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# A STUDY OF THE METALLURGICAL PROPERTIES THAT ARE NECESSARY FOR SATISFACTORY BEARING PERFORMANCE AND THE DEVELOPMENT OF IMPROVED BEARING ALLOYS FOR SERVICE UP TO 1000 F.

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#### FOREWORD

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The use of bearings made from hot work steels and other tool steels in experimental engines has resulted in a few premature engine failures. Unfortunately, very little has been known about the elevated temperature properties such as hot hardness, compressive yield strength, resistance to softening and structural and dimensional stability of these hot work and other tool steels. This report describes the work done to obtain these material properties for 29 steels ranging in type from SAE 52100, its modifications, to hot work and other tool steels. An analysis of the data obtained shows that Halmo, VSM, M50, M10, T1, M2, M1 and two experimental compositions one, Steel B, containing 0.70 carbon, 4.20 chromium, 0.60 vanadium, and 5.30 molybdenum, and the other, Steel G, containing 1.31 carbon, 4.07 chromium, 4.13 vanadium, 5.75 tungsten, and 4.87 molybdenum, are suitable for elevated temperature aircraft bearing application. From a point of view of temperature range of application these steels have been classified as follows:

ABSTRACT

Room Temperature up to 700 F

Room Temperature up to 800 F

Room Temperature up to 900 F

T1, M2, M1 and Steel G

None of the steels investigated appeared suitable for application at 1000 F.

#### PUBLICATION REVIEW

This report has been reviewed and is approved.

FOR THE COMMANDER:

R. R. Kennedy Chief, Metals Branch Materials Laboratory

Approved for Public Release

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#### I. INTRODUCTION

The increase in flight speed and engine power output in jet engines and gas turbine power plants of recent years has forced a steady rise in the engine operating temperatures. These changes, particularly the higher engine operating temperatures, have necessitated the use of materials with superior elevated temperature physical and mechanical properties. For instance, in most present day engines, the maximum bearing temperatures are approaching 500 F. The most commonly used bearing steel, SAE 52100, can no longer be used since it rapidly loses hardness and dimensional stability above 450 F.

Engine operating temperatures are expected to reach 700 F in engines of the near future. Therefore, the need for bearing materials with adequate hot hardness (56 to 58 R<sub>C</sub> at the operating temperature) and other required mechanical and physical properties is very urgent. Numerous attempts are being made by the aircraft industry to use hot work die steels and other tool steels for bearing application with varying degree of success. The main drawback of these presently available steels seems to be their unpredictable fatigue life and somewhat lower mean life in bearing tests when compared with that for SAE 52100. Only scant data are available on the behavior of these steels at elevated temperatures. Furthermore, it has been difficult to analyse properly the causes of premature failure of bearings made from hot work and other tool steels.

This program was, therefore, initiated to obtain data on hot hardness, dimensional stability, compressive strength and resistance to softening for various hot work and other tool steels which have been proposed for bearing application in the temperature range 400 to 1000 F. Since the bearing is a vital part in the aircraft engine, it is also deemed necessary to devise means of accurately predicting fatigue life of bearing steels. A full understanding of the material characteristics and their behavior in simulated and actual bearing performance tests could then lead to the selection of suitable elevated temperature bearing steels and also to the development of new and improved bearing steels.

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#### II. OBJECTIVES OF THIS STUDY

This program comprises three phases. In Phase I, efforts were directed toward, (a) making a literature survey of all available information on high speed rolling contact bearings; (b) collecting in a single report the views expressed by bearing manufacturers and aircraft engine builders on property requirements of high temperature bearing materials; (c) tabulating the steels currently selected for bearing applications above the temperature range covered by SAE 52100 steel; and (d) accumulating all existing data on the results of bearing tests using hot work and other tool steels.

The work on Phase II consisted of a study of the metallurgical properties to include resistance to softening, hot hardness, dimensional stability, and elevated temperature compression behavior of some promising steels for elevated temperature bearing application.

Phase III includes an analysis of the data obtained in Phase II and recommendations of steel compositions which are considered applicable for aircraft engine bearings.

# III. REVIEW OF THE LITERATURE ON AIRCRAFT BEARING STEELS AND INFORMATION OBTAINED FROM BEARING AND AIRCRAFT ENGINE MANUFACTURERS (Phase I)

The bearings used in the turbine type aircraft power plants are of the rolling contact type. Ball thrust bearings are used on the propeller shaft and these are usually heavily loaded. The most pressing problems are connected with the turbine main bearings and compressor thrust bearings. The compressor rear thrust bearing is exposed to hot air leakage from the last stage of the compressor. It is also subjected to a combination of heavy radial and thrust loads.

The four major items of concern to bearing manufacturers are: (1) the engine speed which determines the size of the bearings; (2) the operating and soak back temperatures; (3) the radial loads; and (4) the thrust loads.

The present trend in bearing design is towards higher stresses and much higher operating temperatures up to 1500 F. Operating temperatures of 700 F, thrust loads up to 50,000 pounds and DN values\* of  $3.5 \times 10^6$  are anticipated in aircraft engines of the immediate future

<sup>\*</sup>DN value is bearing bore in mm. times shaft speed in r.p.m.

# Bearing Steel Requirements

The basic requirement of a bearing is complete reliability for a life of 1000 hours. Also, because of severe cyclic stresses exerted on the rolling elements of aircraft bearings, the material used in bearing balls and races must meet very rigid quality standards. Furthermore, the material must possess superior mechanical and physical properties. In metallurgical quality control tests, bearing steels must meet minimum standards. Tests which are applied include: deep etching, fracture rating, taper or hairline tests, and visual, magnetic and ultrasonic inspection methods for hidden and random inclusions. The bearing manufacturers consider it mandatory that consecutive shipments of steel should respond to heat treatment in a uniform manner.

The quality standards have been further defined by ASTM specifications A 295-46T for carbon-chromium ball- and roller-bearing steels. Since these specifications cover only three types of steel, SAE 52100, 51100 and 50100, these may be inapplicable for hot work die steels and other tool steels. The cleanliness specifications proposed by bearing manufacturers require an inclusion rating not to exceed 1.5 based on the thin series ASTM specifications A 295-46T.

In vacuum melted bearing steels, the aim should be to obtain an inclusion rating below 0.5 in the thin series. The austenitic grain size requirement is between 7 and 8 and decarburization in annealed bars must not exceed .030 inch for bar sizes 1 to 3 inches.

Physical and Mechanical Properties: The following material properties are desired by bearing manufacturers for fully heat treated steel of ideal high temperature bearing composition:

- 1. A hardness of Rockwell C 58 at the operating temperature for at least 1000 hours.
- 2. Dimensional stability when exposed to the temperatures developed immediately after a shutdown. (100-200 F higher than the engine operating temperature)
- 3. A high elastic limit.
- 4. Good thermal stability and resistance to oxidation and corrosion in environments encountered in engine application.
- 5. Good wear resistance and the ability to maintain a very highly polished surface.
- 6. Fairly constant coefficient of thermal expansion.

- 7. A low friction coefficient.
- 8. A constant elastic modulus in the range minus 65 F up to the maximum service temperature.
- A uniform distribution of carbides and freedom from stringer type and large randomly distributed inclusions.
- 10. A high resistance to rolling fatigue.
- 11. Freedom from seizing and galling.

At the time this survey was made, the steels listed below along with their nominal composition were being considered for elevated temperature bearing application. Some of these steels were even being used in engine tests.

List of Current Bearing Steels and Their Nominal Composition

Designation	<u>C</u>	Mn	Si	Cr	<u>v</u>	Mo	<u>W</u> _	<u>Ni</u>	<u>Al</u>
Halmo	.60	.30	1.2	4.70	.60	5.25	-	_	-
Rex LA	.85	.30	.35	3.00	1.00	3.25	1.50	-	_
<b>M</b> - 1	.80	.30	.25	4.00	1.00	8.00	1.50	-	-
M-2	.85	.25	.30	4.00	2.00	5.00	6.00	_	-
<b>M</b> - 10	.85	.25	.30	4.00	2,00	8.00		_	_
M-50	.80	.30	.30	4.00	1.00	4.00	_	-	-
T-1	.70	.30	.30	4.00	1.00	_	18.00		-
VSM	.70	.50	1.00	3.00	-	5.25	-	_	· -
8Cr-8W	.55	.60	1.00	8.00	.15	-	8.00	-	_
MHT	1.00	.40	.50	1.50	_	-	-	_	1.25
52100	1.00	.35	.25	1.50	-	-	-	-	_
Bower 315	.15	.50	.25	1.50	-	5.00	-	2.80	_
(carburizing grade)									

Since inclusions were considered undesirable in bearing steels, steelmakers were requested to furnish the above compositions induction melted and cast in vacuum, as well as melted by consumable electrode melting practice.

The scant information made available by bearing manufacturers on these steels is summarized below:

1. Ease of grinding was considered important and a partial listing of the relative grindability of some steels using 52100 as standard material is as

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follows:

Grade	Grindability Index
52100	1
Halmo	1.3/1.5 (depending
	upon carbon content)
<b>T-</b> 1	1.5/1.8
M - 1	1.8/2
M-2	3
Rex LA	3
M-10	4*

2. Dimensional stability and a constant coefficient of thermal expansion are other important considerations. One bearing manufacturer reported that in a ring test used for checking dimensional stability, M-10 and T-1 steels did not give satisfactory results above 600 F. A majority of bearing manufacturers reported that the high speed steels in the above list showed no significant dimensional change up to 600 F.

#### Notes on Bearing Failures

Because of the complexity of the interacting factors it is difficult to determine the cause of bearing failures. Reports show that it is almost impossible to detect the initial cause of failure. After complete failure has occurred, it may be attributed to one of several causes listed below:

- 1) Surface fatigue of rolling elements
- 2) Inadequate lubrication
- 3) Breakdown of lubricant and increase in acidity accompanied by corrosion of the steel surface
- 4) Uneven wear of rolling elements and raceways
- 5) Excessive loading
- 6) Incompatibility of cage and rolling element material

Rolling fatigue in bearings snould be distinguished from fatigue in materials due to simple rotation. Exact information on factors influencing rolling fatigue is

<sup>\*</sup> A higher index number denotes increased difficulty in grinding the steel.

still lacking. Hardness and elastic limit have been correlated with rotating beam fatigue life of steels. A material of high hardness normally exhibits a high endurance limit, but it is not known if rolling fatigue has such a relationship to any of the material properties. Therefore, it is difficult to predict with any degree of certainty the normal life of bearings made from hot work and high speed steels.

A great many failures could be attributed to cage or separator material failures or a lack of lubrication at the contact points of rolling elements. Another major cause of failure is wear of rolling elements and raceways when thrust loads, speeds and bore size are increased. In certain instances premature bearing failures have been associated with the presence of large non-metallic inclusions.

#### Appraisal of Survey Information

This survey indicated that Halmo, MHT, M-1, VSM and M-50 were actively considered for elevated temperature bearing application. Bearing manufacturers reported a somewhat sporadic behavior of these materials in actual bearing tests. Early bearing failures were often associated with the occurrence of large randomly dispersed non-metallic inclusions in the steel. Although the role of inclusions in causing bearing failure was not fully understood there was a widespread desire in industry to use cleaner materials produced either by vacuum induction melting or by vacuum arc remelting. It was also apparent that use of vacuum melted steel increased bearing life by a significant factor. On the other hand, the bearing manufacturers and engine builders pointed out that it is of paramount importance to provide more data on the metallurgical, mechanical and physical properties of steels considered for bearing application in order to facilitate selection of the most suitable compositions.

Hence, this program was initiated to study the properties of bearing steels, namely, their resistance to softening, hot hardness, elastic properties in compression, dimensional stability, in the range room temperature to 1000 F. Furthermore, the size and distribution of carbides in these steels was to be observed in samples given the optimum heat treatment.

It is a further objective of this project to conduct simulated bearing tests on the most promising alloys using testing facilities at SKF Industries, Inc., for final screening of the bearing steels.

#### IV. EXPERIMENTAL WORK (Phase II)

#### Material

The experimental steels included in this evaluation program are listed in

Table I along with their chemical compositions. The steels other than those melted in vacuum were made in 30 lb. induction heats. The ingots were forged first to 1-5/8 in. square billets, ground free of any surface defects, and then reforged to either 5/8 in. square or 3/4 in. square bars. Prior to shipment, the bars were stress relieved at 1250 to 1350 F for one hour. The forging temperatures for the steels have been recorded in Table II. The hot workability was judged good for all steels except those marked with an asterisk; the forgeability of the latter was considered only fair.

The vacuum melted steels were obtained from the warehouse stock of Vacuum Metals Corporation in the form of 1 in. round bars. The gas content of the vacuum melted steel was in every case within the limits specified in Table I.

#### Preliminary Heat Treatment

Annealing: All forged bar stock was annealed in the manner described in Table II. Annealing was done by a continuous cooling procedure and the steels were cooled at a rate of 10 to 25 F per hour. Hardness measurements and microstructural studies were made on all steels to check the effectiveness of the annealing treatment.

Austenitizing for Hardening: The optimum austenitizing procedure for most steels included in the current program had been previously established. However, an austenitizing survey was carried out on those compositions on which information was lacking in regard to a proper austenitizing temperature to obtain an austenitic grain size between ASTM 6 and 8 and an optimum carbide size and distribution. The determination of a suitable austenitizing temperature involved quenching samples from various austenitizing temperatures. Subsequently, a check was made of the response of the quenched specimens in developing a maximum secondary hardness. Tempering temperatures were selected as high as was consistent with hardness requirements. The austenitizing procedures for all steels have been listed in Table III.

#### Resistance to Softening Studies

For this study, 3/4 in. thick specimens were cut from the annealed bar stock, austenitized as specified in Table III, oil quenched and tempered at various temperatures in the range 400 to 1100 F for progressively increasing times. In the case of 52100, MHT and the silicon bearing MHT, tempering above 800 F was not done since their useful range of application is well below this temperature. Tempering up to 800 F for these steels was done merely to develop a useful master tempering curve.

The effect of varying austenitizing temperature upon the peak secondary hardness and the temperature range of occurrence of secondary hardness was studied in .9C Halmo, UC, 440 C and 440 BM. In Tables IV and V it will be seen that the austenitizing temperatures used for these steels are different from those used in the tempering survey. This change in the austenitizining temperature was necessary in order

to extend the useful range of application for these steels by pushing the secondary hardness peak to higher tempering temperatures. This is illustrated in the case of .9C Halmo and UC by two tempering curves (Figs. 10 and 27) shown for two different austenitizing temperatures. The results of a short time tempering survey using various austenitizing procedures for 440 C and 440 BM have not been shown.

#### Hot Hardness Studies

Specimens, approximately 3/4 in. in length, were cut from the annealed bar stock of all steels. These were quenched and tempered as described in Table IV. After the heat treatment, two parallel surfaces were ground along the length of the specimen.

The hardness tester shown schematically in Figure 1 is a modified Rockwell Hardness Machine. A front view of the machine with accessories is shown in Figure 2. At the start of the hot hardness test, the specimen is placed on the anvil; this anvil has a rectangular slot which acts as a guide to facilitate lateral movement of the specimen. Thus it is possible to obtain several readings without removing the sample from the furnace enclosure. The furnace, which surrounds the anvil and the indentor is screwed tightly at the bottom to the anvil. The top opening of the furnace is sealed with liquid Wood's metal. The indentor is a conventional diamond "Brale" fastened with high temperature cement to an 8 in. long shaft attached to the loading mechanism.

A 150 kilogram load is then applied in measuring hot hardness of bearing steels and the hardness data are read on the "C" scale. The specimen temperature is measured and controlled by a thermocouple which is welded to the specimen and connected to an electronic temperature controller. Scaling of the specimens in the furnace is prevented by protecting them with an argon atmosphere throughout the test. The hardness measuring techniques need no further explanation since they are no different from the standard room temperature hardness measuring procedures.

#### Dimensional Stability Studies

The specimens used in this study measured 3/8 in. in diameter by 4.000 plus or minus 0.001 in. long, with ends ground to the contour of a 4 in. diameter sphere. The spherical ends were to prevent errors that might result from a slight tilting of a squared-end cylindrical specimen during the length measurement. A drawing of this specimen is shown in Figure 3. Duplicate samples from each bar were machined slightly oversize to the dimensions given above; these specimens were austenitized, quenched and tempered as specified in Table IV. The samples were then finish ground to the specified dimensional tolerance. Precision length determinations were made on a Johansson Comparator by fastening specimens to a jig to keep them vertical. In this apparatus, the standard gage block is 4.2 inches and the difference in length between the sample and the standard block is made up by inserting small gage blocks.

The precision of the measurement is of the order of 10 x 10<sup>-6</sup> inch per inch. The specimens used in this study were held for 1000 hours at 400 F and 600 F by submerging them in a neutral salt bath. The 1000 hours exposure at 800 F and 1000F originally was also carried out in a neutral salt bath. However, it was found that at the higher temperatures the salt was not sufficiently neutral toward the steel so that scaling occurred on the specimens. Consequently, in later tests, specimens were sealed in evacuated Vycor bulbs and then held for 1000 hours in neutral salt baths.

#### Elevated Temperature Compression Tests

A schematic diagram of the subpress used for elevated temperature compression tests is shown in Figure 4. A photograph of this subpress mounted in position in a Riehle Testing machine is shown in Figure 5. Originally, it had been planned to use high temperature SR-4 strain gages in the compression tests; however, elevated temperature strain gage application technology has not been perfected to the point where the readings are reliable. Since this development work involved considerable time, a compressometer which is standard equipment for measuring compressive strains has been used. The use of a compressometer requires more care, as well as time, in aligning the specimen. Except for this disadvantage it is capable of providing more accurate readings than the high temperature SR-4 strain gages in their present form of development.

#### Metallographic Studies

It is evident that fatigue life of the steel may be affected by the size and distribution of carbides in the microstructure. Fatigue cracks have been associated with flaws in the material. The steels included in this program, therefore, were examined for uniformity of microstructure with respect to size and distribution of carbides produced by different austenitizing temperatures. Photomicrographs are used to illustrate the microstructure of the different steels in a state of optimum heat treatment.

#### V. RESULTS

The hardnesses of the annealed bars are shown in Table II. Microstructures of all annealed steels were checked to ascertain that all steels were in a completely spheroidized condition prior to hardening.

The results of the studies on the resistance to softening of the various bearing steels have been summarized in Table III. These results have also been plotted as master tempering curves for each of the steels in Figures 6 to 34 inclusive. In these curves, the tempering temperatures and time at these temperatures have been combined in a single parameter by using the expression  $T(20 + \log t)$  in which T represents the absolute temperature, degrees Rankine, and t equals the tempering time in hours.

Hot work steels are double tempered in commercial practice (two, 2 hour tempers) to transform retained austenite that is generally present after hardening. At any selected temperature, the two consecutive two hour tempers will result in equivalent hardness to a single four hour tempering operation at the same temperature. Hence, in the master tempering curves for secondary hardening steels, a scale has been added to read directly the hardness for a four hour temper at various temperatures in the range 400 to 1100 F.

The results of hot hardness surveys of quenched and tempered bearing steels are shown in Table IV. This table includes hardness values of various bearing steels at room temperature and at elevated temperatures after a 1000 hour exposure at 400, 600, 800, and 1000 F respectively. These results have also been summarized graphically in Figures 35 to 63 inclusive.

In Table V the results of dimensional stability tests have been tabulated for all bearing steels exposed for 1000 hours at 400, 600, 800 and 1000 F respectively. The parameter used is length change in micro-inches per inch.

Data from compression tests are tabulated in Table VI. This table gives the details of heat treatment and yield strength in compression at 0.1 percent and 0.2 percent offset. Yield strengths in compression at these offset points were determined from the stress-strain curve for each steel tested (except 52100 and modified 52100 steels) at 400, 600, 800, and 1000 F. The hot hardness values at these temperatures have also been recorded.

Typical microstructures illustrating the size and distribution of carbides in steels given the optimum heat treatment, are shown in Figures 64 to 92 inclusive.

#### VI. ANALYSIS OF RESULTS (Phase III)

The effect of elevated temperatures on the hardness of quenched and tempered steels can be predicted from master tempering curves for these steels. Such curves are also useful in planning commercial tempering treatments.

It may be seen from the master tempering curves presented in Figures 11 and 12 that SAE 52100 rapidly loses its hardness above 400 F and MHT begins to lose hardness slowly at 600 F and rapidly above this temperature. Therefore, it is imperative that steels capable of secondary hardening be considered for bearing application above 600 F.

Steels containing large amounts of alloying elements e.g. chromium, molybdenum, tungsten, vanadium, in the quenched condition will consist of highly alloyed tetragonal martensite, highly alloyed retained austenite, and undissolved complex carbides. Master tempering curves, Figures 6 to 9 inclusive, illustrate the change of hardness on

tempering highly alloyed steels. As the tempering temperature is raised, an initial softening occurs due to the decomposition of tetragonal martensite to cubic martensite and a precipitation of cementite in a highly alloyed ferrite matrix. This phenomenon occurs up to tempering temperatures of 750 F with accompanying softening normally amounting to 2 to 4 Rockwell "C" points. The iron carbide, or cementite, which precipitates at these low temperatures, probably disappears either by re-solution in the matrix or by reaction with the alloy content of the matrix to form a complex carbide, if long tempering times and high tempering temperatures are employed. At temperatures above 750 F secondary hardening is encountered. That is, increased tempering increases the hardness. This secondary hardening is the result of a precipitation hardening reaction involving alloy carbides. In the final stage of the tempering process, the alloy carbides coagulate into relatively large particles. This stage, which entails rapid softening of the steel, occurs on tempering for long times at temperatures of about 1000 F or on tempering for relatively short times above 1100 F.

The master tempering curves shown in Figures 30 to 33 inclusive vary in accordance with the alloying elements in the steel, in particular with respect to the shape and size of the secondary hardness peak. By varying the austenitizing temperature this point of maximum secondary hardness can be made to occur at any temperature between 900 and 1150 F. The master tempering curves in this study have also been used to establish the optimum austenitizing temperatures for the various steels. The objective here is to use an austenitizing temperature that would push the secondary hardness peak to higher tempering temperatures. This would tend to minimize an accidental softening of the steel due to minor overshooting of the soak back temperature or long exposure of the bearing at somewhat higher than normal operating temperatures.

These master tempering curves can be used also to select steels that show a flat secondary hardness peak. A steel showing a narrow secondary hardness peak will not be suitable for bearing application. In this case small deviations in austenitizing and tempering temperatures which must be expected in commercial heat treating practice may cause permanent loss of hardness in bearing balls or races. For example, in steel E, Figure 32, without precise control, it would be difficult to obtain the same peak hardness consistently in quenched and tempered steels. In order to insure against these occurrences an ideal high temperature bearing steel must have a flat secondary hardness peak as exemplified by curves for Halmo, M2, M10, M50, T1. (Figures 6, 18, 20, 23, 26) All hot work die steels and high speed tool steels investigated possess adequate resistance to softening.

Bearing steels must also be able to withstand wear at elevated temperatures and for this reason they should have adequate hot hardness. It is also probable that elevated temperature strength may be dependent on hot hardness. Based on bearing fatigue and performance tests, bearing manufacturers have fixed the hot hardness at 56-58 Rockwell "C" at the operating temperature. Using this hot hardness criterion and also the room temperature hardness after 1000 hours exposure at the various

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temperatures, the steels included in this study may be grouped as follows:

Bearing Operation Temperature Range	Group of Steels
Room Temp. up to 400 F	52100
Room Temp. up to 500 F	MHT, MHT + Si, 440 C, 440 BM, UC
Room Temp. up to 700 F	Halmo-1, Experimental Compositions A, C, and E
Room Temp. up to 800 F	Halmo-2, .8C Halmo, .9C Halmo, VSM, M50, M10, Experimental Compositions B, D, and F.
Room Temp. up to 900 F	T1, T5, M2, M1, HiC-M10 and Experimental Composition G

In this arrangement, the specified bearing operation temperatures allow a margin of 100 to 150 F increase in temperature due to soak back heating.

Next the hot hardness drop in various steels was studied as it is affected by temperature increases. The average hardnesses for all hot work die steels and other tool steels were calculated at the various test temperatures from the results obtained in the hot hardness studies. Subsequently, the standard deviation of hardness in these steels was computed. The curve shown in Figure 93 portrays the mean hardness values and the standard deviation from the mean hardness value computed for 22 steels at temperatures varying from room temperature to 1000 F.

It may also be noted that with rising temperature a nonlinear hardness drop was observed in these steels. The average hardness drop in the 22 steels investigated is 4.3 Rockwell "C" points between room temperature and 400 F, 1.5 Rockwell "C" points between 400 and 600 F, 2.3 Rockwell "C" points between 600 and 800 F and 4 Rockwell "C" points between 800 and 1000 F. From these data one can estimate the hot hardness based on the specific room temperature hardness of the quenched and tempered steels.

An inspection of the dimensional stability evaluation for the steels (Table V) indicates that in Halmo, T1, M2, M1, M10, VSM, M50, and Experimental steels B and G, the given heat treatment has established the required dimensional stability. However, much work remains to be done to design optimum heat treatment procedures for the remaining useful compositions in order to obtain good dimensional stability. This is particularly true for the experimental compositions.

Approximate

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"Dimensional stability," as used in this report, refers to the expansion or contraction of steel parts subsequent to hardening and tempering. The differences in dimensional stability between hardened steels can be accounted for on the basis of martensite tempering which results in contraction, and transformation of retained austenite, which results in an expansion. In an unstable steel both these reactions can occur simultaneously. The attainment of dimensional stability in tool and die steels depends on minimizing both the contraction due to martensite decomposition and the expansion due to retained austenite transformation. In high alloy steels retained austenite is minimized by using lower austenitizing temperatures consistent with minimum residual carbides and maximum secondary hardening at the highest tempering temperatures.

An inspection of compression test results in Table VI shows that all the hot work die steels and high speed tool steels possess considerably higher compressive yield strength than that of SAE 52100. It is known that greater flight speed results in exponential increases of bearing loads. Hot work die steels and high speed tool steels, within the suggested temperature range of application for aircraft bearings have compressive yield strengths above 200,000 psi. It is the opinion of bearing manufacturers that these strength levels are sufficient in a high temperature bearing steel.

The microstructures for most steels tested, Figures 64 to 92 inclusive, seem to be characteristic for bearing races or balls, inasmuch as the carbides are evenly distributed and the martensite is uniformly tempered. In Figures 67, 74, 75, and 90 however there is evidence of retained austenite which could not be eliminated without sacrificing some other required properties. From the point of view of microstructure, therefore, .8C Halmo, 440 C, 440 BM, and steel E have been considered unsuitable for bearing application.

Attention of the reader is drawn to commercial grades Halmo, VSM, M50, M10, M2 and M1 and experimental steels B and G. These compositions not only possess optimum metallurgical properties required in high speed aircraft bearings, they are also comparatively lean in alloying elements. In the group of materials suggested for bearings to operate in the range room temperature up to 900 F, (Page 20) the logical choice of steels for further evaluation in bearing tests would be M2, M1 and experimental steel G.

#### VII. CONCLUSIONS

In conclusion it may be stated that for elevated temperature bearings, Halmo-1, T1, M2, M1, M10, and M50 appear to be the most promising materials. Among the experimental grades, the steels with the following compositions seem to have also fulfilled the preliminary requirements of a bearing material:

	<u>C</u>	Mn	Cr	<u>v</u>	<u>w</u>	Mo
Steel B	.0.7	0.29	4.21	0.59	-	5.31
Steel G	1.31	0.29	4.07	4.13	5.75	4.87

These steels could be further classified from the view point of their temperature range of application in the following manner:

Room Temperature up to 700 F Halmo-l

Room Temperature up to 800 F VSM, M50, M10 and Steel B

Room Temperature up to 900 F T1 M2, M1 and Steel G

#### VIII. SUGGESTIONS FOR FUTURE WORK

Data of this investigation show some of the major shortcomings of currently available hot work and other tool steels for elevated temperature aircraft bearing application. Furthermore, none of the steels investigated appear suitable for bearing application above 900 F. There is an urgent need for steels specifically suited for bearings to operate in the range room temperature up to 900 F, (Page 12) the logical choice of steels for further evaluation in bearing tests would be M2, M1 and experimental steel G.

The development of new steel compositions specifically suited for aircraft bearings for service in the range room temperature to 1000 F. To achieve this objective a study of basic metallurgical properties must be continued on a range of experimental alloys. In addition to the properties investigated during the last year, particular emphasis must be placed on corrosion and oxidation resistance and fatigue properties. A better understanding of the effect of microstructure, especially carbide content, size, and shape, on fatigue life must be attained.

#### IX. SELECTED REFERENCES

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- 3. Styri, H., "Fatigue Strength of Ball-Bearing Races" Proceedings of the American Society for Testing Materials, Vol. 51, 1951, p. 682.
- 4. Allen, C. M. and Goldthwaite, W. H. "Research in Bearings" Battelle Technical Review, Nov. 1954.
- 5. Dayton, R. W., Allen, C. M., et al., "A Survey of Rolling Contact Bearings for Aircraft Turbine Power Plants" Battelle Memorial Institute Report, July 1952.
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TABLE I

List of Bearing Steels and their Chemical Analyses

(in percent)

Grade	<u>C</u>	Mn	<u>P</u>	<u>s</u>	<u>Si</u>	Ni	Cr	<u>v</u>	<u>W</u>	Mo	Others
Halmo-l Ferrovac	.58 .56	.28 .28	.00 <b>9</b> .005	.032	1.18 1.18	.08 .04	4.72 4.82	•51 •50	<b>-</b>	5.15 5.12	<b>-</b>
Halmo Halmo-2 .8C Halmo .9C Halmo 52100 Ferrovac	.76	.30 .31 .37 .29	.007 .008 .007 .004	.032 .037 .025 .029	1.06 1.07 1.08 .23	.08 .04 .06	4.60 4.57 4.42 1.52 1.50	•54 •59 •52 -	- - -	5.09 5.06 5.02 .07	- - -
52100 MHT Ferrovac MHT	1.03 1.03	_	•00 <sup>1</sup> 4	.030	.24 .46	.09 .02	1.53 1.49	<b>-</b>	-	•03 <del>-</del>	1.30 Al 1.30 Al
MHT + Si 440 C 440 BM Tl L T5 M2 Ferrovac	.83 .79	_	.004 .004 .001 .017 .011	.030 .036 .019 .032 .033	1.23 .27 .45 .18 .29 .28	.09 .04 .10 .02 .18 .08	1.58 16.17 17.15 3.91 4.20 3.83 4.16	- .14 1.37 2.13 1.79 1.96	- 18.17 18.30 6.40 6.47	.02 .76 .18 .13 4.91 5.01	1.29 Al - - 7.92 Co -
M2 M10 M10-M10 VSM M50 UC	.85 1.06 .66 .79	.46	.008 .011 .008 .011 .008	.038 .041 .033 .036	.33 .28 .30 1.30 .36	.09 .08 .09 .08	3.66 4.10 3.99 2.83 4.01 5.66	1.19 1.68 1.78 .03 1.05	1.32 0.01 - .01 .01	8.51 7.77 7.50 5.02 4.16 1.61	- - - - -
A B C D E F G	•71 •92 •74		.007 .007 .008 .010 .010	.039 .035 .040 .036 .023 .033	.32 .30 .31 1.23 .26	.10 .12 .10 .10 .14	11.22 4.21 8.18 3.42 7.96 3.29 4.07	.59 2.12 2.21 .26 1.12 4.13	- - 7.98 1.56 5.75	4.20 5.31 4.90 4.85 .07 3.42 4.87	

Gas content of vacuum induction melted and cast steels (Ferrovac) were within the limits specified below.

