	-3.8867E-02	0.
						1.8955E+02	5.4058E+02	0.	-3.8867E-02	0.
3	IN	1.8955E+02	5.4058E+02	0.		1.8955E+02	5.4058E+02	0.	-3.8867E-02	0.
						1.8955E+02	5.4058E+02	0.	-3.8867E-02	0.
4	IN	1.8955E+02	5.4058E+02	0.		1.8955E+02	5.4058E+02	0.	-3.8867E-02	0.
						1.7394E+02	4.9570E+02	0.	-3.8867E-02	0.
4.5800	1.9325E-03	1.4856E-06	0.	-4.6831E-07	0.	2.5986E+02	7.4237E+02	0.	-3.8867E-02	0.
						2.5986E+02	7.4237E+02	0.	-3.8867E-02	0.
						2.3122E+02	6.6020E+02	0.	-3.8867E-02	0.
1	IN	2.3122E+02	6.6020E+02	0.		2.3122E+02	6.6020E+02	0.	-3.8867E-02	0.
						2.0830E+02	5.9443E+02	0.	-3.8867E-02	0.
2	IN	2.0830E+02	5.9443E+02	0.		2.0830E+02	5.9443E+02	0.	-3.8867E-02	0.
						1.8955E+02	5.4058E+02	0.	-3.8867E-02	0.
3	IN	1.8955E+02	5.4058E+02	0.		1.8955E+02	5.4058E+02	0.	-3.8867E-02	0.
						1.7394E+02	4.9570E+02	0.	-3.8867E-02	0.
4	IN	1.7394E+02	4.9570E+02	0.		1.7394E+02	4.9570E+02	0.	-3.8867E-02	0.
						2.5996E+02	7.4241E+02	0.	-3.1013E-02	0.
4.7500	1.9326E-03	1.3478E-06	0.	-1.9413E-07	0.	2.5996E+02	7.4241E+02	0.	-3.1013E-02	0.
						2.5996E+02	7.4241E+02	0.	-3.1013E-02	0.
						2.3128E+02	6.6023E+02	0.	-3.1013E-02	0.
1	IN	2.3128E+02	6.6023E+02	0.		2.3128E+02	6.6023E+02	0.	-3.1013E-02	0.
						2.0831E+02	6.6023E+02	0.	-3.1013E-02	0.
2	IN	2.0831E+02	6.6023E+02	0.		2.0831E+02	6.6023E+02	0.	-3.1013E-02	0.

AFFDL-TR-71-1116

47500	1.9225E-03	1.23870E+06	0.	-1.9413E+07	0.	1	IN	2.5996E+02	7.4243E+02	0.	-3.1013E-02	0.
						2	IN	2.3126E+02	6.6023E+02	0.	-3.1013E-02	0.
						3	IN	2.0830E+02	5.9443E+02	0.	-3.1013E-02	0.
						4	IN	1.8952E+02	5.4056E+02	0.	-3.1013E-02	0.
						1	OUT	1.7387E+02	4.9566E+02	0.	-3.1013E-02	0.
						2	OUT	1.8952E+02	5.4056E+02	0.	-3.1013E-02	0.
						3	OUT	2.0830E+02	6.6023E+02	0.	-3.1013E-02	0.
						4	OUT	2.3126E+02	7.4243E+02	0.	-3.1013E-02	0.
5.0000	1.9225E-03	1.2080E+06	0.	-2.3658E+08	0.	1	IN	2.6003E+02	7.4243E+02	0.	-2.2739E-02	0.
						2	IN	2.3130E+02	6.6023E+02	0.	-2.2739E-02	0.
						3	IN	2.0831E+02	5.9443E+02	0.	-2.2739E-02	0.
						4	IN	1.8949E+02	5.4056E+02	0.	-2.2739E-02	0.
						1	OUT	1.7381E+02	4.9566E+02	0.	-2.2739E-02	0.
						2	OUT	2.0831E+02	6.6023E+02	0.	-2.2739E-02	0.
						3	OUT	2.3132E+02	7.4243E+02	0.	-2.2739E-02	0.
						4	OUT	2.6007E+02	8.243E+02	0.	-2.2739E-02	0.
5.2500	1.9225E-03	1.0650E+06	0.	6.9487E-08	0.	1	IN	2.6007E+02	7.4243E+02	0.	-1.5377E-02	0.
						2	IN	2.3132E+02	6.6023E+02	0.	-1.5377E-02	0.
						3	IN	2.0831E+02	5.9443E+02	0.	-1.5377E-02	0.
						4	IN	1.8947E+02	5.4056E+02	0.	-1.5377E-02	0.
						1	OUT	1.7377E+02	4.9566E+02	0.	-1.5377E-02	0.
						2	OUT	2.0831E+02	6.6023E+02	0.	-1.5377E-02	0.
						3	OUT	2.3132E+02	7.4243E+02	0.	-1.5377E-02	0.
						4	OUT	2.6007E+02	8.243E+02	0.	-1.5377E-02	0.
5.5000	1.9225E-03	1.0550E+06	0.	6.9687E-08	0.	1	IN	2.6007E+02	7.4243E+02	0.	-1.5377E-02	0.
						2	IN	2.3132E+02	6.6023E+02	0.	-1.5377E-02	0.
						3	IN	2.0831E+02	5.9443E+02	0.	-1.5377E-02	0.
						4	IN	1.8947E+02	5.4056E+02	0.	-1.5377E-02	0.
						1	OUT	1.7377E+02	4.9566E+02	0.	-1.5377E-02	0.
						2	OUT	2.0831E+02	6.6023E+02	0.	-1.5377E-02	0.
						3	OUT	2.3132E+02	7.4243E+02	0.	-1.5377E-02	0.
						4	OUT	2.6007E+02	8.243E+02	0.	-1.5377E-02	0.
5.7500	1.9225E-03	1.0941E+07	0.	1.0941E+07	0.	1	IN	2.6010E+02	7.4243E+02	0.	-9.5125E-03	0.
						2	IN	2.3134E+02	6.6023E+02	0.	-9.5125E-03	0.
						3	IN	2.0831E+02	5.9443E+02	0.	-9.5125E-03	0.
						4	IN	1.8946E+02	5.4056E+02	0.	-9.5125E-03	0.
						1	OUT	1.7374E+02	4.9566E+02	0.	-9.5125E-03	0.
						2	OUT	2.0831E+02	6.6023E+02	0.	-9.5125E-03	0.
						3	OUT	2.3135E+02	7.4243E+02	0.	-9.5125E-03	0.
						4	OUT	2.6012E+02	8.243E+02	0.	-9.5125E-03	0.
5.9000	1.9225E-03	1.0941E+07	0.	1.0941E+07	0.	1	IN	2.6010E+02	7.4243E+02	0.	-9.5125E-03	0.
						2	IN	2.3134E+02	6.6023E+02	0.	-9.5125E-03	0.
						3	IN	2.0831E+02	5.9443E+02	0.	-9.5125E-03	0.
						4	IN	1.8946E+02	5.4056E+02	0.	-9.5125E-03	0.
						1	OUT	1.7374E+02	4.9566E+02	0.	-9.5125E-03	0.
						2	OUT	2.0831E+02	6.6023E+02	0.	-9.5125E-03	0.
						3	OUT	2.3135E+02	7.4243E+02	0.	-9.5125E-03	0.
						4	OUT	2.6012E+02	8.243E+02	0.	-9.5125E-03	0.