Nitrogen - .0004 Oxygen - .0005

Hydrogen - less than 1 part per million

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TABLE II

Details of Bearing Steel Forging and Annealing Procedures

	Forging	Procedure	Annealing Trea	tment	
Grade	Forging Temp. F	Soaking Time, hrs.	Annealing Temp. & Time	Control Cool	Hardness R <sub>C</sub>
Halmo-l	<b>19</b> 50	2	1550 F-2 hrs	25 F/hr	15
Ferrovac Halmo	1 <b>9</b> 50	2	Mill anne	aled	16
Halmo-2	1 <b>9</b> 50	2	1550 F-2 hrs	25 F/hr	17
.8C Halmo	1975	2-1/2	1550 F-2 hrs	25 F/hr	14
.9C Halmo	1 <b>9</b> 75	2-1/2	1550 F-2 hrs	25 F/hr	18
52100	1975	1	1440 F-4 hrs	10 F/hr	8
Ferrovac 52100	1970	1	Mill annea	aled	6
MHT	1975	1-1/2	1440 F-4 hrs	10 F/hr	16
Ferrovac MHT	1970	1-1/2	1440 F-4 hrs	10 <b>F/</b> hr	11
MHT + Si	1 <b>9</b> 75	1-1/2	1500 F-4 hrs	10 F/hr	19
440 C	1 <b>9</b> 75	2	1650 F-6 hrs	25 F/hr	10
440 BM	1975	2	1650 F-6 hrs	25 F/hr	18
Tl	2050	2-1/2	1650 F-2 hrs	25 F/hr	18
Т5	2025	2-1/2	1650 F-2 hrs	15 F/hr	27
<b>M</b> 2	1950	2	1600 F-2 hrs	25 F/hr	16
Ferrovac M2	1950	2	Mill annes	aled	16
Ml	1950	. 2	1550 F-2 hrs	25 F/hr	16
MIO	1950	2	1550 F-2 hrs	25 F/hr	15
Hic-Mlo	1950	2	1550 F-2 hrs	25 <b>F/</b> hr	17
VSM	1 <b>9</b> 75	2	1600 F-2 hrs	25 F/hr	18

TABLE II (Continued)

Details of Bearing Steel Forging and Annealing Procedures

	Forging	Procedure	Annealing Tre	atment	
Grade	Forging Temp. F	Soaking Time, hrs.	Annealing Temp. & Time	Cool	Hardness R <sub>C</sub>
M 50*	1950	2	1550 F-2 hrs	25 F/hr	15
UC	1925	2	1550 F-2 hrs	25 F/hr	16
A*	1960	2-1/2	1650 F-6 hrs	25 F/hr	21
В	1925	. 2	1650 F-1 hr	15 F/hr	14
C*	1960	2-1/2	1650 F-6 hrs	25 F/hr	21
D	1 <b>9</b> 50	2	1600 F-1 hr	15 F/hr	12
E	1925	2	1650 F-2 hrs	15 F/hr	20
F	1925	2	1600 F-1 hr	15 F/hr	14
G*	1960	2-1/2	1600 F-1 hr	15 F/hr	20

<sup>\*</sup> Forgeability considered fair as compared to other grades.

Contrails

TABLE III

Results of Tempering Survey

				]	IARDI		(R <sub>C</sub> )				RING
	·	A - O	Tempering	7.	10	<u>Tin</u> 30	<u>ie at</u> 60			<u>sture</u> 500	1000
Grade	Hardening Treatment*	As-Quenched Hardness R <sub>C</sub>	Temperature (°F)	hrs	hrs	_				-	hrs
Halmo-1	2100 F 20 min.	64	400 600 800 1000	60 58 60 66	60 58 61 65	58 58 61 64	58 58 62 63	58 58 62 63	58 58 63 61	58 59 64 56	58 59 65 53
Ferrovac Halmo	2100 F 20 min	64	400 600 800 1000	60 60 62 64	_	59 60 62 62	59 60 62 60	59 59 62 59		60 60 64 50	59 61 65 50
Halmo-2	2100 F 20 min	65	400 600 800 1000	60 5 <b>9</b> 61 65	5 <b>9</b> 62			63	5 <b>9</b> 64	64	59 60 65 53
.8C Halmo	2050 F 20 min	63	400 600 800 1000	60 58 59 65	58 59	58 60	61	58 61	5 <b>9</b> 62	59 64	,
•9C Halmo	1850 F 60 min	66	400 600 800 1000	61 59 60 64	60	60	62	60 62	62	60 64	65
521.00	1550 F 60 min	66	400 600 800	60 55 49	55	54	53	53	52	53	52
Ferrovac 52100	1550 F 60 min	66	400 600 800	61 57 47	57	55	55		54	55	54
MHT	1550 F 60 min	. 66	400 600 800	62 60 52	61	. 60	60	60	59	59	59
Ferrovac MHT	1550 F 1 hr	66	400 600 800	62 60 53	60	60		60	) 60	59	59

<sup>\*</sup> All samples were oil quenched

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# TABLE III (Continued)

# Results of Tempering Survey

Tempering   Temp					1	HARDI	VESS	(R <sub>C</sub>	) AF	rer 7	rempi	ERING	
Create   Treatment*   Fardness Rc   (°F)   hrs   hrs					<u> </u>								
MHTT + 81 1600 F 66 400 62 63 61 62 62 62 62 62 62 62 62 62 62 62 62 62	Grade	_	-		4 hrs		_				-		
60 min 600 60 60 60 60 60 60 60 60 60 61 60 60 61 60 60 61 60 60 61 61 60 60 60 60 60 61 61 61 61 61 61 61 61 61 61 61 61 61	Grade	11 Ca chieffo.	Hardness IC			111 5	<del></del>	111 5	111 2	111.2	111 2	111.5	
### 100 C 1850 F 60 min	MHT + Si	1600 F	66	400	62	_		62	62	62	62	62	
### C 1850 F 60 min 62		60 min		600	60	60	60	61	60	60	61	60	
60 min 600 57 56 56 56 57 57 57 57 57 57 57 57 57 57 57 57 57				800	55	54	53	53	53	52	51	51	
60 min 600 57 56 56 56 57 57 57 57 57 57 57 57 57 57 57 57 57	440 C	1850 F	62	400	58	58	57	57	57	57	57	57	
\$\begin{array}{c c c c c c c c c c c c c c c c c c c				600		_	_						
60 min 600 55 55 56 56 57 58 58 59 60 62  T1 2350 F 67 400 63 69 62 62 62 63 63 63 63 60 60 60 60 62 63 64 64 64 64 64 65 66 66 66 60 1000 67 66 65 65 65 64 63 60  T5 2300 F 66 400 63 69 62 62 62 62 63 63 63 60 62 62 62 63 63 63 60 60 60 60 60 60 60 60 60 60 60 60 60				800				_					
60 min 600 55 55 56 56 57 58 58 59 60 62  T1 2350 F 67 400 63 69 62 62 62 63 63 63 63 60 60 60 60 62 63 64 64 64 64 64 65 66 66 66 60 1000 67 66 65 65 65 64 63 60 62 62 62 62 63 63 63 60 60 60 60 60 60 60 60 60 60 60 60 60	440 BM	1 <b>90</b> 0 F	62	400	58	58	57	57	57	56	57	56	
## 10 20 50 50 60 62  ## 10 2350 F  ## 10 2350 F  ## 10 2350 F  ## 10 20 50 56 57  ## 10 2350 F  ## 10 30 60 100 200 500 1000		_			-	-	56			_			
5 min 600 62 63 62 63 62 63 63 63 63 63 63 63 63 63 63 63 63 63							57	58	<b>5</b> 8		-		
5 min 600 62 63 62 63 63 62 63 63 63 63 63 63 63 63 63 63 63 63 63	ητj	2350 F	67	400	63	69	62	62	62	62	63	63	
## 10 20 50 56 65 66 66 65 65 66 66 66 66 66 66 66	<u> </u>	•	- 1		_			_		_			
1000 67 66 65 65 65 64 63 60    4						_		_			-	66	
T5 2300 F 66 400 61 60 59 59 63 66 60 62 62 62 62 62 62 62 62 65 65 400 63 63 63 62 62 62 62 62 62 65 65 65				1000							63	60	
T5 2300 F 66 400 61 60 509 59 59 63 66 60 60 62 62 62 62 62 62 62 62 62 65 65 65 66 600 63 63 63 62 62 62 62 62 62 62 62 62 62 62 62 62													
T5 2300 F 66 400 61 60 59 59 63 700 60 62 62 62 62 62 62 62 62 62 62 62 62 62								•					
5 min 500 600 59 59 59 63 700 600 59 60 59 59 63 700 600 60 60 60 60 60 60 60 60 60 60 60					hrs	<u>hrs</u>	hrs	<u>hrs</u>	hrs	<u>hrs</u>	hrs	$\underline{\mathtt{hrs}}$	hrs
M2   2250 F   66   400   63   63   64   64   65   66   1000   67   67   67   67   67   67   67	<b>T</b> 5	2300 F	66	400		61					60		62
700 800 60 62 62 62 62 62 62 62 62 62 62 62 62 62		5 min		_			60						
Boo   60   62   63   66   1000   67   67   66   59   1050   1050   1100   57						5 <b>9</b>		60			5 <b>9</b>		63
1000 67 67 66 59 1050 64 1100 57  4 10 30 60 100 200 500 1000 hrs				<u> </u>	. 60			_				63	66
1050 64 57  4 10 30 60 100 200 500 1000 hrs		-					67	02		66		03	
1100 57  4 10 30 60 100 200 500 1000 hrs					Οí		<b>V</b> 1	64					79
M2   2250 F   66   400   63   63   62   62   62   61   62   62   600   61   61   61   61   61   61   6				-					57				
M2   2250 F   66   400   63   63   62   62   62   61   62   62   600   61   61   61   61   61   61   6					14	10	30	60	100	200	500	1000	·
M2 2250 F 66 400 63 63 62 62 62 61 62 62 10 min 600 61 61 61 61 61 61 62 62 800 62 63 63 64 64 64 65 66 1000 67 67 65 64 64 63 61 61 61 62 62 M2 10 min 600 62 62 62 62 62 62 62 62 62 62 800 63 63 62 62 62 62 62 62 62 62 800 63 63 63 62 63 64 64 65 65							•				_		
10 min 600 61 61 61 61 61 62 62 800 62 63 63 64 64 64 65 66 1000 67 67 65 64 64 63 61 61 61 62 62 M2 10 min 600 63 63 62 62 62 62 62 62 62 62 62 62 62 62 62				1								_	•
800 62 63 63 64 64 64 65 66 1000 67 67 65 64 64 63 61 61 Ferrovac 2250 F 65 400 63 63 62 62 61 61 62 62 M2 10 min 600 62 62 62 62 62 62 62 800 63 63 62 63 64 64 65 65	WZ	-	66	_		63	62						
1000 67 67 65 64 64 63 61 61  Ferrovac 2250 F 65 400 63 63 62 62 61 61 62 62  M2 10 min 600 62 62 62 62 62 62 62  800 63 63 63 62 63 64 64 65 65		TO WIN											
M2 10 min 600 62 62 62 62 62 62 62 62 62 62 65 800 63 63 64 64 65 65					_	_	•				_	_	
M2 10 min 600 62 62 62 62 62 62 62 62 62 62 65 800 63 63 64 64 65 65	Fernance	2250 F	٨¤	المص	62	62						60	
800 63 63 62 63 64 64 65 65			Q)	_	_		_	_	_	_			
	T-1 <del></del>				-	_						-	
							_				,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		

<sup>\*</sup>All samples were oil quenched.

# TABLE III (Continued)

### Results of Tempering Survey

				HA	RDNE		(R <sub>C</sub> )			MPEF		
	Hardening	As-Quenched	Tempering Temperature	4	10	30		• "		<u>500</u>	1000	
Grade	Treatment*	Hardness R <sub>C</sub>	(°F)	hrs	hrs	hrs	hrs	hrs	hrs	hrs	hrs	-
Ml	2200 F 15 min	64	400 600 800 1000	62 61 62 67	62 62 67	61 60 63 65	61 60 63 65	61 60 63 64	61 60 63 63	61 65 61	61 66 61	
M10	2200 F 15 min	64	400 600 800 1000	60 58 5 <b>9</b> 67	60 5 <b>9</b> 60 67	59 58 60 65	59 58 61 65	59 58 61 64	5 <b>9</b> 58 62 63	58 59 63 61	59 59 65 61	
H1C-M10	2300 F 10 min	65	400 600 800 1000	61 58 5 <b>9</b> 67	61 58 60 68	60 57 60 67	60 58 61 66	60 58 61 66	60 57 62 65	60 5 <b>9</b> 65 64	60 5 <b>9</b> 66 63	
VSM	2050 F 20 min	63	400 600 800 1000	60 59 60 63	60 60 61 63	60 59 61 61	60 59 62 60	60 60 62 58		64	60 64 51	
<b>M</b> 50	2100 F 20 min	64	400 600 800 1000	59 56 57 65	59 57 58 65	57 56 58 64	57 56 59 64	58 56 59 63	60	58 57 62 60	58 58 63 60	
UC	1800 F 60 min	66	400 6 <b>0</b> 0 800 1000	62 59 59 59	61 59 59 57	60 58 5 <b>9</b> 55	60 59 60 54	59 60	59 61	60 59 61 48	60 5 <b>9</b> 61 45	
		EXP	ERIMENTAL ANALYSES	-								
				4 <u>hrs</u>	10 <u>hrs</u>		•	-			200 <u>hrs</u>	1000 hr:
<b>A</b>	2000 F 30 min		400 500 600 700 800 1000	57 64	59 57	57 57	57 60 53	•	53	57 56 57	61	57 58 63
			1050 1100				53 	47				

<sup>\*</sup>All samples were oil quenched.

# TABLE III (Continued)

# Results of Tempering Survey

				HARDNESS (R <sub>C</sub> ) AFTER TEMPERING Time at Temperature								
			Tempering			'l'1n	le an	t Ter	nper	ature	<del></del>	
Grade	Hardening Treatment*	As-Quenched Hardness R <sub>C</sub>	Temperature (°F)	4 hrs	10 <u>hrs</u>	20 hrs	50 <u>hrs</u>	56 hrs				1000 hrs
B	2150 F 15 min	61	400 500 600 700 800 1000 1050 1100	56 65	58 55	56 65	55 59 64	54	63	<b>57</b> 55 55	60	57 55 62 57
C	2100 F 20 min	57	400 500 600 700 800 1000 1050 1100	49 62	53 50	49 65	54 53 64	56	61	51 49 49	54	50 49 56 49
D	2200 F 15 min	65	400 500 600 700 800 1000 1050 1100	61 64	62 60	60	61 63	52	61	61 60 60	63	61 61 64
E	2150 F 15 min	61.	400 500 600 700 800 1000 1050	57 63	58 56	56 55	57 5 <b>9</b> 51		53 49	56 55 56	60	56 56 61
F	2200 F 15 min	63	400 500 600 700 800 1000 1050 1100	57 64	60 57	58 64	47 60 62	56	64	58 57 56	61	58 57 62 60

<sup>\*</sup> All samples were oil quenched WADC TR 57-343

# TABLE III (Continued)

# Results of Tempering Survey

	•			HARDNESS (R <sub>C</sub> ) AFTER TEMPERING Time at Temperature								
Grade	Hardening Treatment*	As-Quenched Hardness R <sub>C</sub>	Tempering Temperature (°F)	4 hrs	10 hrs	20 hrs	50 hrs	56 hrs				1000 hrs
G	2200 F 15 min	66	400 500 600		61 59	60				60 59 59		62 63
			700 800 1000 1050 1100	60 67		67	60 62 64	57	66		63	66 5 <b>9</b>

<sup>\*</sup> All samples were oil quenched

TABLE IV

ssults of Hot Hardness Survey

dness R <sub>C</sub> after O hrs Exposure Temperatures F	Elevated Temp.	59 55 35	25 27 27 27 27 27 27 27 27 27 27 27 27 27	4883	62 59 40	60 57 45 75	52 47 -
;	g o	2555	62 62 48 48 48 48 48 48 48 48 48 48 48 48 48	65 65 53	25 25 26 26 27 27	65 65 57	57 53
Hardness 1000 hrs at Tempe	Room	000 1000 1000	009 800 1000	400 600 1000	400 1000	400 600 1000	7 000 800
Room Temp. Hardness	<u> </u>	· <del>19</del>	63	9	99	9	55
Momnorature	ss R <sub>C</sub> at atures F	53 53	59 57 55 52	60 59 53	62 59 53	60 57 58 53	56 51 42
E to the table		7 600 1000	7000 1000 1000	700 1000 1000	400 1000 1000	400 600 1000	000 400 800
Handrace after	<b>~</b> 1	<del>1</del> 9	63	9	99	65	
	Heat Treatment*	2100 F/20 min. 1050 F/2+2 hrs	2100 F/20 min. 1050 F/2+2 hrs	2100 F/20 min. 1050 F/2+2 hrs	2050 F/20 min. 1050 F/2†2 hrs	2050 F/25 min. 1075 F/2+2 hrs	1550 F/1 hr 400 F/2+2 hrs
	Grade	Halmo-1	Ferrovac Halmo	Halmo-2	.8c Halmo	.9c Halmo	52100

temperature. empers at the indicated temperature. from the austenitizing Two consecutive 2 hr. All Steels were oil quenched Double tempering operation -

TABLE IV (Continued)

Results of Hot Handmess Survey

Hardness Rc after 1000 hrs Exposure at Temperatures F	Room Elevated Temp. Temp.	400 58 53 600 51 45	400 63 58 600 59 53 800 46 38	400 62 57 600 58 52 800 44 36	400 62 57 600 60 54 800 48 40	400 61 57 600 61 55 800 59 51	400 61 56 600 61 53 800 59 51	400 66 61 600 66 60 800 66 57 1000 61 49
Room Temp. Hardness	arter not Hardness Test, R <sub>C</sub>	26	59	62	. 59	61		99
	remperature ss Rc at atures F	57 49	58 52 42	58 -	57 54 -	57 55 47	57 55 53	61 60 57 54
	Elevated Tempe Hardness RC Temperatures	400 600	004 009 800	009 800	00 <del>1</del> 009 800	00 <del>1</del> 900 800	004 900 800	400 600 1000
	Hardness after Quench and Temper, R <sub>C</sub>	62	63	62	62	61	62	. 99
	Heat Treatment*	1550 F/1 hr. 400 F/2 hr.	1550 F/1 hr. 400 F/2 hr.	1550 F/1 hr. 400 F/2 hrs	1600 F/1 hr. 400 F/2 hrs	1950 F/1 hr. 350 F/1 hr. Refrigerated 900 F 2+2 hrs	1950 F/1 hr. 350 F/1 hr. Refrigerated 900 F 2+2 hrs	2350 F/5 min. 1050 F/2+2 hrs
	Grade	Ferrovac 52100	MHT	Ferrovac	MHT + Si	ρ. ο.	hto BM	11

indicated temperatures. All steels were oil quenched from the austenitizing temperature. Double Tempering operation - Two consecutive 2 hr. tempers at the

TABLE IV (Continued)

Results of Hot Hardness Survey

Hardness Rc after 1000 hrs Exposure at Temperatures F	Room Elevated Temp. Temp.	400 67 64 600 67 63 800 67 61 1000 61 51	400. 66 62 600 66 60 800 65 57 1000 61 49	1000 66 61 800 66 59 1000 60 49	400 66 62 600 66 60 800 66 57 1000 62 52	400 66 61 600 66 61 800 66 57 1000 57 45	400 67 62 600 67 60 800 66 58 1000 64 53
Room T Hardne	e after Hot Hardness Test, R <sub>C</sub>		99	99	99	99	2.9
	Temperature ss R <sub>C</sub> at atures F	65 64 62 57	62 53 55	66 57 55	61 53 55	85 27 27 27 27 27 27	62 59 59
	Elevated Tempe Hardness R <sub>C</sub> Temperatures	400 600 1000	400 600 1000	400 600 1000	1000 1000	400 600 1000	1000 1000 1000
	Hardness after Quench and Temper, R <sub>C</sub>		99	99	99	99	29
	Heat Treatment*	2300 F/5 min. 1000 F/2+2 hrs	2250 F/10 min. 1050 F/2+2 hrs	2250 F/10 min. 1050 F/2+2 hrs	2200 F/15 min. 1050 F/2+2 hrs	2200 F/15 min. 1050 F/2+2 hrs	2200 F/15 min. 1050 F/2+2 hrs
	Grade	T5	· Z	Ferrovac M2	Ţ.	MILO	H1C-M10

All steels were oil quenched from the austenitizing temperature. Double tempering operation - Two consecutive 2 hr. tempers at the indicated temperatures.

(Continued) TABLE IV

Survey Results of Hot Hardness

R <sub>C</sub> after exposure atures F	Elevated Temp.	85 75 4 4	61 57 55 46	55 53 53 53 53 54 54 54 54 54 54 54 54 54 54 54 54 54	57.75 3.45 3.45 3.45 3.45 3.45 3.45 3.45 3.4	67 28 4 4 4	60 57 36
된		333%	<del>4</del> 4 6 8 8 9 8 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9	62 62 54 54 54 54 54 54 54 54 54 54 54 54 54	# 65 65 65 65 65 65	96 55 55	2888
Hardness 1000 hrs at Tempe	Room	400 600 1000	600 1000 1000	000 1000	400 400 1000	000 1000 1000	000 1000 1000
n dne	after Hot Hardness Test, R <sub>C</sub>	<del>†</del> 9	<del>1</del> 9	63	<del>1</del> 79	99	99
	Temperature ss R <sub>C</sub> at ttures F	62 60 57 53	62 57 52	53. 53. 53.	59 53	61 59 53	61 59 57 53
	Elevated Temp Hardness R <sub>C</sub> Temperature	400 600 1000	400 600 1000	400 600 1000	7000 1000	400 600 1000	400 600 1000
	Hardness after Quench and Temper, R <sub>C</sub>	49	<del>1</del> 79	63	<del>1</del> 79	99	99
	Heat Treatment*	2100 F/20 min. 1000 F/2+2 hrs	2100 F/20 min. 1050 F/2+2 hrs	1950 F/45 min. 1000 F/2†2 hrs	2000 F/30 min. 1000 F/2+2 hrs	2150 F/15 min. 1000 F/2+2 hrs	2100 F/20 min 1000 F/2+2 hrs
	Grade	VSM	M50	a A	⋖	m	<b>U</b>

temperatures. temperature. tempers austenitizing scutive 2 hr. t Two consecutive quenched from the All steels were oil quenche Double tempering operation oil

\*

TABLE IV (Continued)

Results of Hot Hardness Survey

O hrs exposure Temperatures F	Elevated Temp.	59 57 45	53 35	55 55 148	63 59 45
ì H I	• 1	3308	63	65.65	67 67 57 57
1000 hrs	Room	600 1000 1000	000 1000 1000	000 1000 1000	400 600 1000
Room Temp. Hardness after Hot	OU I	<del>1</del> 9	63	65	· 64
Temperature	s Rc at tures F	58 53 53	58 53 49	55 57 57	63 59 56
Elevated	Hardness R <sub>C</sub> Temperature	600 1000	400 600 1000	400 600 1000	400 600 1000
Hardness after	Quench and Temper, Rc	<del>1</del> 79	63	9	29
	Heat Treatment*	2200 F/15 min. 1000 F/2+2 hrs	2100 F/20 min. 1000 F/2+2 hrs	2200 F/15 min. 1000 F/2†2 hrs	2200 F/15 min. 1000 F/2+2 hrs
	Grade	D	E .	[≖₁	٠ ت

temperature. austenitizing cutive 2 hr. t consecutive from the Two conse quenched eration - ' tempering operation All steels were oil Double tempering ope

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TABLE V

Dimensional Stability Test Results for Bearing Steels

Length Change Micro inch/inch in Specimens Exposed for 1000 hours at temperatures

				<del></del>	
Grade	Heat Treatment*	400 F	600 F	800 F	1000 F
Halmo-l	2100 F/20 min 1050 F/2+2 hrs	No Change	No Change	+ 10	- 36
Ferrovac Halmo	2100 F/20 min 1050 F/2+2 hrs	No Change	No Change	+ 12	- 47
Halmo-2	2100 F/20 min 1050 F/2+2 min	+ 15	+ 12	<b>+</b> 35	<del>-</del> 62
.8c Halmo	2050 F/25 min 1050 F/2+2 hrs	+ 37	- 7	<b>+</b> 3 <b>9</b> 5	- 57
.9C Halmo	2050 F/25 min 1075 F/2+2 hrs	+245	+ 50	+ 501	+1257
52100	1550 F/1 hr 400 F/2 hrs	+ 55	-742	-1030	-
Ferrovac 52100	1550 F/1 hr 400 F/2 hrs	+ 40	-730	- 8 <b>9</b> 5	
MHT	1550 F/1 hr 400 F/2 hrs	+ 20	+ 20	-1502	-
Ferrovac MHT	1550 F/1 hr 400 F/2 hrs	+242	+ 12	-1440	-
MHT + Si	1600 F/1 hr 400 F/2 hrs	+280	+120	<b>-</b> 1375	-57
440 C	1950 F/1 hr 350 F/1 hr Ref. 900 F/2+2 hrs	+ 22	+ 15	+1195	+847
440 BM	1950 F/1 hr 350 F/1 hr Ref. 900 F/2+2 hrs	<b>-</b> 52	- 52	+1137	+458
Tl	2350 F/5 min 1050 F/2+2 hrs	+ 7	- 75	+ 30	- 7

<sup>\*</sup> All specimens were oil quenched from the austenitizing temperature.

Double tempering operation involved two consecutive 2 hour tempers at the indicated temperature.

# TABLE V (Continued)

#### Dimensional Stability Test Results for Bearing Steels

Length Change Micro inch/inch in Specimens Exposed for 1000 hours at temperatures:

Grade	Heat Treatment*	400 F	600 F	800 F	1000 F
<b>T</b> 5	2300 F/5 min 1000 F/2+2 hrs	+ 1	+ 27	+ 46	+ 270
<b>M</b> 2	2250 F/10 min 1050 F/2+2 hrs	25	- 17	- 10	+ 32
Ferrovac M2	2250 F/10 min 1050 F/2+2 hrs	- 2	- 12	No Change	+ 15
MI	2200 F/15 min 1050 F/2+2 hrs	- 10	+ 20	- 7	<b>- 2</b> 5
<b>M</b> 10	2200 F/15 min 1050 F/2+2 hrs	- 25	- 17	- 10	No Change
HiC-MlO	2200 F/15 min 1050 F/2+2 hrs	- 10	No Change	- 28	+ 76
VSM	2050 F/20 min 1000 F/2+2 hrs	+ 15	No Change	+ 35	- 1
M50	2100 F/20 min 1050 F/2+2 hrs	+ 22	+12	+ 27	<del>-</del> 50
UC	1950 F/45 min 1000 F/2+2 hrs	+ 90	No Change	+ 37	- 132
A	2000 F/30 min 1000 F/2+2 hrs	-168	-120	+315	+1535
В	2150 F/15 min 1000 F/2+2 hrs	+ 1	- 16	+ 5	+ 60
C	2100 F/20 min 1000 F/2+2 hrs	<b>-</b> 65	- 55	+247	+ 467
D	2200 F/15 min 1000 F/2+2 hrs	_ 4	- 8	+ 75	+ 447
<b>E</b>	2100 F/20 min 1000 F/2+2 hrs	- 50	- 20	+ 96	+ 100

<sup>\*</sup> All specimens were oil quenched from the austenitizing temperature. Double tempering operation involved two consecutive 2 hour tempers at the indicated temperature.

## TABLE V (Continued)

## Dimensional Stability Test Results for Bearing Steels

Length Change Micro inch/inch in Specimens Exposed for 1000 hours at temperatures:

Grade	Heat Treatment*	400 F	600 F	800 <b>F</b>	1000 F
F	2200 F/15 min 1000 F/2+2 hrs	+ 105	- 10	+ 25	+ 523
G	2200 F/15 min 1000 F/2+2 hrs	+ 31	- 17	- 20	- 12

<sup>\*</sup> All specimens were oil quenched from the austenitizing temperature. Double tempering operation involved two consecutive 2 hour tempers at the indicated temperature.

TABLE VI

Compression Test Results for Bearing Steels

		Testing	Hot	Yield Strength, psi		
Grade	Heat	Treatment*	Temperature (F)	Hardness R <sub>C</sub>	0.10% Offset	0.2% Offset
Halmo-l	2100	F/20 min.	400	60	364,000	392,000
	1050	F/2+2 hrs	600	5 <del>9</del>	318,000	344,000
		•	800	56	268,000	300,000
			1000	53	244,000	268,000
Halmo	2100	F/20 min.	400	5 <b>9</b>	368,000	398,000
Ferrovac	1050	F/2+2 hrs	600	57	303,000	327,000
			800	55	290,000	312,000
			1000	52	220,000	236,000
Halmo-2	2100	F/20 min.	400	60	372,000	396,000
		F/2+2 hrs	600	59	332,000	362,000
	/-	-,-,-	800	57	298,000	322,000
			1000	53	267,000	273,000
.8C Halmo	2050	F/25 min.	400	62	388,000	400,000
		F/2+2 hrs	600	49	366,000	390,000
	,_	-, -,	800	56	283,000	329,000
			1000	53	273,000	296,000
.9C Halmo	2050	F/25 min.	400	60	3 <b>9</b> 2,000	406,000
		F/2+2 hrs	600	57	344,000	388,000
		-/	800	56	248,000	380,000
			1000	5 <b>3</b>	168,000	204,000
52100	1550	F/1 hr	400			•
72100		F/2 hrs		56	188,000	232,000
	400	r/2 nrs	600	51	137,000	159,000
			800	42	104,000	119,000
52100	1550	F/1 hr	400	57	196,000	240,000
Ferrovac		F/2 hrs	600	49	152,000	177,000
MHT	1550	F/l hr	400	58	222,000	303,000
		F/2 hrs	600	52	167,000	193,000
		-,	800	42	135,000	157,000
MHT	1550	F/1 hr.	400	58	256,000	294,000
Ferrovac		F/2 hrs	600	52	187,000	208,000
MHT+ Si	1600	F/l hr	400	57	240,000	276,000
	400	F/2 hrs	600	52	176,000	216,000
· · · · · · · · · · · · · · · · · · ·		-/		<i></i>		

<sup>\*</sup> All Steels were oil quenched from the austenitizing temperature.

Double tempering operation - Two consecutive 2 hr. tempers at the indicated temperatures.