AFFDL-TR-71-116

67777

5.7500	1.9224E-03=7.6969E=07 0.	1.1366E=07 0.	OUT 2.3135E+02 6.6022E+02 0.	-5.2569E-03 0.
			IN 2.3135E+02 6.6022E+02 0.	-5.2569E-03 0.
2			OUT 2.0831E+02 5.9440E+02 0.	-5.2569E-03 0.
			IN 2.0831E+02 5.9440E+02 0.	-5.2569E-03 0.
3			OUT 1.8945E+02 4.9512E+02 0.	-5.2569E-03 0.
			IN 1.8945E+02 4.9512E+02 0.	-5.2569E-03 0.
4			OUT 1.7372E+02 4.9559E+02 0.	-5.2569E-03 0.
			IN 1.7372E+02 4.9559E+02 0.	-5.2569E-03 0.
6.0000	1.9224E-03=6.1824E=07 0.	1.0277E=07 0.	OUT 2.3135E+02 6.6021E+02 0.	-2.4586E-03 0.
			IN 2.3135E+02 6.6021E+02 0.	-2.4586E-03 0.
2			OUT 2.0831E+02 5.9439E+02 0.	-2.4586E-03 0.
			IN 2.0831E+02 5.9439E+02 0.	-2.4586E-03 0.
3			OUT 1.8945E+02 4.9511E+02 0.	-2.4586E-03 0.
			IN 1.8945E+02 4.9511E+02 0.	-2.4586E-03 0.
4			OUT 1.7372E+02 4.9558E+02 0.	-2.4586E-03 0.
			IN 1.7372E+02 4.9558E+02 0.	-2.4586E-03 0.
6.2500	1.9224E-03=4.8504E=07 0.	8.0340E=08 0.	OUT 2.3135E+02 6.6020E+02 0.	-8.4246E-04 0.
			IN 2.3135E+02 6.6020E+02 0.	-8.4246E-04 0.
2			OUT 2.0831E+02 5.9439E+02 0.	-8.4246E-04 0.
			IN 2.0831E+02 5.9439E+02 0.	-8.4246E-04 0.
3			OUT 1.8945E+02 4.9511E+02 0.	-8.4246E-04 0.
			IN 1.8945E+02 4.9511E+02 0.	-8.4246E-04 0.
4			OUT 1.7372E+02 4.9558E+02 0.	-8.4246E-04 0.
			IN 1.7372E+02 4.9558E+02 0.	-8.4246E-04 0.
6.5000	1.9224E-03=3.1064E=07 0.	5.4167E=08 0.	OUT 2.3135E+02 6.6020E+02 0.	-9.5594E-05 0.
			IN 2.3135E+02 6.6020E+02 0.	-9.5594E-05 0.
2			OUT 2.0831E+02 5.9438E+02 0.	-9.5594E-05 0.
			IN 2.0831E+02 5.9438E+02 0.	-9.5594E-05 0.
3			OUT 1.8945E+02 4.9509E+02 0.	-9.5594E-05 0.
			IN 1.8945E+02 4.9509E+02 0.	-9.5594E-05 0.
4			OUT 1.7372E+02 4.9558E+02 0.	-9.5594E-05 0.
			IN 1.7372E+02 4.9558E+02 0.	-9.5594E-05 0.

6.7500	1.9323E-03=1.5549E-07	0.	2.7079E-08	0.	OUT	1.7372E+02	4.9558E+02	0.	-9.5594E-05	0.
					IN	2.6013E+02	7.4240E+02	0.	8.9210E-05	0.
					OUT	2.3136E+02	6.6019E+02	0.	8.9210E-05	0.
					IN	2.3136E+02	6.6019E+02	0.	8.9210E-05	0.
					OUT	2.0831E+02	5.9438E+02	0.	8.9210E-05	0.
					IN	2.0831E+02	5.9438E+02	0.	8.9210E-05	0.
					OUT	1.8945E+02	5.4050E+02	0.	8.9210E-05	0.
					IN	1.8945E+02	5.4050E+02	0.	8.9210E-05	0.
					OUT	1.7372E+02	4.9558E+02	0.	8.9210E-05	0.
6.7500	1.9323E-03=1.5549E-07	0.	2.7079E-08	0.	IN	2.6013E+02	7.4240E+02	0.	8.9210E-05	0.
					OUT	2.3136E+02	6.6019E+02	0.	8.9210E-05	0.
					IN	2.3136E+02	6.6019E+02	0.	8.9210E-05	0.
					OUT	2.0831E+02	5.9438E+02	0.	8.9210E-05	0.
					IN	2.0831E+02	5.9438E+02	0.	8.9210E-05	0.
					OUT	1.8945E+02	5.4050E+02	0.	8.9210E-05	0.
					IN	1.8945E+02	5.4050E+02	0.	8.9210E-05	0.
					OUT	1.7372E+02	4.9558E+02	0.	8.9210E-05	0.
7.0000	1.9323E-03=3.9680E-17	0.	6.9082E-17	0.	IN	2.6013E+02	7.4240E+02	0.	1.6735E-11	0.
					OUT	2.3136E+02	6.6019E+02	0.	1.6735E-11	0.
					IN	2.3136E+02	6.6019E+02	0.	1.6735E-11	0.
					OUT	2.0831E+02	5.9438E+02	0.	1.6735E-11	0.
					IN	2.0831E+02	5.9438E+02	0.	1.6735E-11	0.
					OUT	1.8945E+02	5.4050E+02	0.	1.6735E-11	0.
					IN	1.8945E+02	5.4050E+02	0.	1.6735E-11	0.
					OUT	1.7372E+02	4.9558E+02	0.	1.6735E-11	0.

IBRM=0 WHICH INDICATES END OF JOB. EXIT IS CALLED

AFFDL-TR-71-116

EXTERNAL PRESSURE LOADING
CASE III

REFERENCES

-----UNSATISFIED EXTERNALS-----

THICK

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
BRIGHT PATTERSON AIR FORCE BASE FLIGHT DYNAMICS LABORATORY VERSION: 22 JULY 1968
MODIFIED PROGRAM TO ACCOUNT FOR TRANSVERSE SHEAR DEFORMATION --WRITTEN BY R.L. CITERLEY
ANAMEY LABORATORIES INC. BERKELEY, CALIFORNIA 1 OCT 1969

STATIC ANALYSIS PARTS= 1 BRANCHES= 0 NUMBER OF SUBCASES= 0

ANGLES OF ROTATION OF BOUNDARY CONDITIONS ARE ACFL=0. ALFRS=0.

PART NO 1

SIN 0. SX= 7.00000E+00 IPAR= 2# ING= 1 SHELL TYPE 2 NTP= 0 LAYERS HLY= 4

CYLINDRICAL SHELL NO 2 R= 6.65000E+01 RA= 1.66750E+00 PHI= 90.000 DEGREES

LAYER NO 1 FROM Z= 3.32500E-01 TO Z= 1.66250E-01
CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.50000E+05 POISSONS RATIO NU= 3.50000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI= 0.0. ATHETA= 0.0. MASS DENSITY RHOD= 0.0.

LAYER NO 2 FROM Z= 1.66250E-01 TO Z= 0.
CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.50000E+05 POISSONS RATIO NU= 3.50000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI= 0.0. ATHETA= 0.0. MASS DENSITY RHOD= 0.0.

LAYER NO 3 FROM Z= 0.
CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.50000E+05 POISSONS RATIO NU= 3.50000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI= 0.0. ATHETA= 0.0. MASS DENSITY RHOD= 0.0.

LAYER NO 4 FROM Z= 1.66250E-01 TO Z= 3.32500E-01
CONSISTS OF ISOTROPIC MATERIAL, YOUNGS MODULUS E= 4.50000E+05 POISSONS RATIO NU= 3.50000E-01
COEFFICIENTS OF THERMAL EXPANSION AFI= 0.0. ATHETA= 0.0. MASS DENSITY RHOD= 0.0.

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
WRIGHT PATTERSON AIR FORCE BASE FLIGHT DYNAMICS LABORATORY VERSION: 22 JULY 1968
MODIFIED PROGRAM TO ACCOUNT FOR TRANSVERSE SHEAR DEFORMATION --WRITTEN BY R.L. GITERLEY
ANAMEY LABORATORIES INC. BERKELEY, CALIFORNIA 1 OCT. 1969

SUBCASE NO 1 FOR FOURIER HARMONIC COS=0 THETA

BOUNDARY CONDITIONS AT STARTING EDGE 2 0. 4 0. 6 0.

BOUNDARY CONDITIONS AT FINAL EDGE 2 0. 3 0. 5 0.