TABLE VI (Continued)

Compression Test Results for Bearing Steels

		Testing	Hot	Yield Strength, psi	
Grade	Heat Treatment*	Temperature (F)	Hardness R <sub>C</sub>	0.10% Offset	0.2% Offset
440 C	1950 F/1 hr 350 F/1 hr Refrigerated 900 F/2+2 hrs	400 600 800	57 55 47	288,000 252,000 216,000	300,000 282,000 234,000
440 BM	1950 F/1 hr 350 F/1 hr Refrigerated 900 F/2+2 hrs	400 600 800	57 55 53	270,400 212,000 160,000	297,600 248,000 163,000
Tl	2350 F/5 min 1050 F/2+2 hrs	400 600 800 1000	61 60 57 54	384,000 323,000 268,000 264,000	410,000 350,000 288,000 280,000
<b>T</b> 5	2300 F/5 min 1000 F/2+2 hrs	400 600 800 1000	65 64 62 57	426,000 402,000 380,000 332,000	440,000 410,000 384,000 356,000
M2	2250 F/10 min 1050 F/2+2 hrs	400 600 800 1000	62 61 5 <b>9</b> 55	384,000 352,000 287,000 224,000	392,000 367,000 300,000 248,000
M2 Ferrovac	2250 F/10 min 1050 F/2+2 hrs	400 600 800 1000	61 60 58 55	376,000 372,000 308,000 248,000	400,000 3 <b>92</b> ,000 336,000 256,000
Ml.	2200 F/15 min 1050 F/2+2 hrs	400 600 800 1000	61 60 58 . 55	344,000 288,000 364,000 228,000	368,000 372,000 364,000 252,000
Mlo	2200 F/15 min 1050 F/2+2 hrs	400 600 800 1000	62 61 58 54	376,000 372,000 260,000 270,000	424,000 396,800 328,000 292,000
HiC-MlO	2200 F/15 min 1050 F/2+2 hrs	400 600 800 1000	62 61 59 56	392,000 328,000 368,000 312,000	404,000 372,000 376,000 320,000

<sup>\*</sup> All steels were oil quenched from the austenitizing temperature.

Double tempering operation - Two consecutive 2 hr. tempers at the indicated temperatures.

TABLE VI (Continued)

Compression Test Results for Bearing Steels

		Testing	Hot	Yield Strength, psi.	
Grade	Heat Treatment*	Temperature (F)	Hardness R <sub>C</sub>	0.10% Offset	0.2% Offset
VSM	2050 F/20 min 1000 F/2+2 hrs	400 600 800 1000	62 60 57 53	326,000 312,000 292,000 268,000	346,000 339,000 316,000 286,000
<b>M</b> 50	2100 F/20 min 1050 F/2+2 hrs	400 600 800 1000	62 59 57 52	347,500 280,000 256,000 248,000	358,000 328,000 288,000 268,000
UC	1950 F/45 min 1000 F/2+2 hrs	400 600 800 1000	59 56 53 48	248,000 232,000 216,000 176,000	286,000 264,000 288,000 188,000
Experiment	tal				·
Steel A	2000 F/30 min 1000 F/2+2 hrs	400 600 800 1000	59 59 57 53	248,000 228,000 204,000 144,000	268,000 244,000 220,000 160,000
Steel B	2150 F/15 min 1000 F/2+2 hrs	400 600 800 1000	61 59 57 53	384,000 332,000 272,000 180,000	384,000 360,000 288,000 204,000
Steel C	2100 F/20 min 1000 F/2+2 hrs	400 600 800 1000	61 59 57 53	292,000 224,000 190,000 184,000	336,000 252,000 216,000 204,000
Steel D	2200 F/15 min 1000 F/2+2 hrs	400 600 800 1000	60 58 57 53	320,000 298,000 268,000 196,000	342,000 320,000 304,000 210,000
Steel E	2100 F/20 min 1000 F/2+2 hrs	400 600 800 1000	58 57 53 49	335,000 320,000 288,000 176,000	348,000 324,000 304,000 200,000

<sup>\*</sup> All steels were oil quenched from the austenitizing temperature.

Double tempering operation - Two consecutive 2 hr. tempers at the indicated temperatures.

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## TABLE VI (Continued)

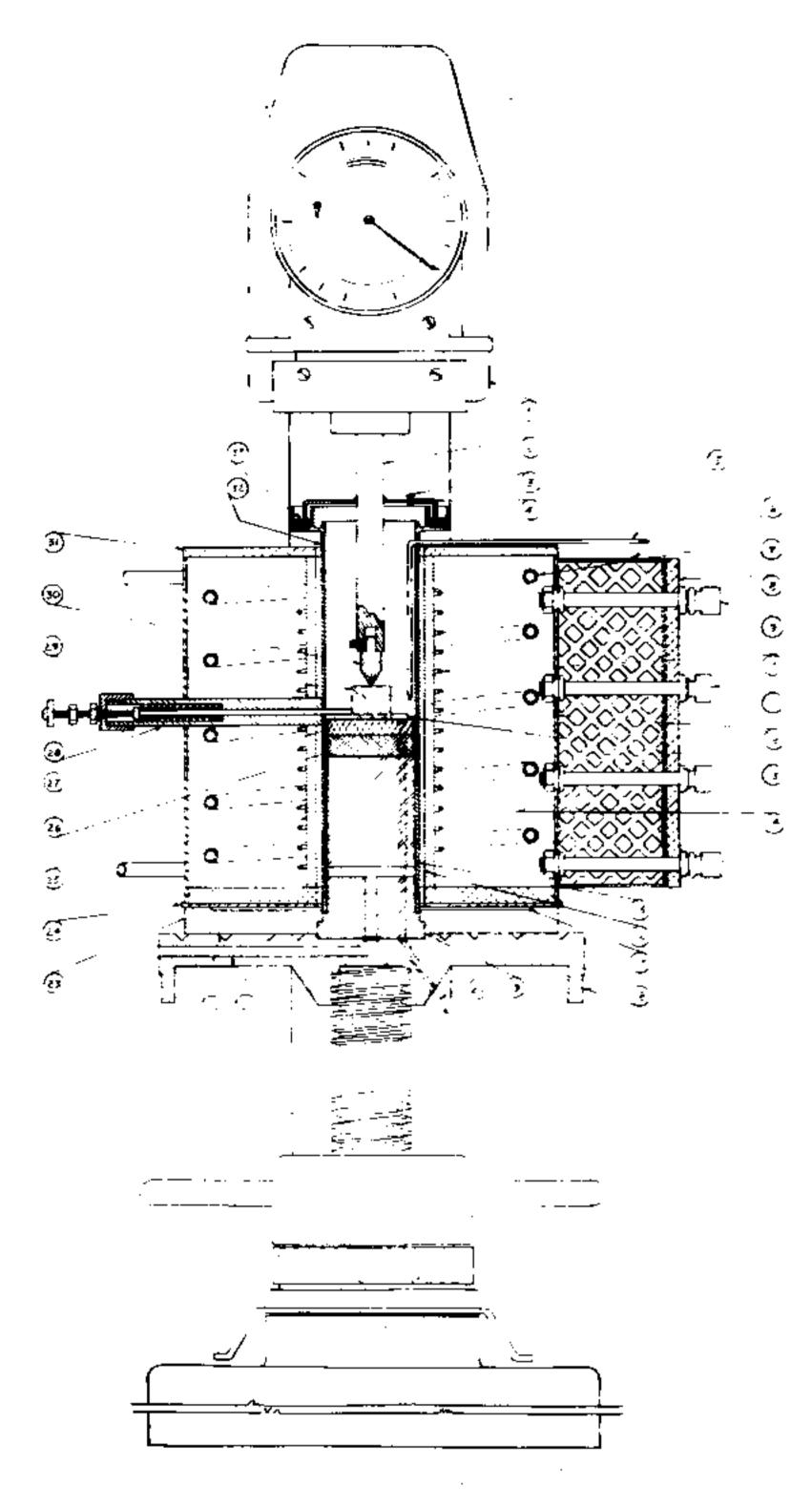
# Compression Test Results for Bearing Steels

Grade	Heat Treatment*	Testing Temperature (F)	Hot Hardness R <sub>C</sub>	Yield Strength, psi.				
				0.10% Offset	0.2% Offset			
Experimental								
Steel F	2200 F/15 min 1000 F/2+2 hrs	400 600 800 1000	60 58 55 51	268,000 256,000 240,000 228,000	304,000 288,000 256,000 251,000			
Steel G	2200 F/15 min 1000 F/2+2 hrs	400 600 800 1000	63 62 59 56	400,000 348,000 328,000 272,000	416,000 380,000 344,000 312,000			

<sup>\*</sup> All steels were oil quenched from the austenitizing temperature.

Double tempering operation - Two consecutive 2 hr. tempers at the indicated temperatures.





#### **LEGEND**

- l. Hardness Tester
- 2. Indentor Extension Rod (Nichrome)
- 3. Gas Outlet Screw
- 4. Seal Cup (Nichrome)
- 5. Control Thermocouple
- 6. Holder Box
- 7. Cooling Coil (Stainless)
- 8. Insulation (Transite)
- 9. Heating Power Terminal
- 10. Heater Windings
- 11. Shunt Terminal
- 12. Furnace Shell
- 13. Specimen Guides
- 14. Furnace Insulation (Sil-O-Sel)
- 15. Air Space
- 16. Air Space
- 17. Air Space

- 18. Stage of the Rockwell Tester
- 19. Anvil (S-816)
- 20. Thermocouple (Anvil)
- 21. Gas Inlet
- 22. Air Space
- 23. Furnace Support (Transite)
- 24. Plate (Stainless)
- 25. Muffle (Ceramic)
- 26. Spacer (Soapstone)
- 27. Plate (S-816)
- 28. Specimen Positioning Screw
- 29. Specimen
- 30. High Temperature Indentor
- 31. Cover (Transite)
- 32. Muffle (Nichrome)
- 33. Liquid Metal Seal (Wood's Metal)

FIGURE 1. SCHEMATIC DRAWING OF THE ROCKWELL TYPE HOT-HARDNESS TESTING ASSEMBLY.

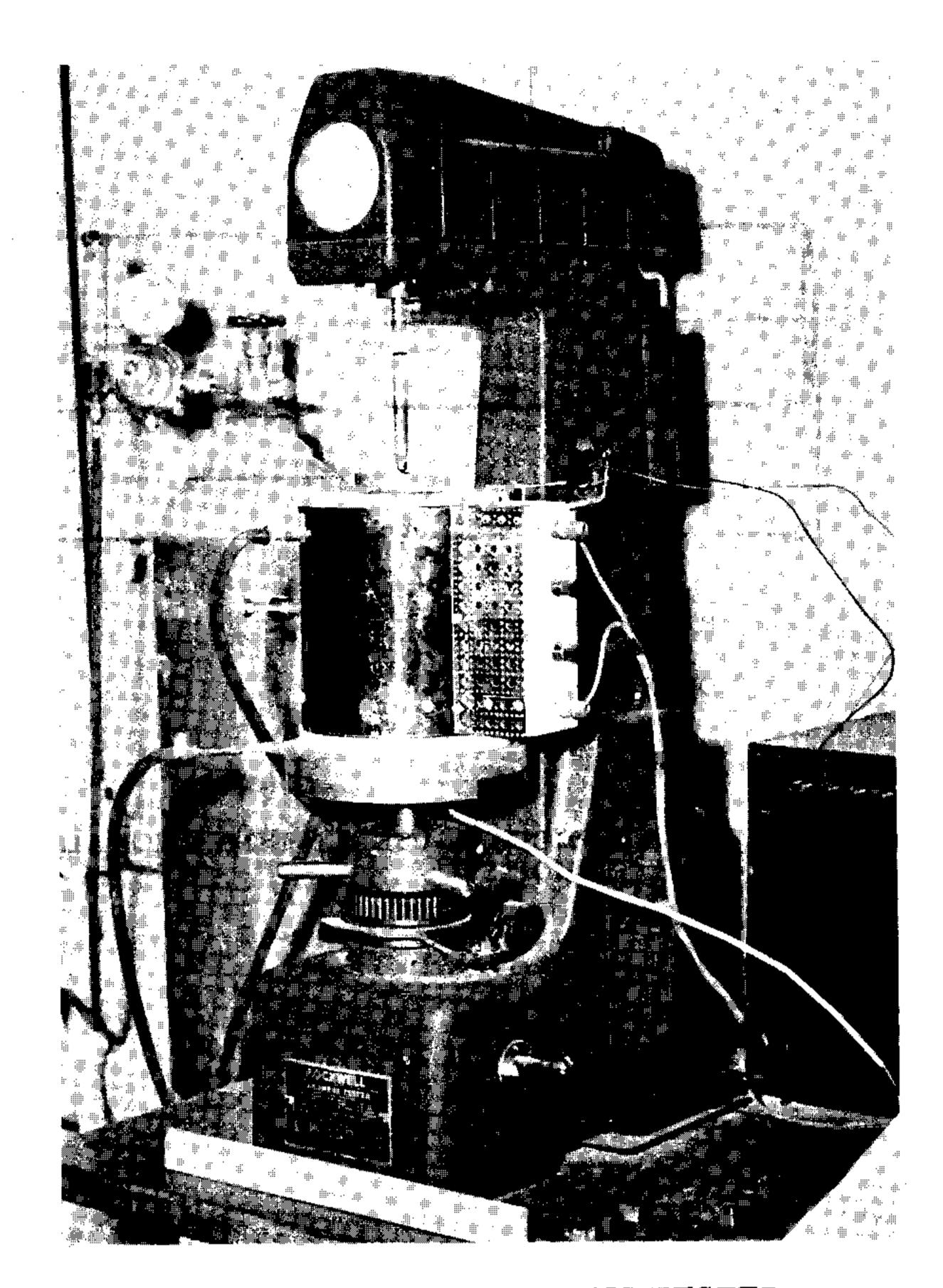
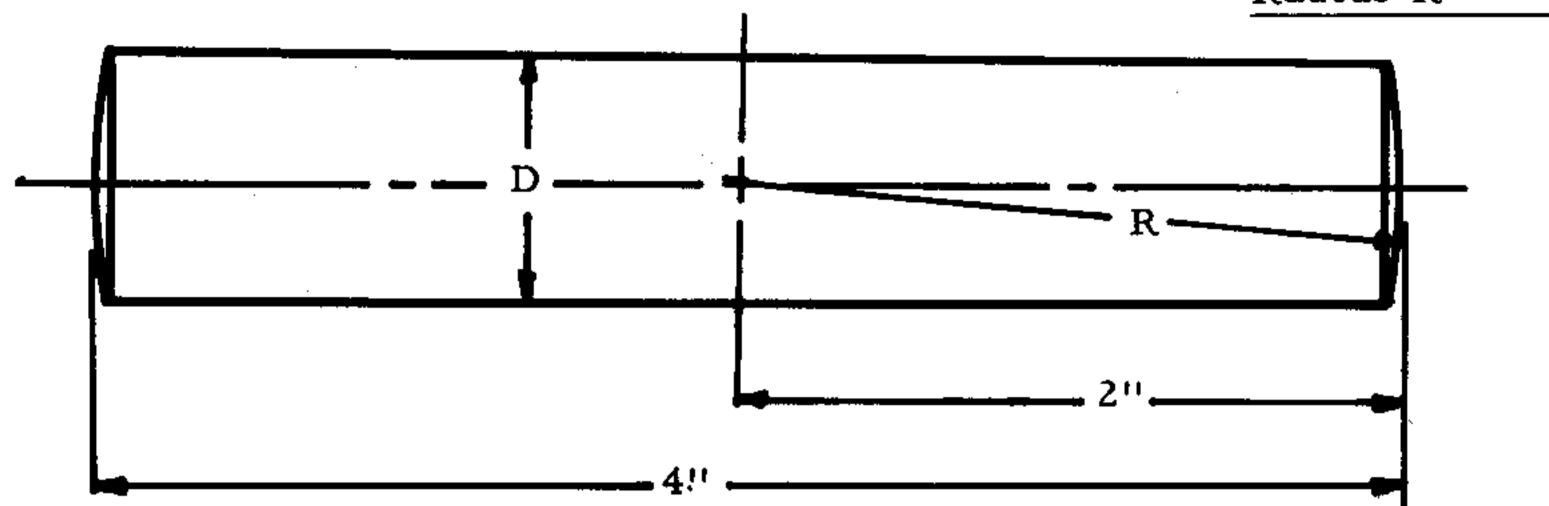


FIGURE 2. HOT HARDNESS TESTER

Both ends ground spherical to Radius R



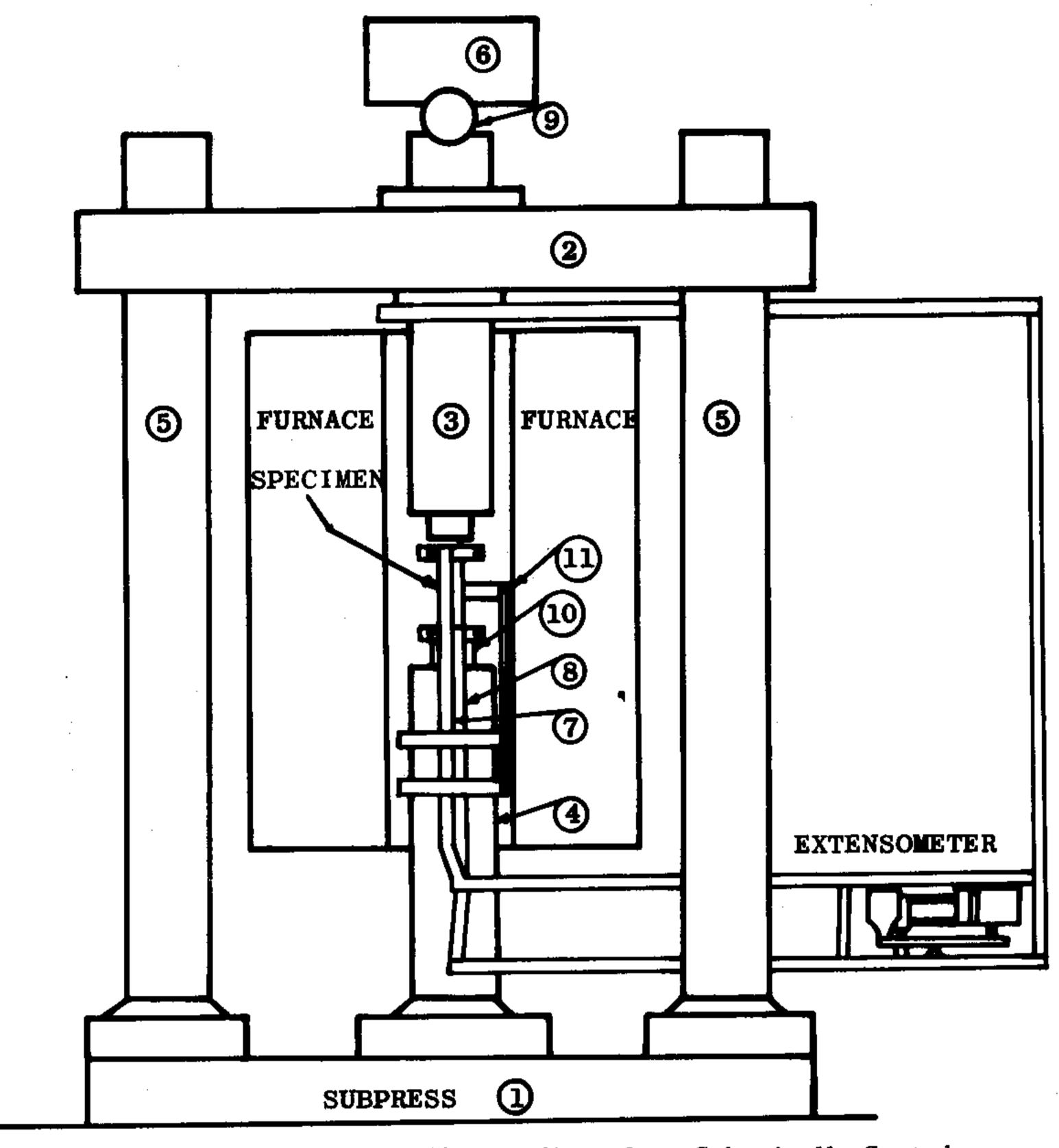
R = 2 in.

D = 3/8 in.

FIGURE 3. SPECIMEN USED IN DIMENSIONAL STABILITY STUDIES.

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- Bottom Plate
- (2) Top Plate (Fixed)(3) Upper Compression Rod
- (4) Lower Compression Rod
- (5) Subpress Guide Rods (3)
- (6) Loading Plate-Spherically Seated
- (7) Extensometer Upper Assembly
- (8) Extensometer Lower Assembly
- (9) Load Alignment Ball
- (10) Bearing Plate
- (11) Specimen Centering Device

COMPRESSION TEST FIXTURE FIGURE 4. (Schematic Drawing)

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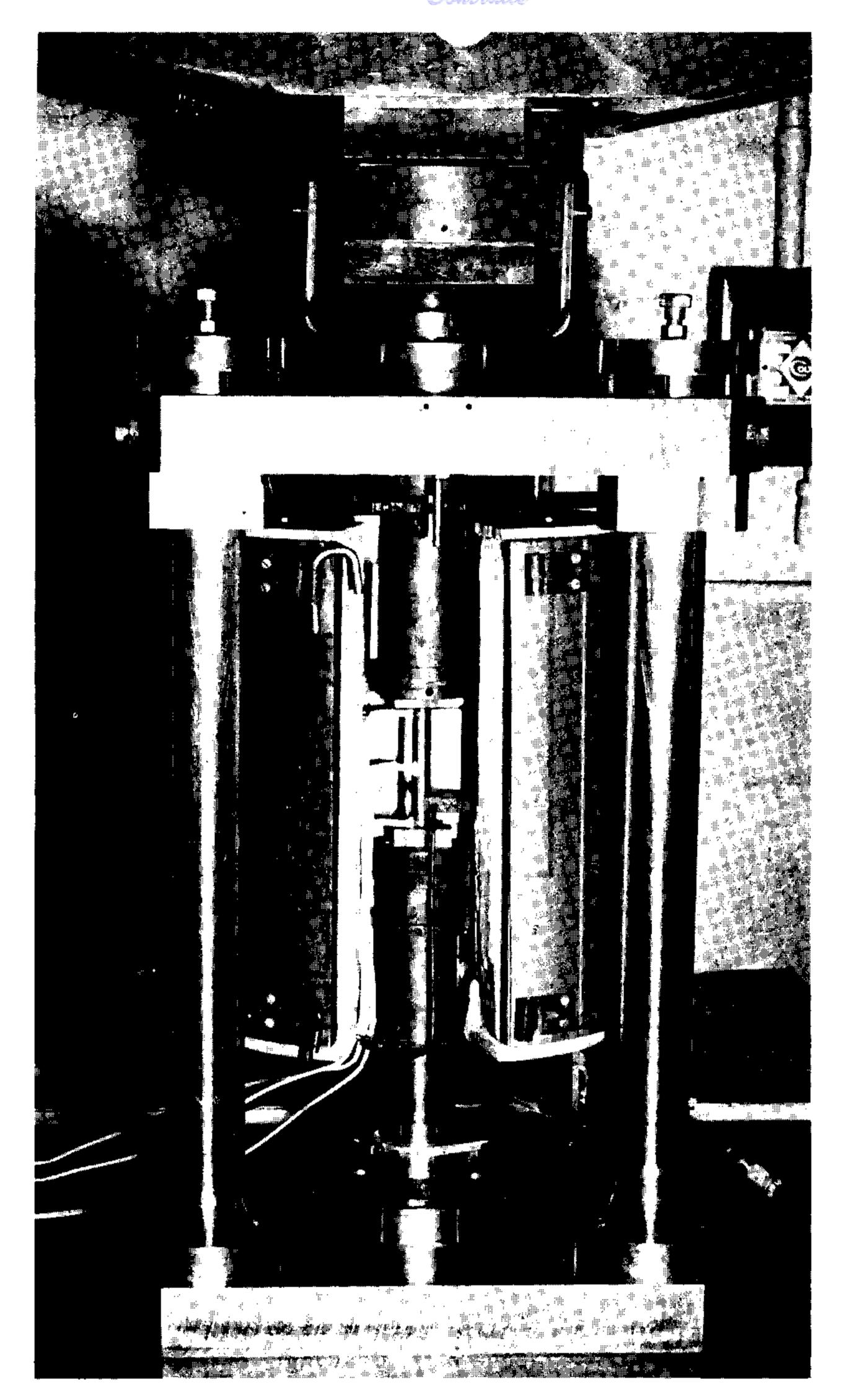
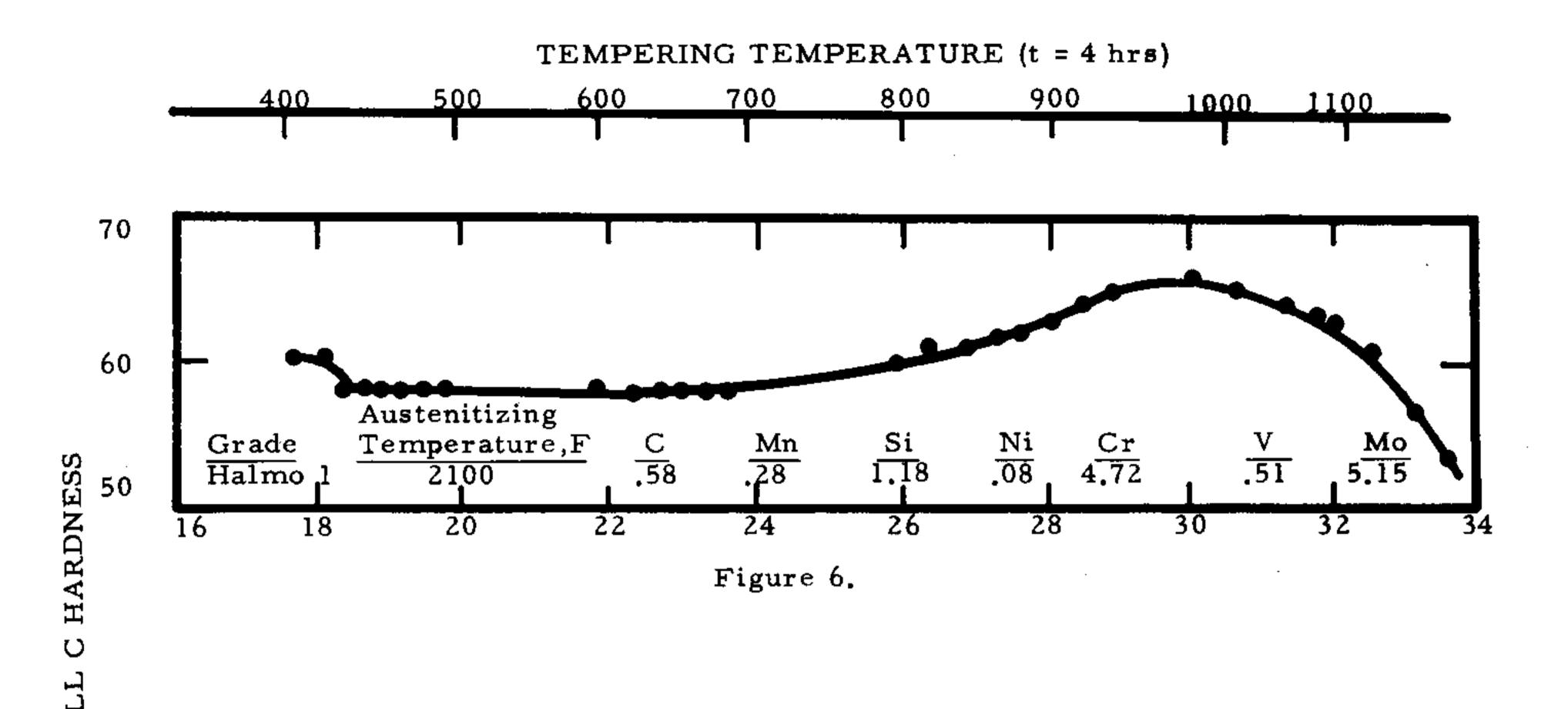


FIGURE 5. SUBPRESS USED IN ELEVATED TEMPERATURE COMPRESSION TESTS.



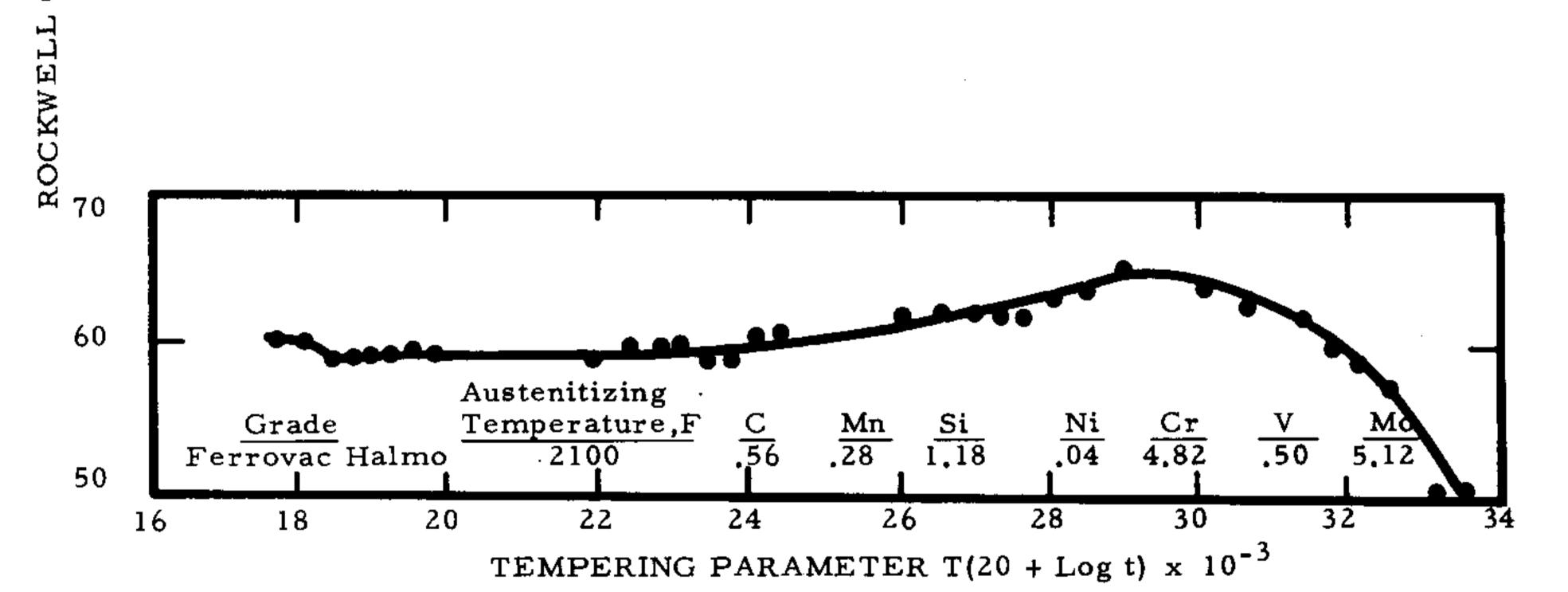


Figure 7.

Figures 6 and 7. Master Tempering Curves for Bearing Steels.