LOADS FOR PART NO 1 SURFACE NO 1

RING LOADS AT END OF THIS PART ARE Q=0. MPHIE=0. NPHIE=0. NTP=0. NTP=0.

SURFACE AND TEMP LOADS ARE P=3.00000E+02 PFI=0. PTMETHA=0. T1=0. TU=0.

AFFDL-TR-71-116

REFERENCE SHEET

STRESS ANALYSIS PROGRAM OF SHELLS OF REVOLUTION UNDER STATIC LOADS
WRIGHT PATTERSON AIR FORCE BASE FLIGHT DYNAMICS LABORATORY WENSON, 22 JULY 1968
MODIFIED PROGRAM TO ACCOUNT FOR TRANSVERSE SHEAR DEFORMATION --WRITTEN BY R.L. CITERLEY
ANARNEY LABORATORIES INC. BERKELEY, CALIFORNIA 1 OCT. 1969

STATIC SOLUTION AT POINTS ALONG MERIDIAN FOR WAVE NUMBER NKB = 0

MAIN SHELL PART NO 1

S	M	U-PHI	U-THETA	BETA-PHI	BETA-THETA	U-CYR	U-T	U-R	U-TAU	U-THETA	U-TAU	U-TAU
0.0000	3.1013E-03	1.0374E-04	0.	1.1249E-04	0.	1	IN	1.1201E+01	1.0493E+03	0.	0.	0.
						2	IN	1.2451E+00	9.2919E+02	0.	0.	0.
						3	IN	1.2451E+00	9.2919E+02	0.	0.	0.
						4	IN	1.2451E+00	9.2919E+02	0.	0.	0.
						1	IN	1.0193E+00	7.6070E+02	0.	0.	0.
						2	IN	1.0193E+00	7.6070E+02	0.	0.	0.
						3	IN	1.0193E+00	7.6070E+02	0.	0.	0.
						4	IN	1.0193E+00	7.6070E+02	0.	0.	0.
2500	3.0547E-03	4.9873E-10	0.	2.0600E-04	0.	1	IN	2.1502E+01	1.0372E+03	0.	0.	0.
						2	IN	4.3665E+00	9.1718E+02	0.	0.	0.
						3	IN	3.7601E+00	8.2304E+02	0.	0.	0.
						4	IN	3.7601E+00	8.2304E+02	0.	0.	0.
						1	IN	5.2843E+00	7.4775E+02	0.	0.	0.
						2	IN	5.2843E+00	7.4775E+02	0.	0.	0.
						3	IN	5.2843E+00	7.4775E+02	0.	0.	0.
						4	IN	5.2843E+00	7.4775E+02	0.	0.	0.
2500	3.0547E-03	0.0793E+00	0.	2.0600E-04	0.	1	IN	4.7089E+02	1.1700E+03	0.	0.	0.
						2	IN	3.5217E+02	1.0399E+03	0.	0.	0.
						3	IN	3.1251E+02	9.3373E+02	0.	0.	0.
						4	IN	2.7945E+02	8.4743E+02	0.	0.	0.
						1	IN	2.5135E+02	7.7529E+02	0.	0.	0.
						2	IN	4.1548E+02	1.1554E+03	0.	0.	0.
						3	IN	3.1391E+02	9.1835E+02	0.	0.	0.
						4	IN	2.7346E+02	8.3102E+02	0.	0.	0.
						1	IN	2.7346E+02	8.3102E+02	0.	0.	0.
						2	IN	2.7346E+02	8.3102E+02	0.	0.	0.
						3	IN	2.7346E+02	8.3102E+02	0.	0.	0.
						4	IN	2.7346E+02	8.3102E+02	0.	0.	0.
2500	2.9964E-03	6.0793E-06	0.	1.0389E-04	0.	1	IN	4.1548E+02	1.1554E+03	0.	0.	0.
						2	IN	3.6095E+02	1.0242E+03	0.	0.	0.
						3	IN	3.1391E+02	9.1835E+02	0.	0.	0.
						4	IN	2.7346E+02	8.3102E+02	0.	0.	0.
						1	IN	4.1548E+02	1.1554E+03	0.	0.	0.
						2	IN	3.6095E+02	1.0242E+03	0.	0.	0.
						3	IN	3.1391E+02	9.1835E+02	0.	0.	0.
						4	IN	2.7346E+02	8.3102E+02	0.	0.	0.
7500	2.9922E-03	9.1445E-06	0.	1.0748E-04	0.	1	IN	4.1966E+02	1.1420E+03	0.	0.	0.
						2	IN	3.6237E+02	1.0117E+03	0.	0.	0.
						3	IN	3.1385E+02	9.0653E+02	0.	0.	0.
						4	IN	2.7111E+02	8.1956E+02	0.	0.	0.
						1	IN	2.7111E+02	8.1956E+02	0.	0.	0.
						2	IN	2.7111E+02	8.1956E+02	0.	0.	0.
						3	IN	2.7111E+02	8.1956E+02	0.	0.	0.
						4	IN	2.7111E+02	8.1956E+02	0.	0.	0.
7500	2.9922E-03	9.1445E-06	0.	1.0748E-04	0.	1	IN	5.1968E+02	1.1420E+03	0.	0.	0.
						2	IN	3.6237E+02	1.0117E+03	0.	0.	0.
						3	IN	3.1385E+02	9.0653E+02	0.	0.	0.
						4	IN	2.7111E+02	8.1956E+02	0.	0.	0.
						1	IN	5.1968E+02	1.1420E+03	0.	0.	0.
						2	IN	3.6237E+02	1.0117E+03	0.	0.	0.
						3	IN	3.1385E+02	9.0653E+02	0.	0.	0.
						4	IN	2.7111E+02	8.1956E+02	0.	0.	0.