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### TEMPERING TEMPERATURE (t = 4 Hours)

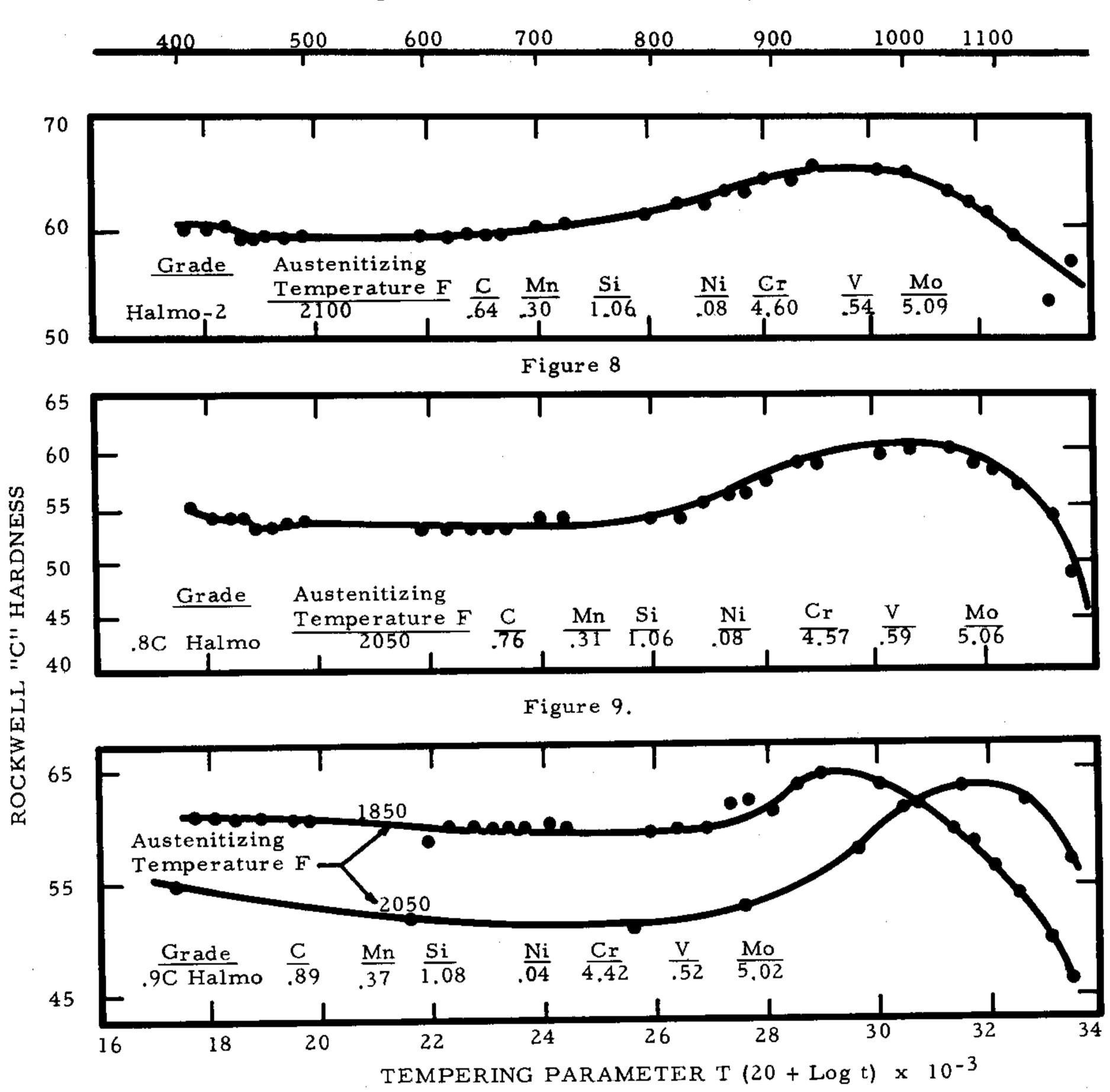


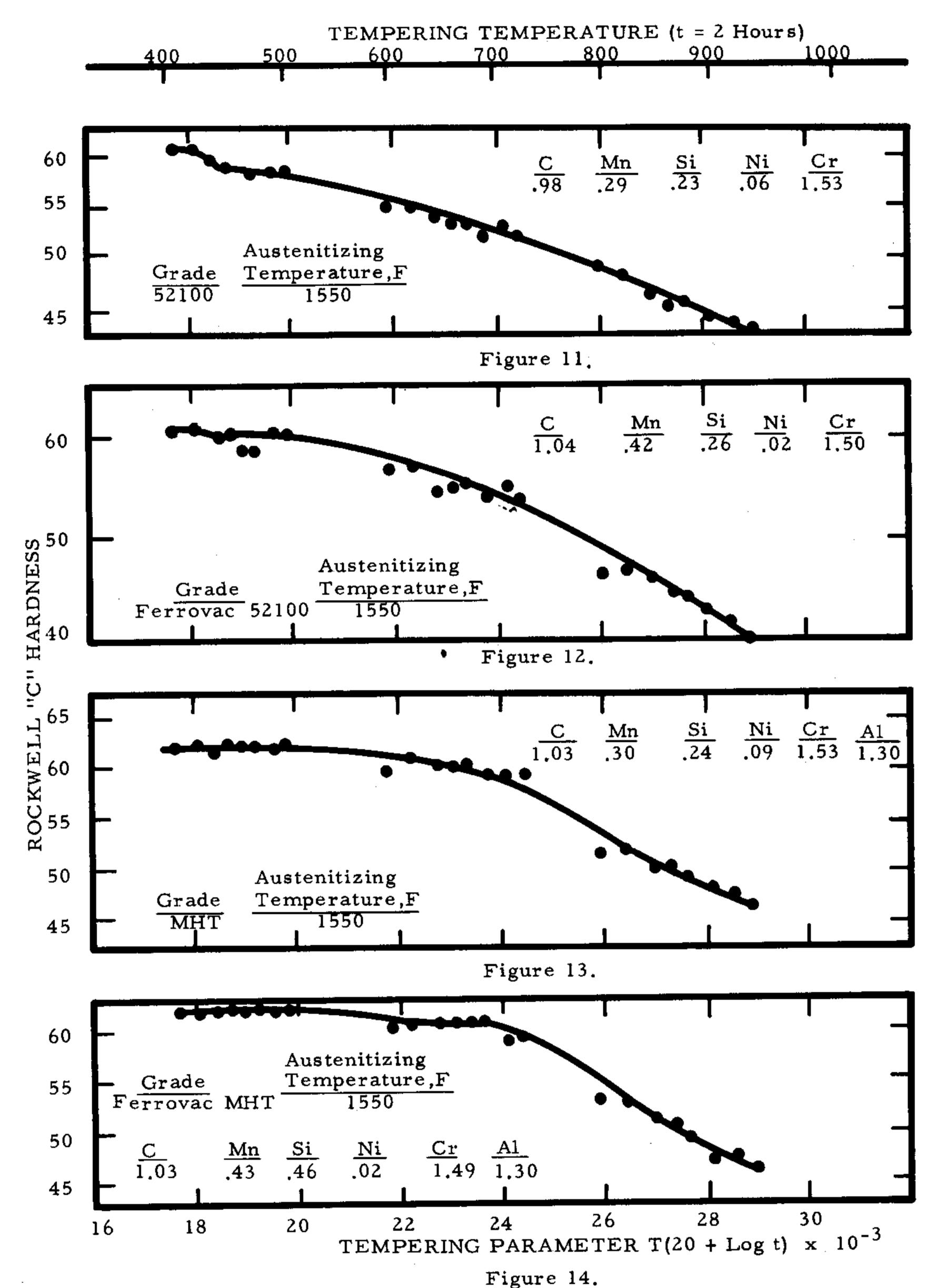
Figure 10

Figures 8 to 10. Master Tempering Curves for Bearing Steels

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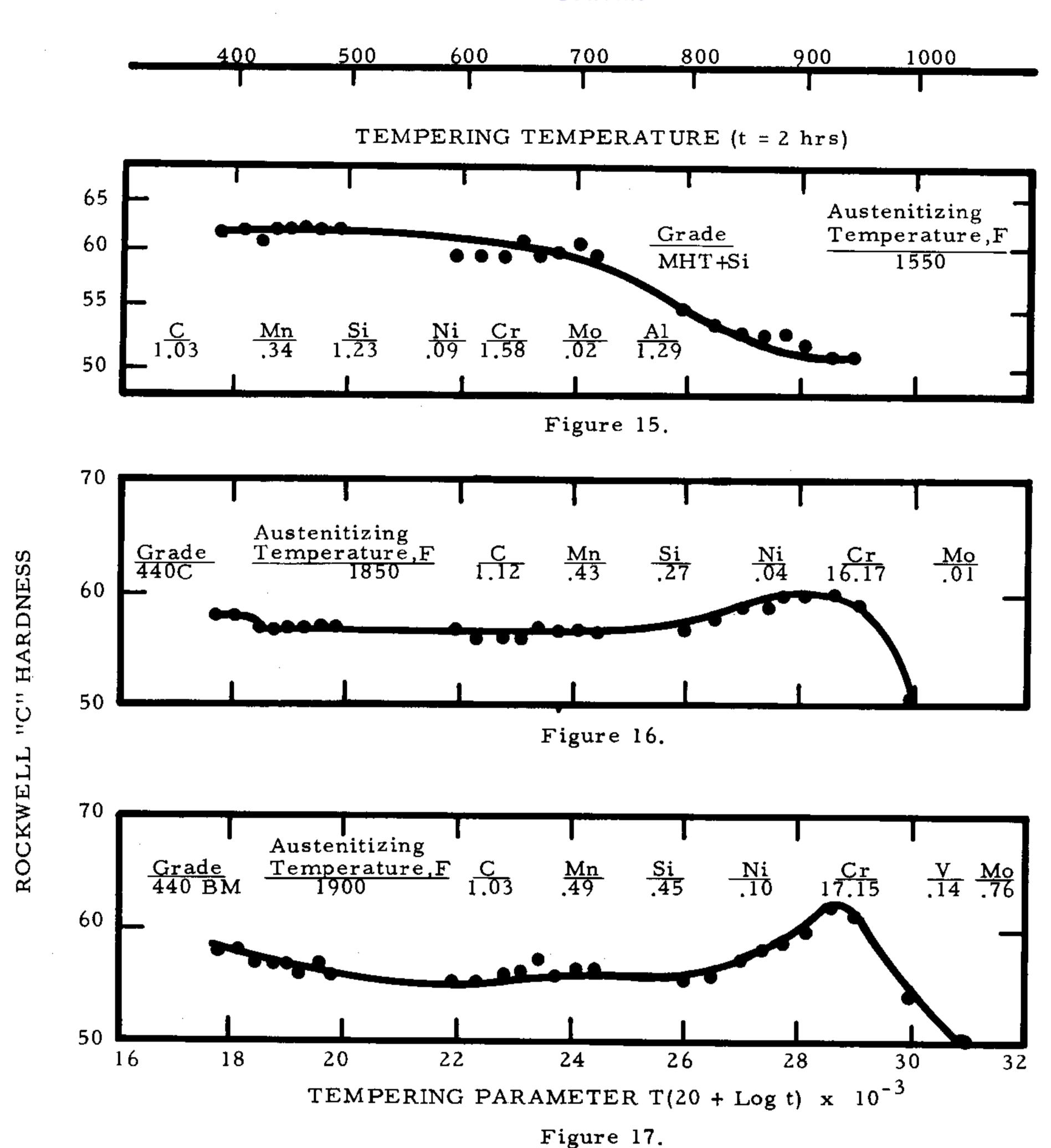
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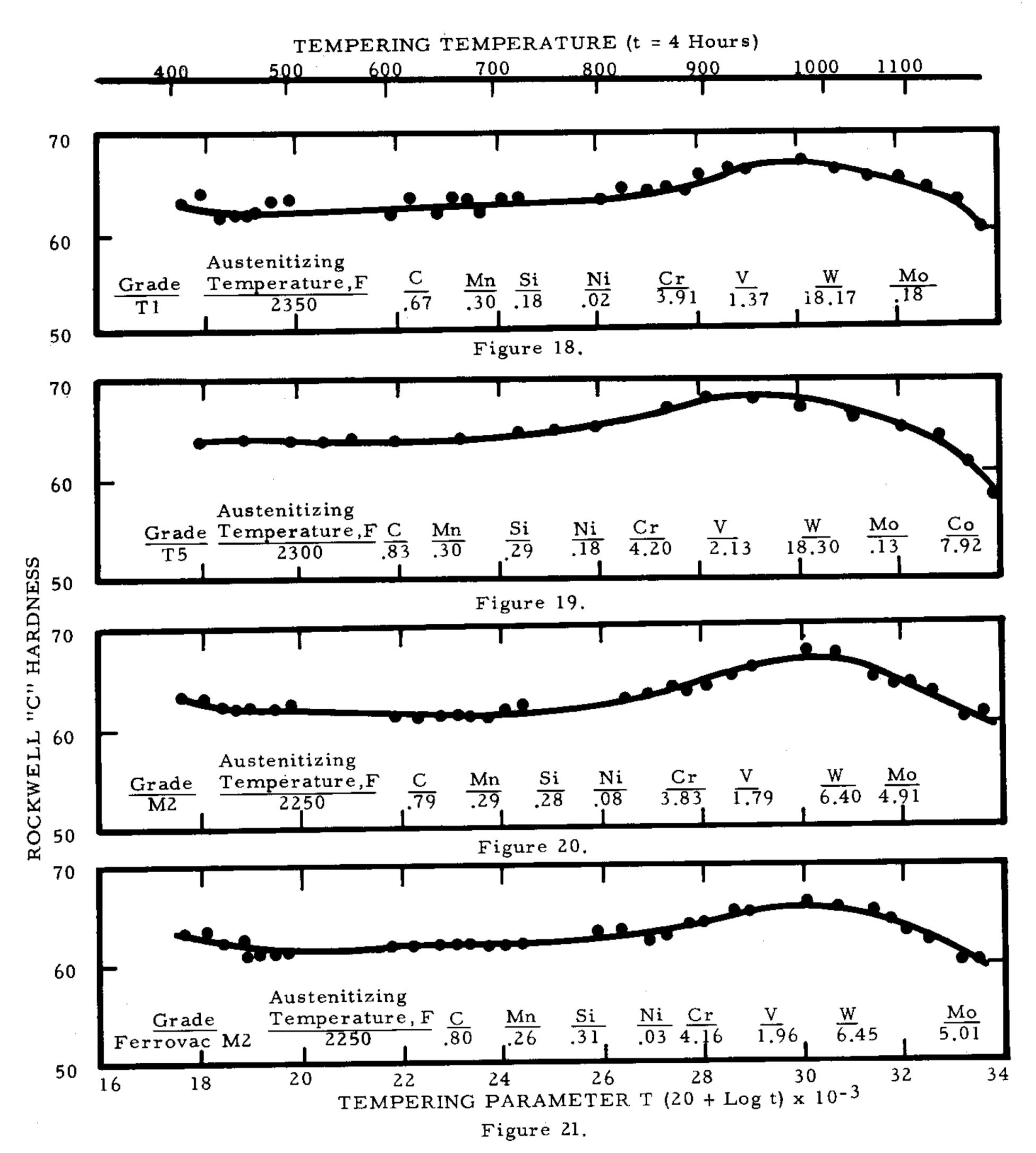


Figures 11 to 14. Master Tempering Curves for Bearing Steels 43 WADC TR 57-343





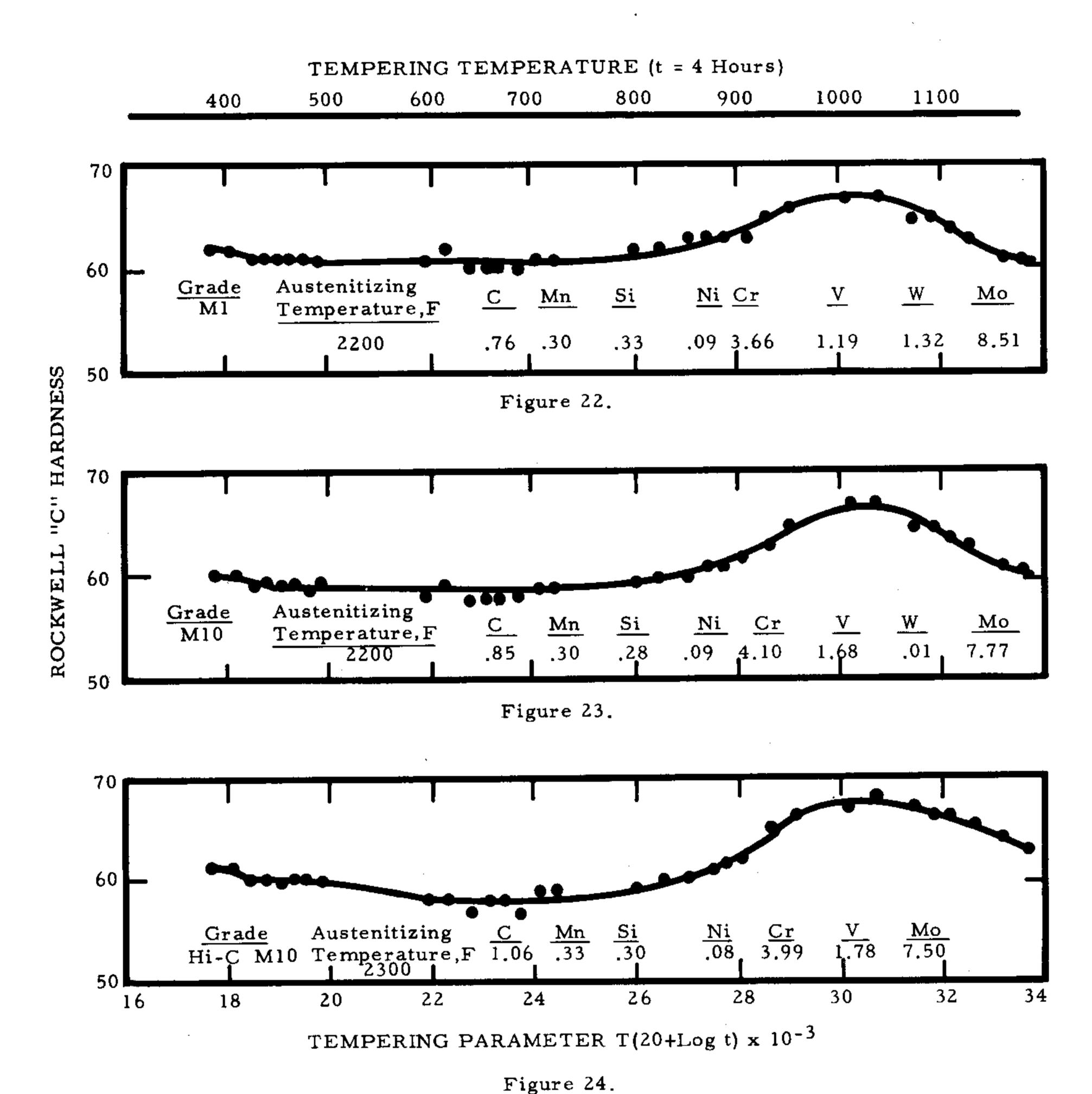
Figures 15 to 17. Master Tempering Curves for Bearing Steels



Figures 18 to 21. Master Tempering Curves for Bearing Steels

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Figures 22 to 24. Master Tempering Curves for Bearing Steels
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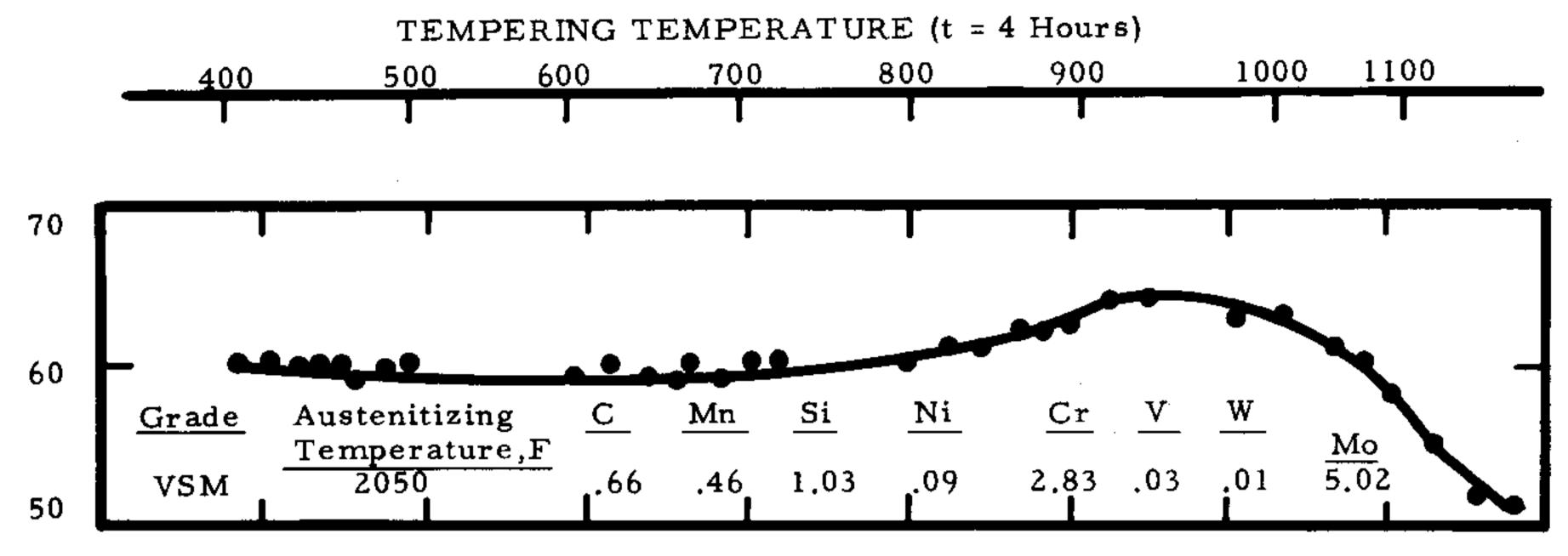
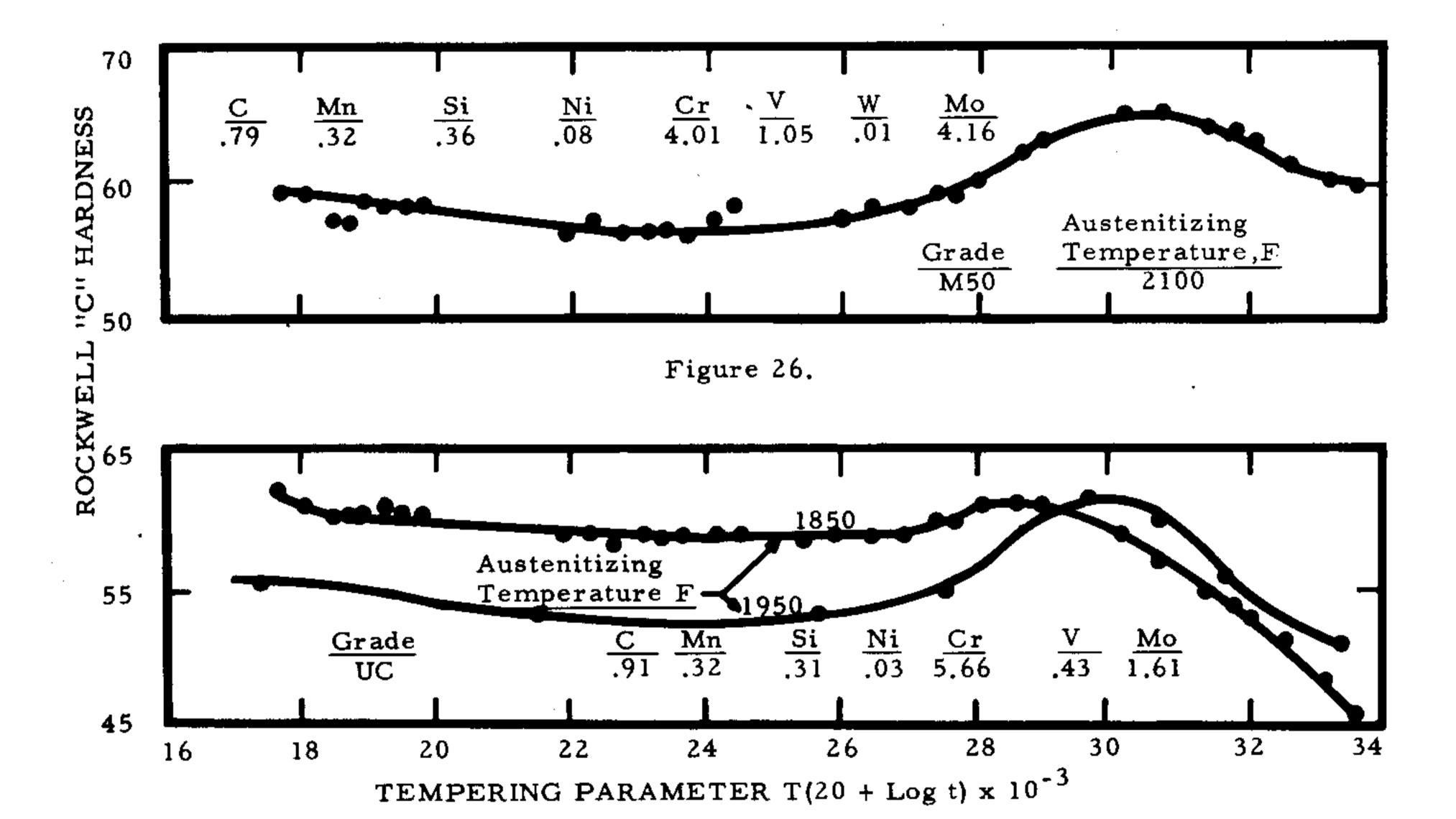
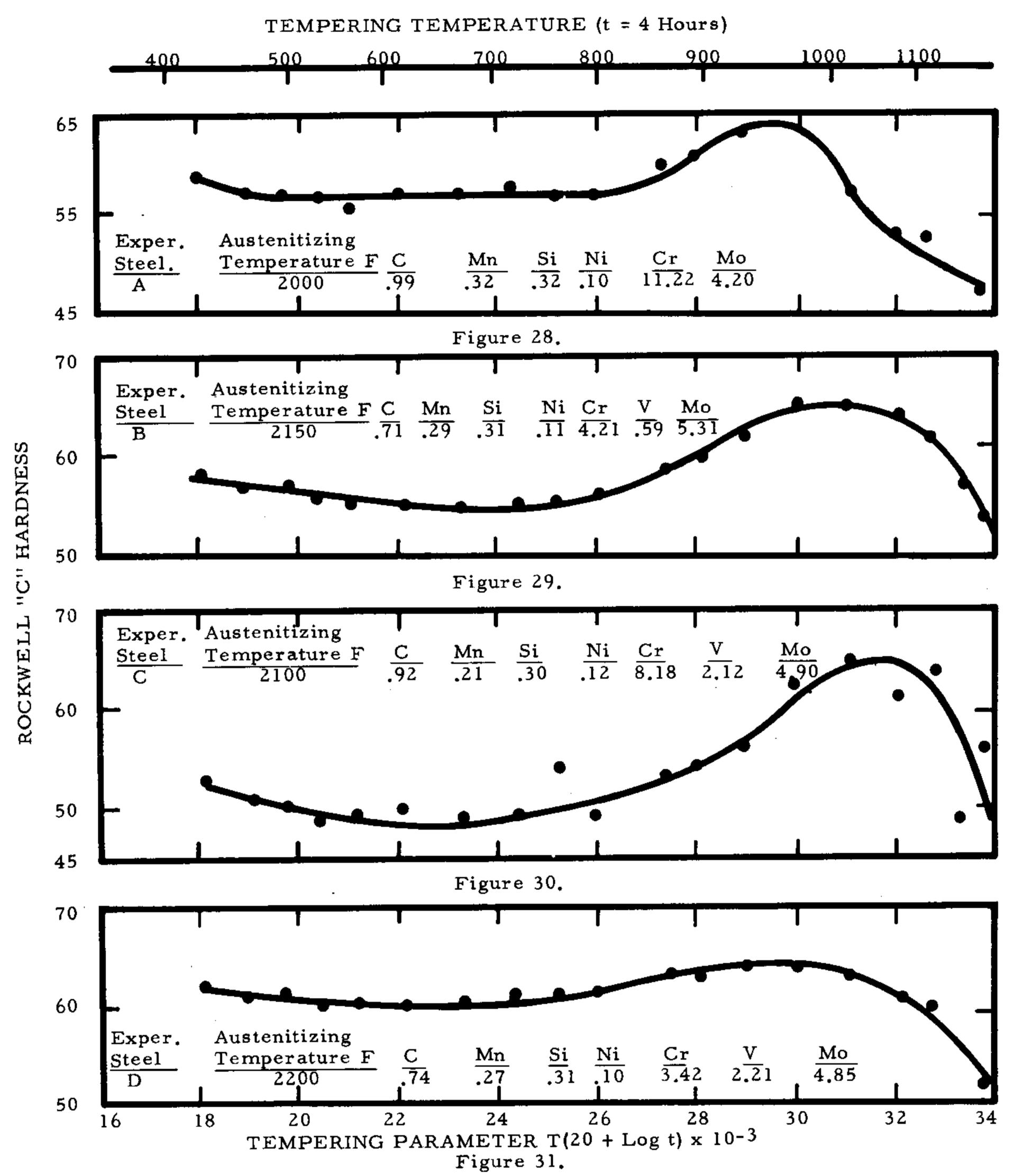


Figure 25.



Figures 25 to 27. Master Tempering Curves for Bearing Steels

Figure 27.



Figures 28 to 31. Master Tempering Curves for Bearing Steels

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### TEMPERING TEMPERATURE (t = 4 Hours)

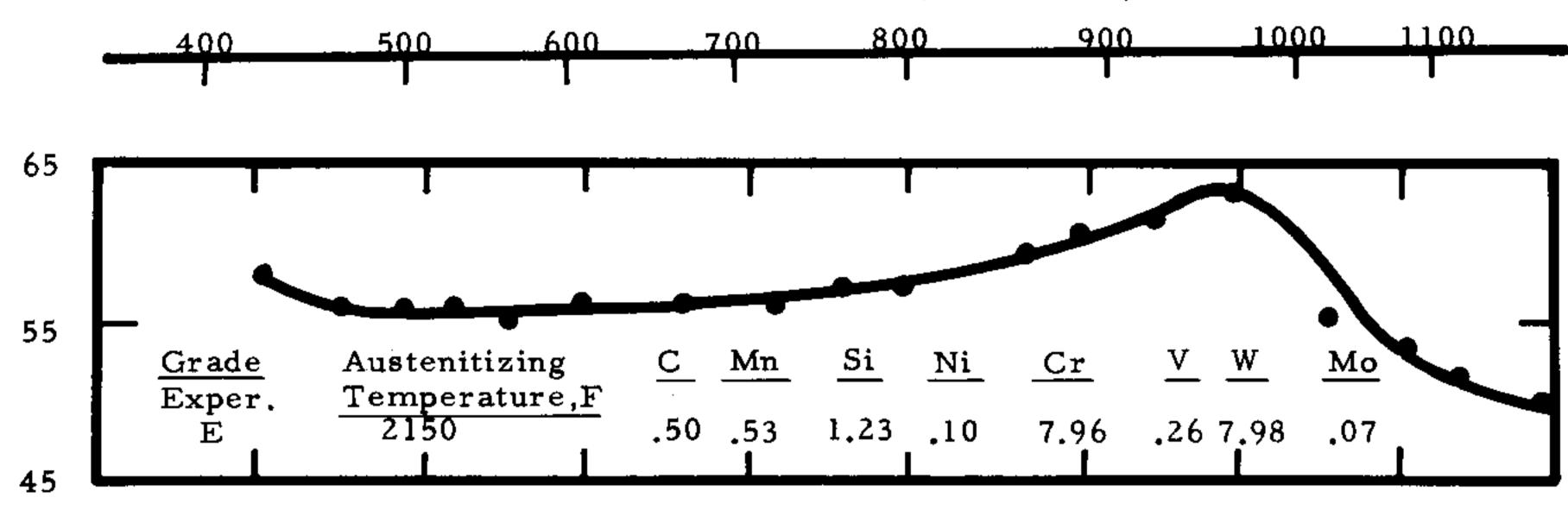
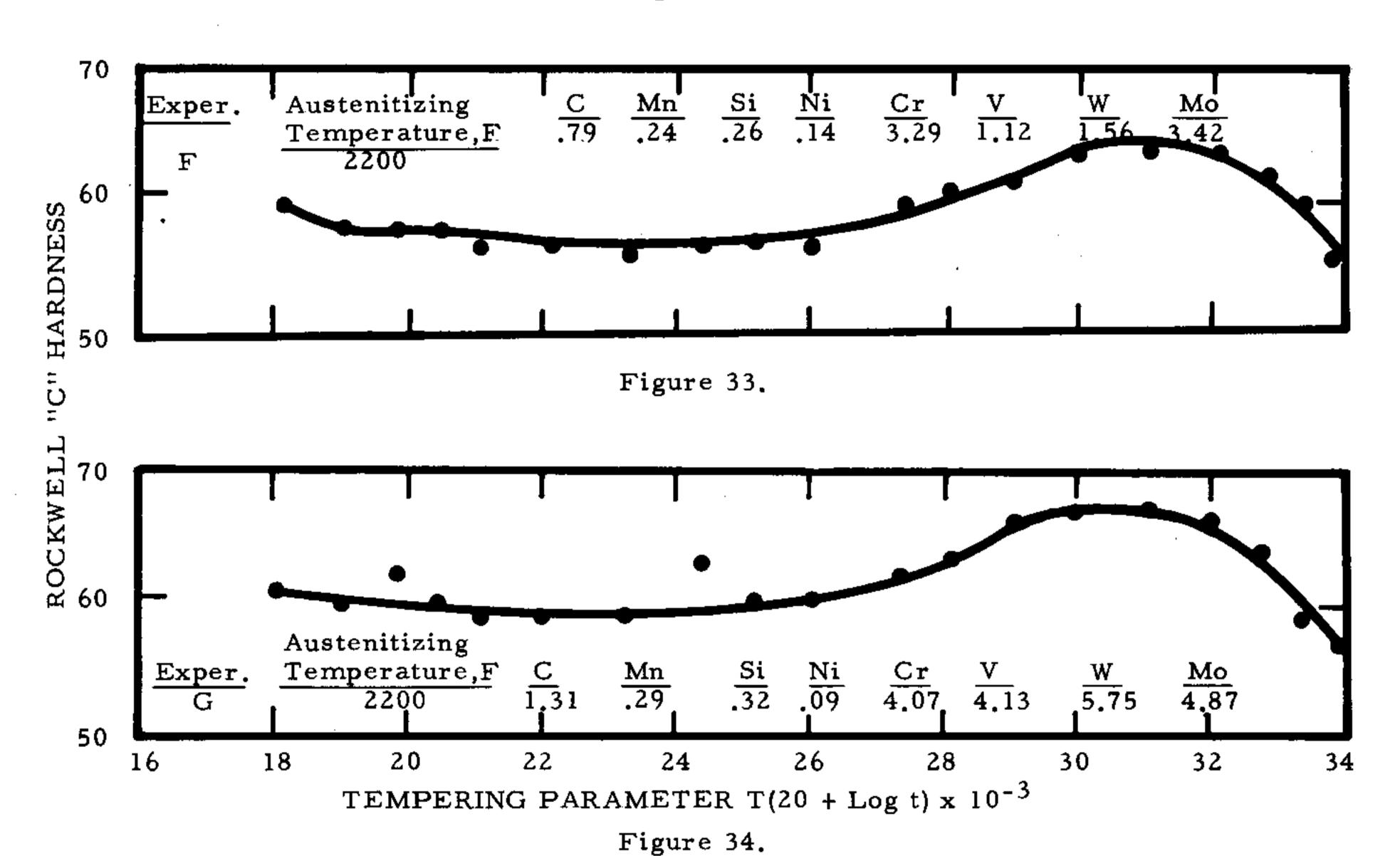
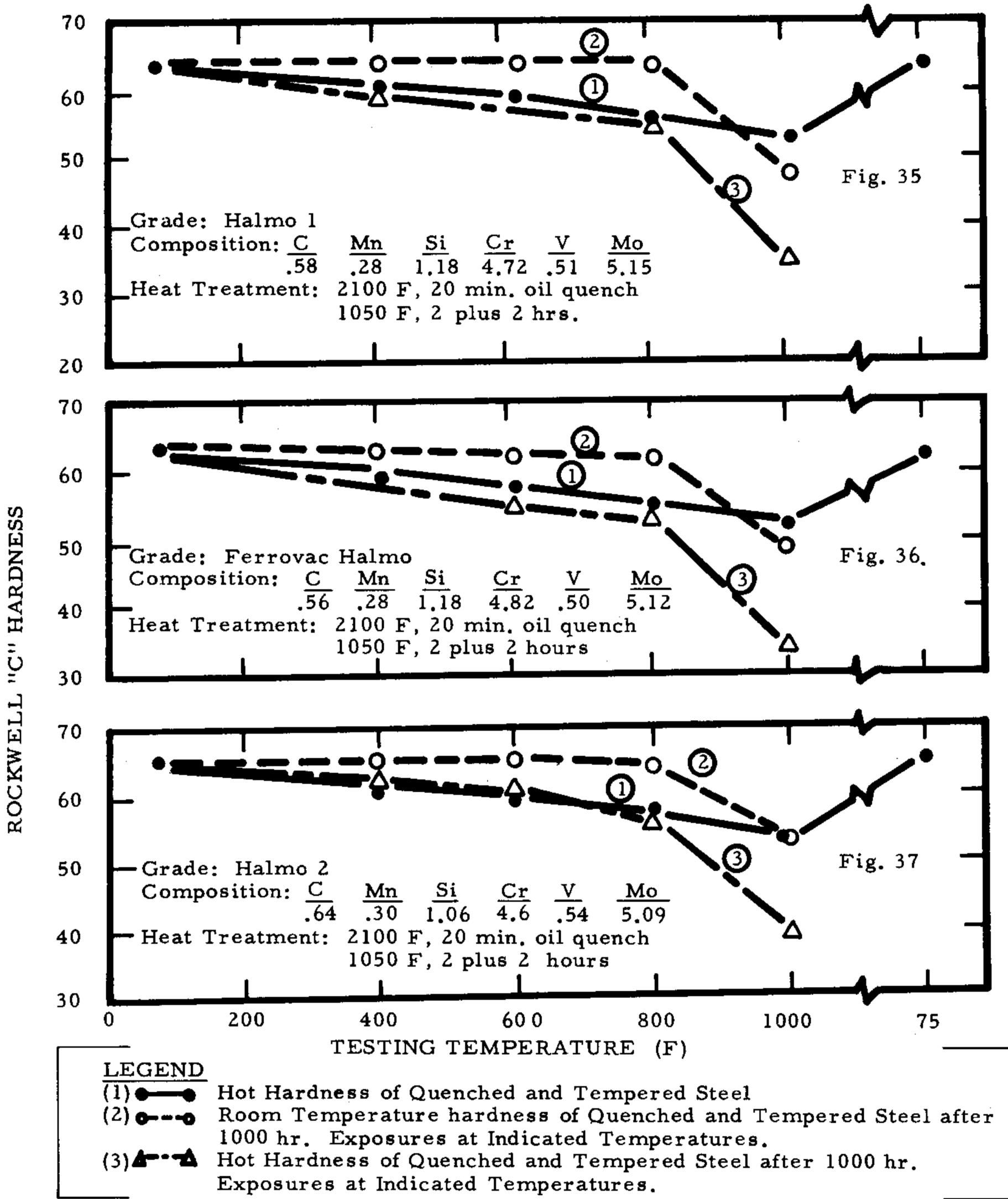


Figure 32.

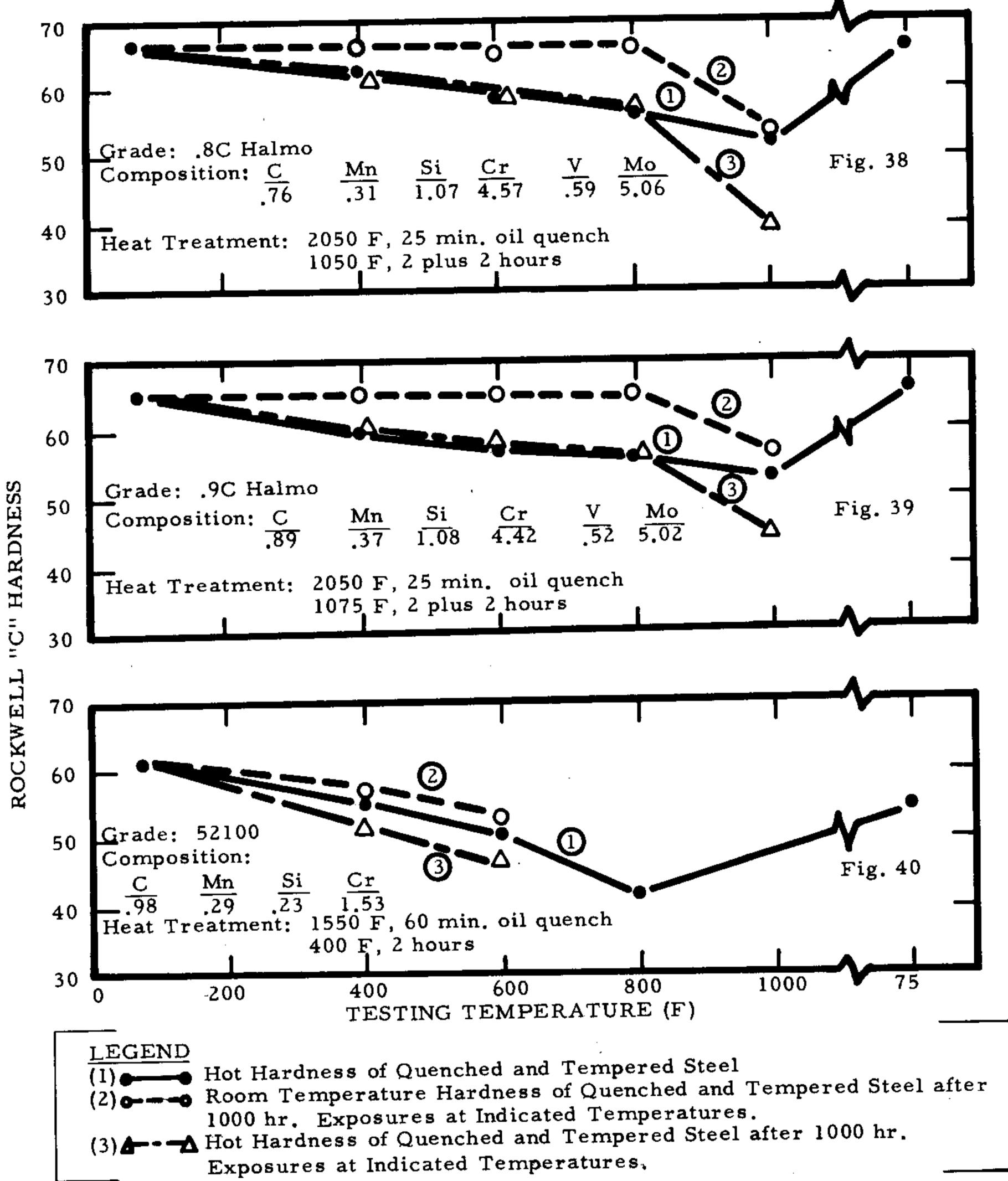


Figures 32 to 34. Master Tempering Curves for Bearing Steels.



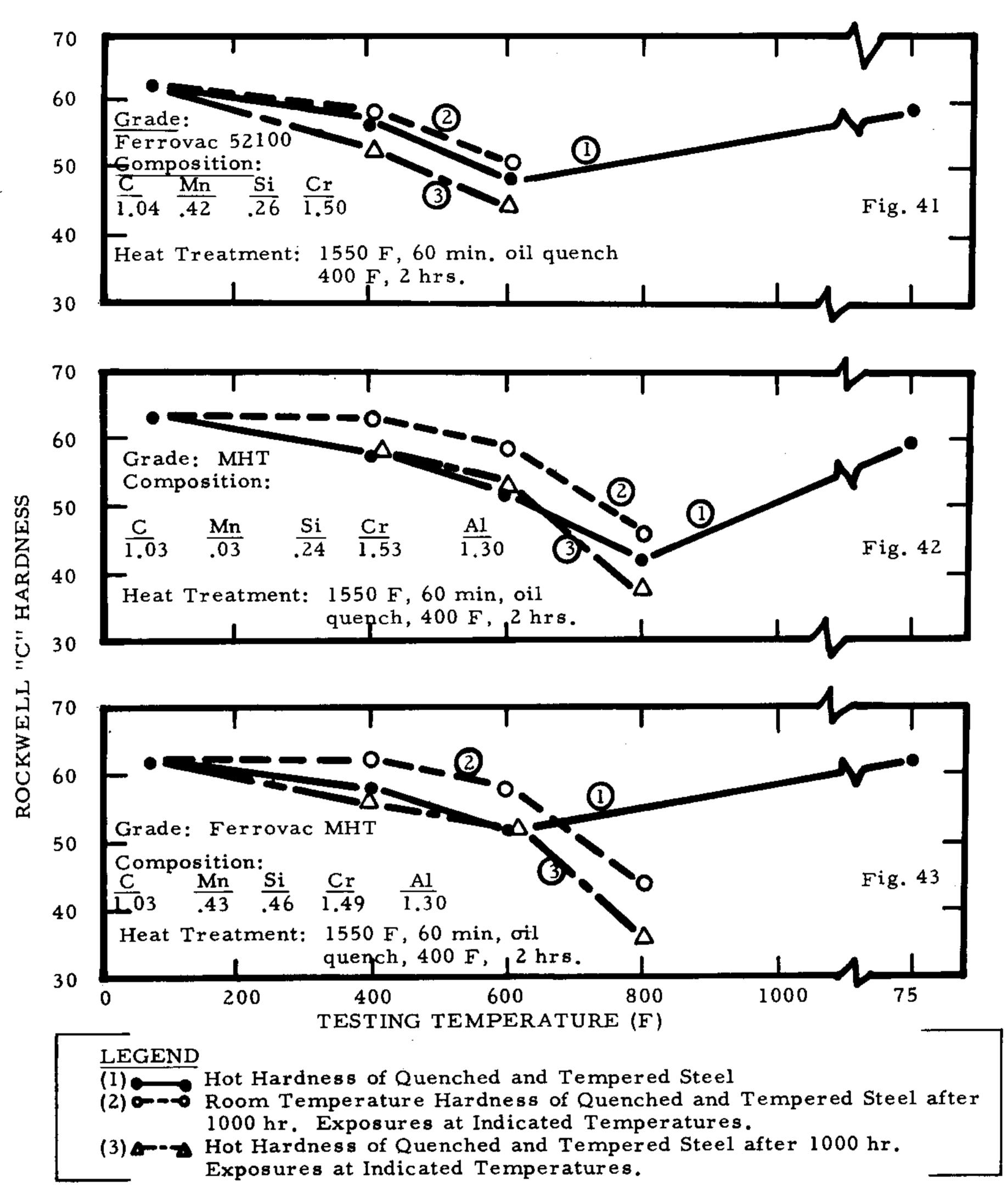


Figures 35 to 37. Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).

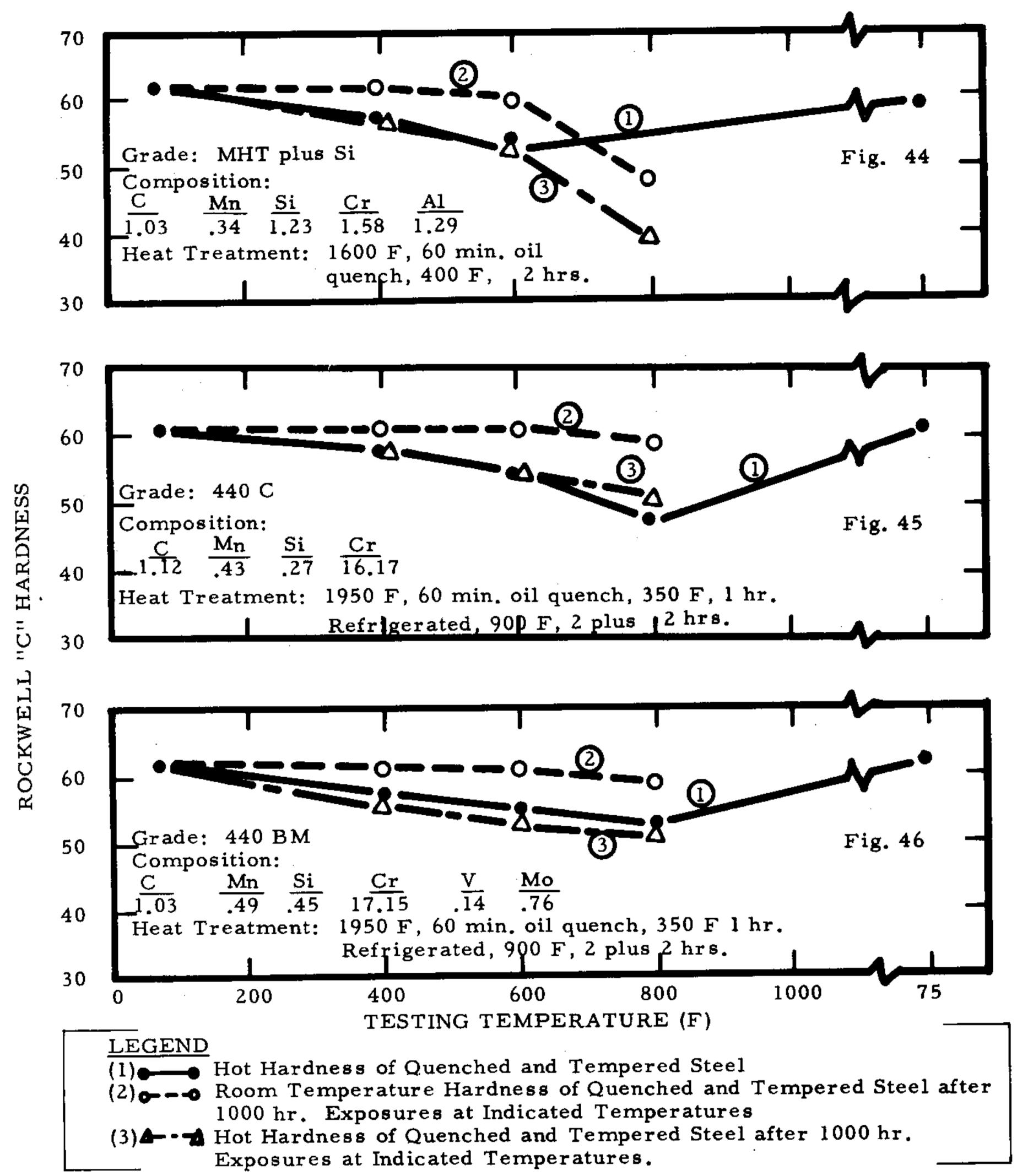


Figures 38 to 40. Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).

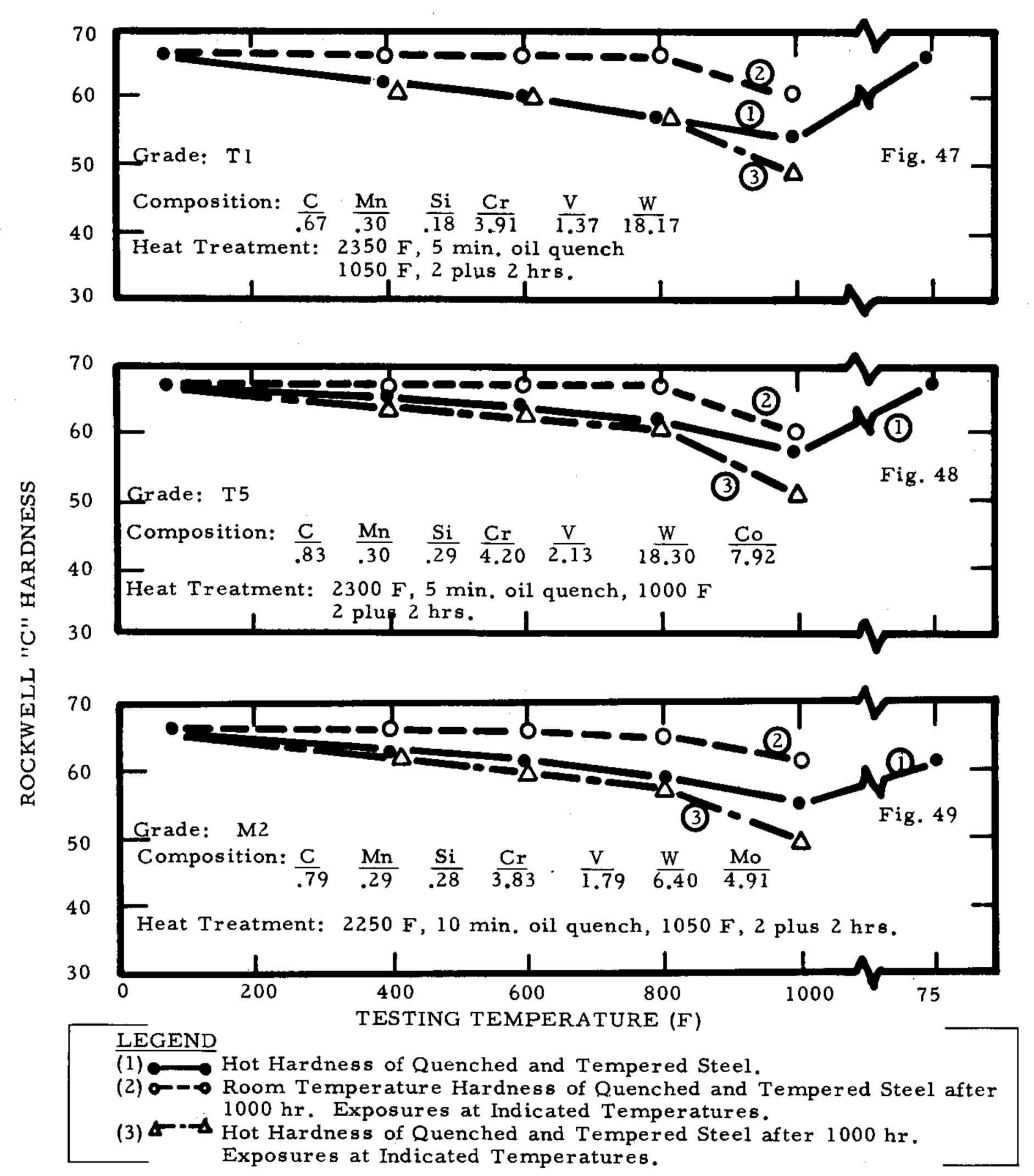
51



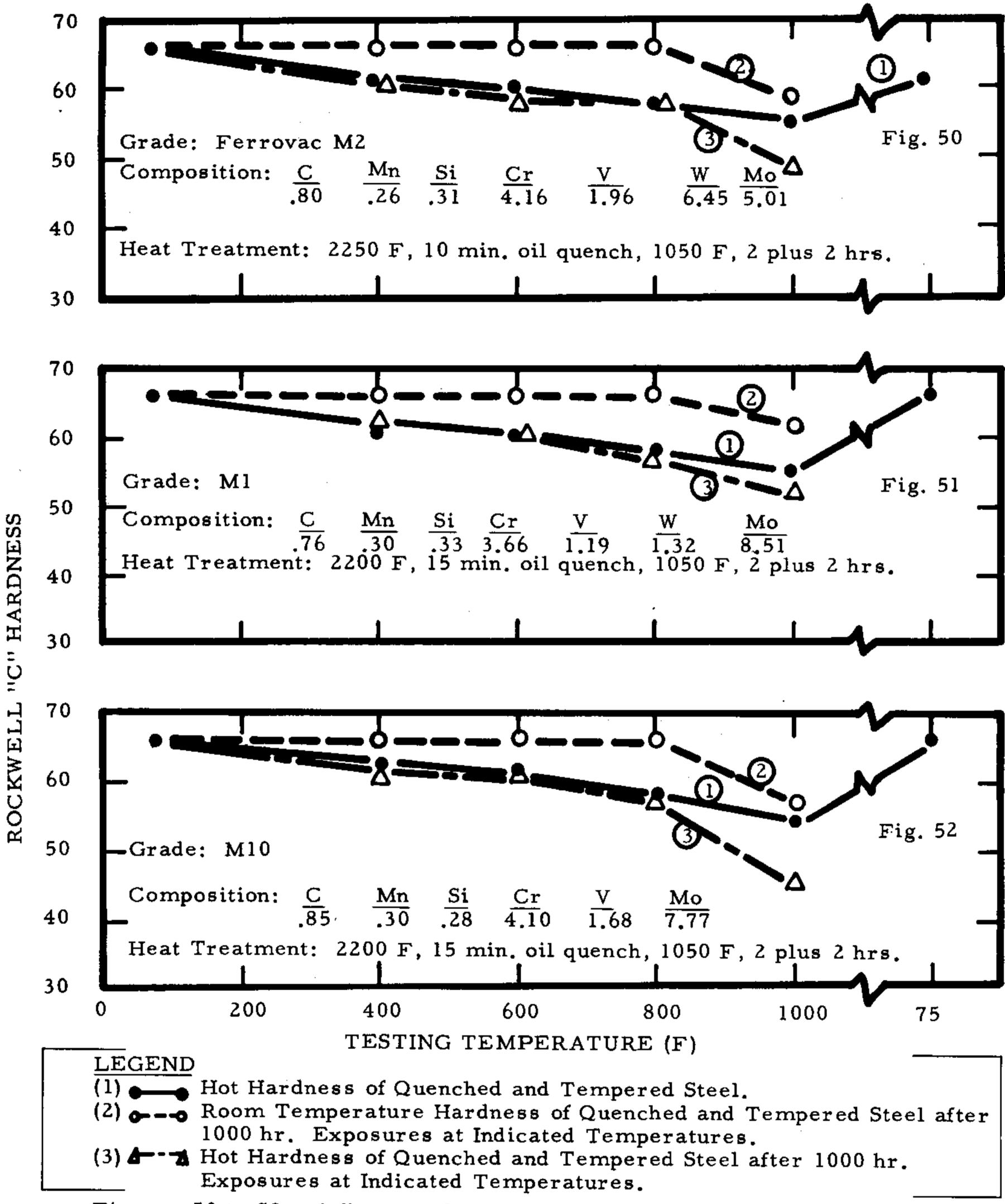
Figures 41 to 43. Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).



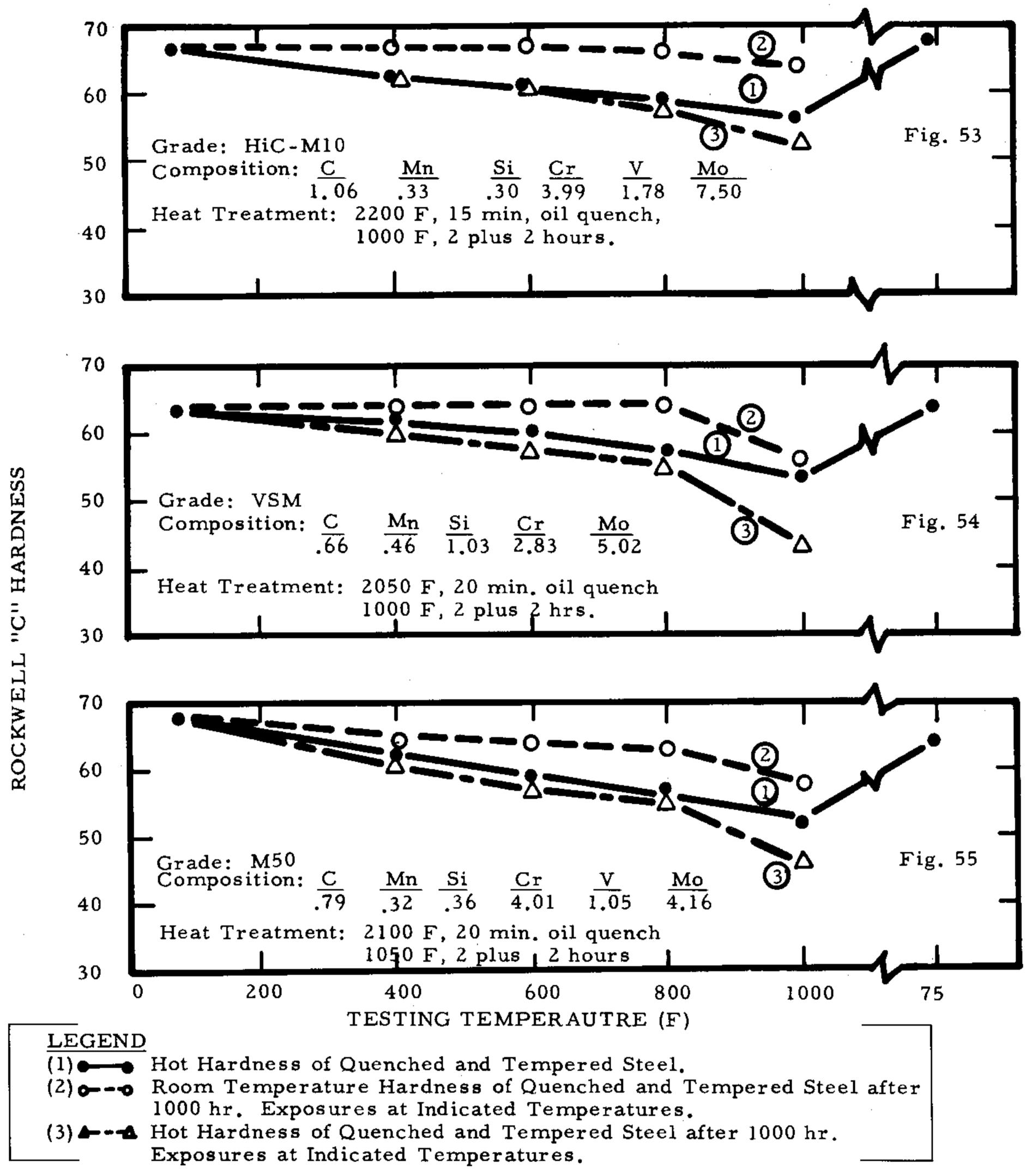
Figures 44 to 46. Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).