2.7500	2.8903E-03=4.9220E-06	0.	-5.3793E-06	0.	OUT	2.5973E+02	7.4077E+02	0.	4.8700E-01	0.	
		1	IN	3.9026E+02	1.1102E+03	0.			4.8700E-01	0.	
			OUT	3.4690E+02	9.8718E+02	0.			4.8700E-01	0.	
		2	IN	3.4690E+02	9.8718E+02	0.			4.8700E-01	0.	
			OUT	3.1212E+02	8.8649E+02	0.			4.8700E-01	0.	
		3	IN	3.1212E+02	8.8649E+02	0.			4.8700E-01	0.	
			OUT	2.8339E+02	8.0804E+02	0.			4.8700E-01	0.	
		4	IN	2.8339E+02	8.0804E+02	0.			4.8700E-01	0.	
			OUT	2.5973E+02	7.4077E+02	0.			4.8700E-01	0.	
3.0000	2.8903E-03=3.8903E-06	0.	-5.8968E-06	0.	1	IN	3.8937E+02	1.1105E+03	0.	2.5862E-01	0.
						OUT	3.4643E+02	9.8762E+02	0.	2.5862E-01	0.
		2	IN	3.4643E+02	9.8762E+02	0.			2.5862E-01	0.	
			OUT	3.1207E+02	8.8921E+02	0.			2.5862E-01	0.	
		3	IN	3.1207E+02	8.8921E+02	0.			2.5862E-01	0.	
			OUT	2.8395E+02	8.0868E+02	0.			2.5862E-01	0.	
		4	IN	2.8395E+02	8.0868E+02	0.			2.5862E-01	0.	
			OUT	2.6022E+02	7.4150E+02	0.			2.5862E-01	0.	
3.2500	2.8920E-03=3.6760E-06	0.	-5.2714E-06	0.	1	IN	3.8895E+02	1.1110E+03	0.	1.0040E-01	0.
						OUT	3.4620E+02	9.8804E+02	0.	1.0040E-01	0.
		2	IN	3.4620E+02	9.8804E+02	0.			1.0040E-01	0.	
			OUT	3.1204E+02	8.8965E+02	0.			1.0040E-01	0.	
		3	IN	3.1204E+02	8.8965E+02	0.			1.0040E-01	0.	
			OUT	2.8412E+02	8.0913E+02	0.			1.0040E-01	0.	
		4	IN	2.8412E+02	8.0913E+02	0.			1.0040E-01	0.	
			OUT	2.6090E+02	7.4201E+02	0.			1.0040E-01	0.	
3.5000	2.8932E-03=3.1497E-06	0.	-4.2339E-06	0.	1	IN	3.8883E+02	1.1113E+03	0.	8.9734E-03	0.
						OUT	3.4613E+02	9.8839E+02	0.	8.9734E-03	0.
		2	IN	3.4613E+02	9.8839E+02	0.			8.9734E-03	0.	
			OUT	3.1203E+02	8.8999E+02	0.			8.9734E-03	0.	
		3	IN	3.1203E+02	8.8999E+02	0.			8.9734E-03	0.	
			OUT	2.8417E+02	8.0966E+02	0.			8.9734E-03	0.	
		4	IN	2.8417E+02	8.0966E+02	0.			8.9734E-03	0.	
			OUT	2.6101E+02	7.4233E+02	0.			8.9734E-03	0.	
3.7500	2.8941E-03=2.8907E-06	0.	-3.1168E-06	0.	1	IN	3.8888E+02	1.1117E+03	0.	-4.2081E-02	0.
						OUT	3.4616E+02	9.8867E+02	0.	-4.2081E-02	0.
		2	IN	3.4616E+02	9.8867E+02	0.			-4.2081E-02	0.	
			OUT	3.1203E+02	8.9023E+02	0.			-4.2081E-02	0.	
		3	IN	3.1203E+02	8.9023E+02	0.			-4.2081E-02	0.	