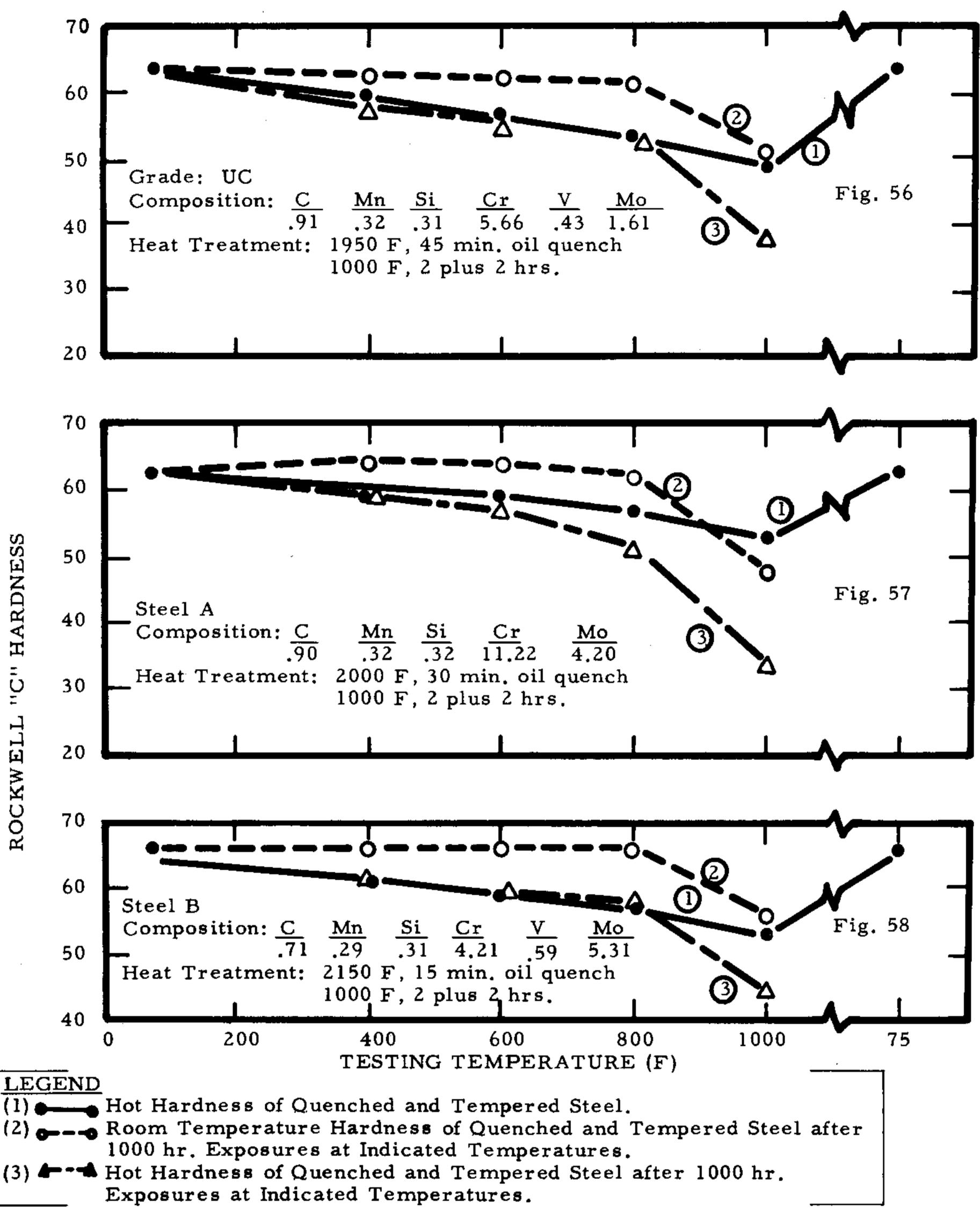
Figures 47 to 49. Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).



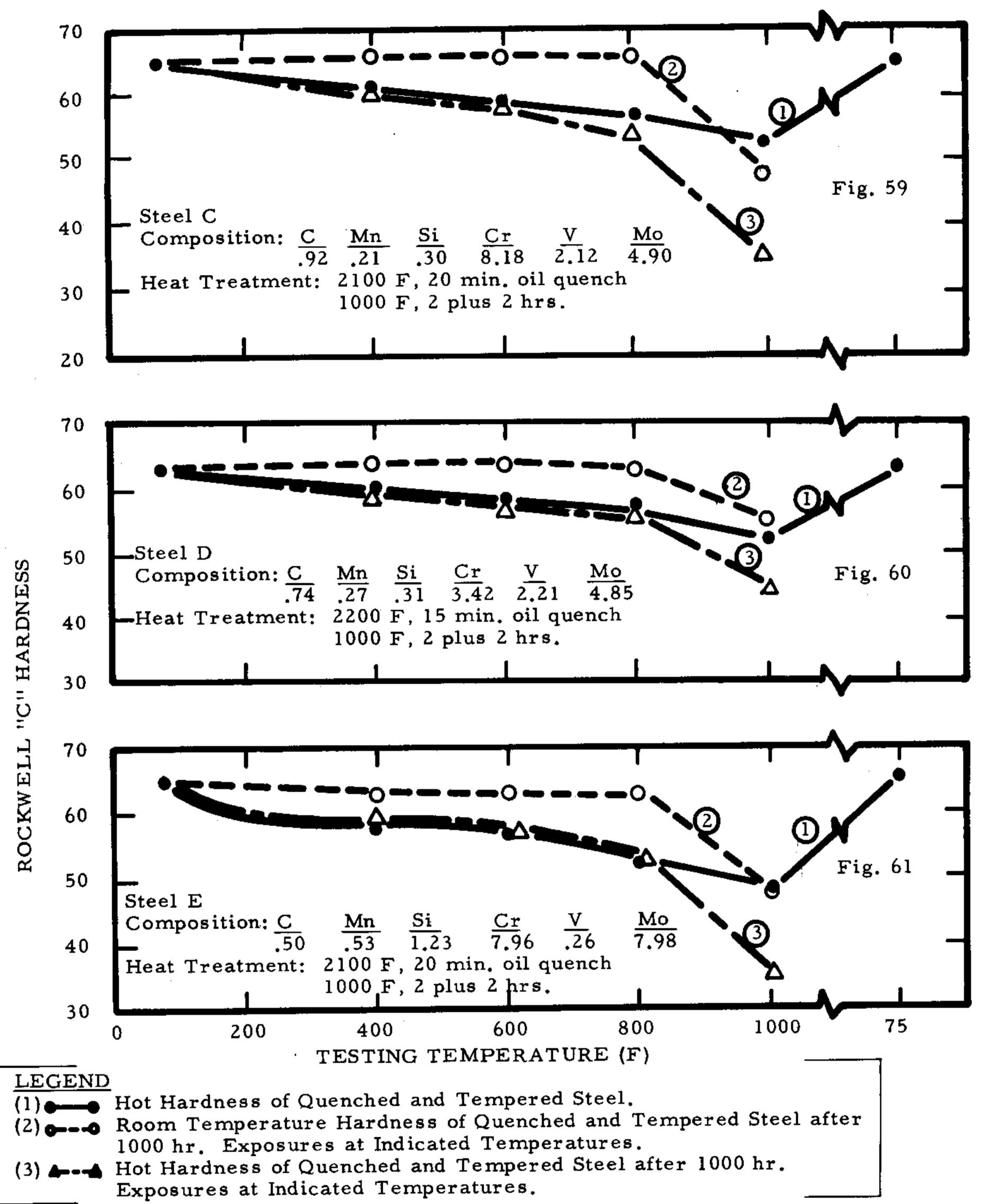
Figures 50 to 52. Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).



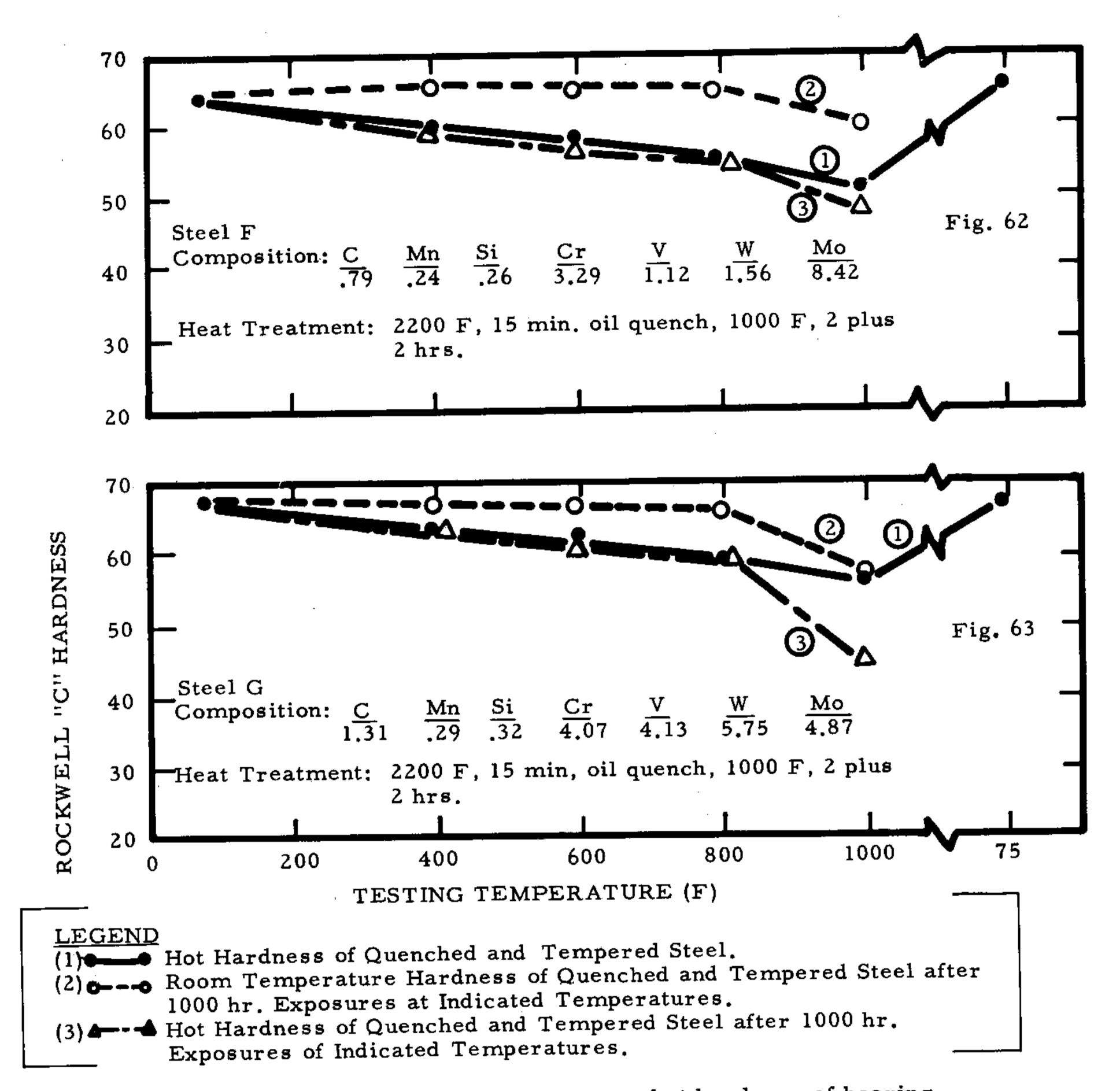
Figures 53 to 55. Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).



Figures 56 to 58: Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).



Figures 59 to 61. Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).



Figures 62 and 63. Influence of temperature on hot hardness of bearing steels (Curve 1). Also shown is the effect of 1000 hour exposure at different temperatures on the room temperature hardness (Curve 2) and hot hardness of the steels (Curve 3).

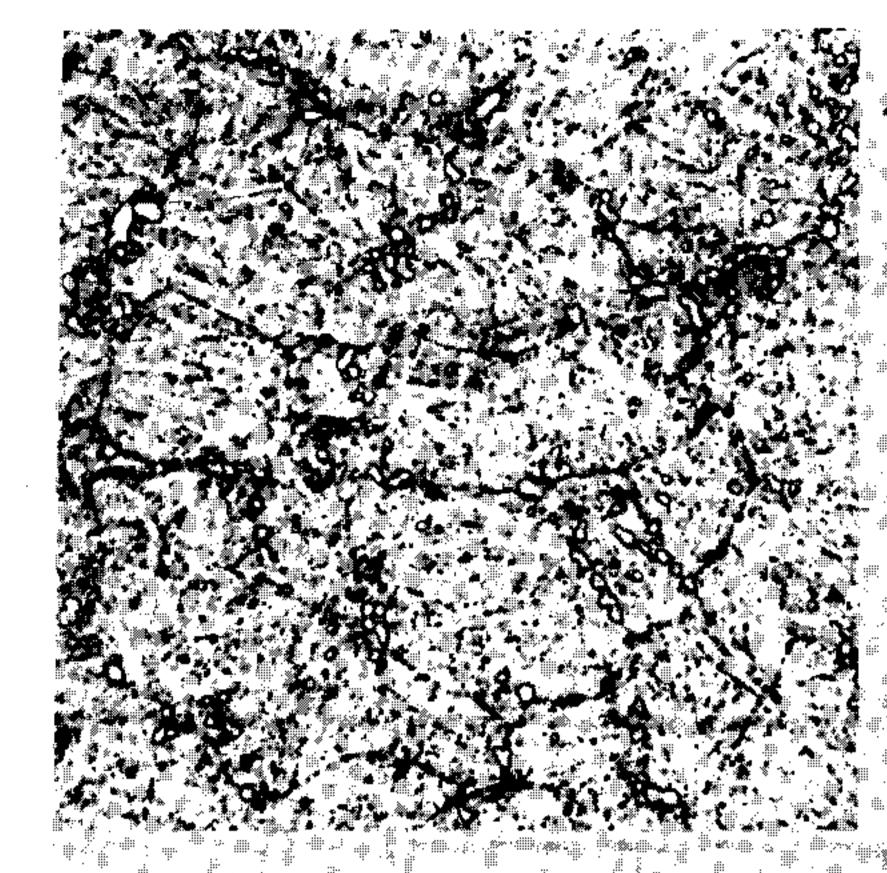


Fig. 64 Halmo-1, austenitized Fig. 6
2100 F. 20 min. oil
quench tempered 1050 F.
2 plus 2 hours.

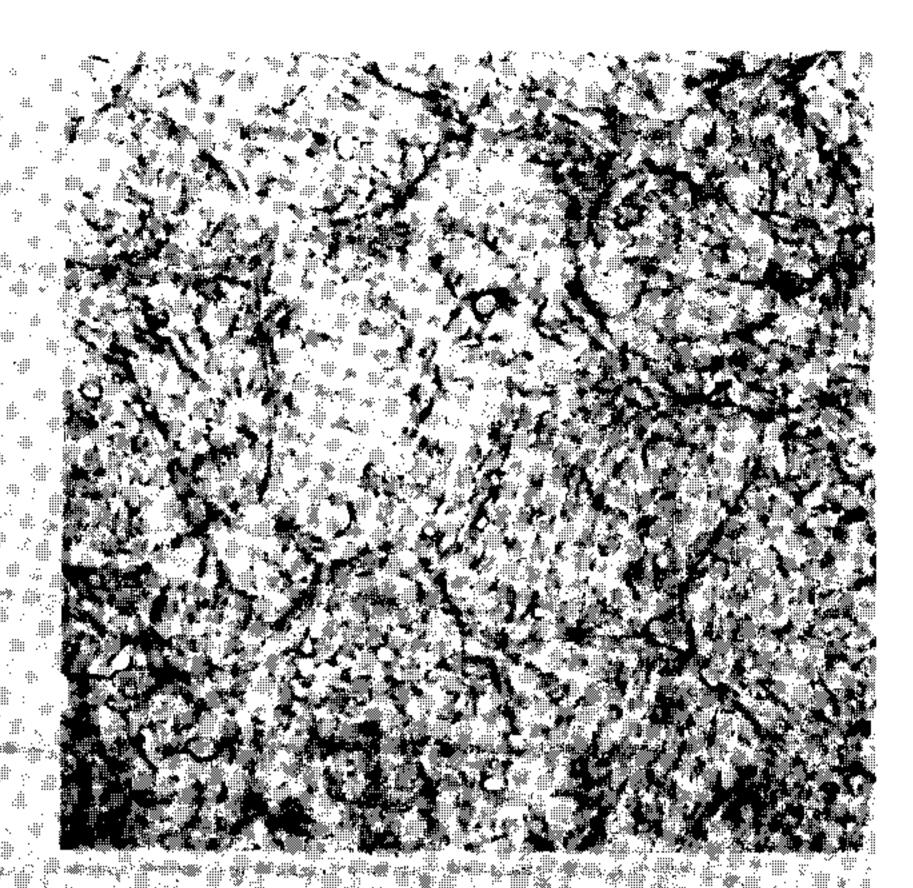


Fig. 65 Reviews: Halmo austenthred 2100 F. 20 min. oil quench, tempered 1050 F. 2 plus 2 hours.

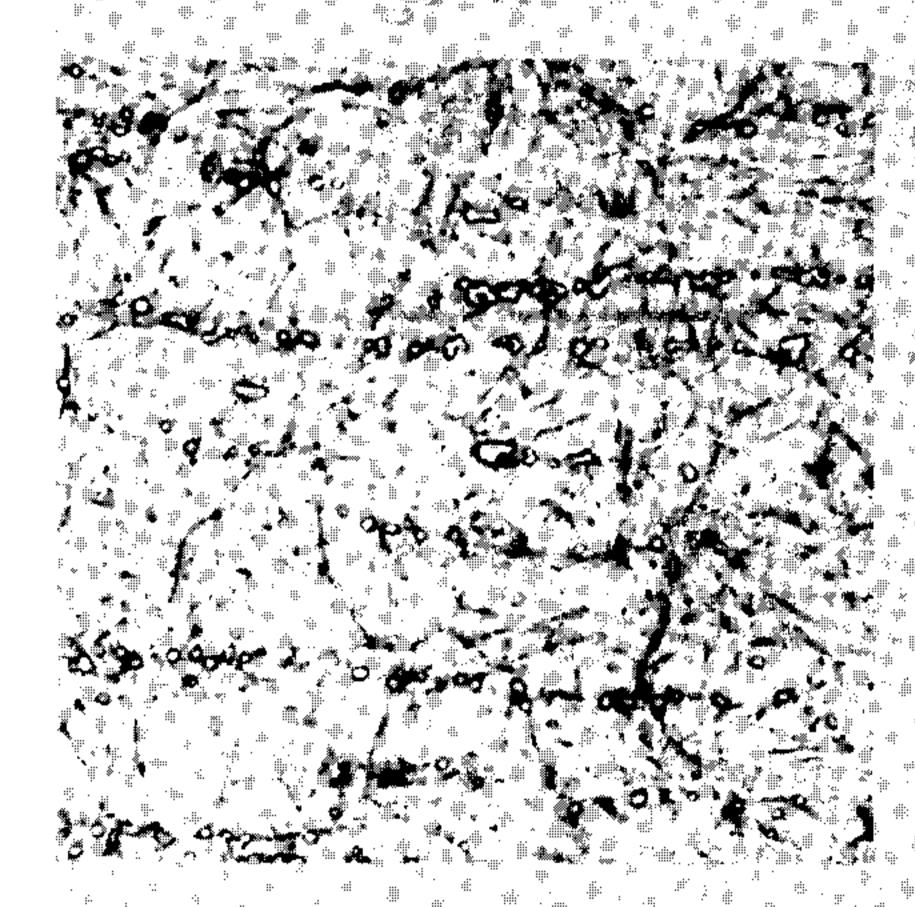


Fig. 66. Halmo-2, austenitized

2100 F, 20 min. oil

quench, tempered 1050 F,

2 plus 2 hours.

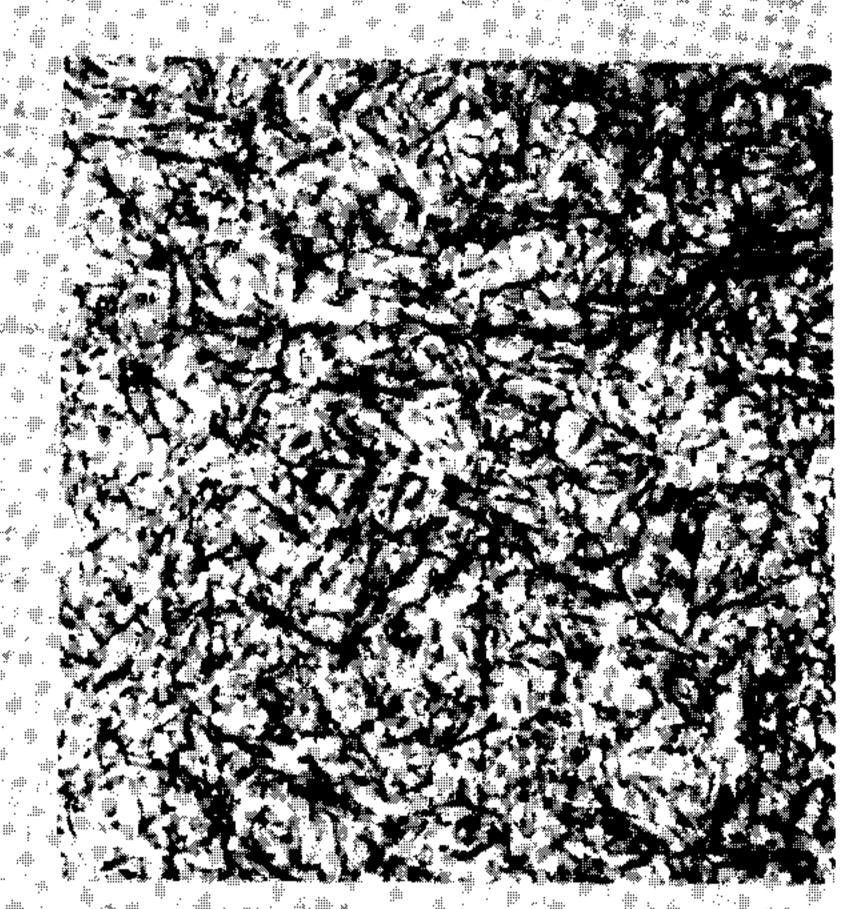


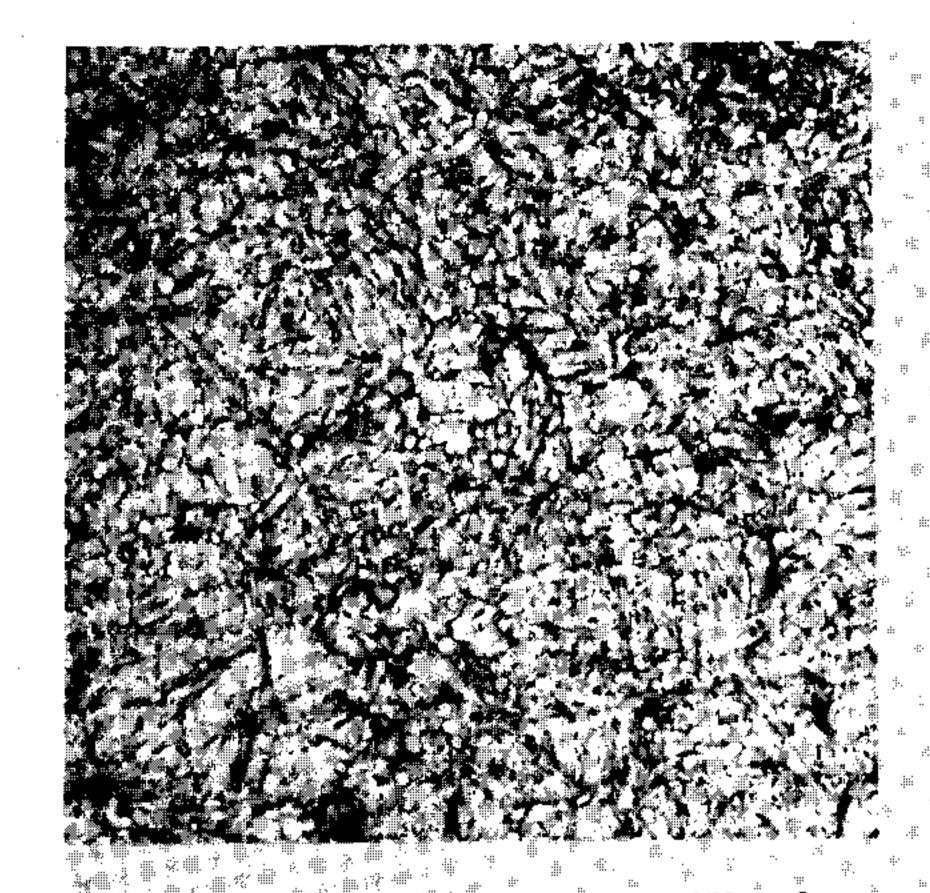
Fig. 67 .8C Halmo, austenitized

2050 F, 25 min. oil
quench, tempered 1050 F

2 plus 2 hours.

FIGURES 64 to 67 MICROSTRUCTURES SHOWING CARBIDE SIZE AND DISTRIBUTION IN BEARING STEELS.

Picral + 0.2% HCl Etch



g. 68 .9C Halmo, austenitized 2050 F, 25 min. oil guench, tempered 1075 F 2 plus 2 hours.

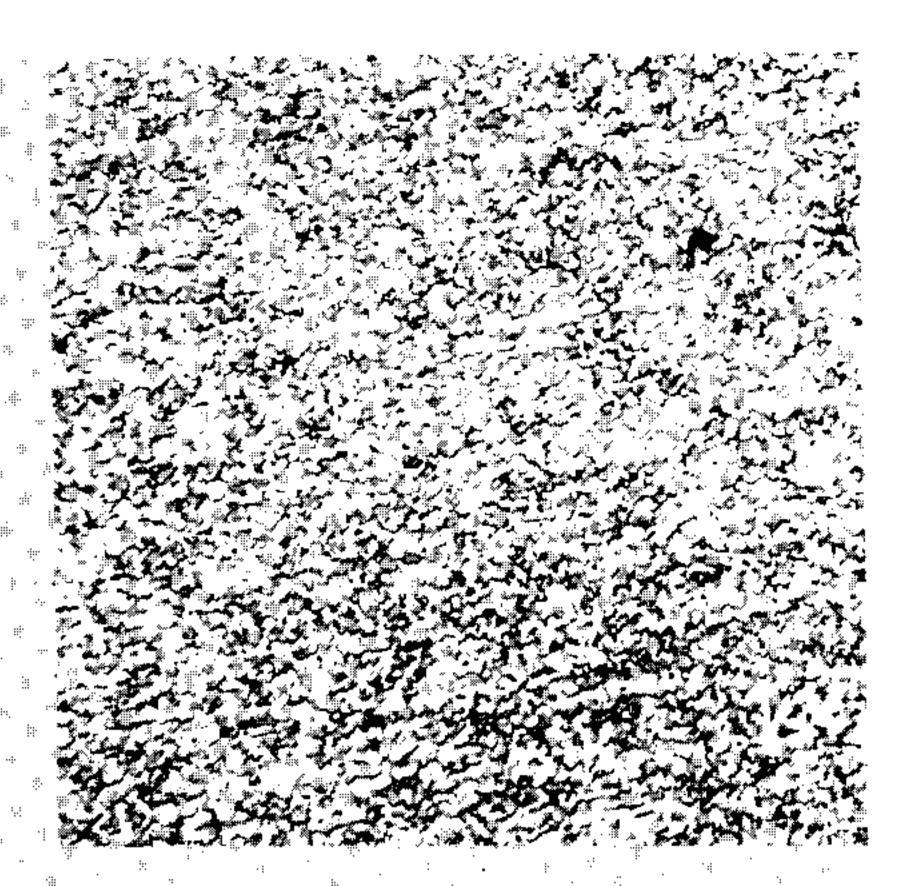
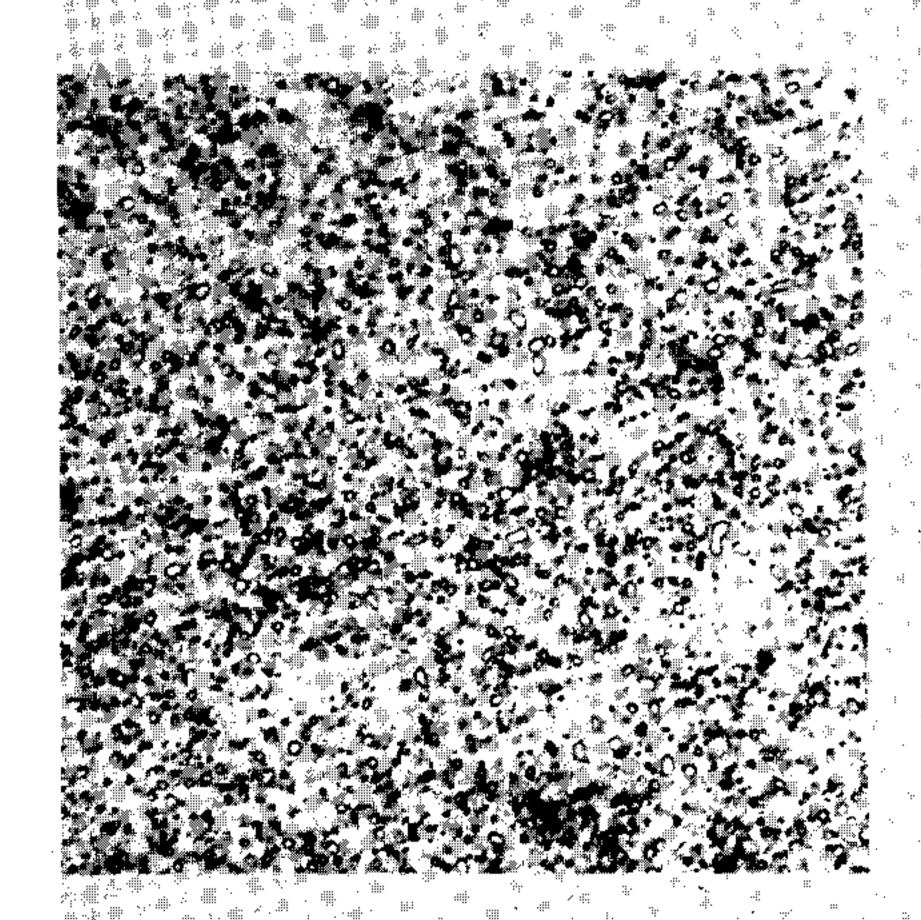
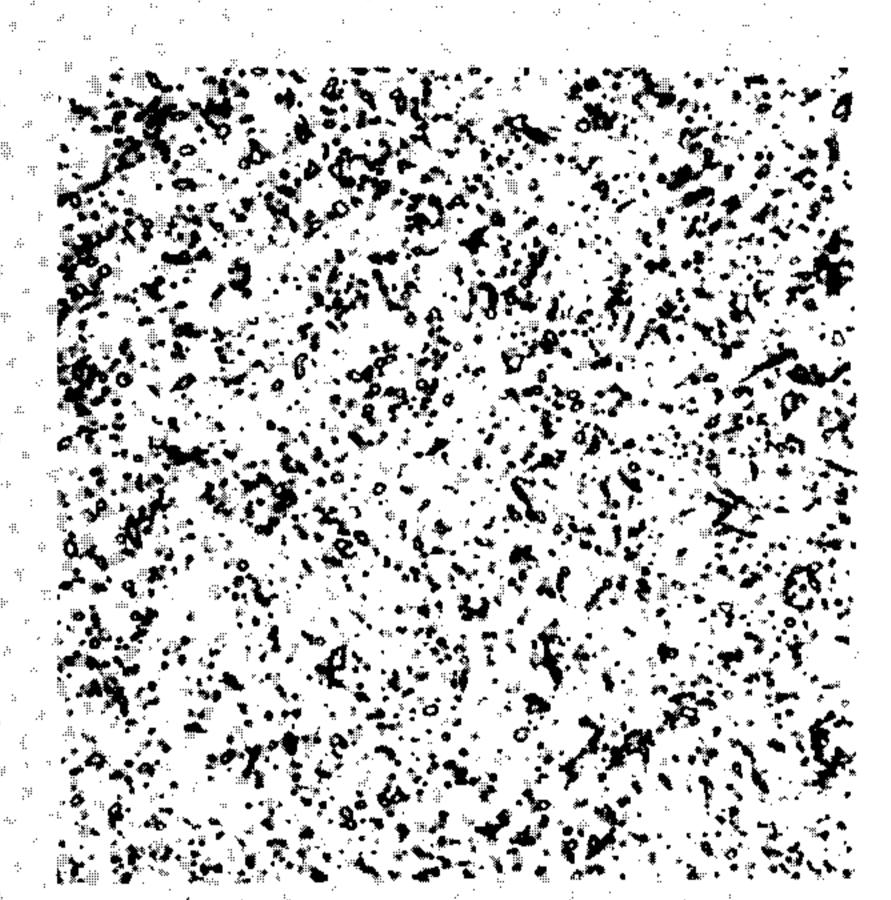


Fig. 69 52100, austenitized 1550 F, 1 hr., oil quench, tempered 400 F 2 hours.



70 Ferrovac 52100, austen-Fig. 71 MHT, austenitized 1550 F, 1 tized, 1550 F, 1 hr., I hour, oil quench, cil quench, tempered tempered, 400 F, 2 hours.



FIGURES 68 to 71 MICROSTRUCTURES SHOWING CARBIDE SIZE AND DISTRIBUTION IN BEARING STEELS.

Picral + 0.1% HCl Etch

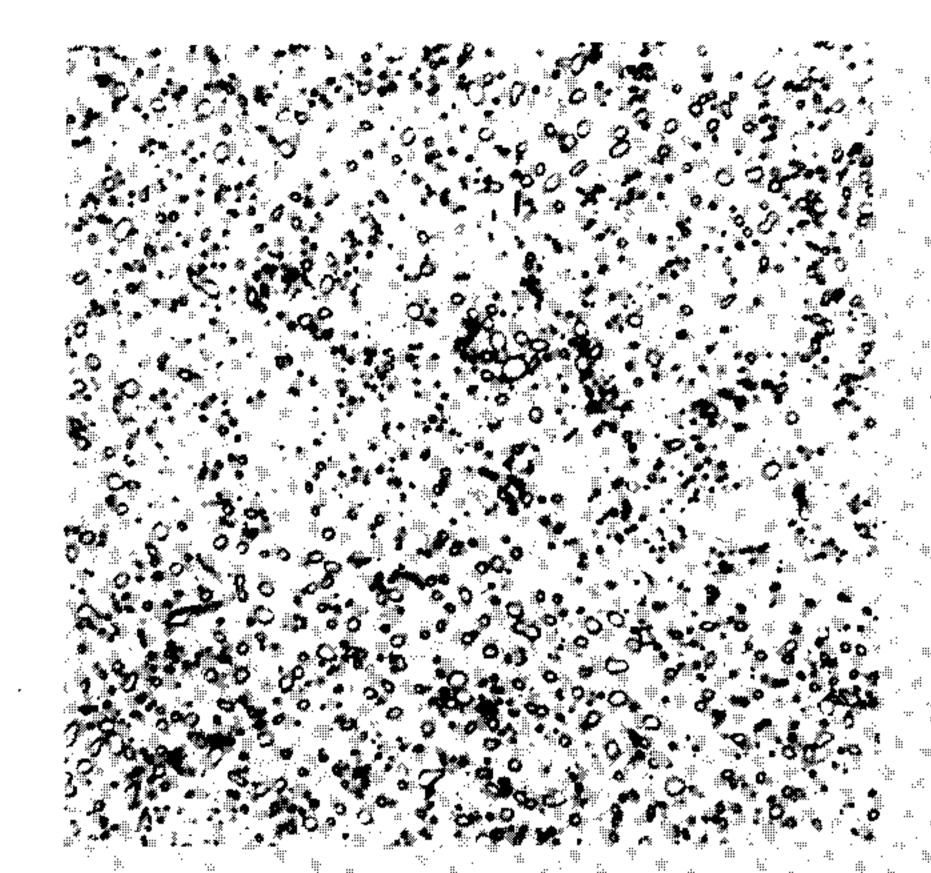
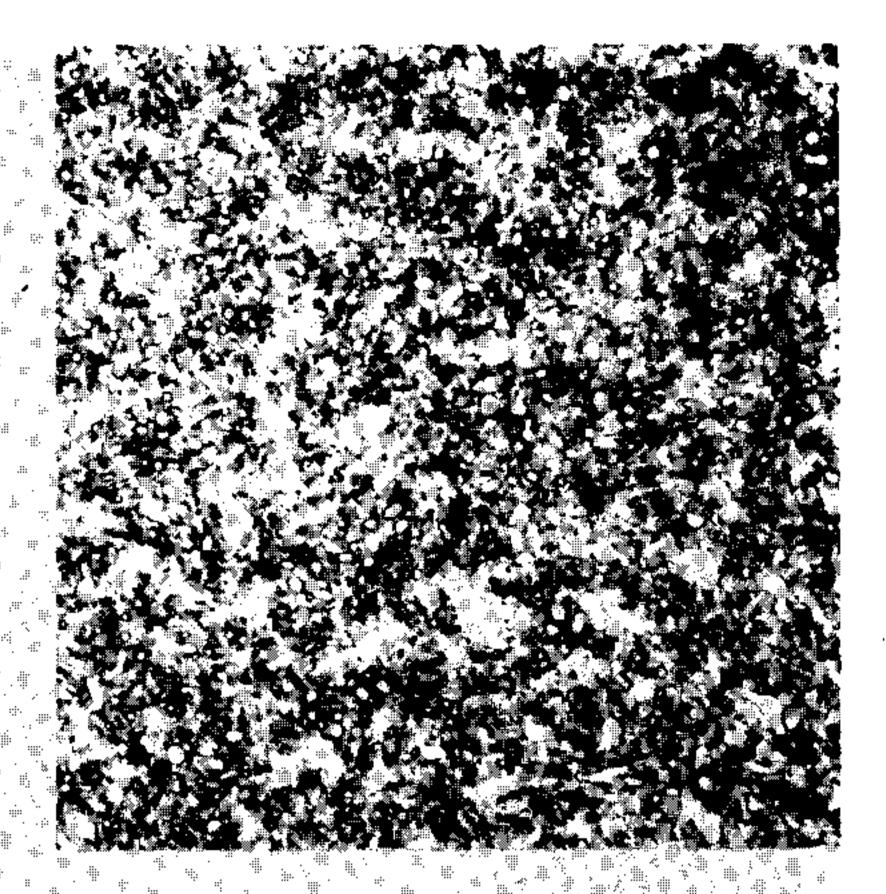


Fig. 72 Ferrovac MHT, austenitized 1550 F, 1 hour,

oil quench, tempered

400 F, 2 hours

Picral + 0.1% HCl Etch Picral + 0.1% HCl Etch



73 MHT plus St, austen-. Itized 1600 F, I hour,

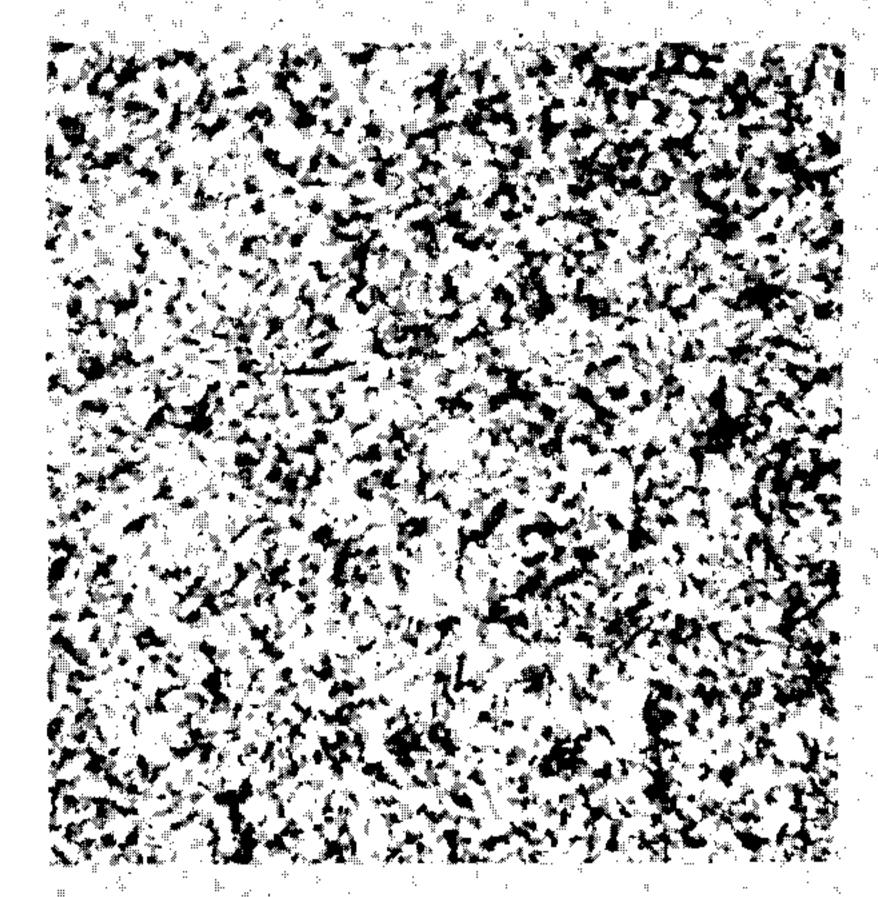


Fig. 74 440 C, austenitized 1950 F, 1 hour, oil quench, heated to 350 F 1 hour, refrigerated and tempered 900 F, 2

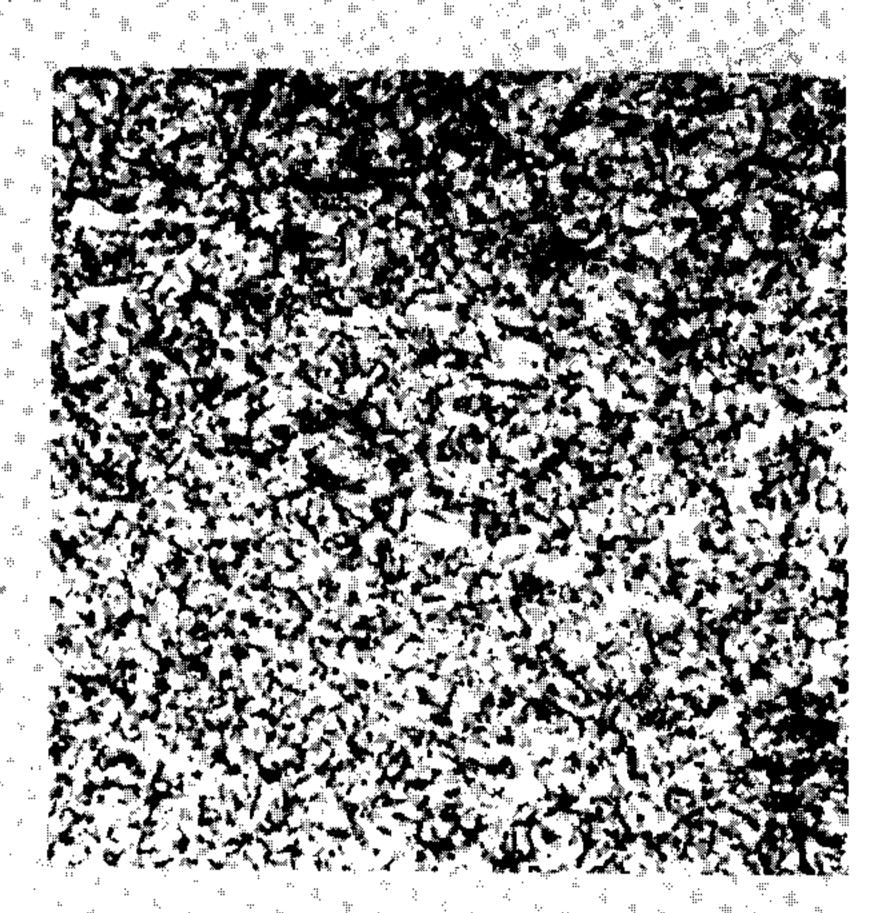


Fig. 75 440 BM, austenitized 1950 F, I hour, oil quench, heated to 350 F I hour, refrigerated and tempered 900 F, 2 plus 2 hours. plus 2 hours.

FIGURES 72 to 75 MICROSTRUCTURES SHOWING CARBIDE SIZE AND DISTRIBUTION IN BEARING STEELS

Magnification X 750 Alcoholic 10% HCl Etch (440C and 440BM) 62 WADC TR 57-343

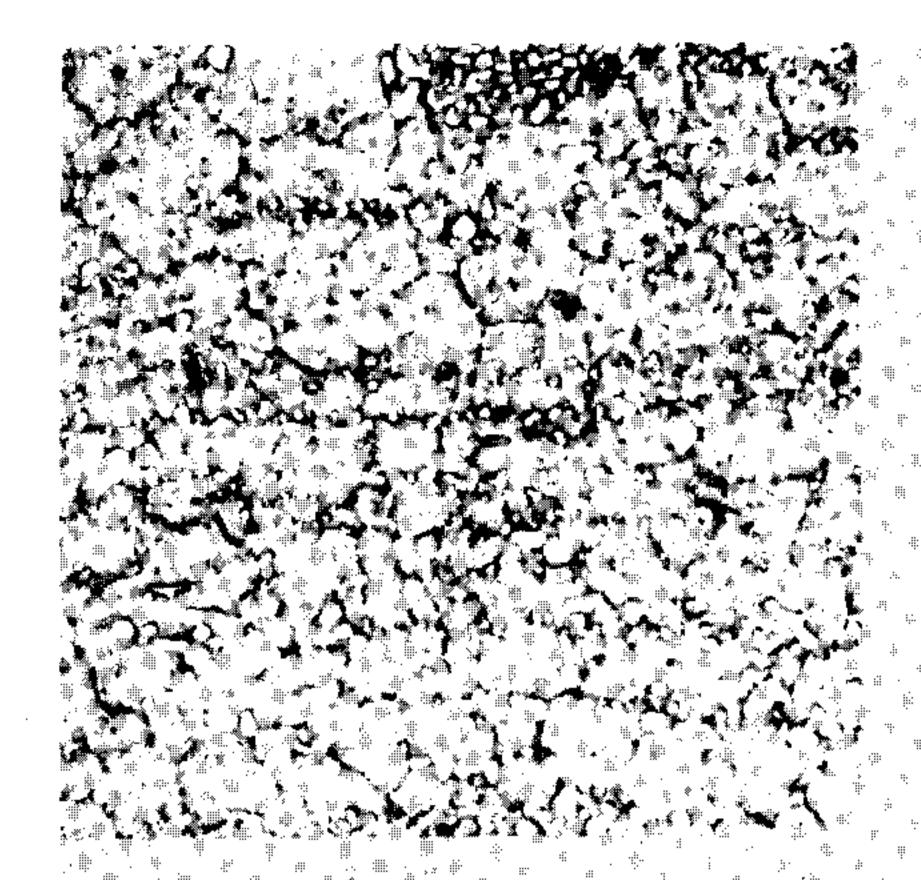


Fig. 76 Ti, austenitized
2350 F, 5 min. oil
quench, tempered 1050 F
2 plus 2 hours

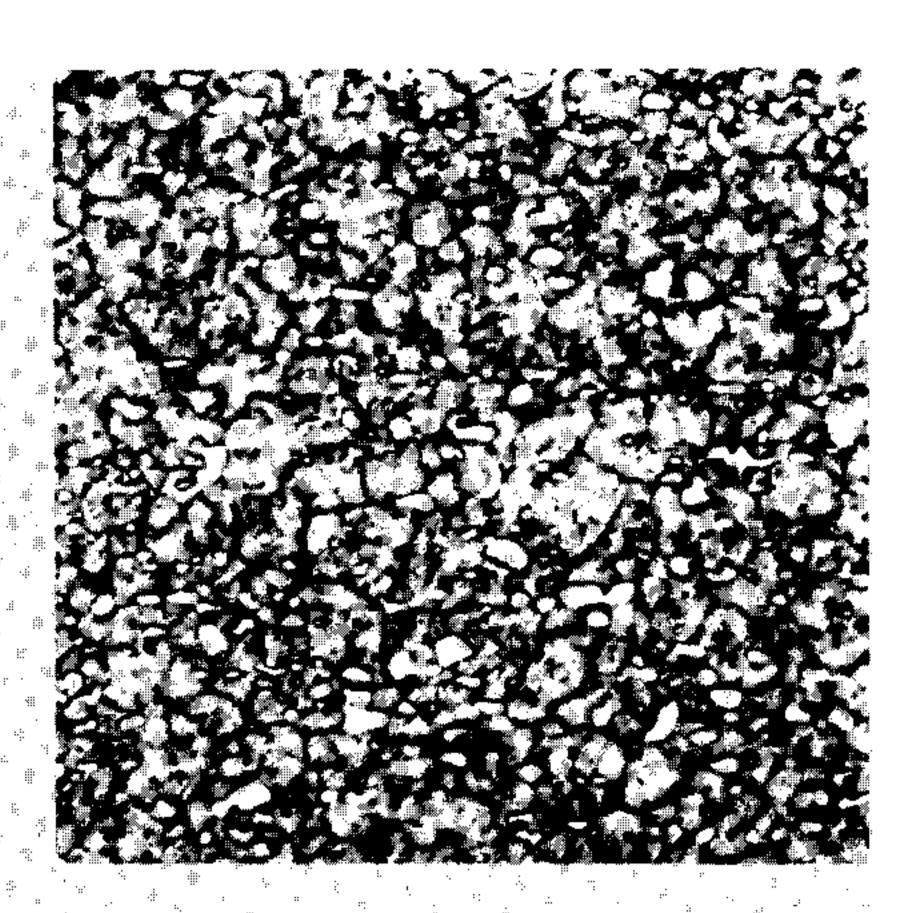


Fig. 77 T5, austenitized
2300 F, 5 min. oil
quench, tempered 1000 F
2 plus 2 hours.

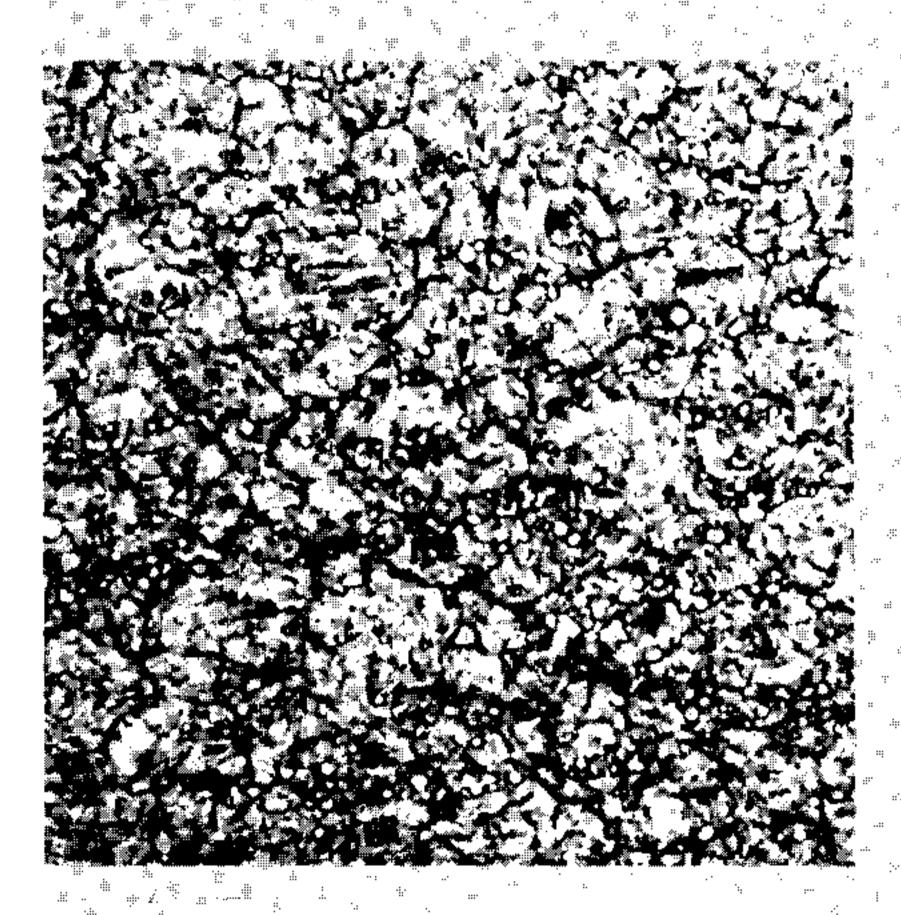


Fig. 78 M2, austenitized

2250 F, 10 min. oil
quench, tempered

1050 F, 2 plus 2 hours.

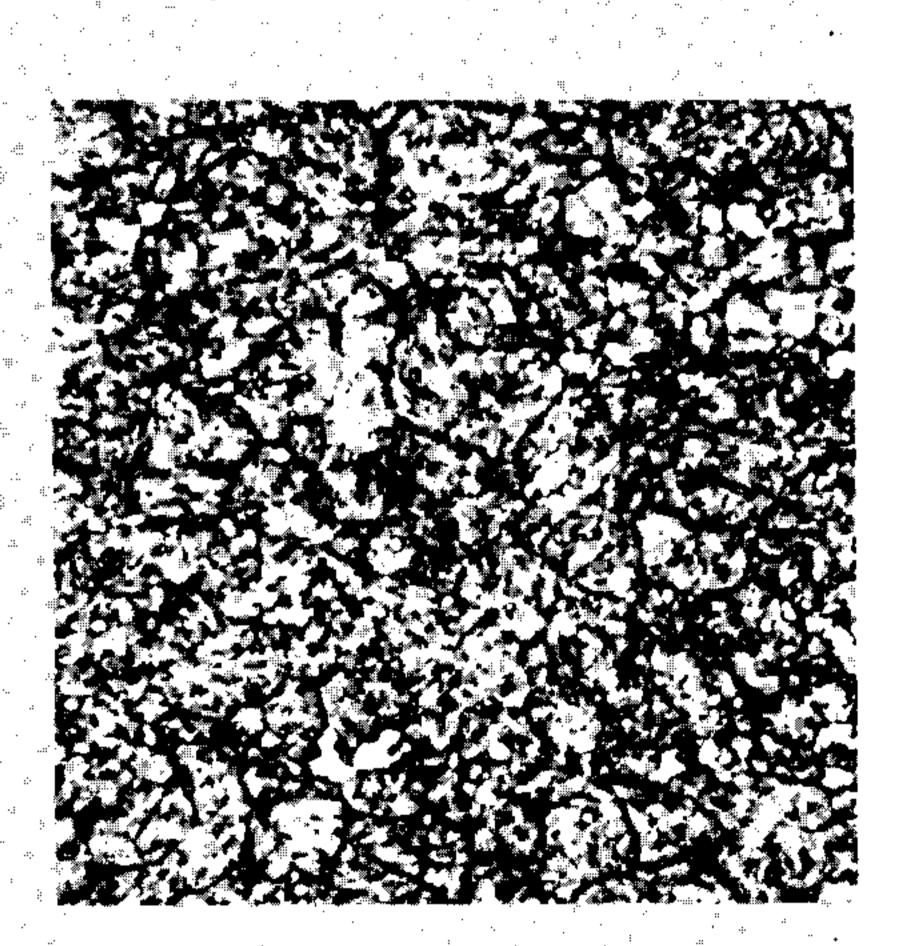


Fig. 79 Ferrovac-M2, austenitized 2250 F, 10 min. oil quench, tempered 1050 F, 2 plus 2 hours.

FIGURES 76 to 79 MICROSTRUCTURES SHOWING CARBIDE SIZE AND DISTRIBUTION IN BEARING STEELS.

Picral + 0.1% HCl Etch

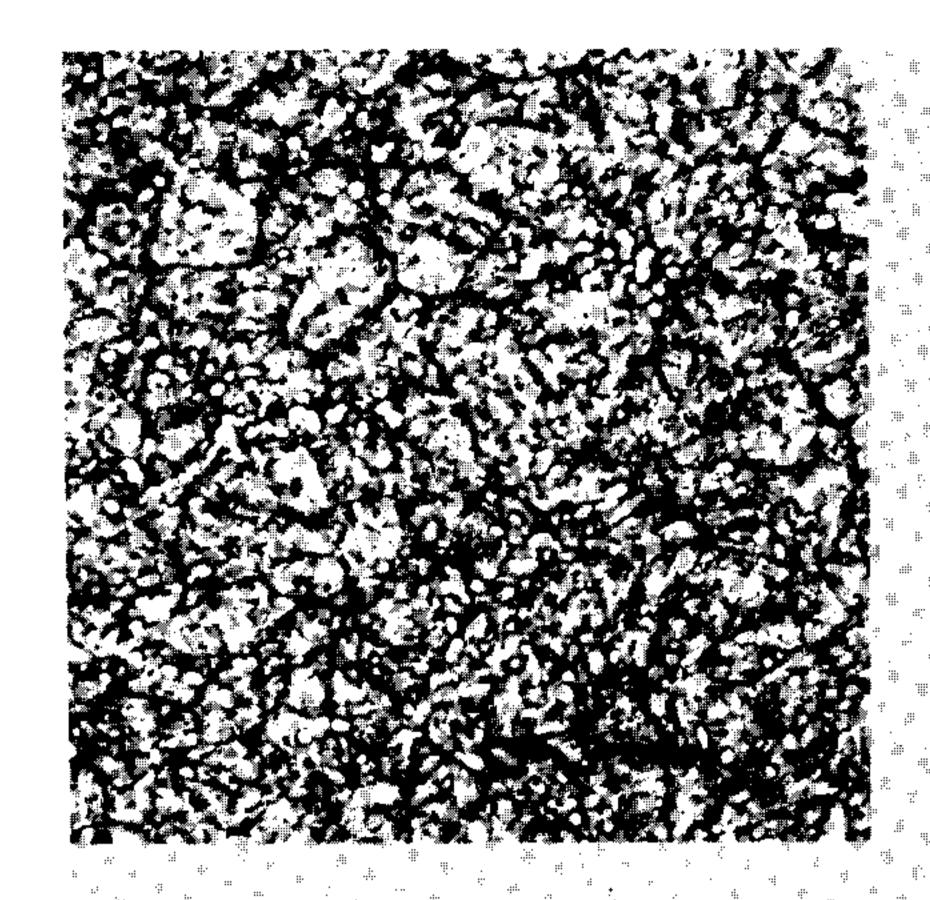
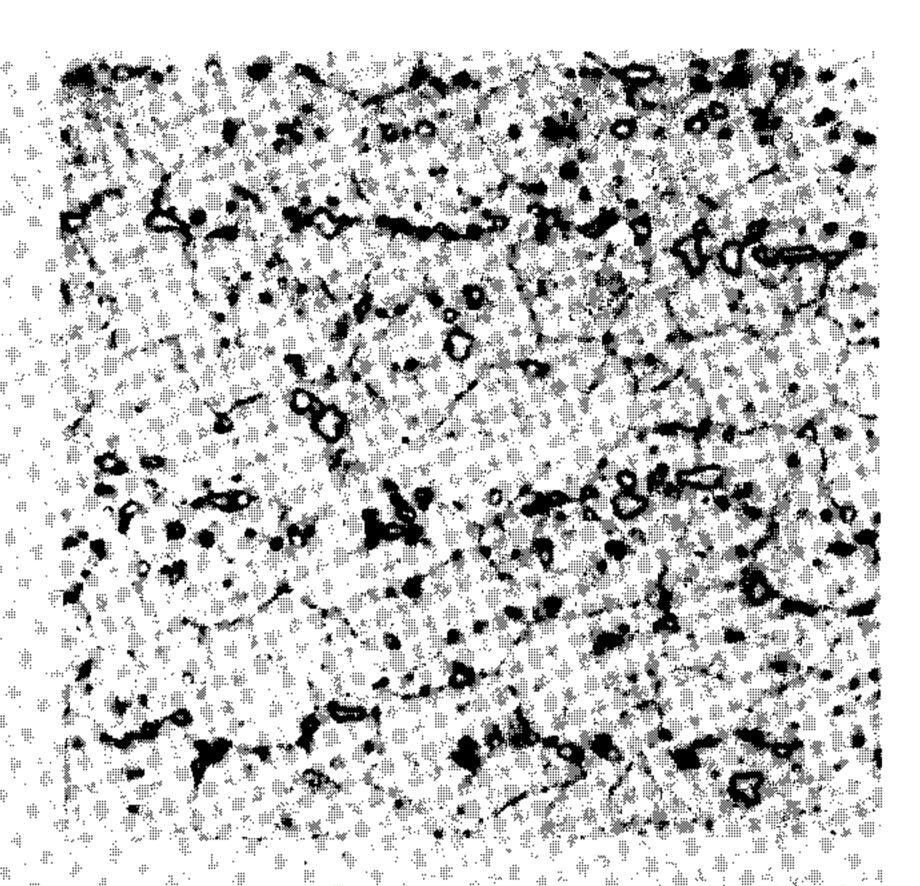
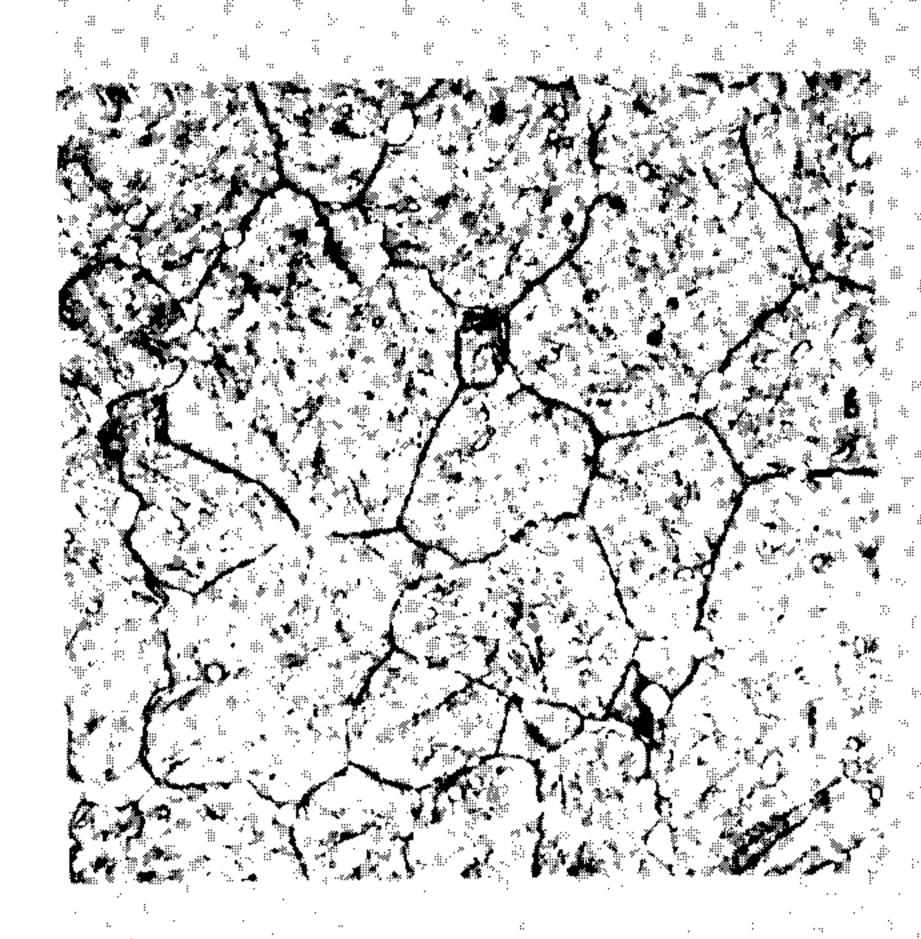


Fig. 80 Ml, austentized . 



MIO, austenitized 2200 F 15 min, oil quench, tempered 1050 F, 2 plus 2 hours.



2 plus 2 hours.