AFFDL-TR-71-116

6.7500	2.8949E-03-2.3293E-07	0.	4.0567E-08	0.	IN	2.6025E+02	7.4244E+02	0.	-1.4321E-04	0.
					OUT	3.4660E+02	1.1122E+03	0.	1.3365E-04	0.
2					IN	3.4660E+02	9.8905E+02	0.	1.3365E-04	0.
					OUT	3.1208E+02	8.9046E+02	0.	1.3365E-04	0.
3					IN	3.1208E+02	8.9046E+02	0.	1.3365E-04	0.
					OUT	2.6381E+02	8.0974E+02	0.	1.3365E-04	0.
4					IN	2.6381E+02	8.0974E+02	0.	1.3365E-04	0.
					OUT	2.6025E+02	7.4244E+02	0.	1.3365E-04	0.
6.7500	2.8949E-03-2.3293E-07	0.	4.0567E-08	0.	IN	3.8971E+02	1.1122E+03	0.	1.3365E-04	0.
					OUT	3.4660E+02	9.8905E+02	0.	1.3365E-04	0.
2					IN	3.4660E+02	9.8905E+02	0.	1.3365E-04	0.
					OUT	3.1208E+02	8.9046E+02	0.	1.3365E-04	0.
3					IN	3.1208E+02	8.9046E+02	0.	1.3365E-04	0.
					OUT	2.6381E+02	8.0974E+02	0.	1.3365E-04	0.
4					IN	2.6381E+02	8.0974E+02	0.	1.3365E-04	0.
					OUT	2.6025E+02	7.4244E+02	0.	1.3365E-04	0.
7.0000	2.8949E-03-1.2353E-16	0.	1.3223E-16	0.	IN	3.8971E+02	1.1122E+03	0.	3.1480E-11	0.
					OUT	3.4660E+02	9.8905E+02	0.	3.1480E-11	0.
2					IN	3.4660E+02	9.8905E+02	0.	3.1480E-11	0.
					OUT	3.1208E+02	8.9046E+02	0.	3.1480E-11	0.
3					IN	3.1208E+02	8.9046E+02	0.	3.1480E-11	0.
					OUT	2.6381E+02	8.0974E+02	0.	3.1480E-11	0.
4					IN	2.6381E+02	8.0974E+02	0.	3.1480E-11	0.
					OUT	2.6025E+02	7.4244E+02	0.	3.1480E-11	0.

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