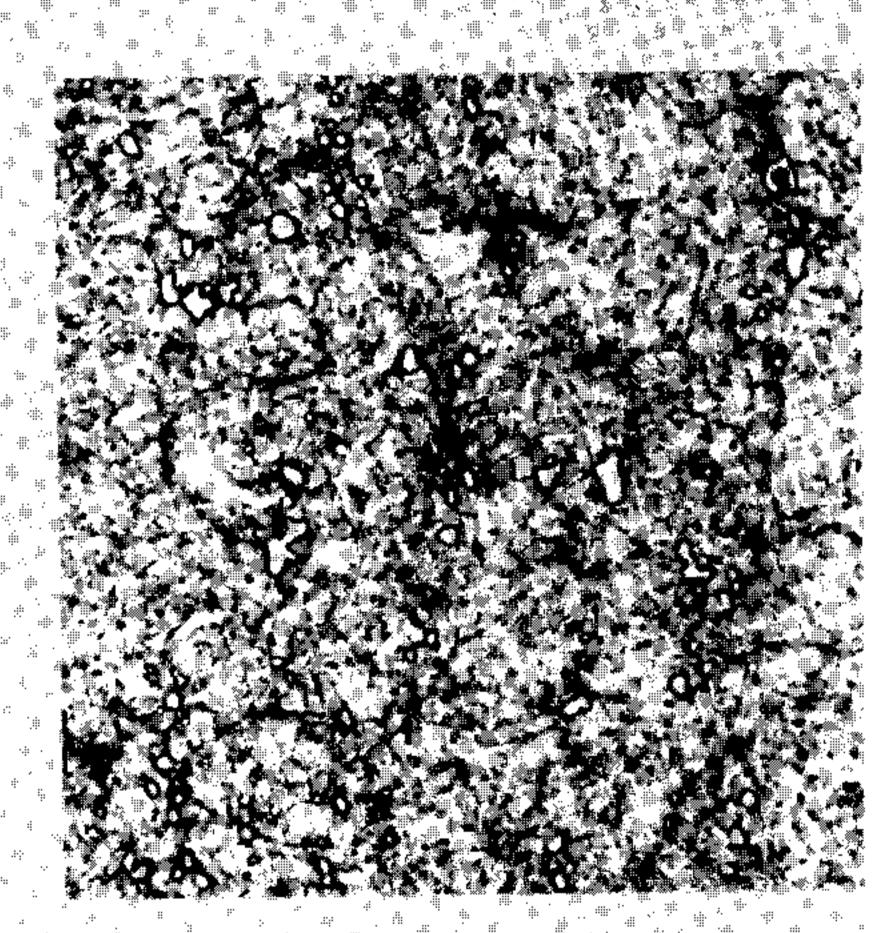
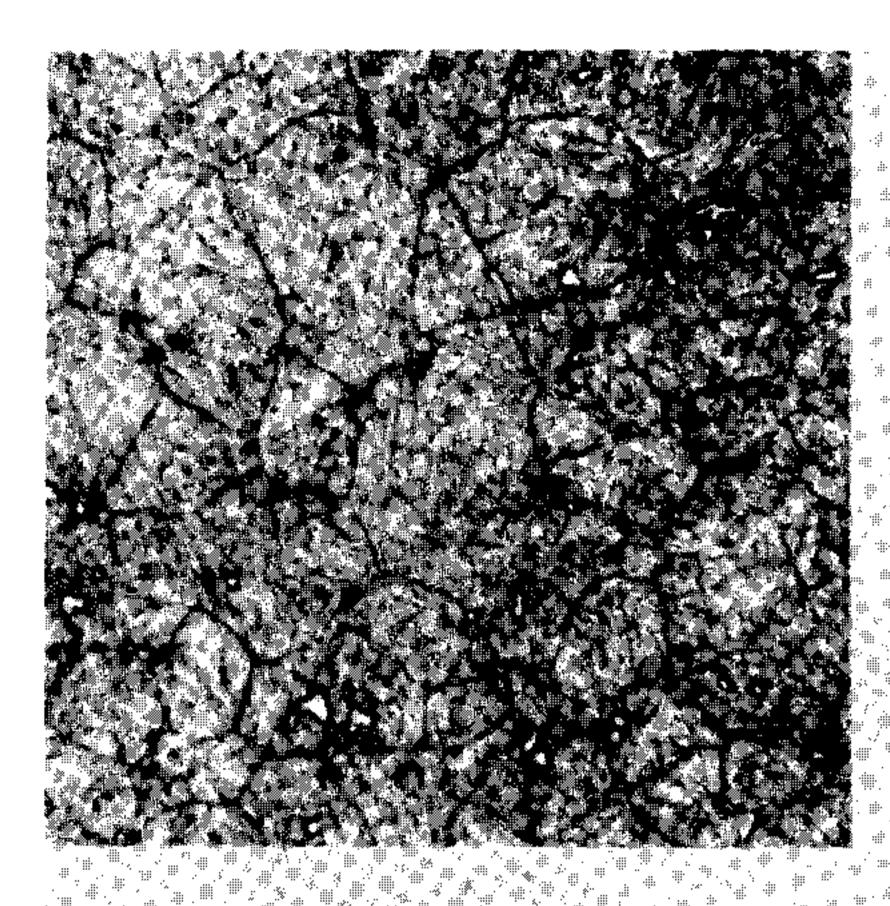


Fig. 82 HiC-MlO, austenitized Fig. 83 VSM, austenitized 2050 F 2200 F, 15 min. oil quench, quench, tempered 1050 F, 2 plus 2 hours.

FIGURES 80 to 83 MICROSTRUCTURES SHOWING CARBIDE SIZE AND DISTRIBUTION IN BEARING STEELS.

Picral + 0.2% HCl Etch

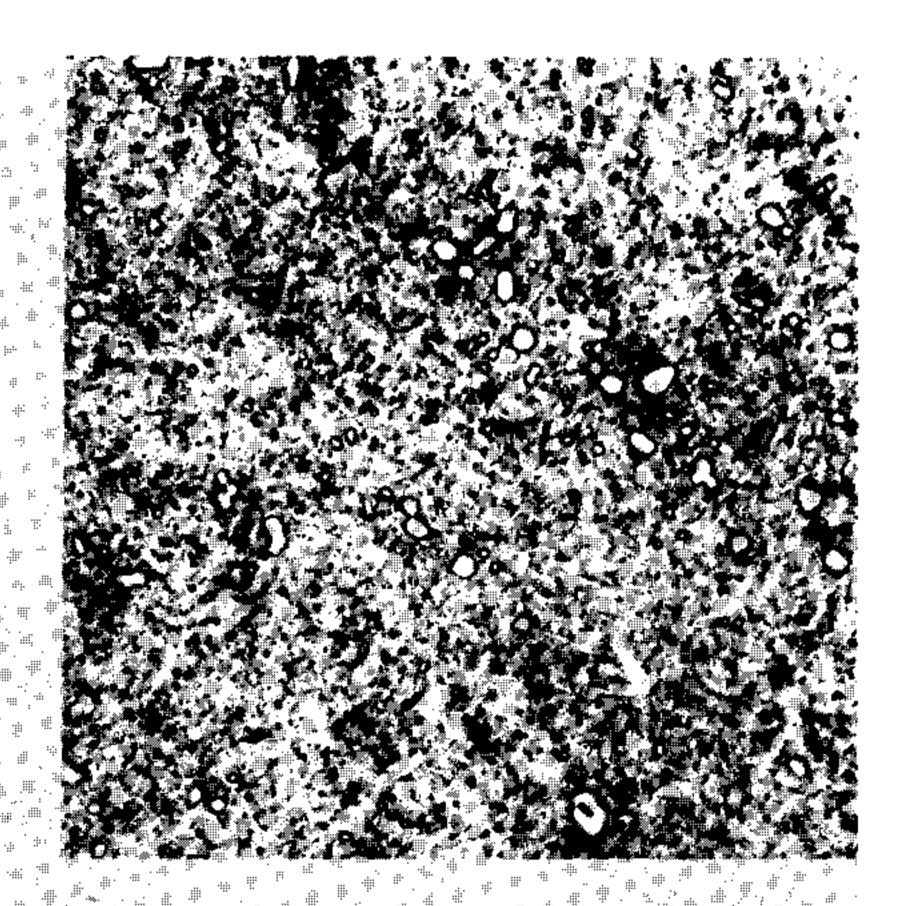


Hig. 84 MSO and entitled 2100 F.

20 min. oil quench.

tempered 1050 F. 2 plus

2 hours.



FIGURES 84 and 85 MICROSTRUCTURES SHOWING CARBIDE SIZE AND DISTRIBUTION IN BEARING STEELS.

Picral + 0.2% HCl Etch

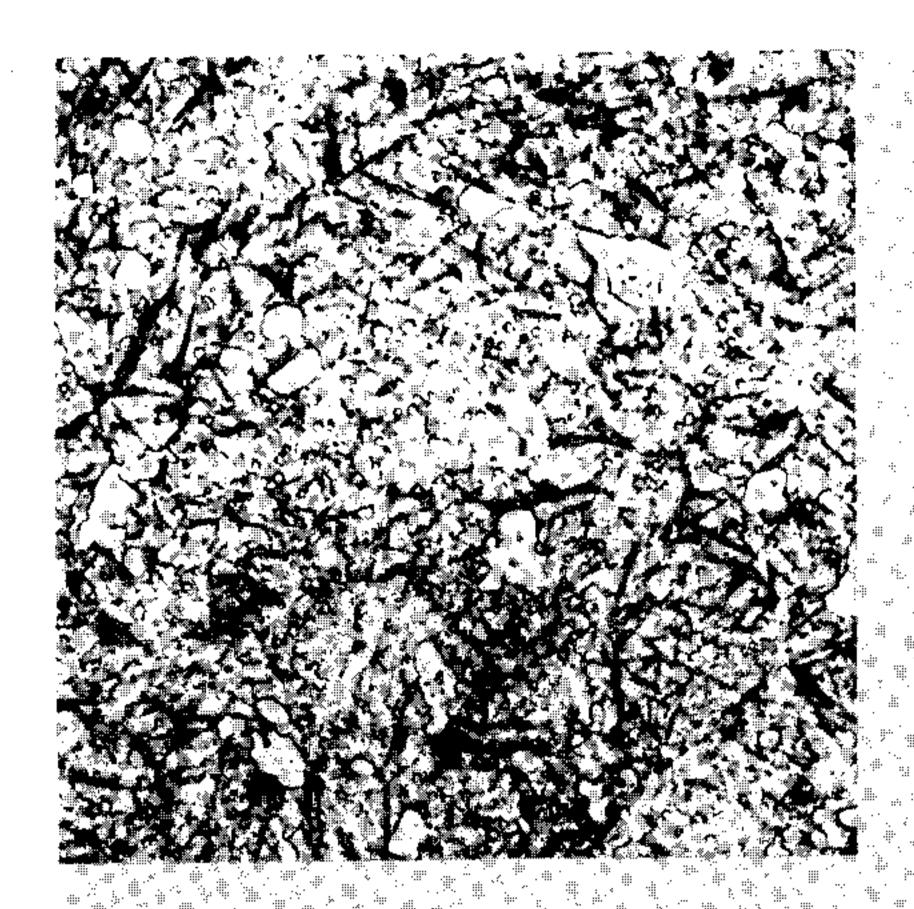


Fig. 86 Steel-A, austenitized

2000 F. 30 min. oil

quench, tempered 1000 F

20149 2 hours

Alcoholic 10% HCI Etch

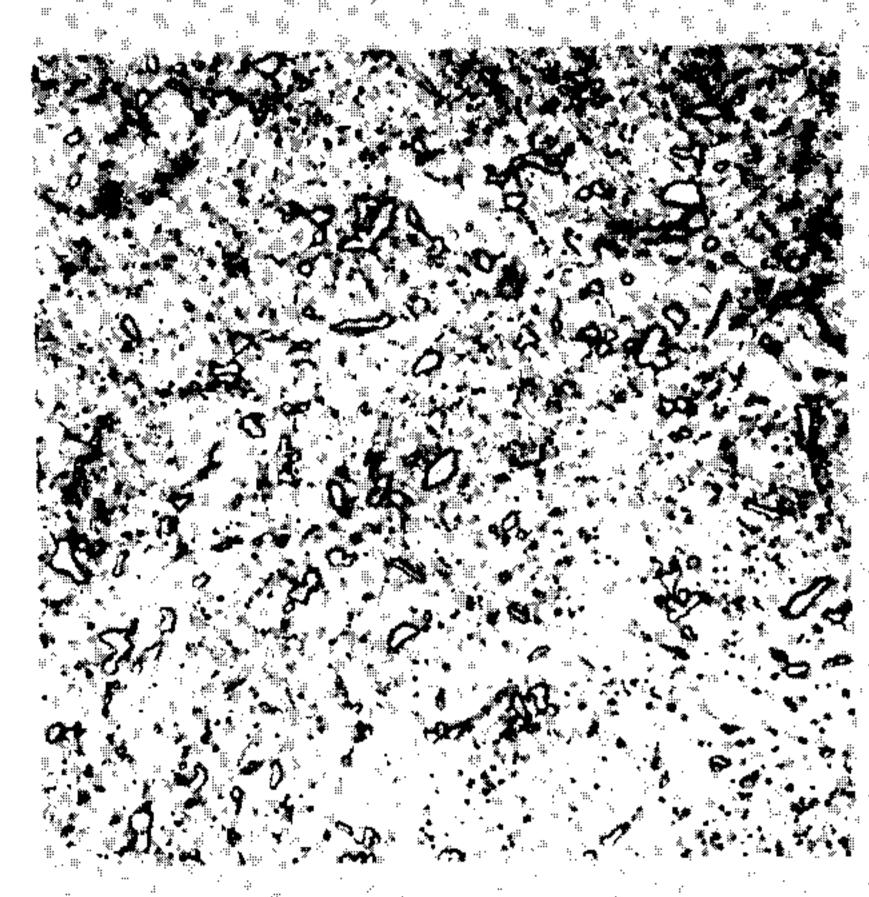


Fig. 88 Steel C, austenitized Fig. 89
2100 F, 20 min. oil
quench, tempered 1000 F
2 plus 2 hours.
Alcoholic 10% HCl Etch

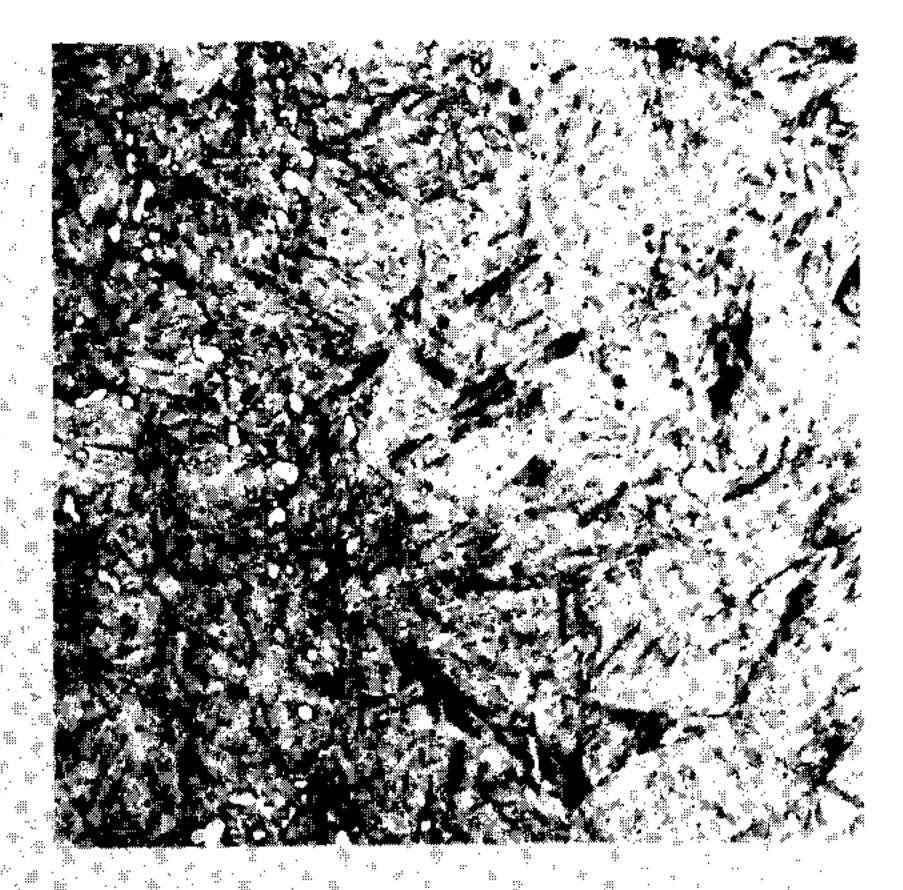


Fig. 87 Steel-B, austenitized

2150 F, 15 min. oil

quench, tempered 1000 F

2 plus 2 hours.

Alcoholic 10% HCl Etch

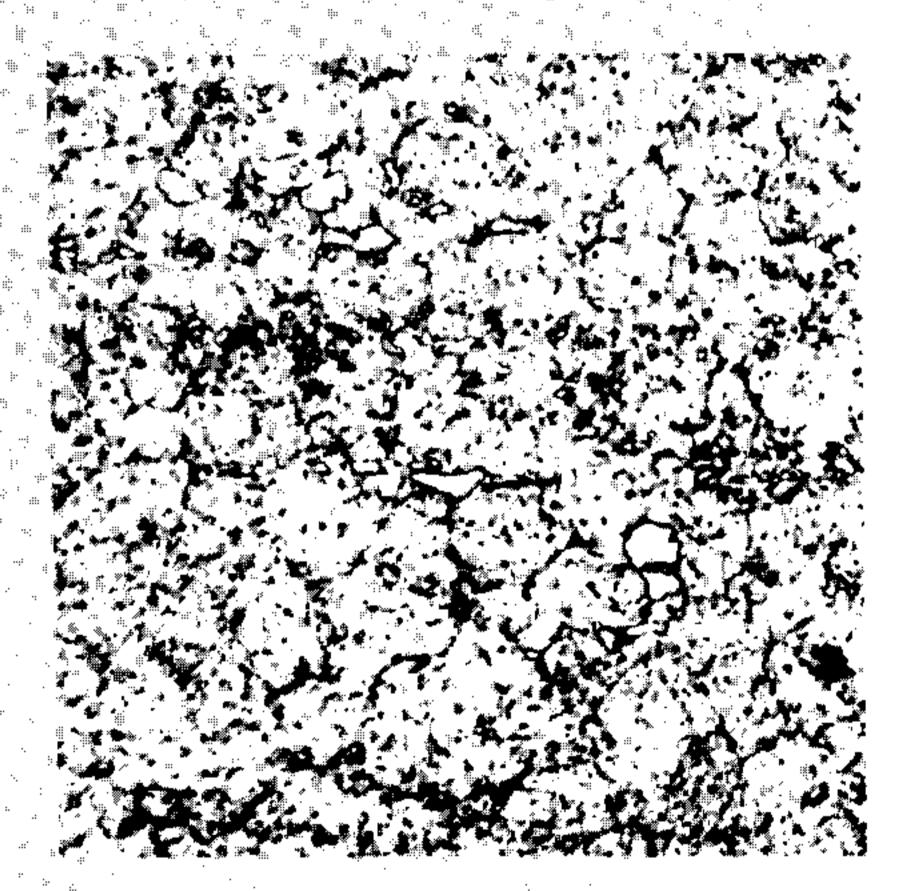


Fig. 89 Steel-D, austenitized
2200 F, 15 min. oil
quench, tempered 1000 F
2 plus 2 hours.
Picral Etch

FIGURES 86 to 89 MICROSTRUCTURES SHOWING CARBIDE SIZE AND DISTRIBUTION IN BEARING STEELS.

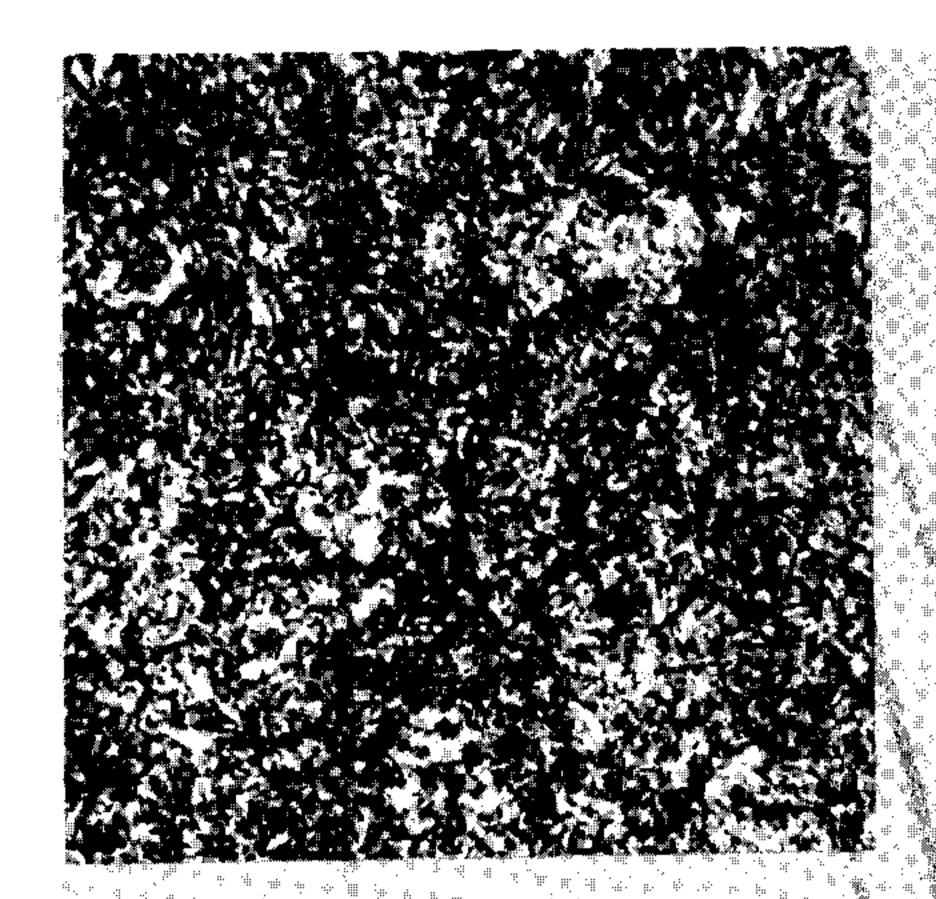


Fig. 90 Steel-E, austenitized 2 plus 2 hours.

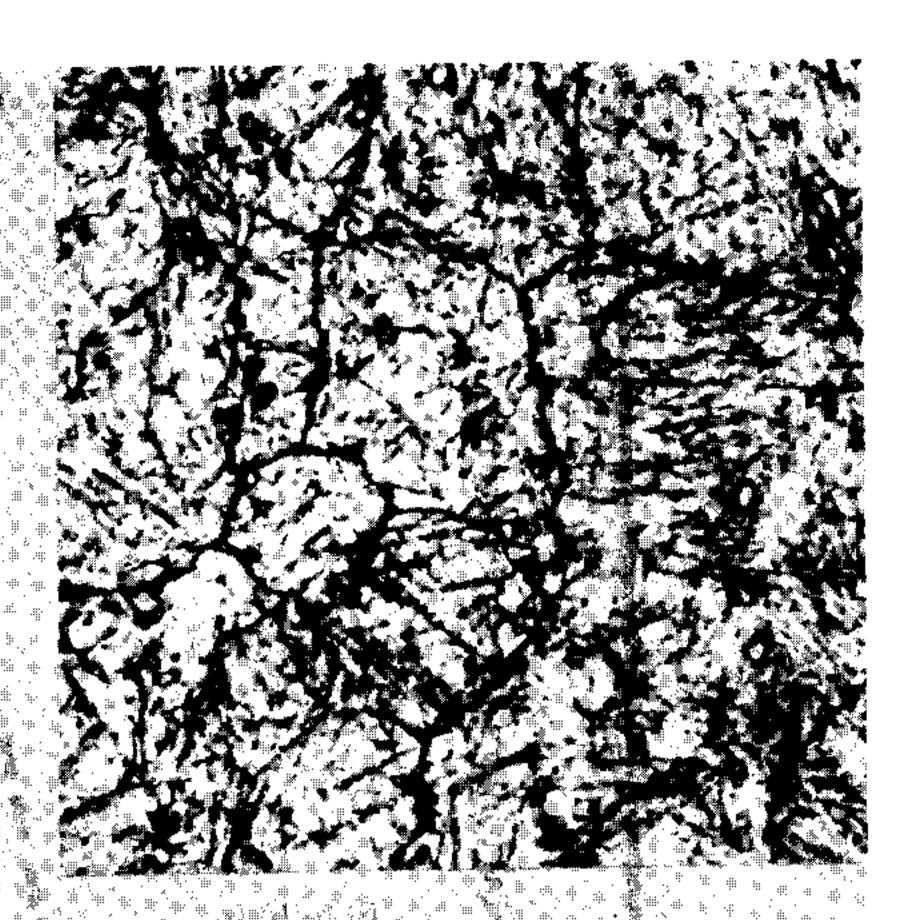


Fig. 91 Steel-F, austenlitzed
2200 F, 15 min. oil 2100 F. 20 min. oil. A series quench semperous que semperous que semperous quench semperous que semperous q

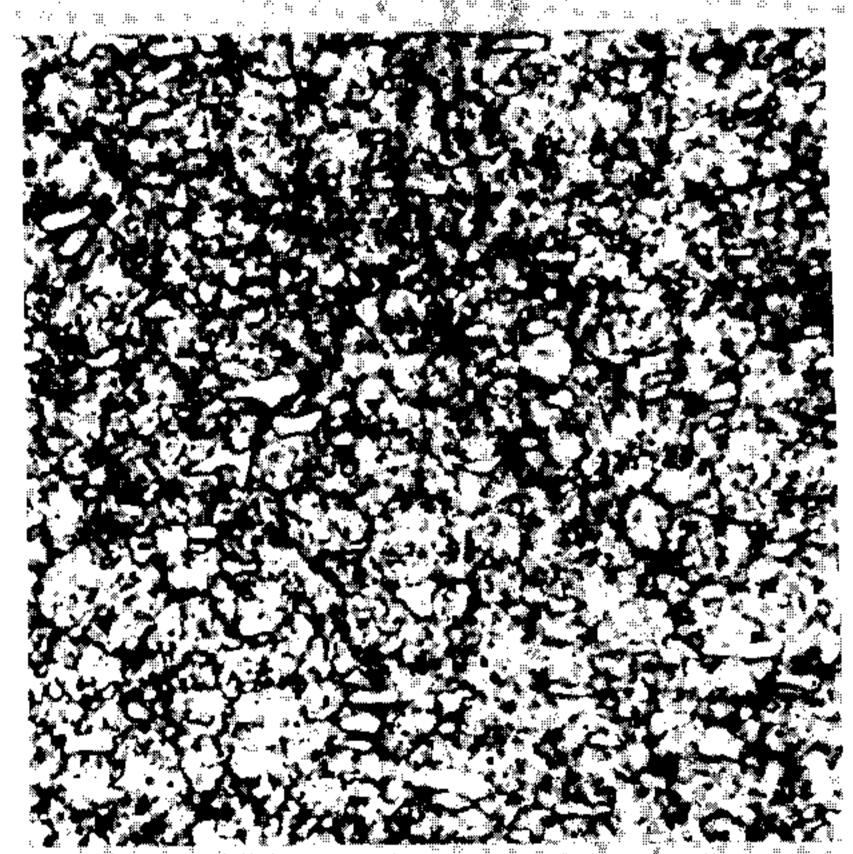


Fig. 92 Steel-G, austenitized

2200 F, 15 min., oil

quench, tempered 1000 F,

2 plus 2 hours.

FIGURES 90 to 92 MICROSTRUCTURES SHOWING CARBIDE SIZE AND DISTRIBUTION IN BEARING STEELS.

Picral + 0.3% HC1 Etch

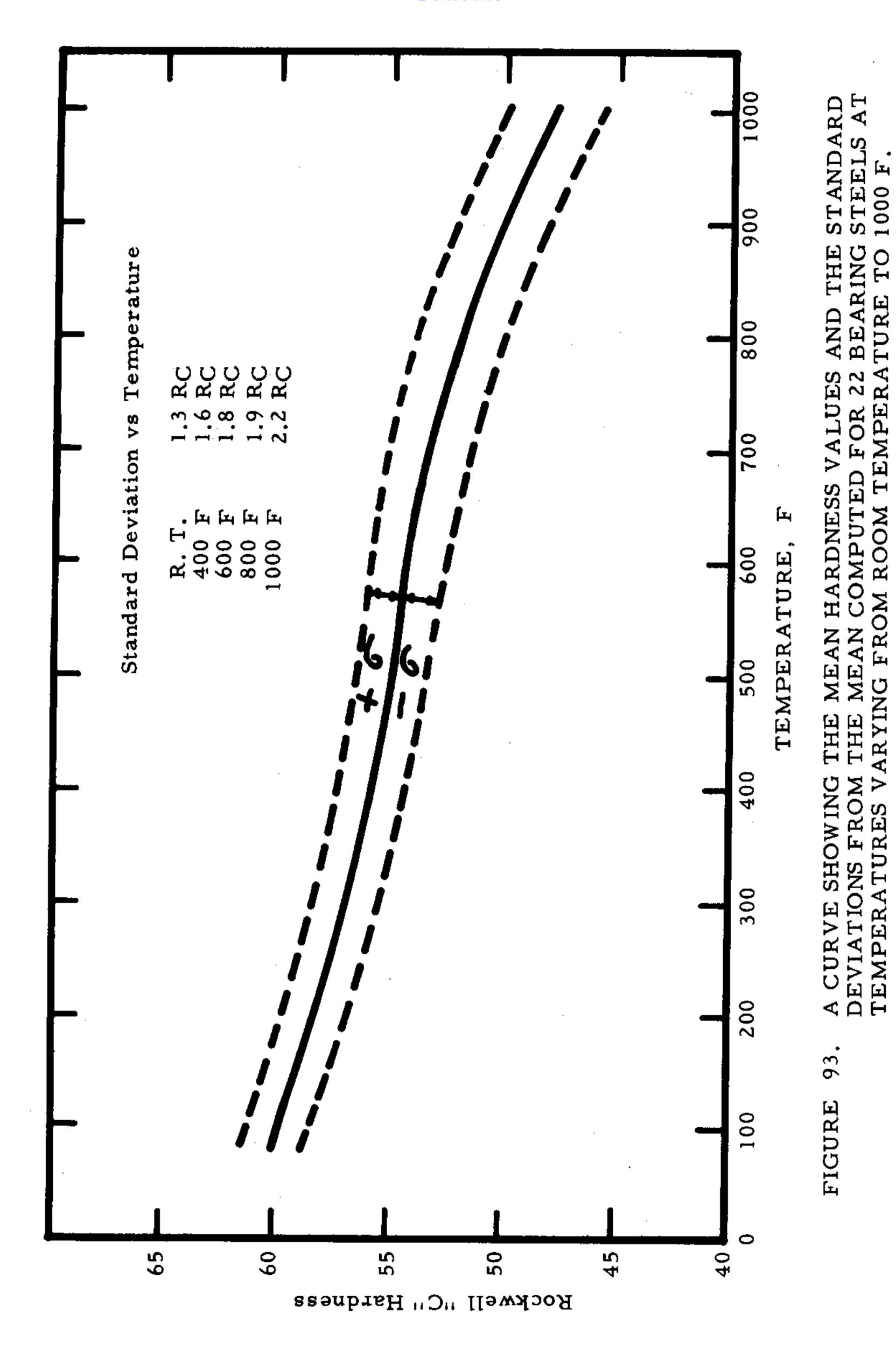
BEARING

VARYING FROM ROOM TEMPERATURE

COMPUTED

MEAN

DEVIATIONS FROM TEMPERATURES V.



WADC TR 57-343